

Bolt Shear Impact Test on In-Pavement Light Fixture Bolt Configurations

For MCB Industries Inc.

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P.O. Number Credit Card Date Tested Test Personnel **Test Documents**

November 8, 2021 Kate Fanning

Instant Shear Test(s) Rev. 2

Environmental Procedure: FAA EB83A Bolt Shear

Impact rev1.0

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1. REPORT REVISION HISTORY

Revision	Date	Description		
_	November 17, 2021	Initial release		



2. Introduction

This document presents the results of an environmental (ENV) test that was performed on four (4) In-Pavement Light Fixture Bolt Configurations (hereinafter referred to as the Device Under Test (DUT)). The DUTs were identified as follows:

Part Description	S/N
SAE J429 Grade 5 Bolt (uncoated) – 3.00"	S*
ASTM F593C Bolt (coated with integrated locking washer) – 3.00"	T*, X*, Z*
2-part ramp washers	N/A
HDG (Galvanized) Impact Plate	Red
HDG (Galvanized) Spacers – (1) 3/4"; (1) 1/2"; (1) 1/4"; (1) 1/8"; (2) 1/16"	Red
HDG (Galvanized) Mounting Bar	Red
#21 Impact Plate	Yellow
#21 Spacers – (1) 3/4"; (1) 1/2"; (1) 1/4"; (1) 1/8"; (2) 1/16"	Yellow
#21 Mounting Bar	Yellow
B50 (RCSC Class B Coating) Impact Plate	Green
B50 (RCSC Class B Coating) Spacers – (1) 3/4"; (1) 1/2"; (1) 1/4"; (1) 1/8"; (2) 1/16"	Green
B50 (RCSC Class B Coating) Mounting Bar	Green

3. TEST DOCUMENTS

The test was performed in accordance with the Customer Provided Document "Instant Shear Test(s) Rev. 2" and the Environmental Procedure: FAA EB83A Bolt Shear Impact rev1.0.

4. Modifications Made to DUT and/or Deviations to the Test Documents During Testing

No modifications were made to the DUTs during the testing. No deviations from the test documents were made during the testing.

5. SUMMARY

The following ENV test was performed, and the results are shown below:

Test Description	Document Section	Test Results	S/N	Date Tested
Bolt Shear Impact	Customer Provided Document "Instant Shear Test(s) Rev. 2" and Environmental Procedure: FAA EB83A Bolt Shear Impact rev1.0	See Section 11.1.4	S*, T*, X*, Z*	11/08/2021

6. OPERATION STATES

The ENV test was performed with the DUTs operating in one or more of the test modes described below.

6.1. Unpowered

The DUTs were unpowered throughout the test.

7. Performance/Monitoring

The DUTs were visually monitored and video recorded during testing. A torque wrench was used to measure the breakaway torque of the DUT's bolt following the test.



8. ACCEPTANCE CRITERIA

The DUTs shall withstand the specified shear impact force without breakage or excessive deformation. The DUTs shall have the ability to be fully removed from the fixture after the test is complete.

9. CERTIFICATION

Elite Electronic Engineering Incorporated certifies that the information contained in this report was obtained under conditions which meet or exceed those specified in the test documents, except if noted otherwise. The data presented in this test report pertains to the DUTs at the test date as operated and monitored if required. Any electrical or mechanical modification made to the DUTs subsequent to the specified test date will serve to invalidate the data and void this certification.



10. DEVICE UNDER TEST PHOTOGRAPHS





DUT Test Bolts (SAE J429 Grade 5 Bolt – Top and ASTM F593C Bolt – 3.00" – Bottom)





2-part Ramp Washers





B50 (RCSC Class B Coating) Samples: Spacers – (1) 3/4"; (1) 1/2"; (1) 1/4"; (1) 1/8"; (2) 1/16" Impact Plate and Mounting Bars





HDG (Galvanized) Samples: Spacers – (1) 3/4"; (1) 1/2"; (1) 1/4"; (1) 1/8"; (2) 1/16" Impact Plate and Mounting Bar





#21 Samples: Spacers – (1) 3/4"; (1) 1/2"; (1) 1/4"; (1) 1/8"; (2) 1/16" Impact Plate and Mounting Bars



11. TEST SECTIONS

11.1. Bolt Shear Impact

11.1.1. Requirements:

The DUTs shall withstand the specified shear impact force without breakage or excessive deformation. The DUTs shall have the ability to be fully removed from the fixture after the test is complete.

11.1.2. Test Procedure:

The test was conducted in accordance with the Customer Provided Document "Instant Shear Test(s) Rev. 2" and the Environmental Procedure: FAA EB83A Bolt Shear Impact rev1.0.

FAA EB83A, Appendix A, specifies the aircraft configuration and assumptions used to determine the shear (maximum traction) force imparted on in-pavement light fixtures by an Airbus A380-800 aircraft. This is the maximum potential shear force present on U.S. airports and was used to set the two target shear impact forces and impact durations for testing:

Description	Total Shear Force (lb)	Shear Force per Bolt (lb)	Target Shear Impact Force (lb)	
Maximum Traction Force	19,760	3,293	3,300	
Differential Force Required by Bolts	10,620	1,770	1,770	

- A380-800 Landing Speed ≈ 170 mph = 2992 in/sec.
- Time for Tire to Roll Over 12" Light Fixture = 12 / 2992 = 4 ms
- Target Impact Duration <= 4 ms

Target bolt torque was specified by MCB Industries, Inc. to achieve the following clamping forces:

Bolt Type	Specified Torque (ft-lb)	Clamping Force (lb)
SAE J429 Grade 5 (uncoated)	62 to 64	4,900
ASTM F593C (coated with integrated locking washer)	42 to 44	3,800

The DUTs consisted of an assembly of washers, spacers, and/or shims in a specific order, referred to as a "stack," and held in place with a 3/8"-16 bolt of a specified length. The DUT stack was then assembled and mounted to the impact tester as specified by MCB Industries, Inc. personnel. The DUT impact plate and stack were aligned to be perpendicular to the hammer impact path. Using a torque wrench, the DUT test bolt was tightened to the specified torque. The impact hammer was adjusted as necessary to align with the center of the DUT impact plate. The number of impacts was defined by MCB Industries, Inc. personnel.

Several stack orientations were defined by MCB Industries, Inc. personnel, to evaluate the performance of different surface coatings. The orientations are as follows:

Orientation	Components	Number of Hits	Target Force (lbf)
S*	Grade 5 3.00" Bolt 2-part lock washer HDG (Galvanized) Impact Plate HDG (Galvanized) Spacers – (1) 3/4"; (1) 1/2"; (1) 1/16" HDG (Galvanized) Mounting Bar	1	1770



Т*	F593C 3.00" Bolt HDG (Galvanized) Impact Plate HDG (Galvanized) Spacers – (1) 3/4"; (1) 1/2"; (1) 1/16" HDG (Galvanized) Mounting Bar	1	1770
X*	F593C 3.00" Bolt B50 (RCSC Class B Coating) Impact Plate B50 (RCSC Class B Coating) Spacers – (1) 3/4"; (1) 1/2"; (1) 1/16" B50 (RCSC Class B Coating) Mounting Bar	1	1770
Z*	F593C 3.00" Bolt #21 Impact Plate #21 Spacers – (1) 3/4"; (1) 1/2"; (1) 1/16" #21 Mounting Bar	1	1770

A pendulum hammer with a one (1) meter swing arm and a minimum 7.4-kilogram head was used and swung from a height of approximately one (1) meter. An accelerometer was attached to the impact hammer. A 1/2" thick impact-resistant PVC plate was attached to the impact plate for each orientation. A GoPro camera was used to capture high speed video of each test.

Following the test, the breakaway torque required to loosen the bolt on the stack was recorded, and the DUT stack was removed from the test area. A visual examination of the DUT was conducted.

11.1.3. Description of Test Apparatus:

Eq ID	Equipment Description	Manufacturer	Model No.	Serial No.	Frequency Range	Cal Date	Due Date
CGP2	GOPRO CAMERA	GOPRO	HERO8 BLACK	C3331324678283		CNR	
ELF2	VIBRATION VIEW I/O BOX	VIBRATION RESEARCH	VIBRATION VIEW	0B3C8843		12	
XAE2	SHOCK ACCELEROMETER	PCB PIEZOTRONICS	350B03	20158	2HZ-10KHZ	4/1/2021	4/1/2022
XWN2	TORQUE WRENCH	PROTO	JH5-100FRB	01163500	100FT/LB	4/28/2021	4/28/2022
XZJ2	IK IMPACT TEST MACHINE	ELITE	IEC 60068-2-75			9/2/2021	9/2/2023

I/O: Initial Only N/A: Not Applicable CNR: Calibration Not Required

Note 1: For the purpose of this test, the equipment was calibrated prior to the test or monitored by a calibrated instrument.

11.1.4. Test Results:

The DUTs were subjected to and completed the bolt shear impact test per the Customer Provided Document "Instant Shear Test(s) Rev. 2" and the Environmental Procedure: FAA EB83A Bolt Shear Impact rev1.0. The peak force applied to each DUT and post-test observations can be seen below. The DUTs were returned to the customer for further evaluation.

The observations and comparisons of each DUT orientation led to the following conclusions about the behavior of the in-pavement light fixture bolted connection described in FAA EB83A:

- 1. Grade 5 bolts at 4900 lb and F593C bolts at 3800 lb clamping force demonstrated similar performance without significant slipping or bending for all faying surface coatings evaluated.
- 2. Higher friction coefficients on the faying surfaces reduce slipping under the same clamping force.
- 3. The actual friction coefficients of each faying surface coating were not evaluated. The target peak impact force of 1770 lb is based on the EB83A Appendix A calculations assuming a friction coefficient of 0.37 for hot-dip galvanized coatings. Higher friction coefficients would lower the peak impact force requirement.



Orientation	Pretest Torque (ftlb)	Breakaway Torque (ftlb)	Peak Force (lbf)	Duration (ms)	Corrected Duration (ms)	Observations
S*	63.1	58.2	1878.4	6.40	0.95	Little to no bending.
T*	42.3	25.3	1858.8	7.30	0.75	Little to no bending.
X*	43.4	26.0	1948.8	6.70	1.0	Little to no bending.
Z*	43.2	23.4	1867.6	6.65	0.70	Little to no bending.



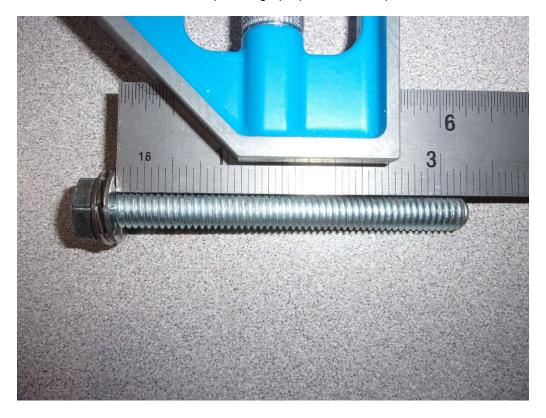


Test Setup Photograph



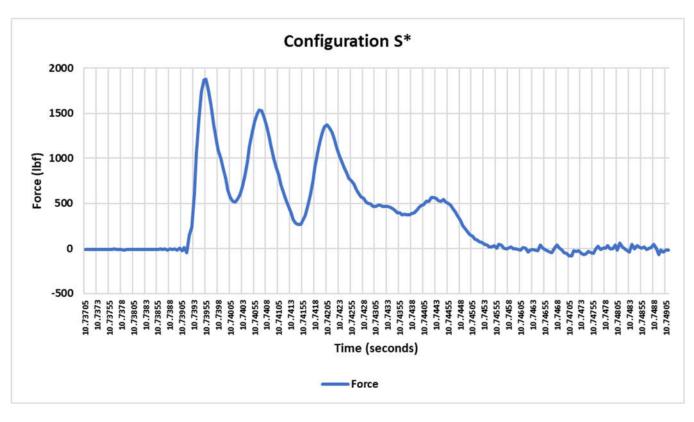


Test Setup Photograph (Orientation S*)



Post-test Photograph (Orientation S*)



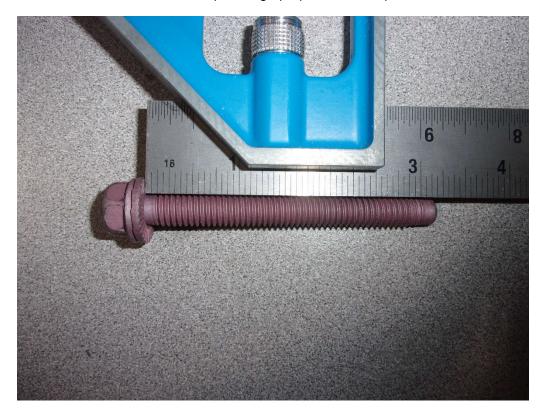


Force Graph



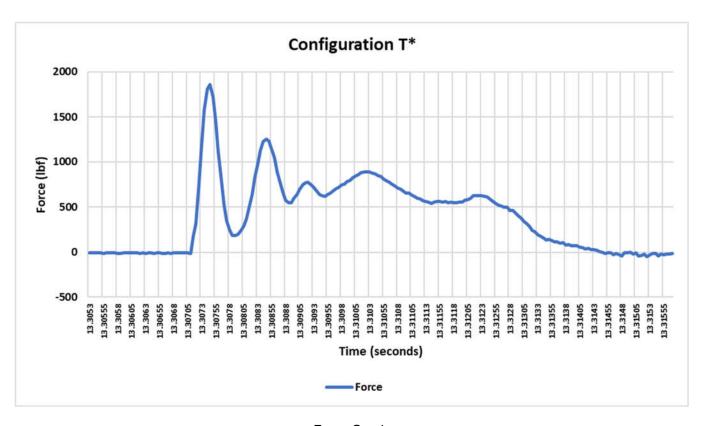


Test Setup Photograph (Orientation T*)



Post-test Photograph (Orientation T*)





Force Graph



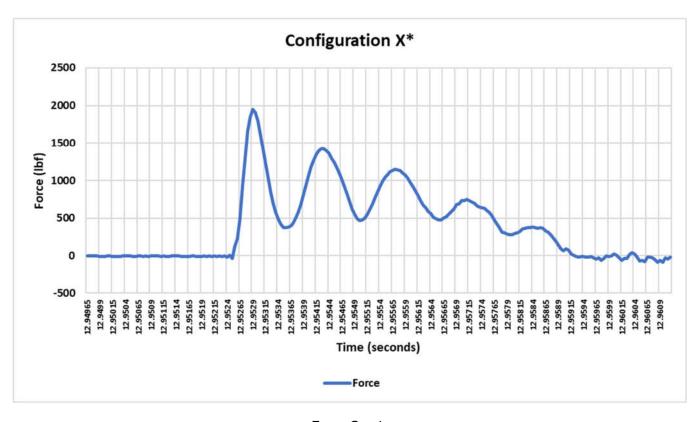


Test Setup Photograph (Orientation X*)



Post-test Photograph (Orientation X*)





Force Graph



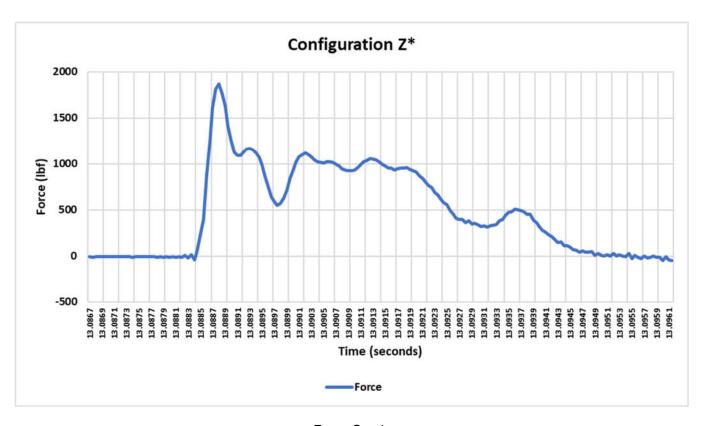


Test Setup Photograph (Orientation Z*)



Post-test Photograph (Orientation Z*)





Force Graph