



Communications IPT
Ku-Band Antenna UAV Optimization Trade Study – Preliminary
Rev D
February 17, 2009

Advanced Tech Engineering, Inc.
Frank A. Lucchesi

Advanced Tech Engineering, Inc
This document contains financial, business, scientific, technical or engineering information. Disclosure to others, use,
or copying, without the prior written authorization of Advanced Tech Engineering, Inc. is strictly prohibited.

UNCLASSIFIED

Contents

- Purpose of the Study
- Caveats / Tasks
- Top-Level Take-away
- UAV to GDT Geometry
- Coordinate Reference Diagram
- Ku-Band Ideal Free Space Antenna Pattern
- Nose Fuselage Placement
- Sled Placement
- Top Placement
- Aft Fuselage Placement
- Tail Boom Placement
- Antenna Isolation aka “RF Coupling” Table
- Coordinate Frame Reference
- Radar / Comm. Unobstructed LOS
- Radar / Comm. Horizon Reference

Purpose of the Study

- Evaluate alternative locations for Ku-Band & UHF Antennas
- Rev D study addresses Ku-band antenna
- UHF trade study in progress

Caveats & Tasks

- The overall EM gain/coverage trades & coupling analyses are considered preliminary pending completion and commencement of the following tasks:
 - 1) **Incorporate realistic UAV EM material properties**

Narrative: Present model assume perfect EM PEC conductor for UAV material which results in high number of multipath reflections and strong antenna-to-antenna coupling. This is worse case and provides pessimistic results. Need to update material EM properties ASAP.
 - 2) **Validate Ku-band Antenna Model**

Narrative: An ideal Ku-band Omni-directional antenna is currently modeled. The Ku-band antenna provided by vendorX requires near field (Xfdtd) validation prior to importing into the overall far field (Xgtd) model. The antenna has greater directivity in theta than the theoretical model. However, the trend is valid which shows relative performance of locating the antenna at various locations
 - 3) **Quantitative Gain / Coverage Assessment**

Narrative: The present PBS states unobstructed coverage which is unrealizable. We expect that the final PBS will quantify the requirement. Pending completion of 1 and 2 above, Gain / coverage Vs geometry will be assessed to close the link at specified ranges
 - 4) **Locating other data link antennas**

Narrative: Trades to select location for the other antenna are pending the outcome of selecting Ku-band location. The antenna gain patterns over desired coverage (geometry driven) must support link closure at 50 nmi, 3 kft, in clear and rain environmental conditions for Ku-band and 100 nmi, etc. for UHF band per MEUAS RFP response.
 - 5) **Complete coupling and commence co-site analysis**

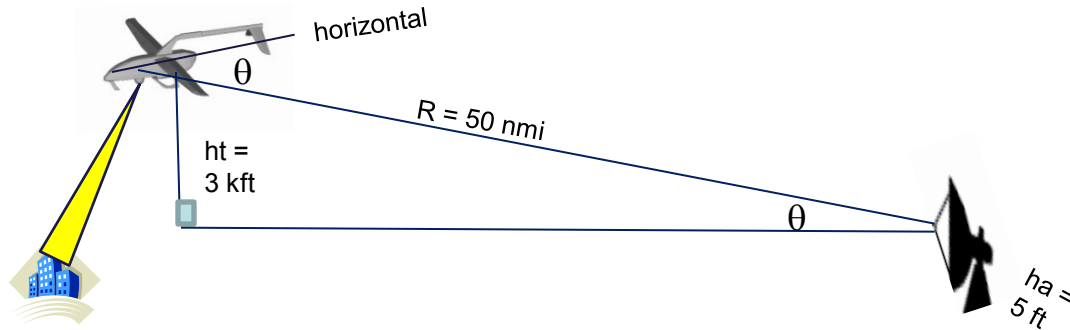
Narrative: Coupling and co-site analysis is pending the selection of all data link antenna locations. This outcome of this analysis drives link analysis and data link design (e.g., filters, materials, etc.)

Top-level Take-Away

- The location recommended for Ku-Band antenna provides the best antenna gain over desired coverage volume for the Spiral-1 UAV configuration
- Additional information as defined in the caveat/tasks slide are required improve model fidelity and predictions for demo UAV and Tier II UAV
- Based on current EM analyses, the Spiral-1 UAV configuration does not support Tier II UAS link closure over required /assumed geometries at range and AGL altitude
- Path forward for Spiral-1 Demo UAV:
 - Choose acceptable location and carefully construct scenario based on physical and performance constraints
- Path forward for Tier II UAV:
 - Continue improving the fidelity of the EM models
 - Consider UAV modifications that supports data link apertures and closure requirements (e.g., two switched apertures)
 - Consider CDL increment III transmit power improvements
 - Custom electronically steerable directional antenna

UAV to GDT Geometry

50nmi slant range, 3 kt AGL altitude



Finding Target Elevation θ_t (Flat Earth Calculation)

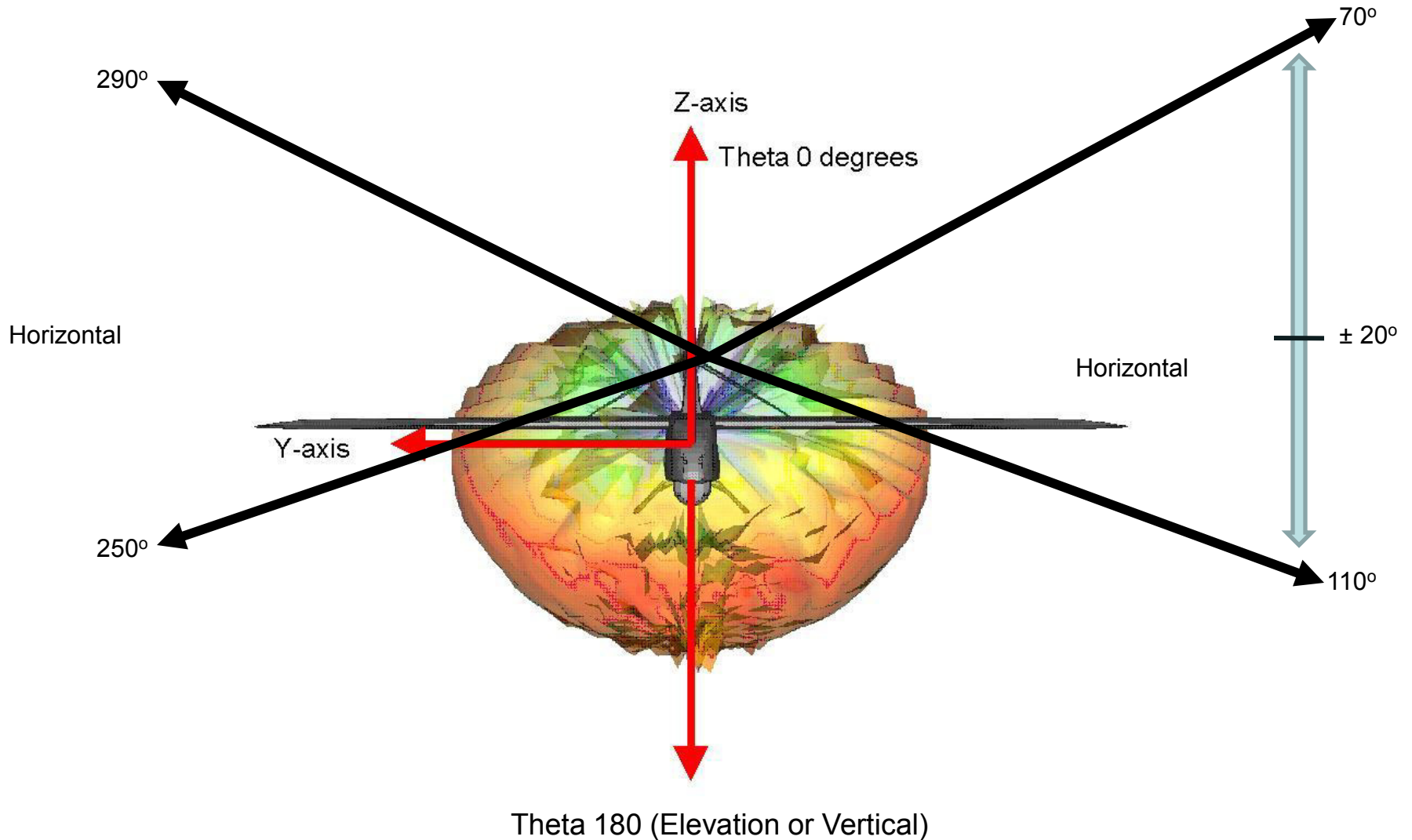
$$\sin \theta = (h_t - h_a)/R$$

$$\begin{aligned} \theta &= \text{Arcsin} (3000 \text{ ft}/6080 \text{ ft /nmi})/50 \text{ nmi} \\ &= 0.57 \text{ degrees} \end{aligned}$$

Aircraft level flight (e.g., no roll or pitch) @ 50 nmi slant range and 3 kft Altitude AGL
= - 0.57° below horizontal axis

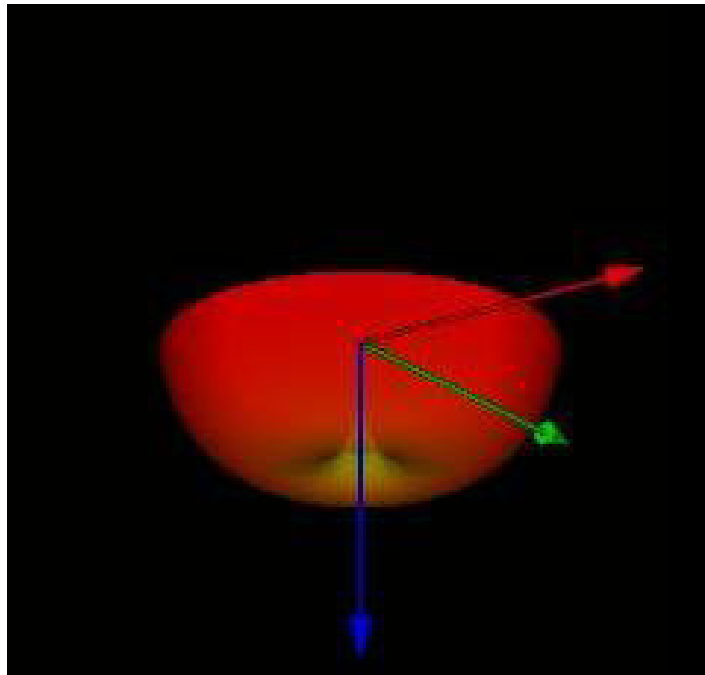
Aircraft normal flight (assumes 20° max roll angle during orbits or flight conditions)
= +19.43° (above horizontal axis) to 20.57° (below horizontal axis)

Reference Diagram

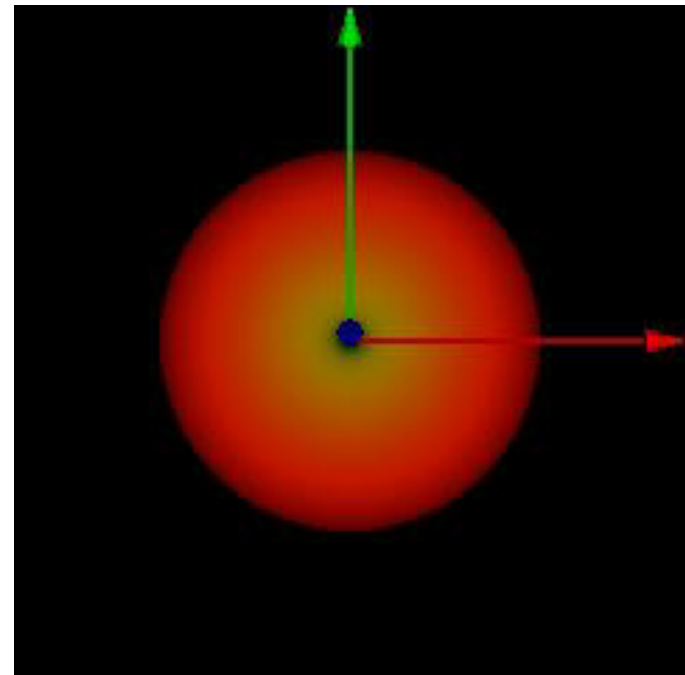


Theta 180 (Elevation or Vertical)

Ideal Monopole Transmitter in Free Space “No Platform EM Interaction”

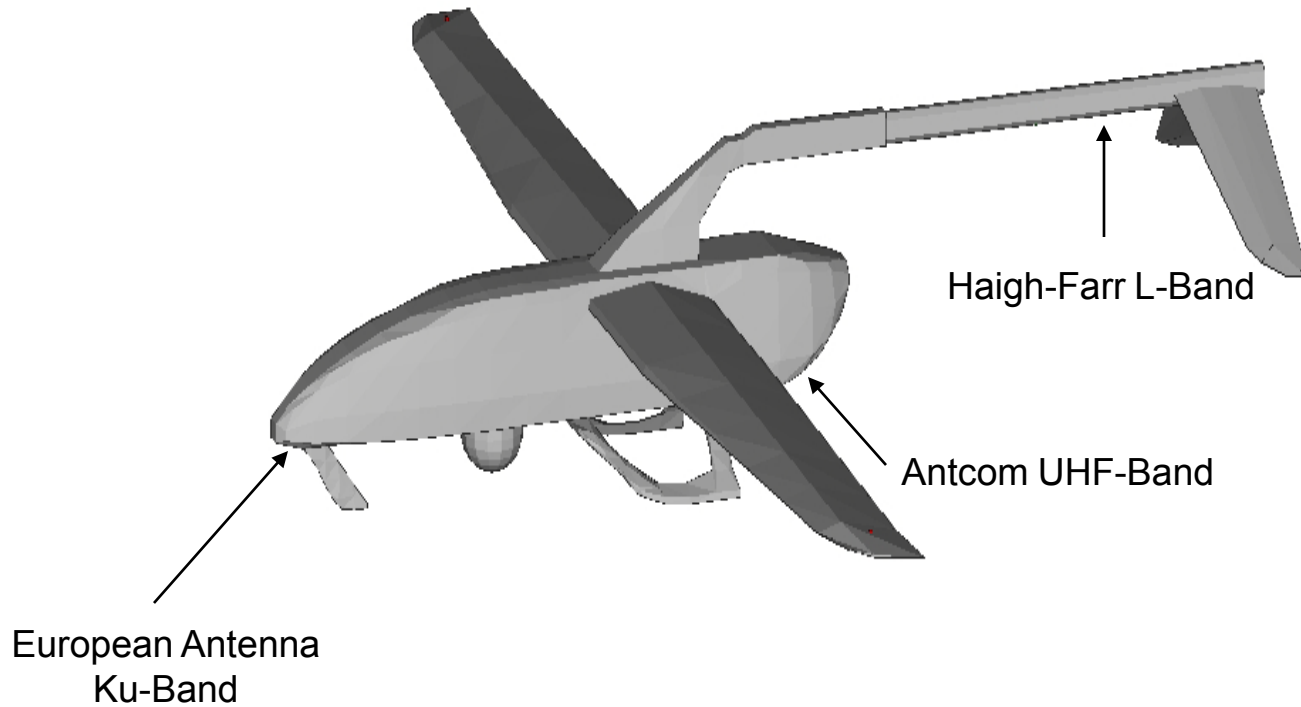


Side view of monopole
radiation pattern.

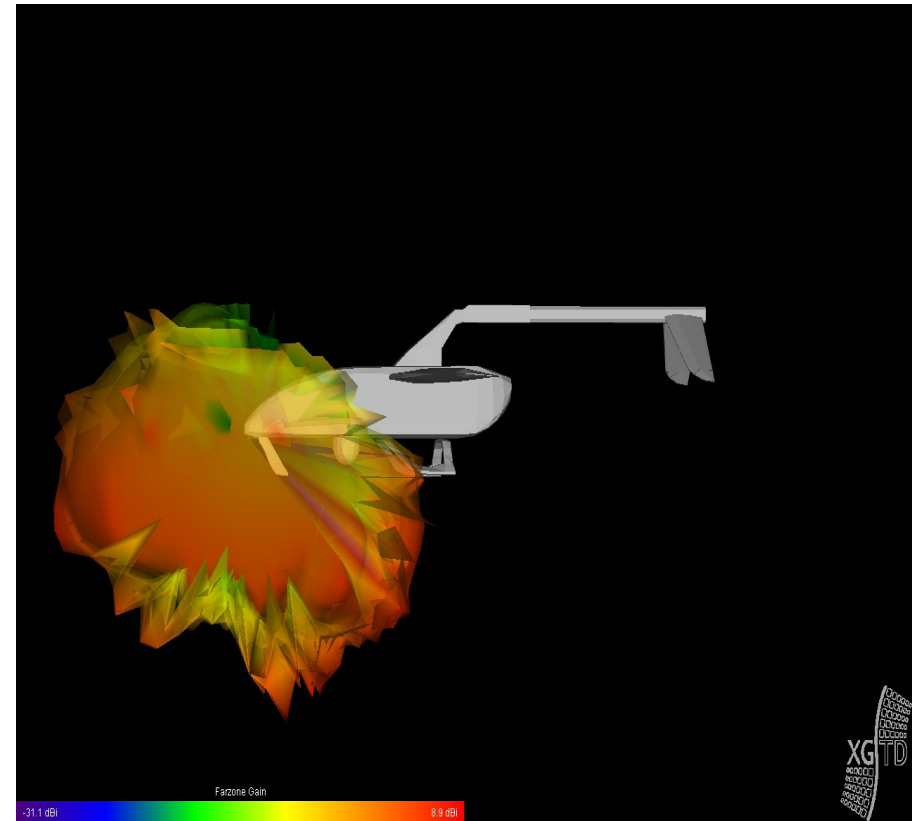
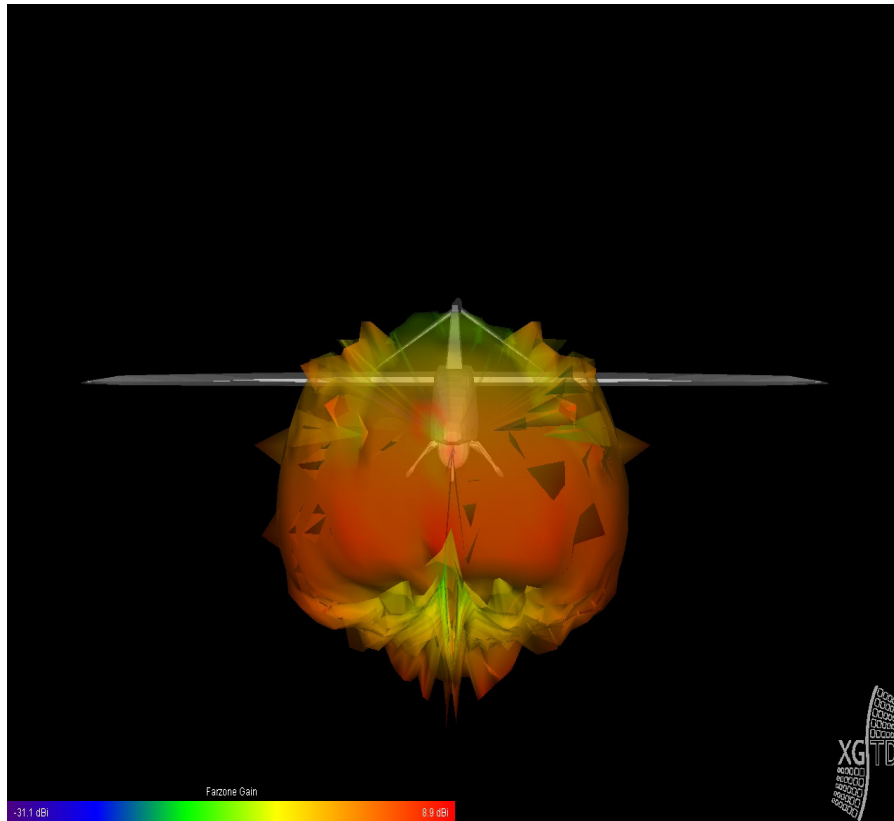


Bottom view of monopole
radiation pattern.

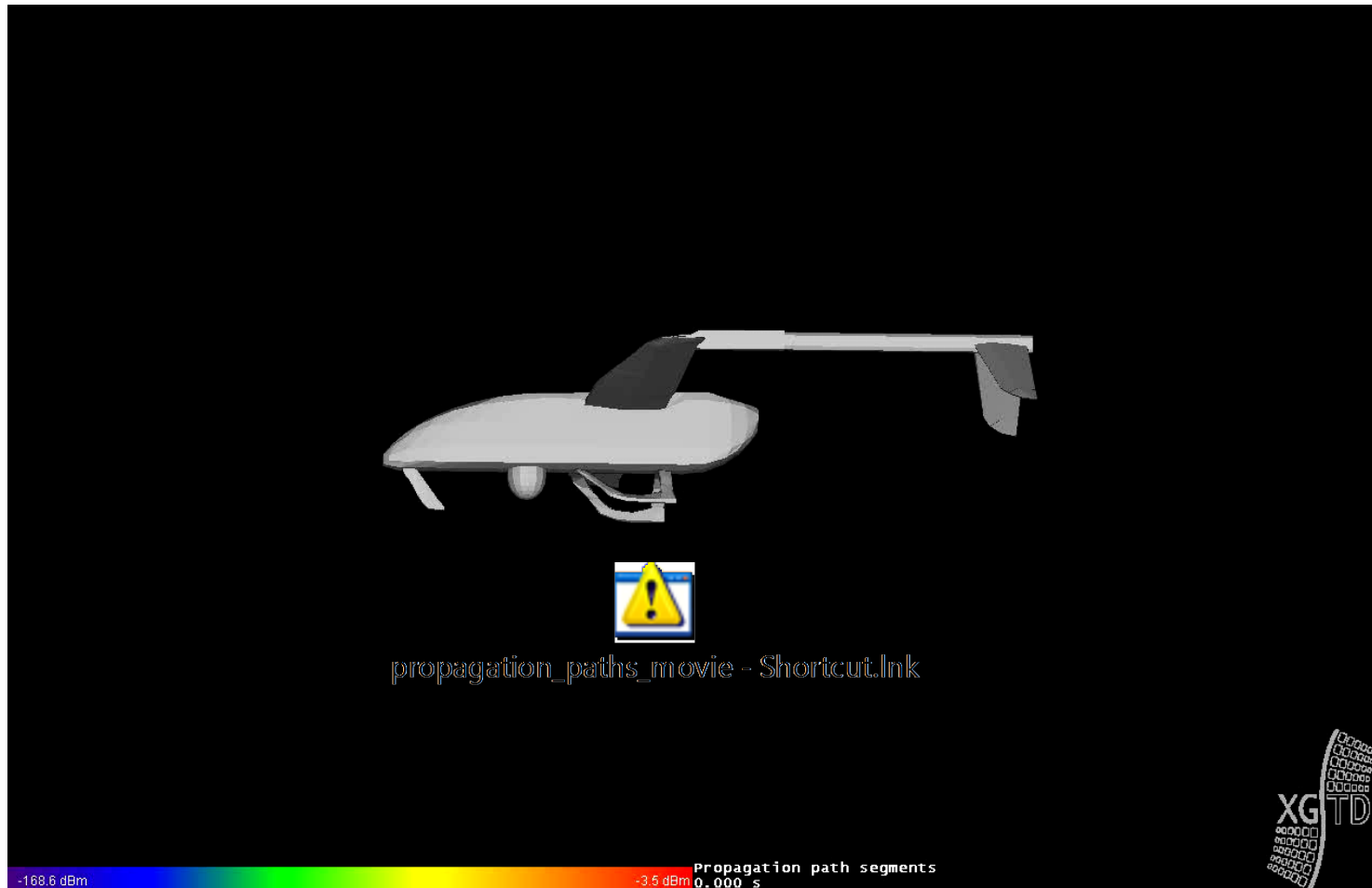
Recommended Antenna Placements



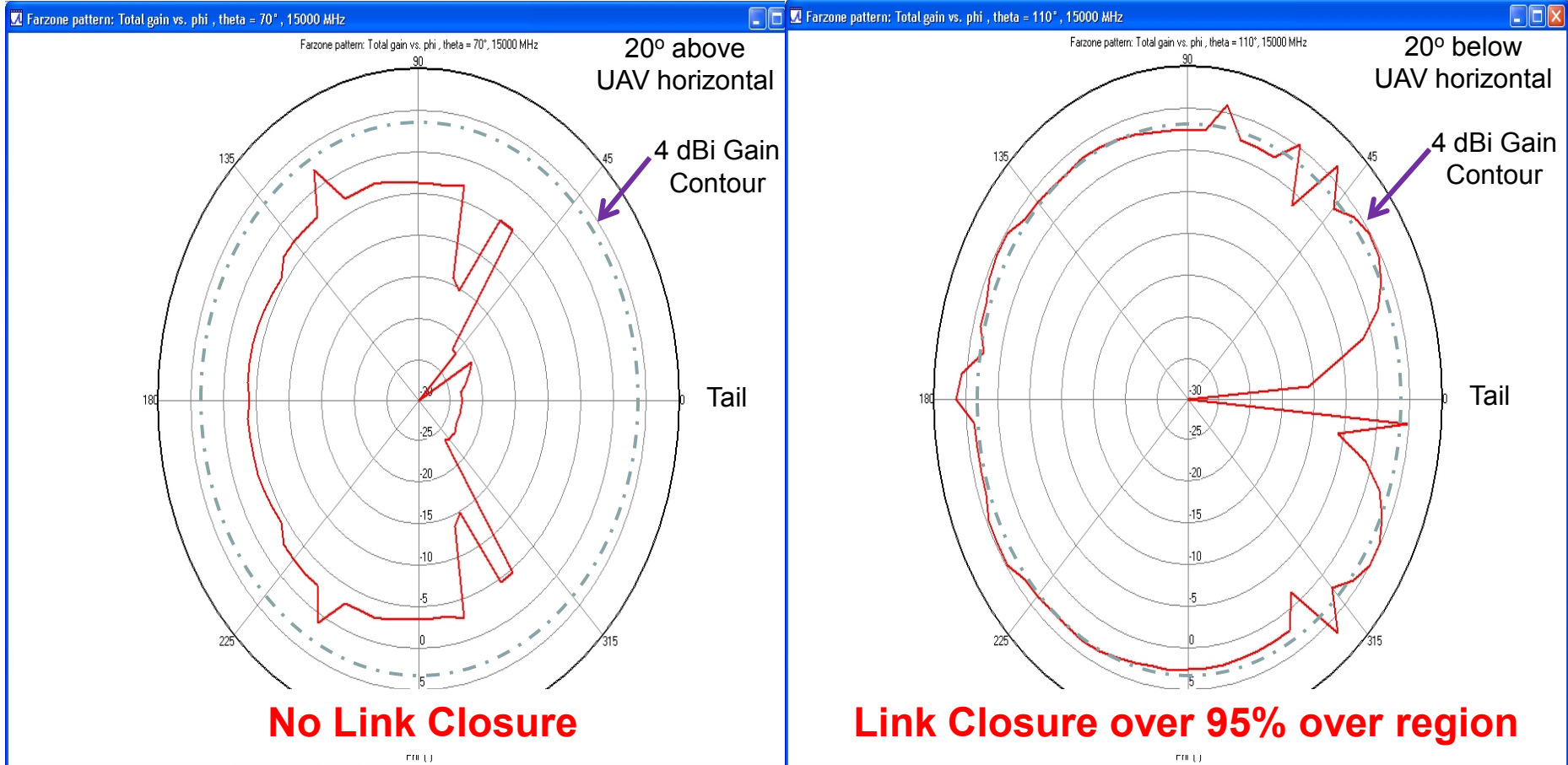
Ku-Band Antenna Placement on Front



Movie Illustrating RF Propagation Paths

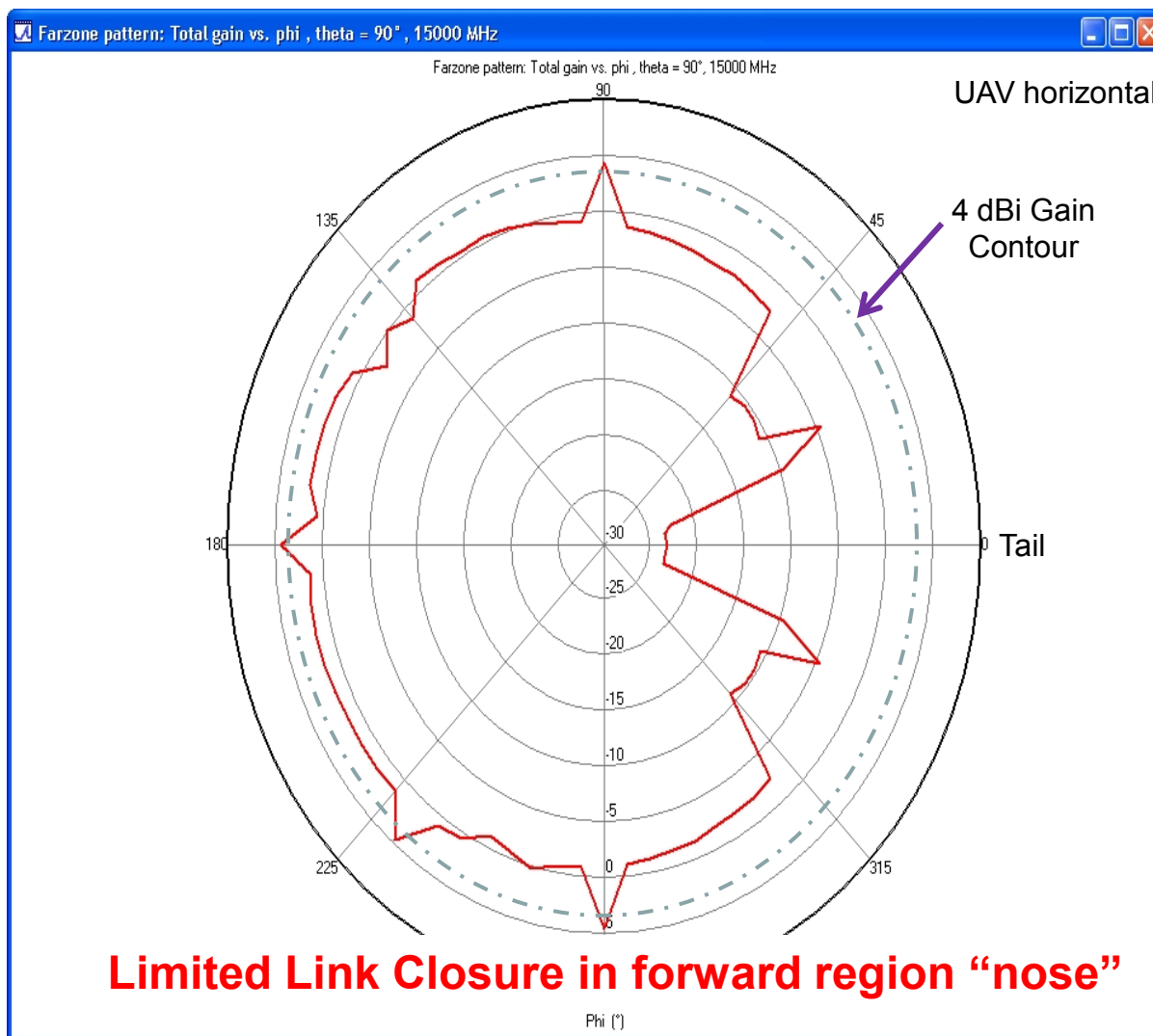


Ku-Band Antenna Placement on Front – Pitch / Roll 20° Cut



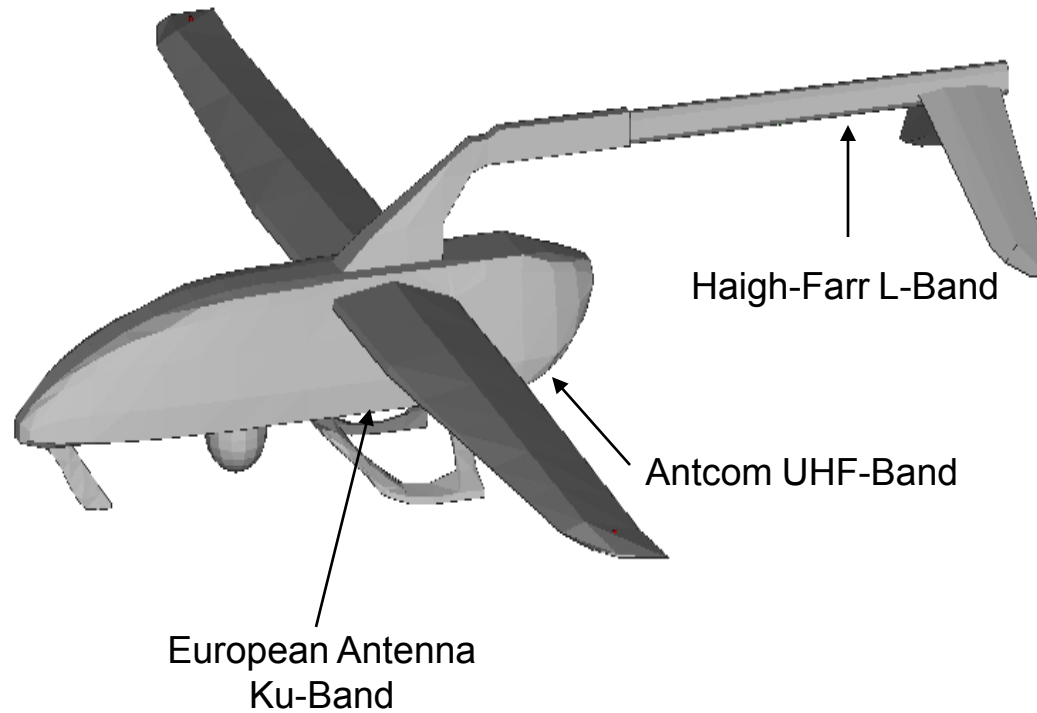
Antenna Gain > 3.4 dBi to close links at 50 nmi and 3 kft (95% availability, clear environmental conditions)

Ku-Band Antenna Placement on Front – Horizontal Cut

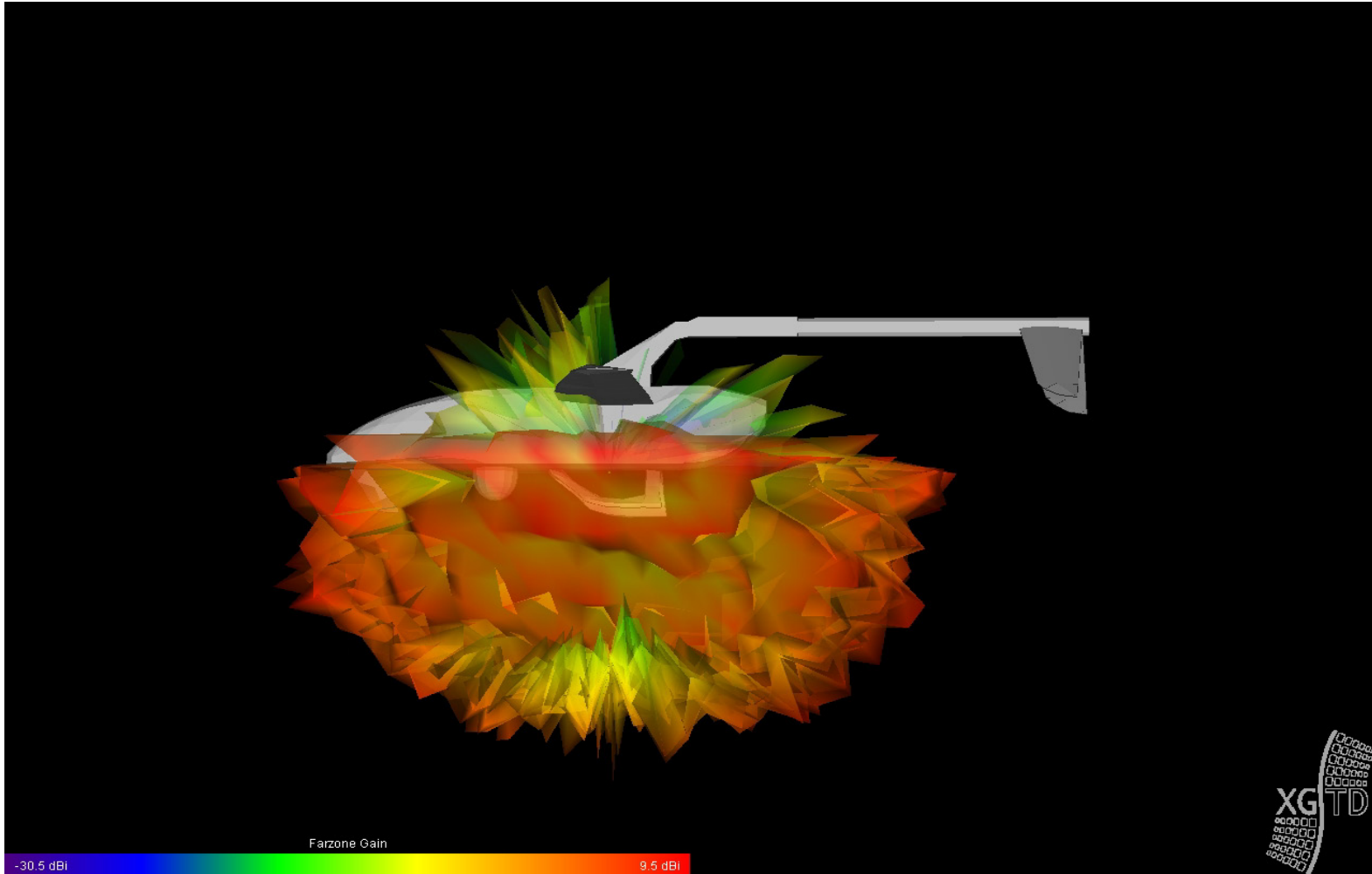


Cautionary Note:
Gain > 4 dBi implies multiple platform reflections occurs which results in greater inter-symbol interference

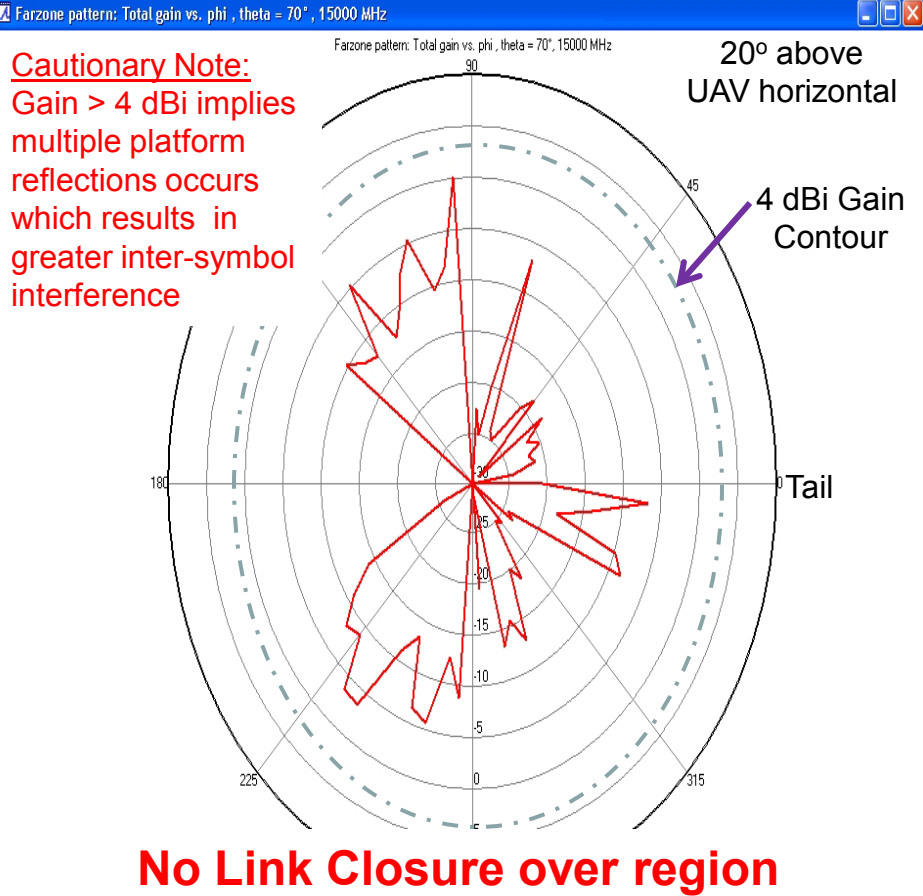
Alternative Ku-Band Antenna Placement Between the Skids



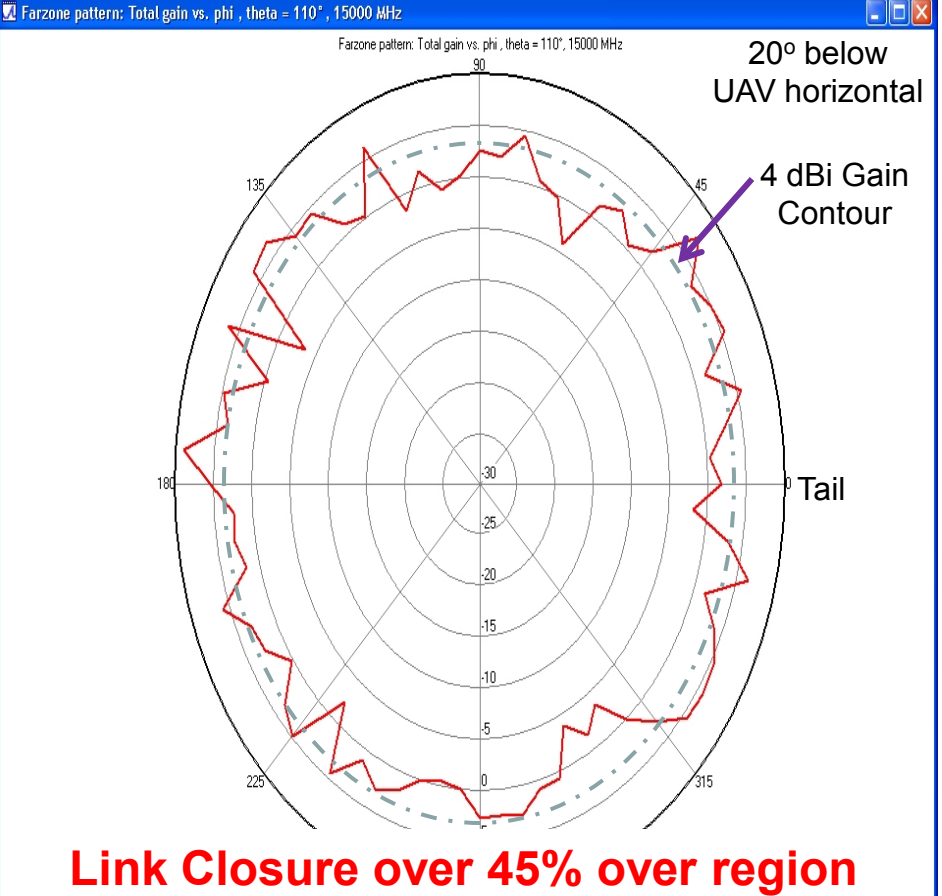
Ku-Band Antenna Placement Between Skids



Ku-Band Antenna Placement Between Skids – Pitch / Roll 20° Cut

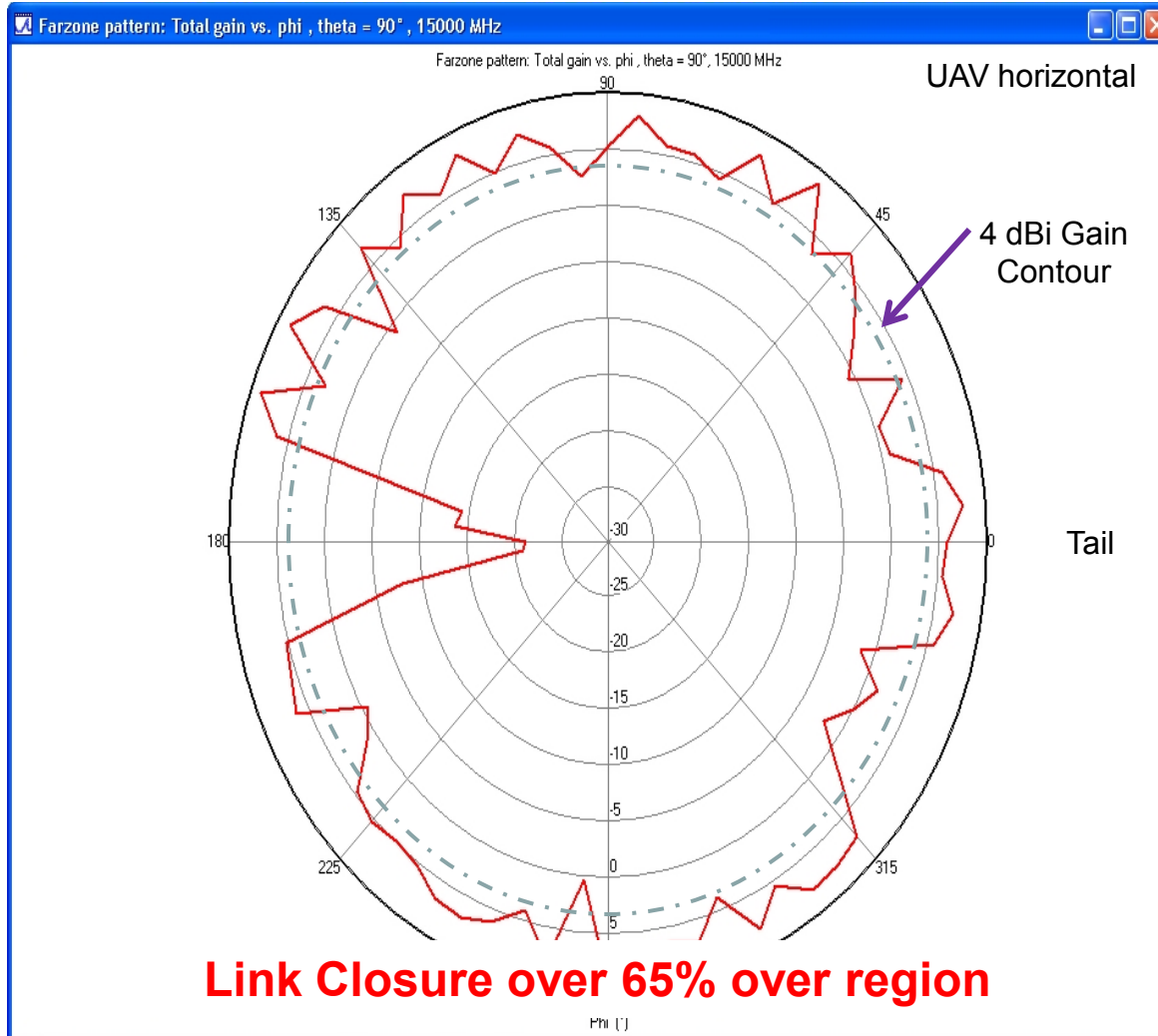


Cautionary Note:
Gain > 4 dBi implies multiple platform reflections occurs which results in greater inter-symbol interference



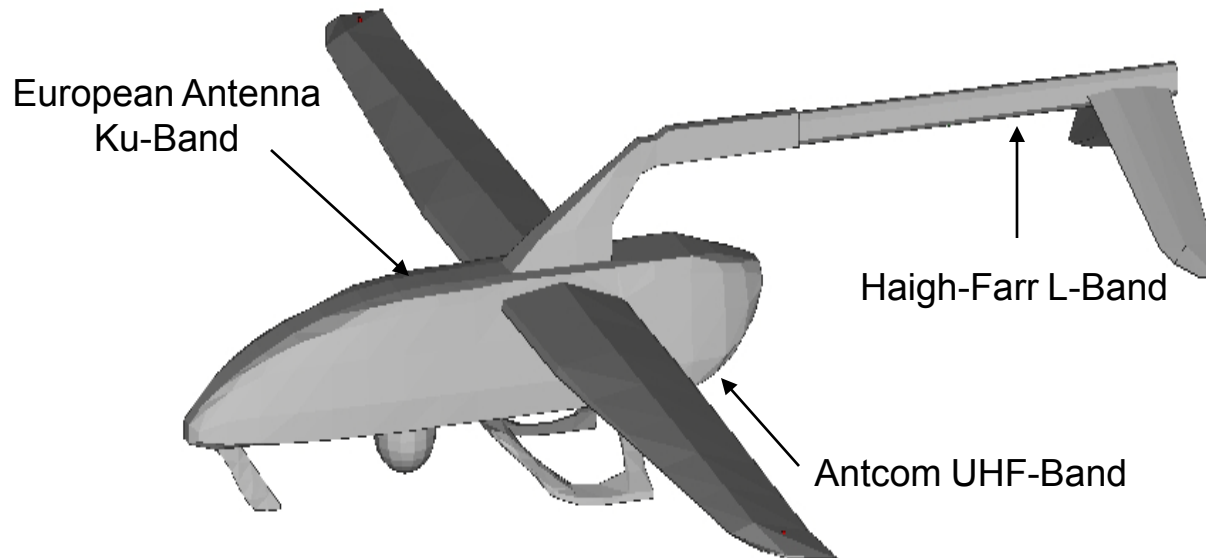
Antenna Gain > 3.4 dBi to close links at 50 nmi and 3 kft (95% availability, clear environmental conditions)

Ku-Band Antenna Placement Between Skids – Horizontal Cut

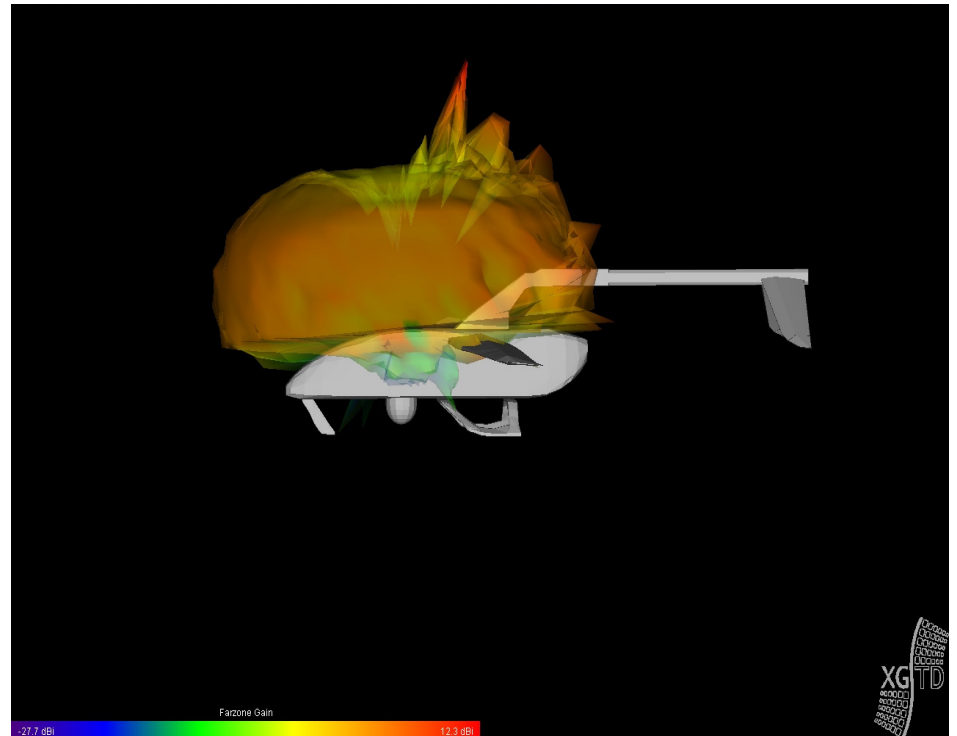
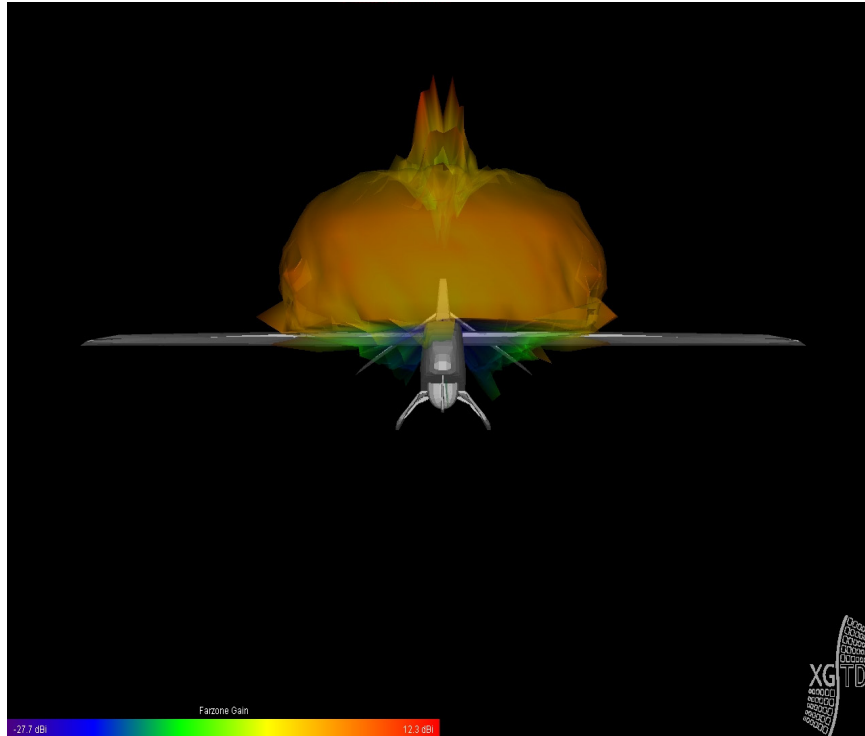


Cautionary Note:
Gain > 4 dBi implies multiple platform reflections occurs which results in greater inter-symbol interference

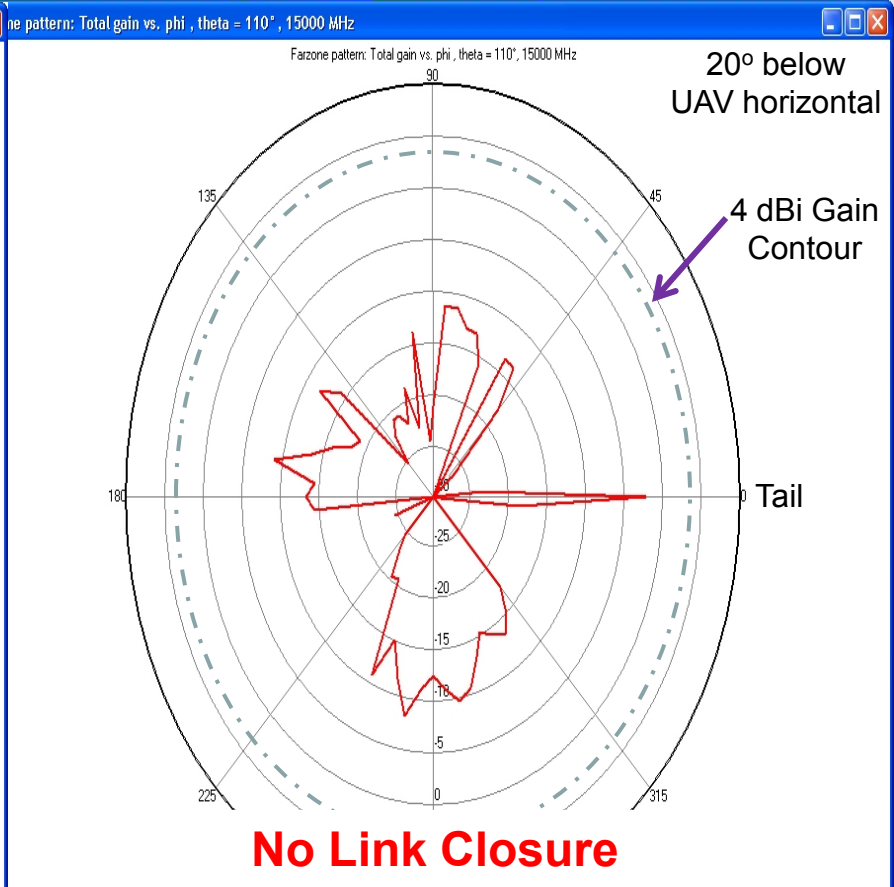
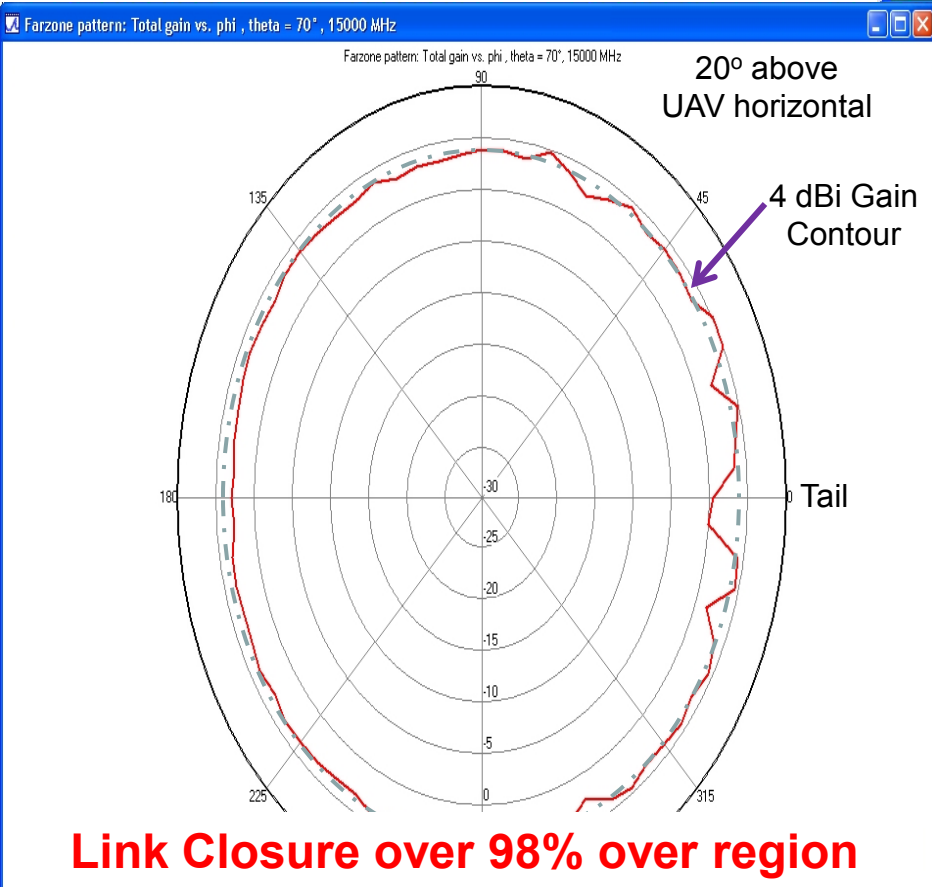
Alternative Ku-Band Antenna Placement on Top of Aircraft Fuselage



Ku-Band Antenna Placement on Top of Aircraft Fuselage

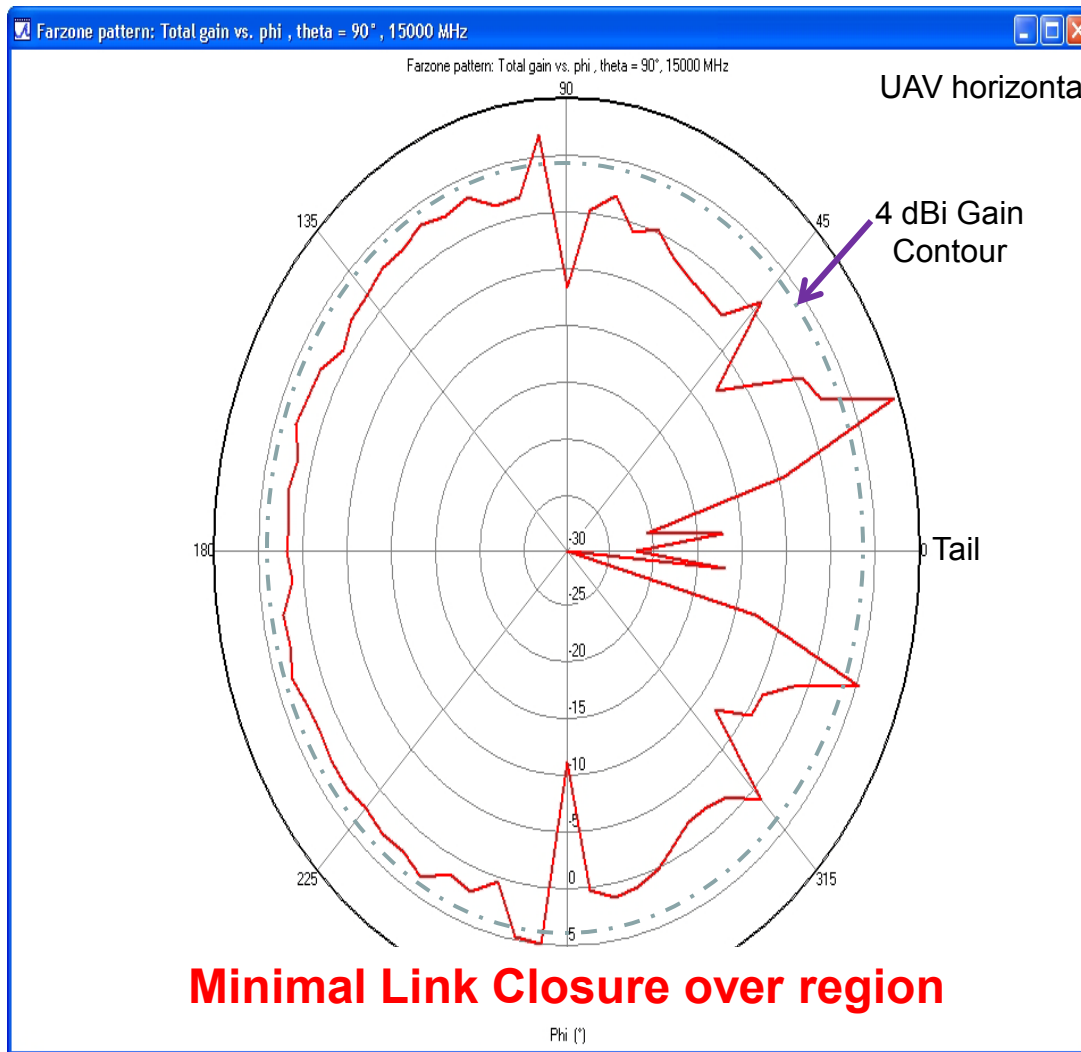


Ku-Band Antenna Placement on Top of Aircraft Fuselage – Pitch / Roll 20° Cut



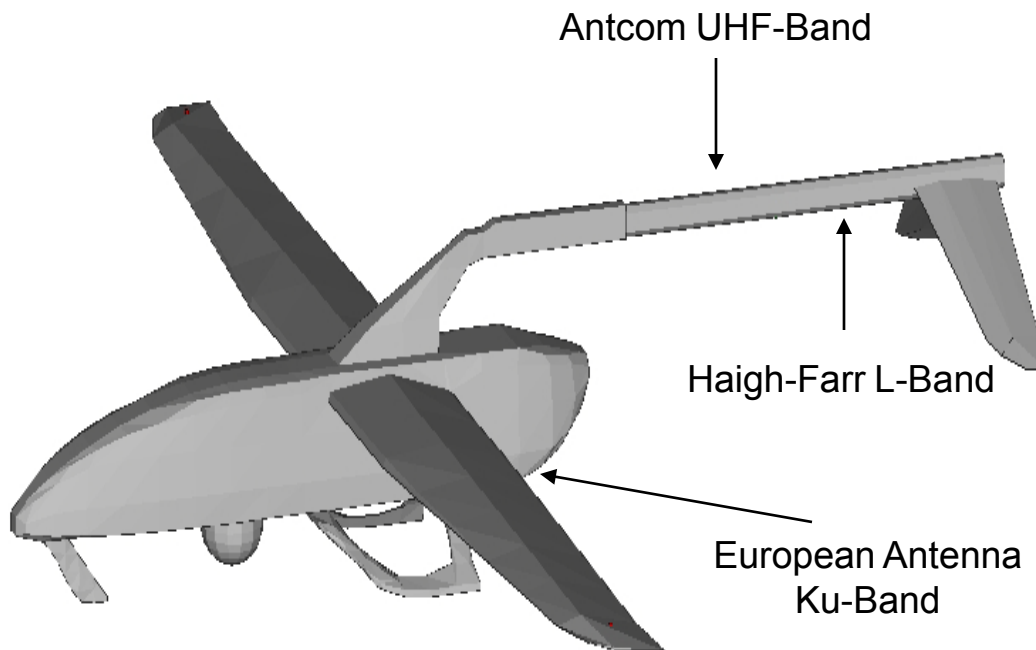
Antenna Gain > 3.4 dBi to close links at 50 nmi and 3 kft (95% availability, clear environmental conditions)

Ku-Band Antenna on Top of Aircraft Fuselage – Horizontal Cut

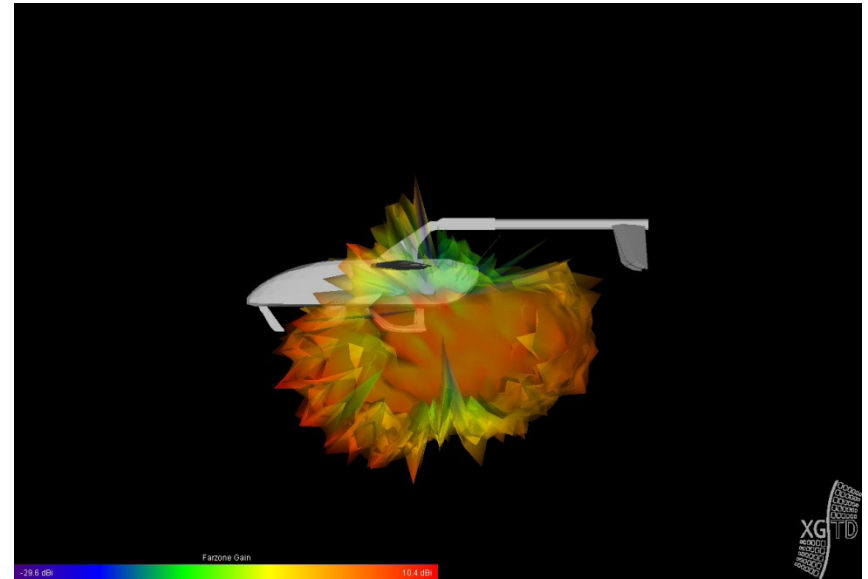
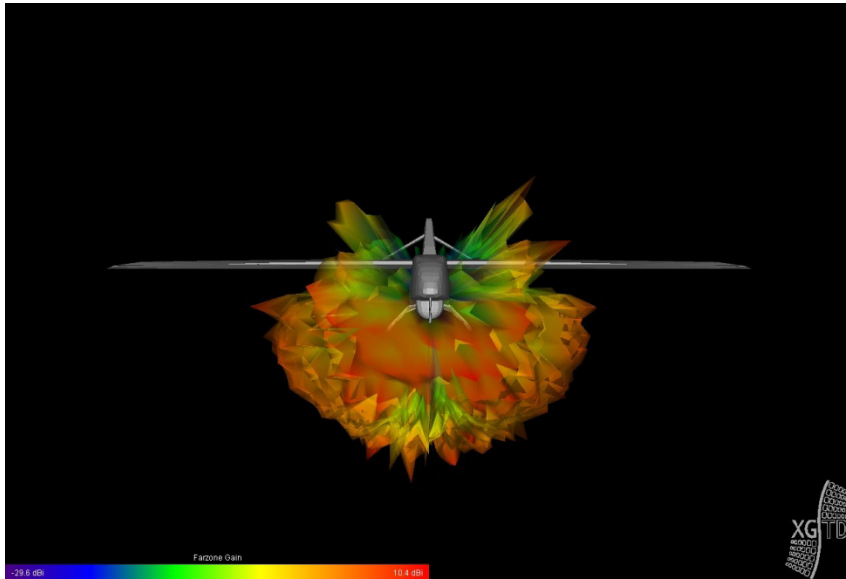


Cautionary Note:
Gain > 4 dBi implies multiple platform reflections occurs which results in greater inter-symbol interference

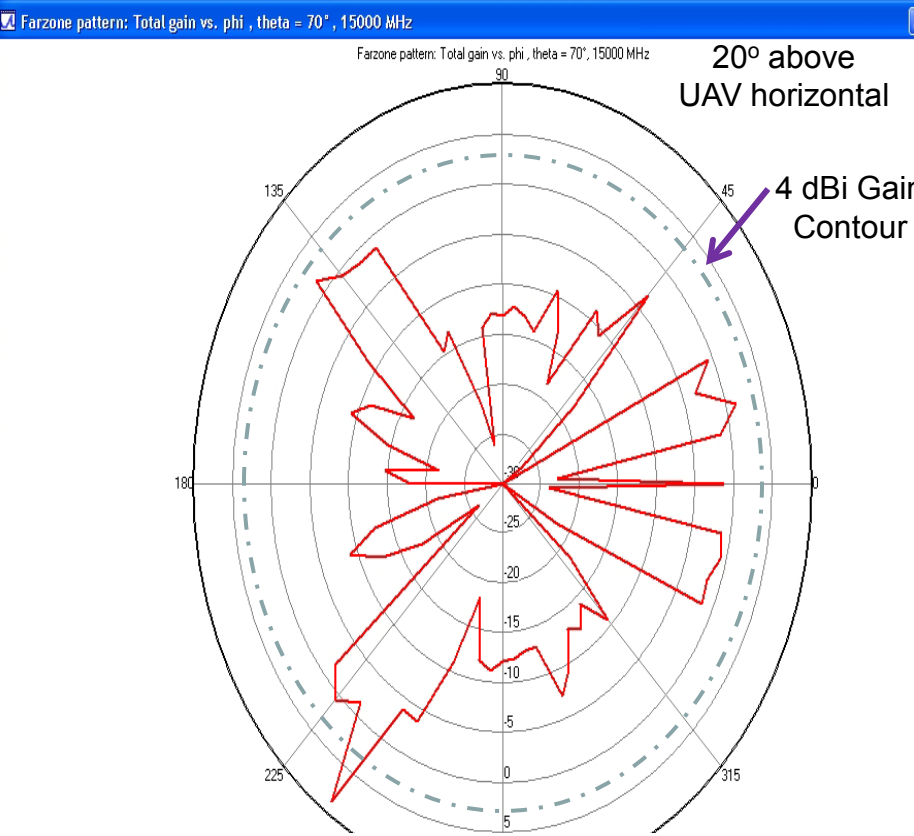
Alternative Antenna Placement Near Aft Fuselage



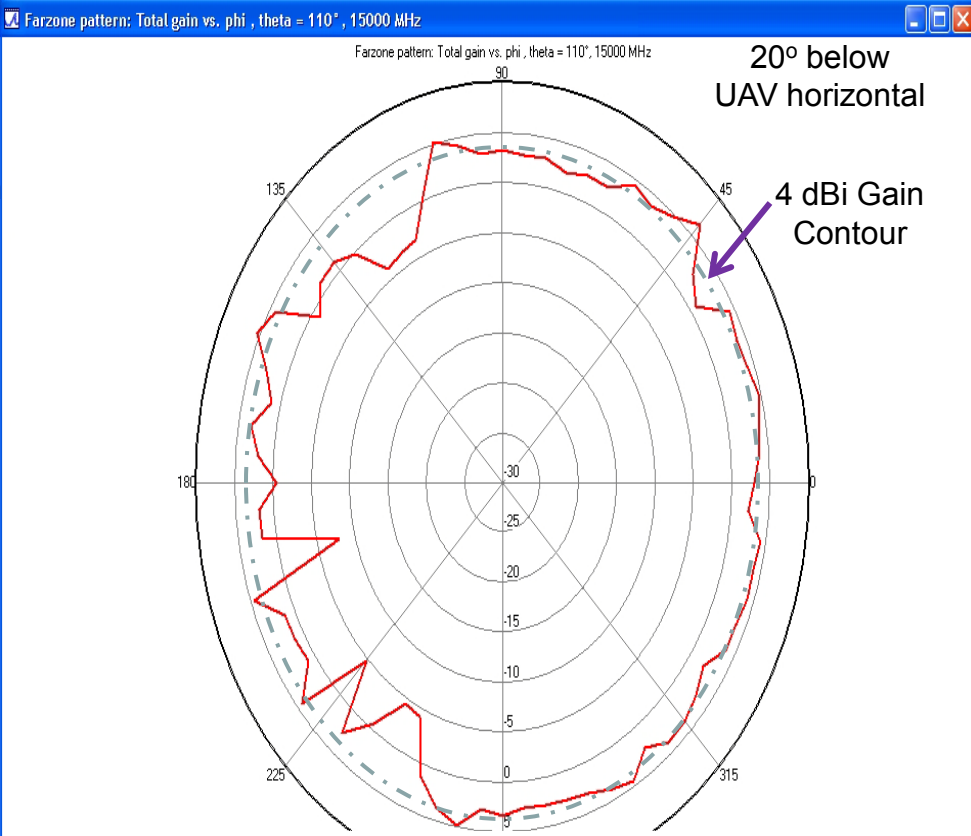
Ku-Band Antenna Placement Aft Fuselage



Ku-Band Antenna Placement Aft Fuselage- Pitch / Roll 20° Cut



Link Closure over 1% over region

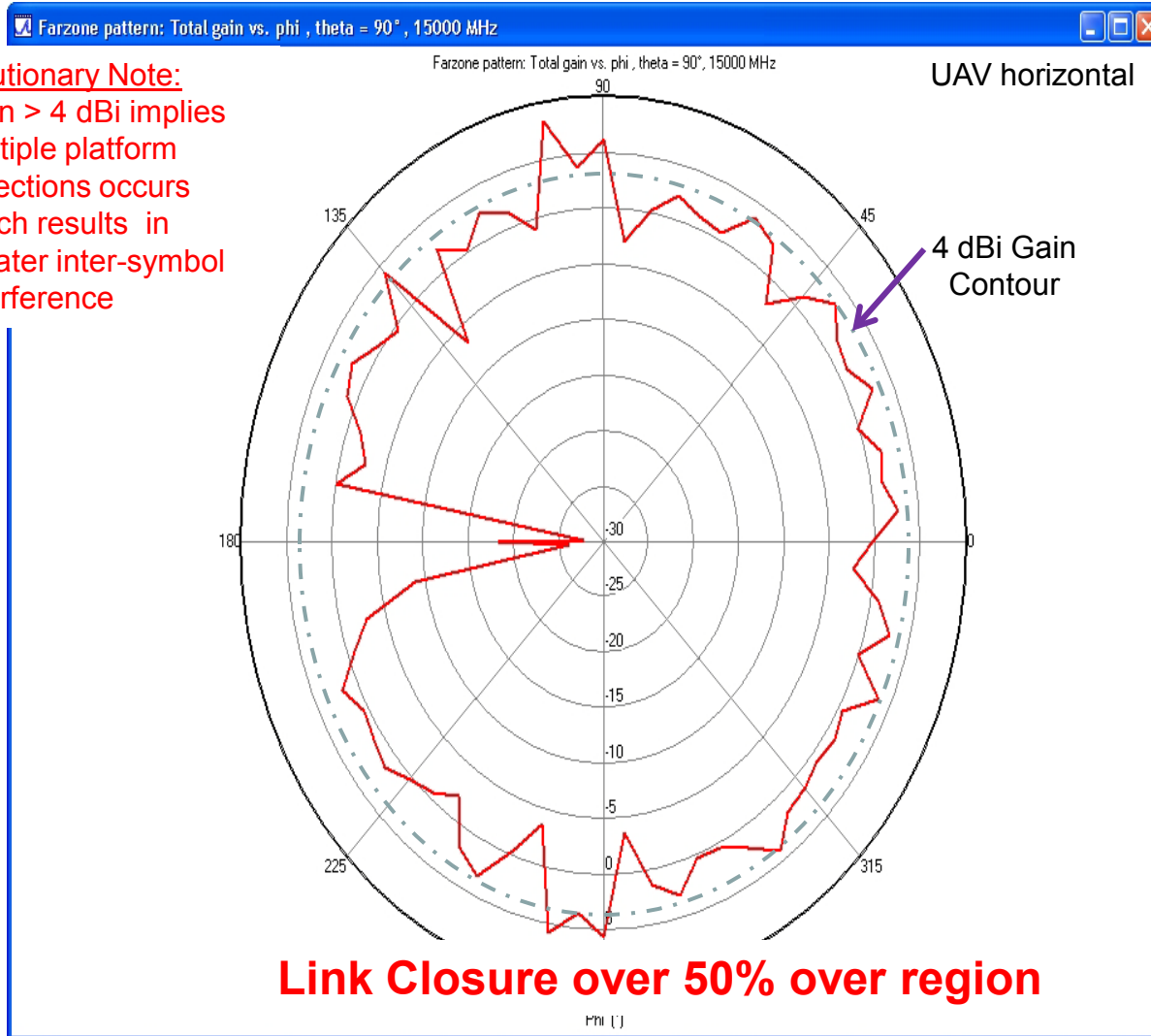


Link Closure over 89% over region

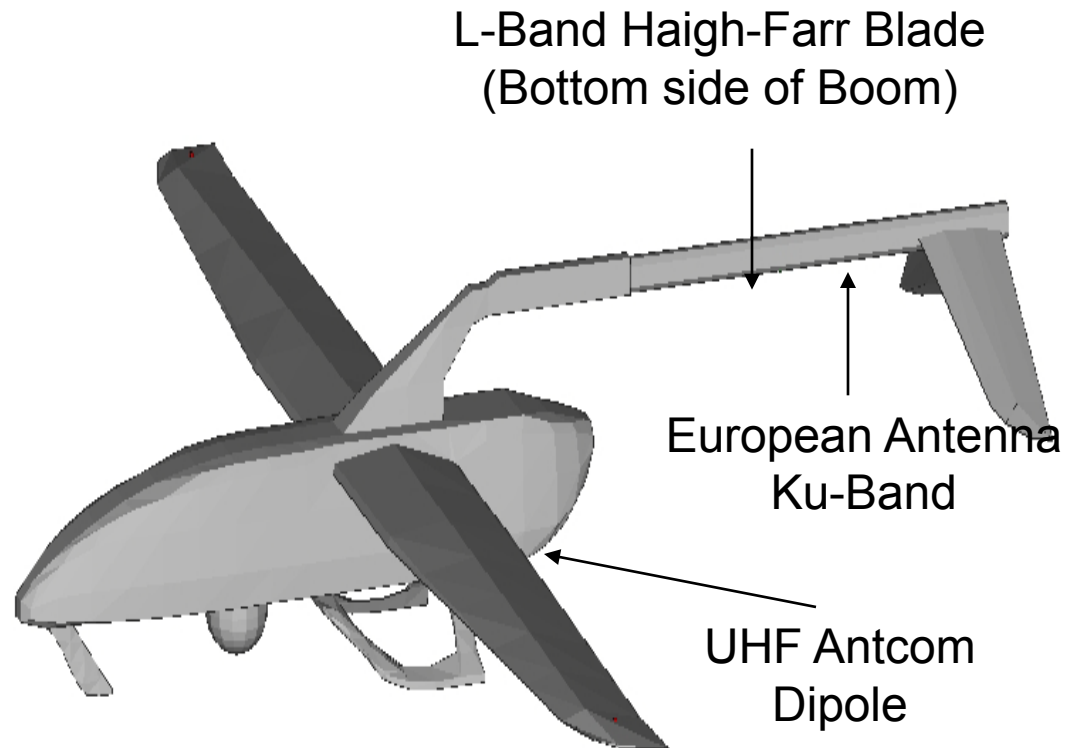
Antenna Gain > 3.4 dBi to close links at 50 nmi and 3 kft (95% availability, clear environmental conditions)

Ku-Band Antenna Placement Aft Fuselage Horizontal Cut

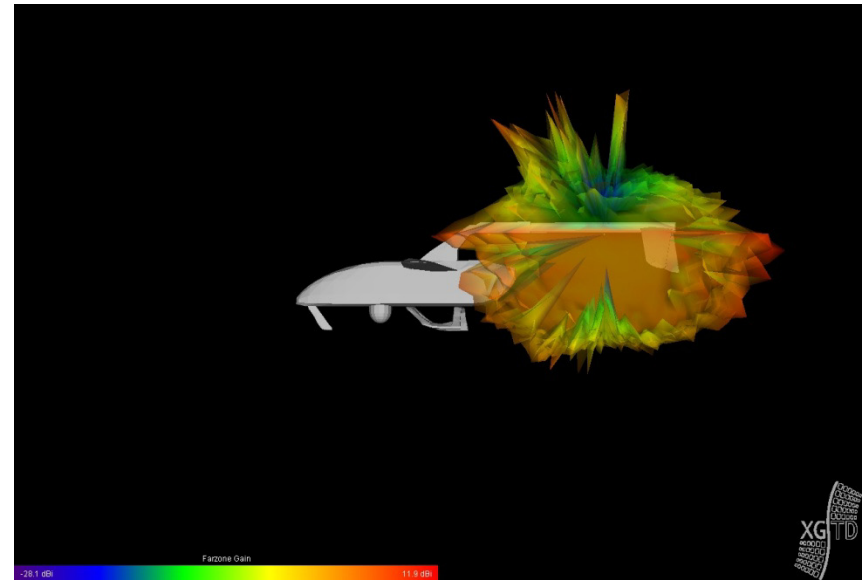
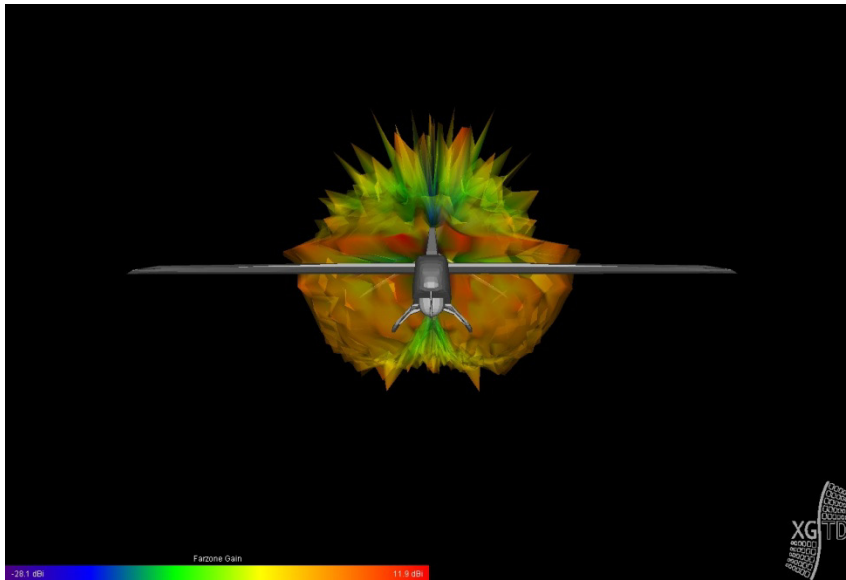
Cautionary Note:
Gain > 4 dBi implies
multiple platform
reflections occurs
which results in
greater inter-symbol
interference



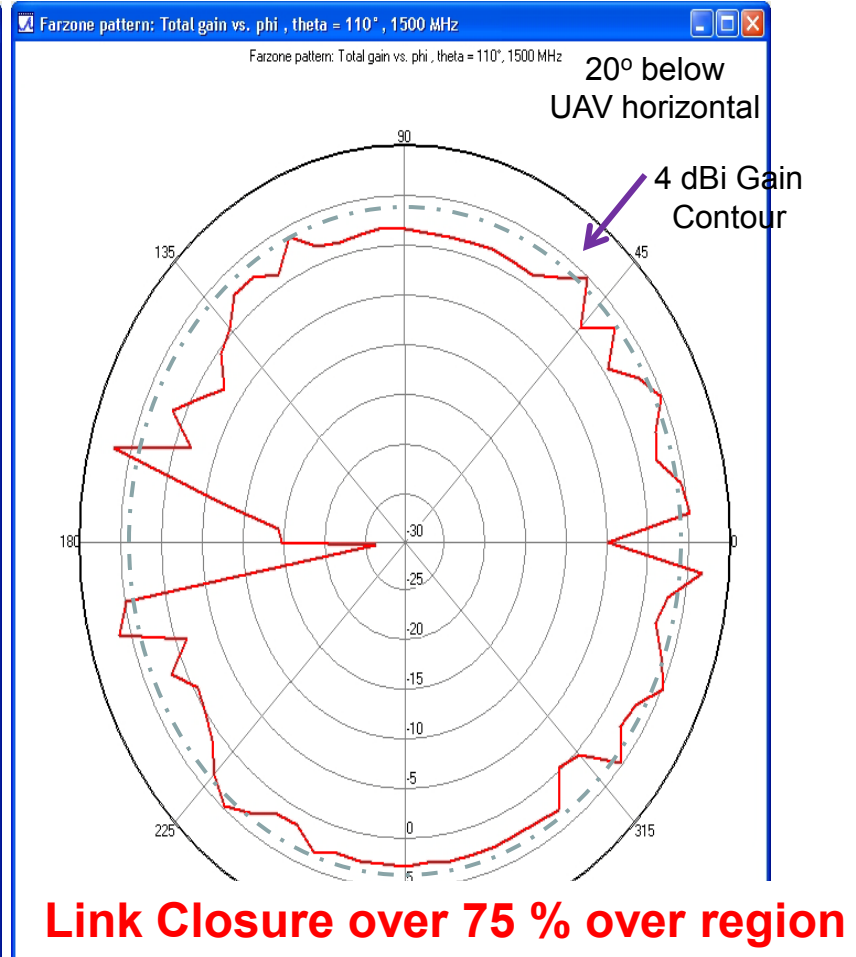
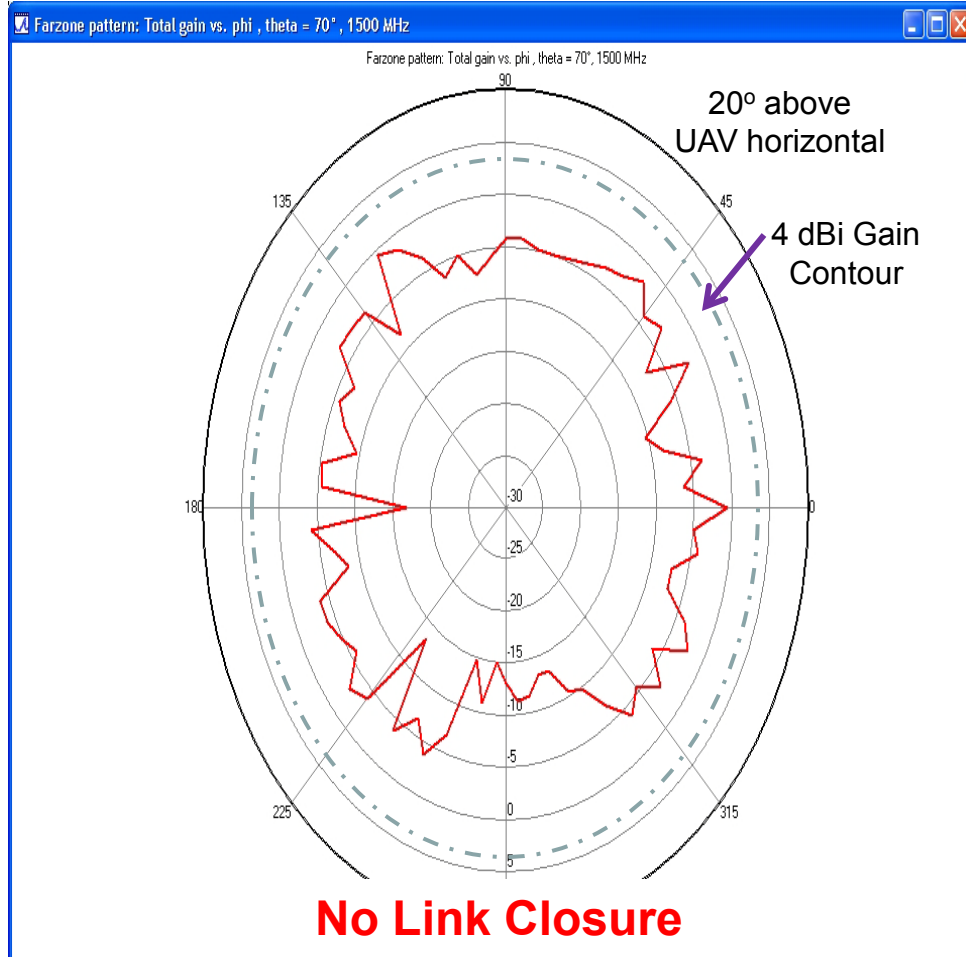
Alternative Antenna Placement Bottom Tail Boom



Ku-Band Antenna Placement Bottom Tail Boom



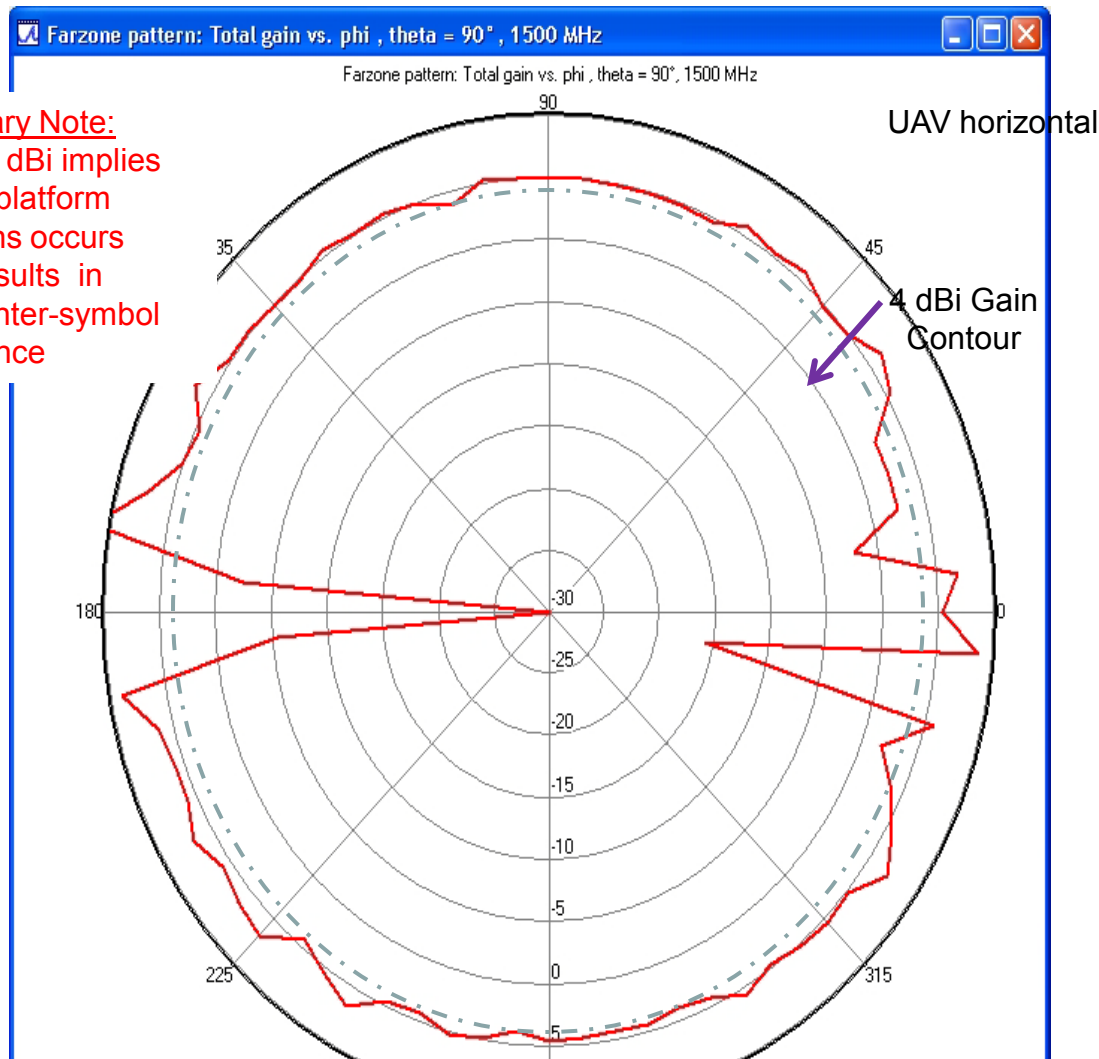
Ku-Band Antenna Placement Bottom Tail Boom - Pitch / Roll 20° Cut



Antenna Gain > 3.4 dBi to close links at 50 nmi and 3 kft (95% availability, clear environmental conditions)

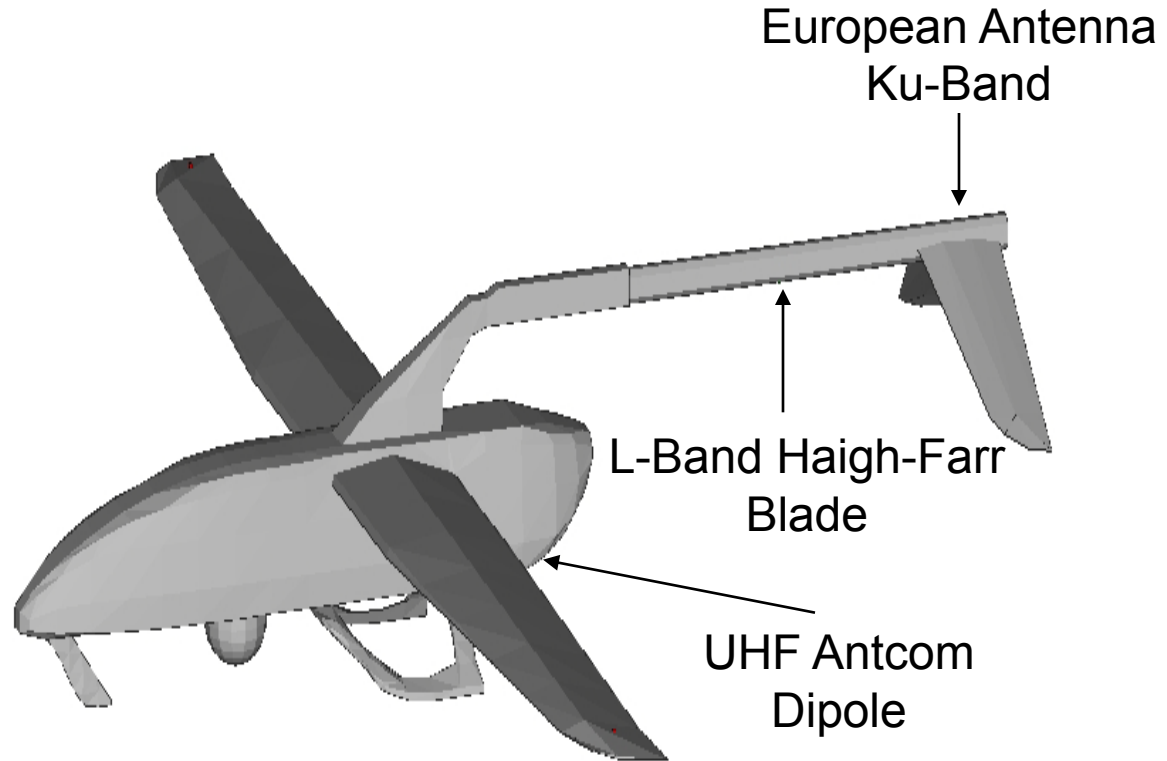
Ku-Band Antenna Placement Bottom Tail Boom– Horizontal Cut

Cautionary Note:
Gain > 4 dBi implies
multiple platform
reflections occurs
which results in
greater inter-symbol
interference



Link Closure over 80% over region (See Cautionary Note)

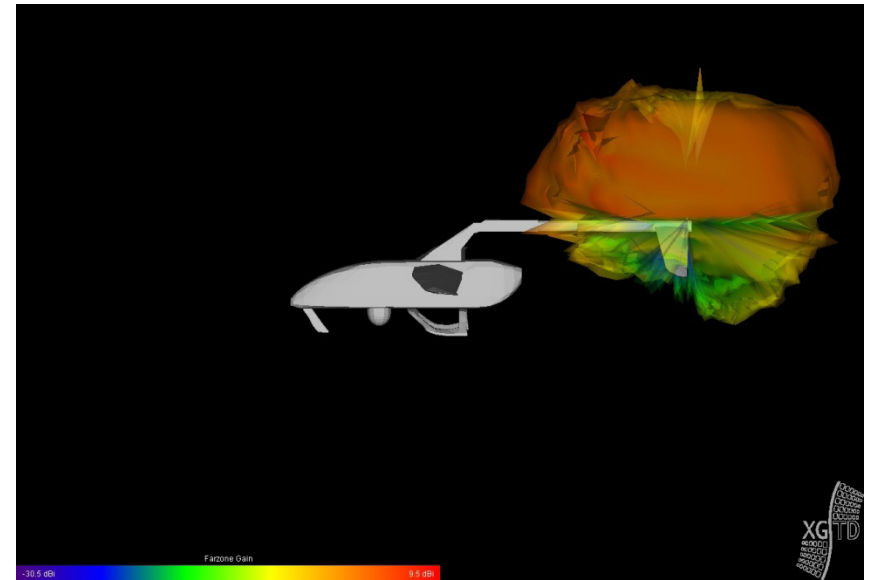
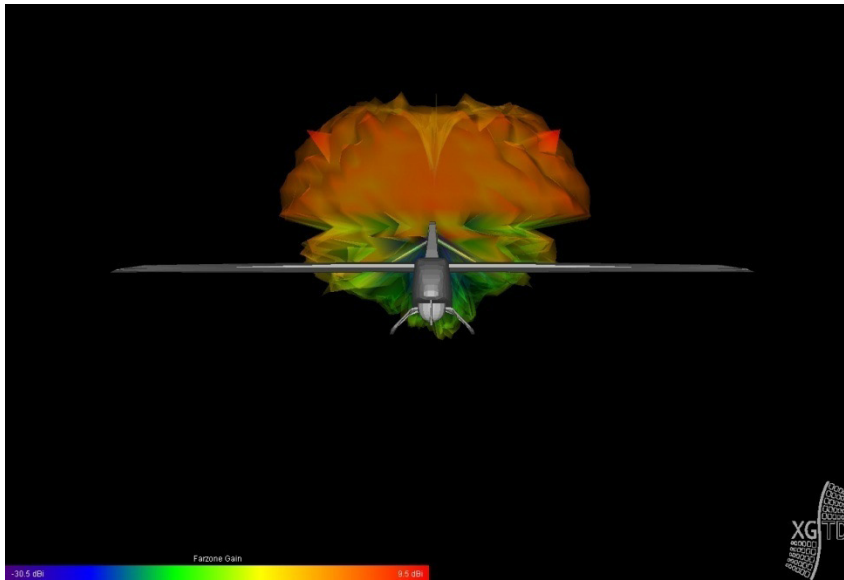
Antenna Placement Top Tail Boom (1.5 inches from end)



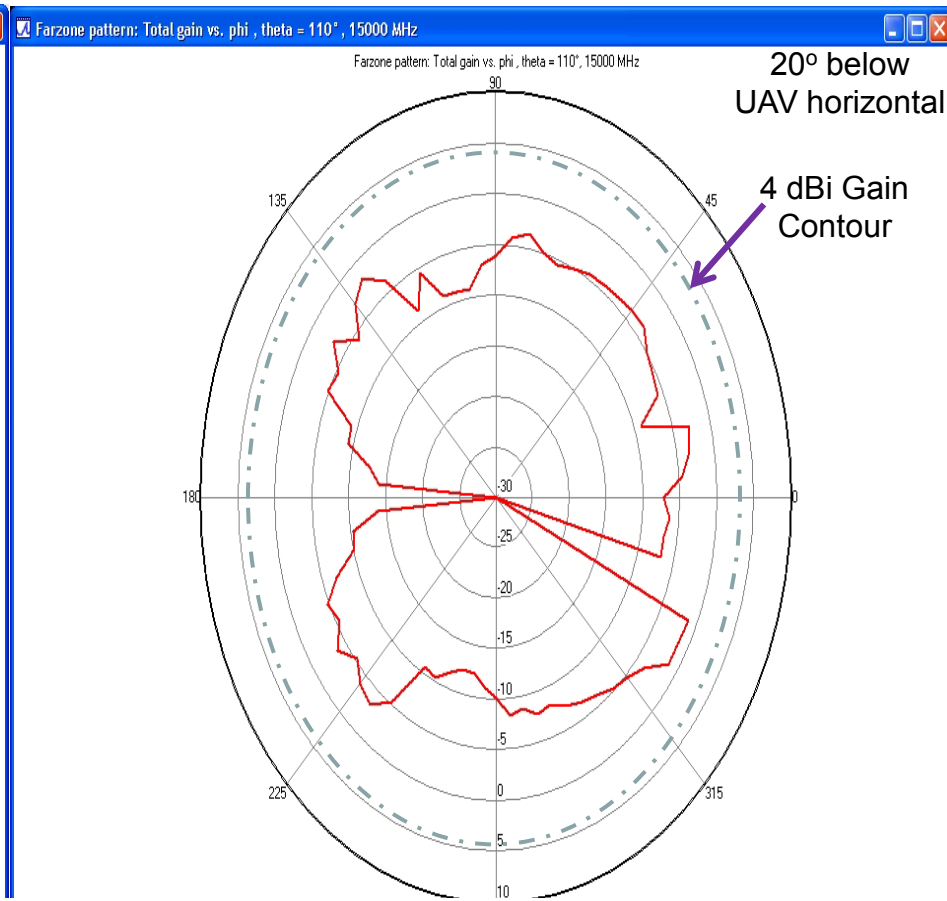
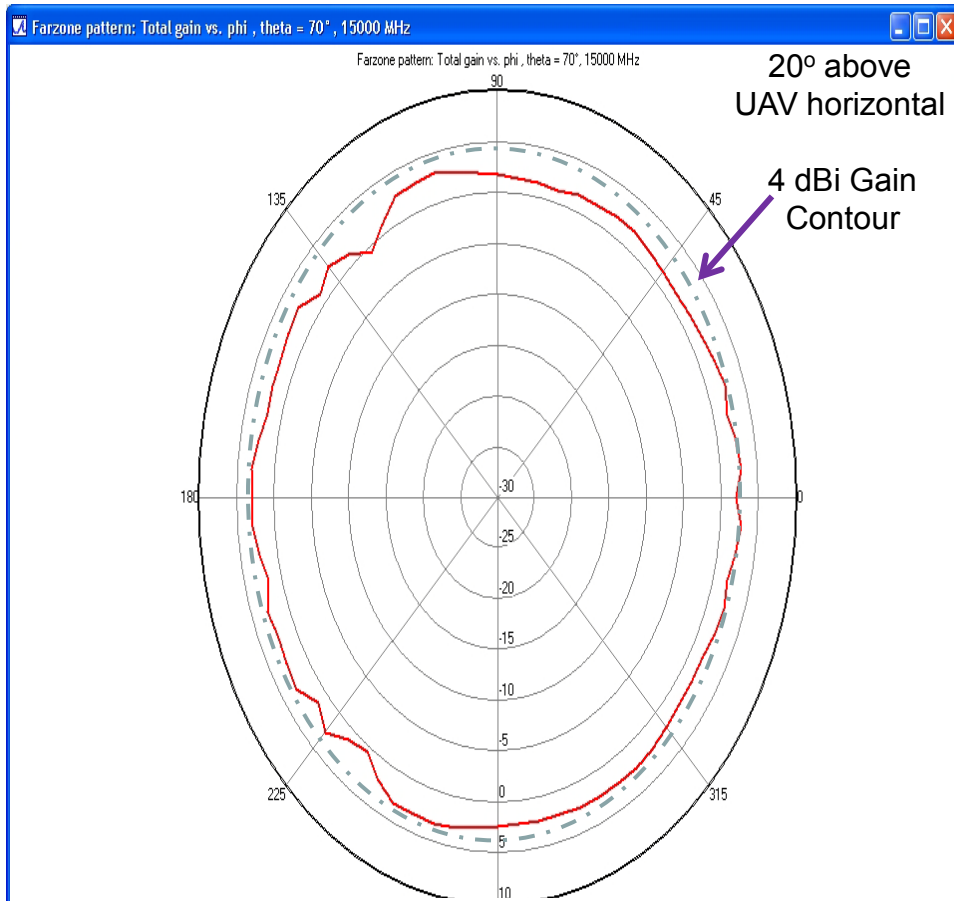
Movie Illustrating RF Propagation Paths and Significant Multipath Reflections from Tail Boom Design



Ku-Band Antenna Placement Top Tail Boom (1.5 inches from end)



Ku-Band Antenna Placement Top Tail Boom (1.5 inches from end)- Pitch / Roll 20° Cut

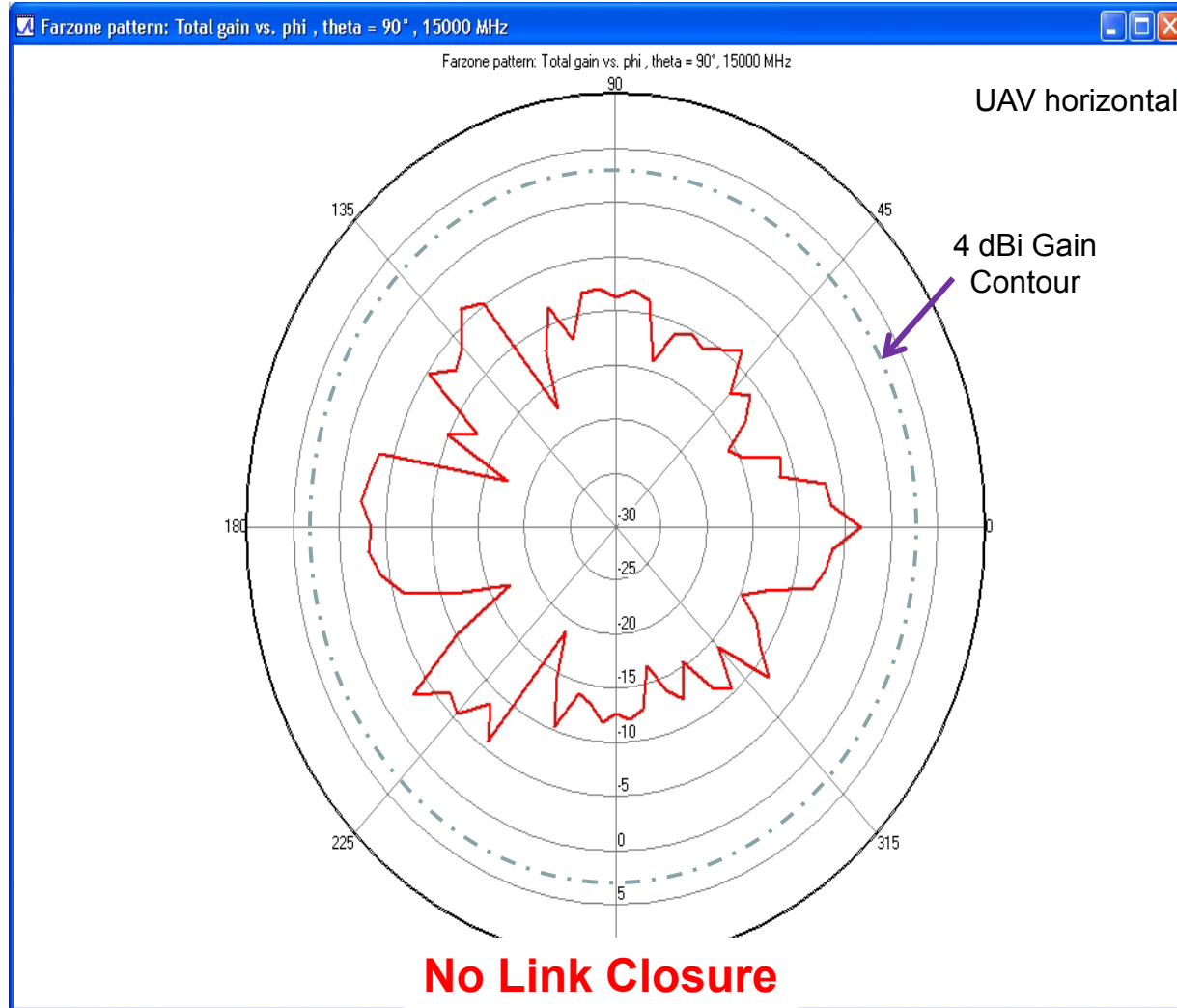


Link Closure less than 95% over region

No Link Closure

Antenna Gain > 3.4 dBi to close links at 50 nmi and 3 kft (95% availability, clear environmental conditions)

Ku-Band Antenna Placement Top Tail Boom (1.5 inches from end)- Horizontal Cut



Isolation Results Between Ku-Band Transmitter and UHF & L-Band Receivers

Ku-Band Antenna Placement on nose of aircraft	
Ku-Band Transmitter Received Power (dBm) at victim receivers	
UHF Antenna Receiver	-250
L Band Antenna Receiver	-250
Ku-Band Antenna Placement in between skids	
Ku-Band Transmitter Received Power (dBm) at victim receivers	
UHF Antenna Receiver	-24.8
L Band Antenna Receiver	-250
Ku-Band Antenna Placement on top of aircraft	
Ku-Band Transmitter Received Power (dBm) at victim receivers	
UHF Antenna Receiver	-63.55
L Band Antenna Receiver	-28.14
Ku-Band Antenna Placement on bottom of aircraft near aft fuselage	
Ku-Band Transmitter Received Power (dBm) at victim receivers	
UHF Antenna Receiver (top of boom)	-250
L Band Antenna Receiver (bottom of boom)	-50.06

Isolation Results Between Ku-Band Transmitter and UHF & L-Band Receivers

Ku-Band Antenna Placement on bottom of tail boom 12" from tail	
Ku-Band Transmitter Received Power (dBm) at victim receivers	
UHF Antenna Receiver (top of boom)	1.74
L Band Antenna Receiver (bottom of boom)	7.68
Ku-Band Antenna Placement on top of tail boom 1.5" from tail	
Ku-Band Transmitter Received Power (dBm) at victim receivers	
UHF Antenna Receiver (top of boom)	-22.82
L Band Antenna Receiver (bottom of boom)	-10.86



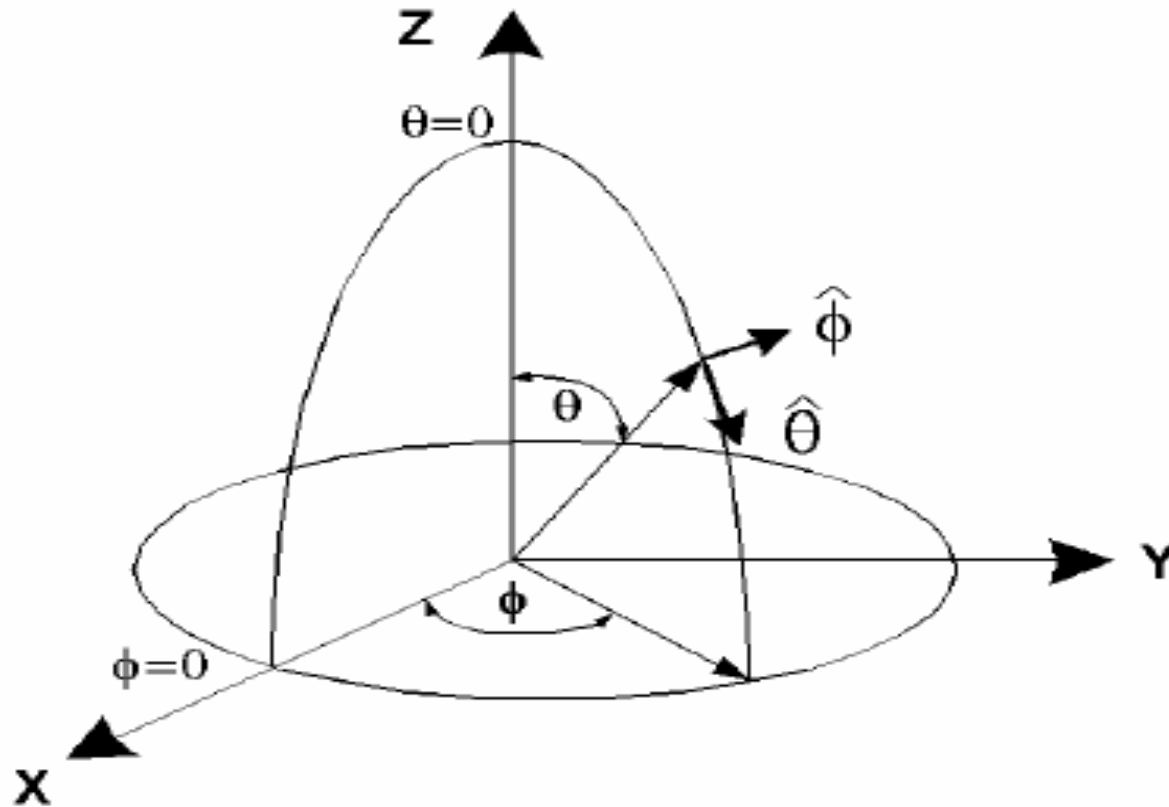
Coordinate System Reference Diagrams

Advanced Tech Engineering, Inc

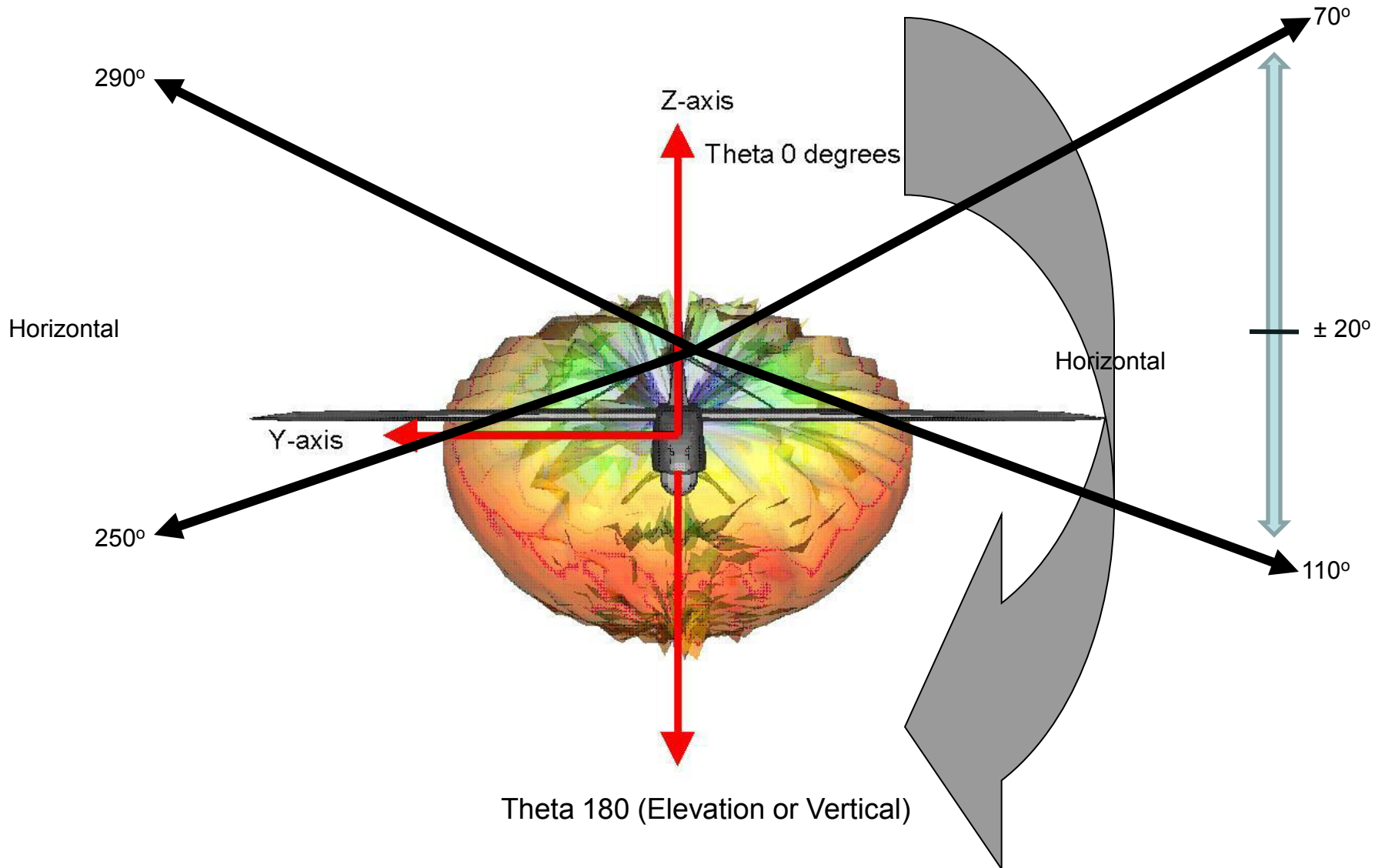
This document contains financial, business, scientific, technical or engineering information. Disclosure to others, use, or copying, without the prior written authorization of Advanced Tech Engineering, Inc. is strictly prohibited.

UNCLASSIFIED

Coordinate Frame Reference

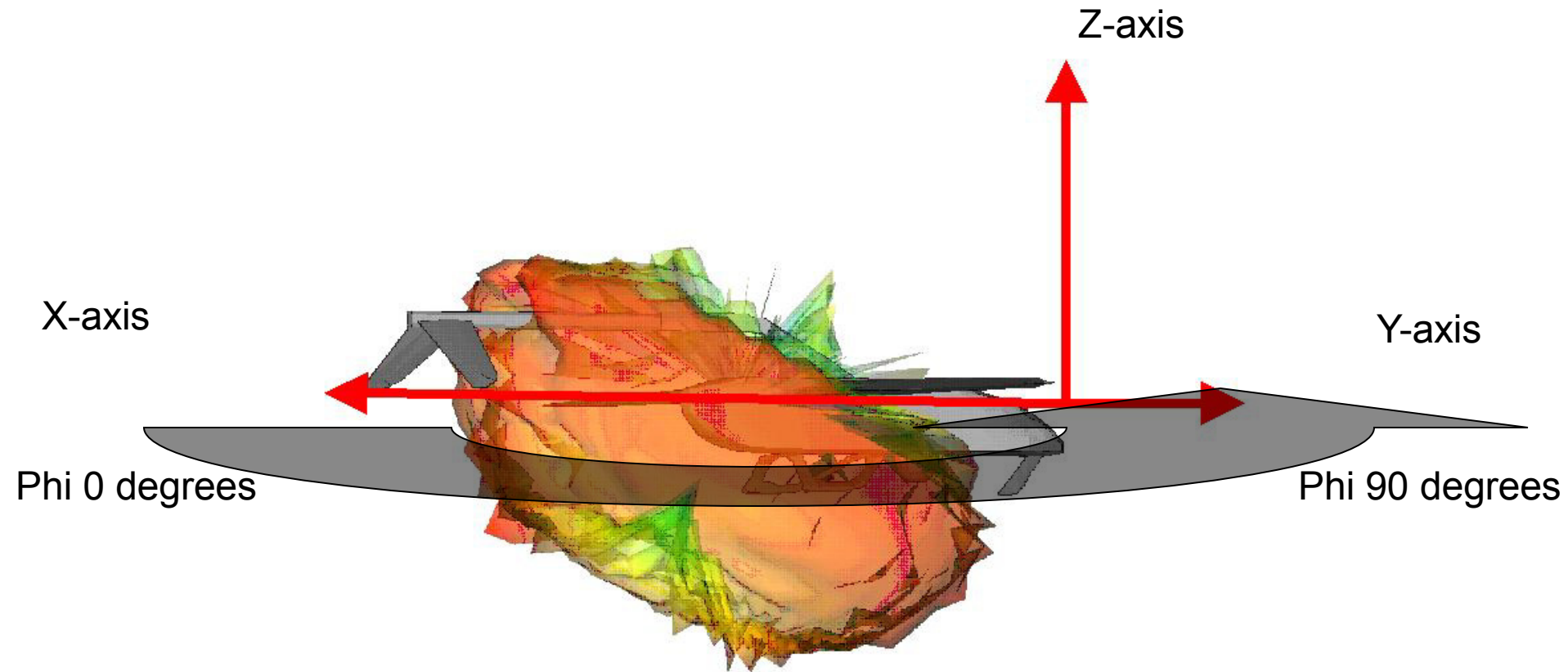


Reference Diagram

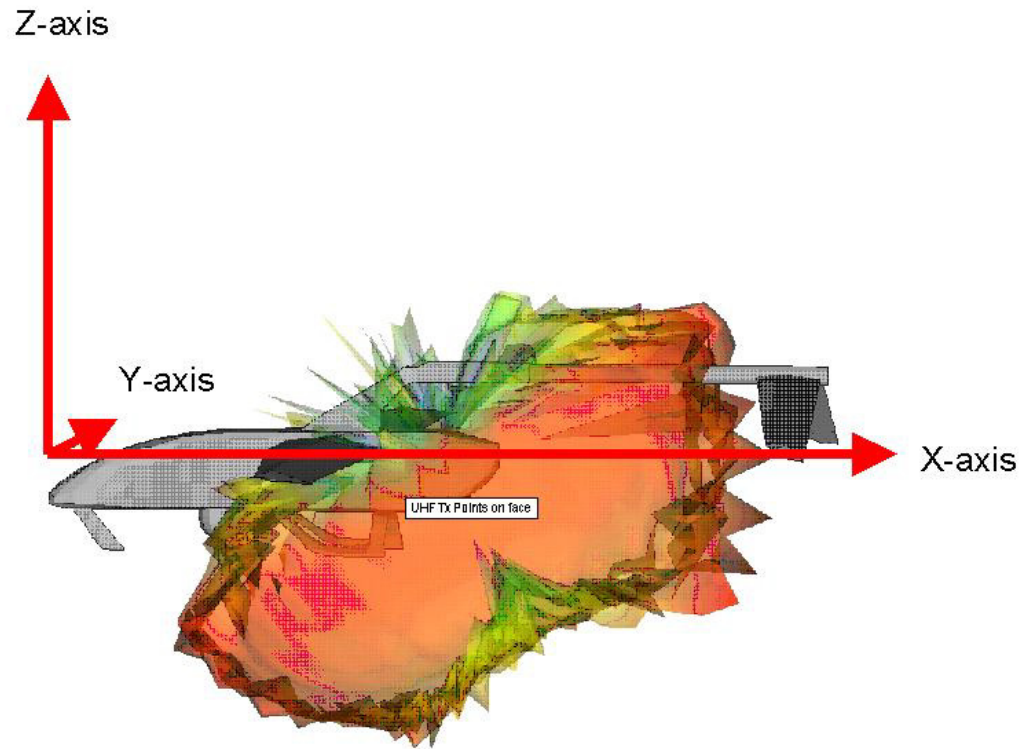


Theta 180 (Elevation or Vertical)

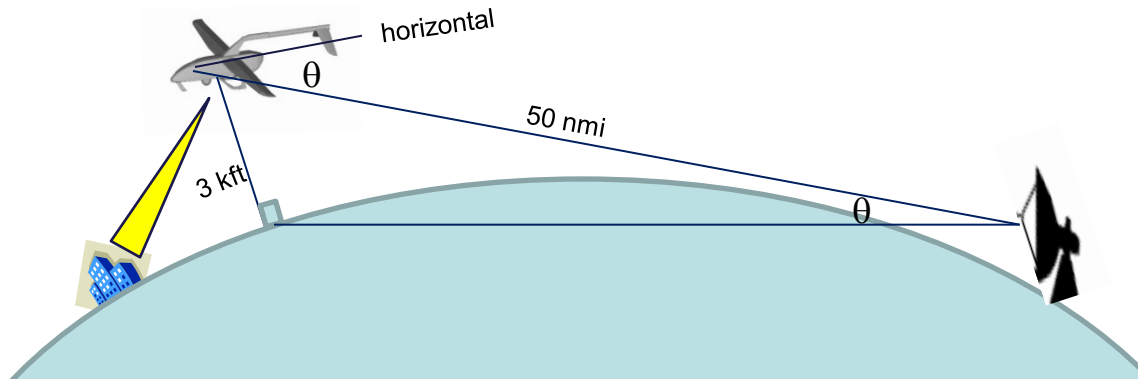
Reference Diagram



Reference Diagram



UAV to GDT Geometry @ 50 nmi



Finding θ

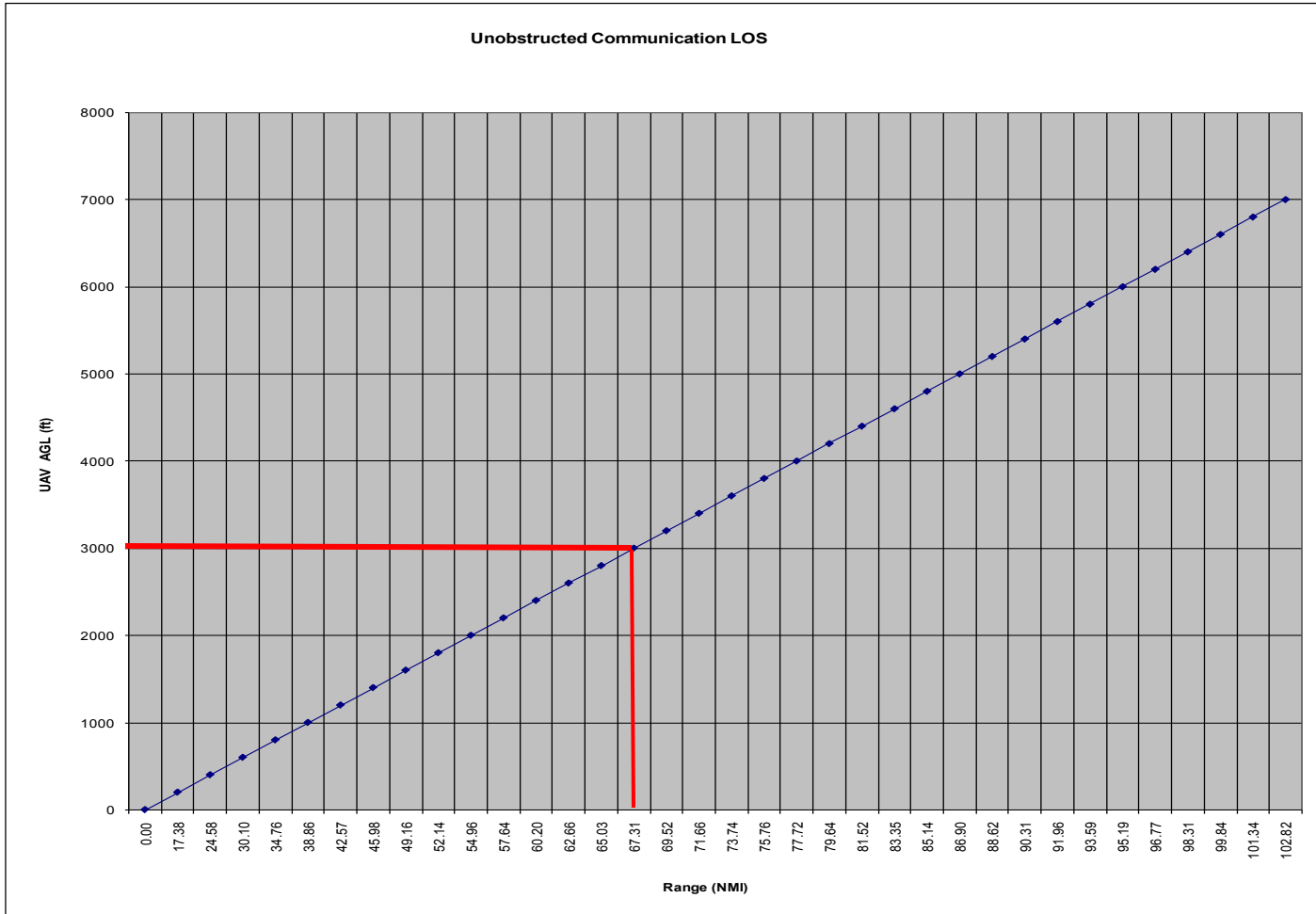
$\sin \theta = \text{Opposite} / \text{hypotenuse}$

$$\begin{aligned}\theta &= \text{Arcsin} (3000 \text{ ft} / 6080 \text{ ft} / \text{nmi}) / 50 \text{ nmi} \\ &= 0.57 \text{ degrees}\end{aligned}$$

Aircraft level flight (e.g., no roll or pitch) @ 50 nmi slant range and 3 kft Altitude AGL
= - 0.57° below horizontal axis

Aircraft normal flight (assumes 20° max roll angle during orbits or flight conditions)
= +19.43° (above horizontal axis) to 20.57° (below horizontal axis)

Radio / Radar Unobstructed LOS



Radio / RADAR Horizon Reference

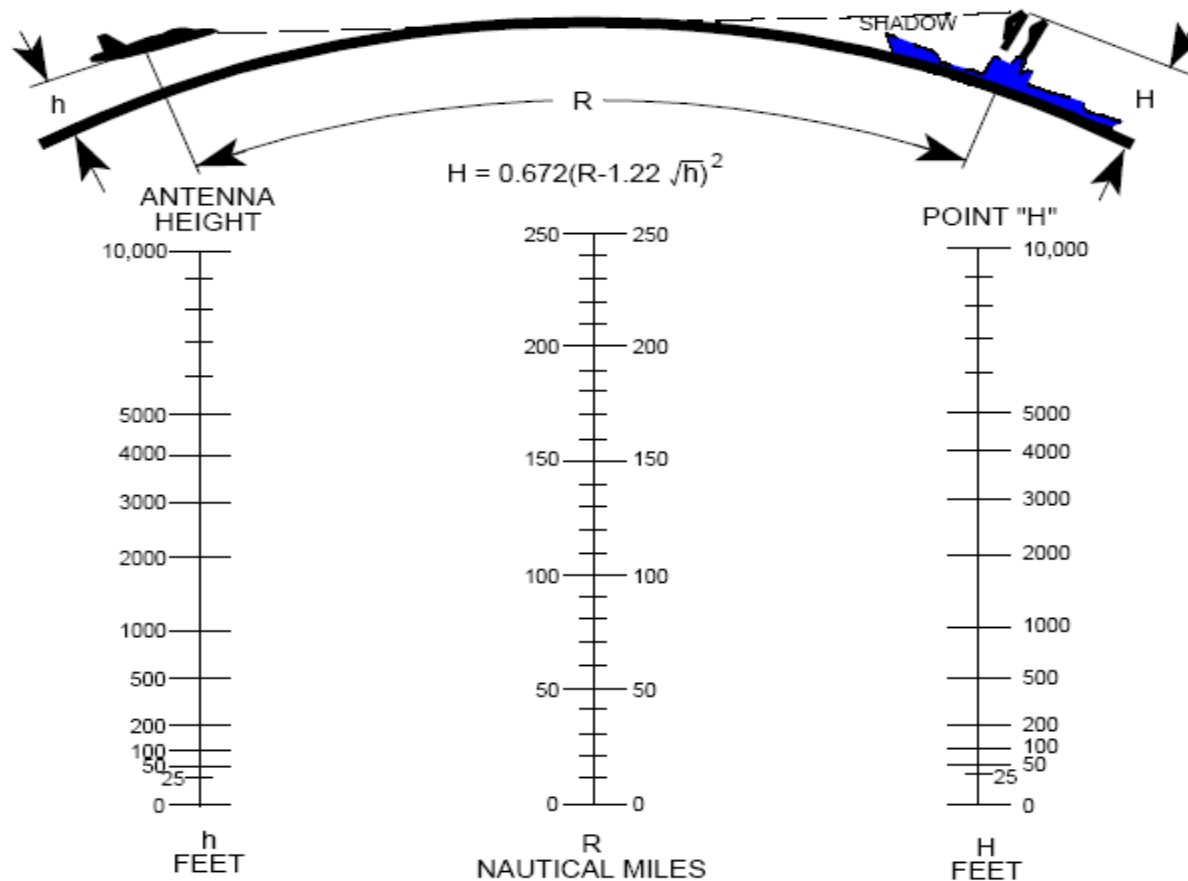


Figure 2. Earth Curvature Nomograph