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Aug 14/2010

THEORY OF MACHINES

ANALYSIS OF PLANE MECHANISMS

Sreenag. r 4-col

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MECHANISM:-

Converts available motion into desired motion.
Eg: slider crank

MACHINE:-

Converts energy from one form to another.
Machine consists of a no. of individual mechanisms.

If all the links are constrained to move in a single plane or parallel plane, it is called a PLANE MECHANISM.

ANALYSIS

Forward (Routine)

Reverse (Inverse)

MECHANISM: Inter connected rigid bodies.

MECHANICS:- study of rigid bodies under forces.

MECHANICS

STATICS
(Body under rest)

DYNAMICS
(Body in motion)

STATICS:- Body in equilb.
Net force acting = 0.

DYNAMICS:-

1) Kinematics: It deals with the Geometric variables of motion without considering the causes of motion.

Geometric variables: vel, accn, position, disp, jerk etc...

2) Kinetics : study along with forces that cause motion.

LINKS :-

Kinematic elements which are the individual parts of a mechanism.

In a kinematic pair, there should be constrained relative motion b/w the 2 bodies.

CLASSIFICATION OF kinematic pairs :-

(1) Based on nature of the relative motion

a) turning pair / Revolute / Rotary / Hinge

b) sliding pair / prismatic pair / Translation.

Combination of turning & sliding pair gives cylindrical pair.

- Spherical pair.

(2) Based on Nature of contact

a) Lower pairs - surface to surface contact

b) Higher pairs - point or line contact.

Higher pairs {
Eg: Pairs of gear meshed : Line contact
Cam & follower : Point contact

(3) Based on nature of closure

a) Form closure

b) Force closure

Contact by form / shape is called form closure
if contact is by an external contact,

axis

which is responsible for maintaining contact, is called force contact.

deal

(A) Based on Constraint.

ined

(S) Based on the Degree of Freedom.

For any given system, the no. of individual body's reqd. to specify the location of the body is called D.O.F.

DOF for a rigid body in space = 6

body in a plane, DOF = 3 (2 translation + 1 rotation)

SPATIAL KINEMATIC PAIRS

ives

When 2 links form a pair, it loses some of the individual DOF.

CONSTRAINED

DOF

ALLOWED DOF

5

1

(CLASS-I)

4

2

(CLASS-II)

3

3

(CLASS-III)

2

4

(CLASS-IV)

1

5

(CLASS-V)

PLANAR KINEMATIC PAIRS

C. DOF

A. DOF

2

1

(CLASS I)

1

2

(CLASS II)

KINEMATIC CHAIN

- closed kinematic chains are most stable than open " " .

Eg: Mechanisms of Robot manipulator is an open kinematic chain.

$J_1 \rightarrow$ NO. of class I pairs

$I_3, I_4 \rightarrow$ comp class III, IV pairs

Q. = links.

$$\text{DOF} = (6 \times 1) - 5J_1 - 4J_2 - 3J_3 - 2J_4 - J_5$$

Each class I pair loose 5 degrees of freedom.
class II " " 4 degrees " " etc.

def for a spatial mechanism

$$\det = 6(1-1) - 5J_1 - 4J_2 - 3J_3 - 2J_4 - J_5.$$

dot for a PLANAR KINEMATIC CHAIN

$$dof = \underbrace{31-2J_1}_{I_1} - \underbrace{1J_2}_{J_2}$$

dot in a PLANE MECHANISM = $3(l-1) - 2J - 3$

MECHANISM WITH SINGLE dof.

$$l, J_1 = J, J_2 = 0$$

$$1 = 3(l-1) - 2J$$

$$J = \frac{3}{2}l - 2$$

If "l" is odd, J becomes fractional.

So "l" has to be even, then "J" becomes a whole no.

- For $l=2, J=1$

- $l=4, J=4$

- $l=6, J=7$

- $l=8, J=10$



FOUR BAR CHAIN OR QUADRIC CYCLE CHAIN.

(1) Length of longest link < sum of the other 3

↳ NECESSARY condition for a 4 bar chain.

(2) Also,

sum of longest + smallest link < sum of other 2

$$l + s < p + q$$

The above is called the GRASHOFF'S condition

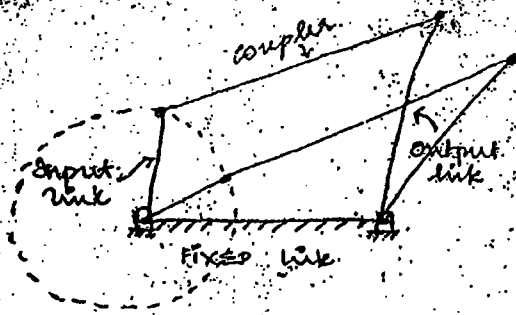
If $l + s > p + q$ ∴ Non Grashoff's kinematic chain

of $l + s = p + q$ ∴ Special " " "

INVERSION

Process of obtaining diff mechanisms by fixing diff links of a kinematic chain.

$1 - I_2$

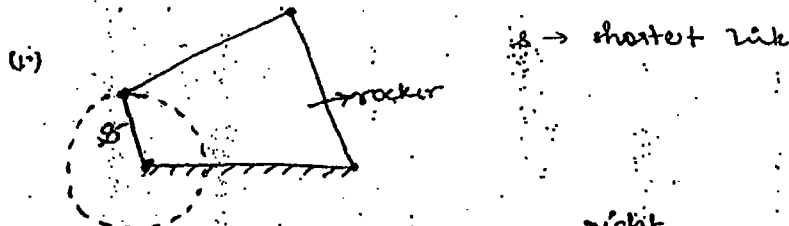


If input link is capable of complete rotation it's called a CRANK.

If output link isn't capable of complete revolution, it's called a Rocker / oscillatory link / lever.

	Input link	Output link
1.	Crank	Crank
2.	Crank	Rocker
3.	Rocker	Crank
4.	Rocker	Rocker

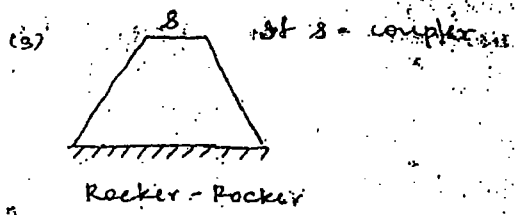
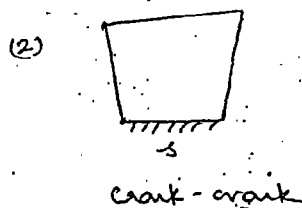
Inversions (Grashoff's kinematic chain)



By fixing the link on ^{right} ~~other~~ side of the shortest link, we get the shortest link as crank and the opp link as rocker.

If we fix the link on the left of shortest link we get ~~shortest~~ 's' as rocker & opp as crank.

So 2 mechanisms: Crank - rocker
& Rocker - Crank.



INVERSIONS { GRASHOF'S }

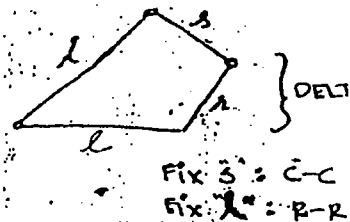
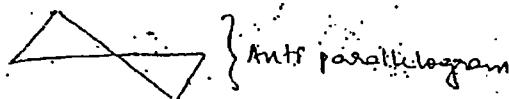
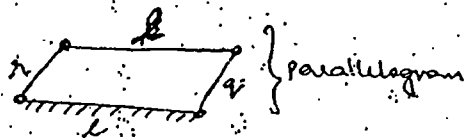
All inversions are double rockers { R-R's }

INVERSIONS { special Grashof's }

$$l + s = p + q$$

if $l = p, s = q$: parallelogram linkage

in a 4-bar linkage : all inversions are double cranks.

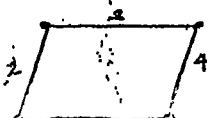


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EQUIVALENT CHAIN

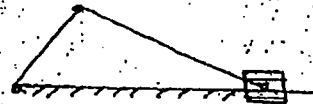
Replacing a turning pair with a sliding pair in a 4-bar chain we get a Equit chain.

we get a SLIDER crank chain.

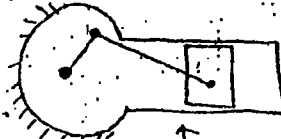


INVERSIONS OF SLIDER CRANK

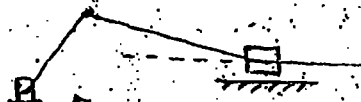
(1)



FIRST INV. ↑



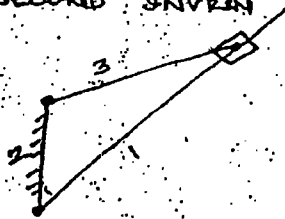
Application of 1st Inversion of S.C.R. engine



OFFSET SLIDER CRANK MECHANISM

(2)

SECOND INVERSION (double crank)



As Timed Angular disp

Time for FORWARD STROKE

$$180 + 2\alpha$$

$$180 - 2\alpha$$

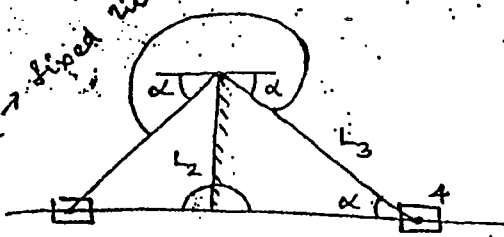
Time for REVERSE STROKE

Quick Return Ratio,

$$QRR = \frac{180 + 2\alpha}{180 - 2\alpha}$$

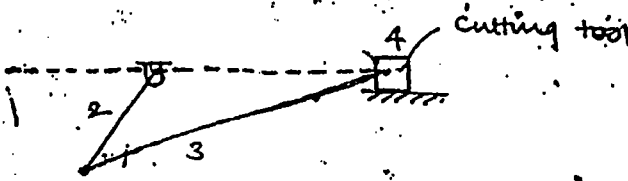
$$\alpha = \sin^{-1} \frac{l_2}{l_3}$$

l2 → fixed link



Quick Return mechanism

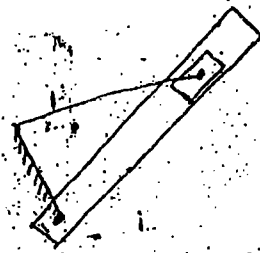
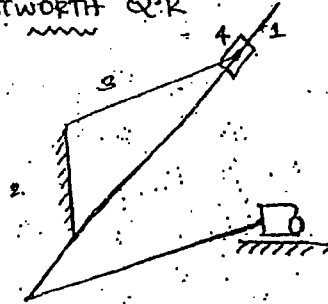
Application: In shaper, planing and cutting operations



Cutting tool

This could have been used as the quick return mechanism but doesn't can not be given on link 2. so we combine 1st inv & 2nd inversion.

WHITWORTH Q.R



Q.C engine

Whitworth Q.R mechanism is a combination of 2nd inversion and 1st inversion.

ROTARY CYLINDER MECHANISM.

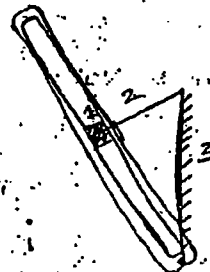
Whitworth + Rotary → application of 2nd inversion

(3) m

3rd INVERSION

FORWARD stroke

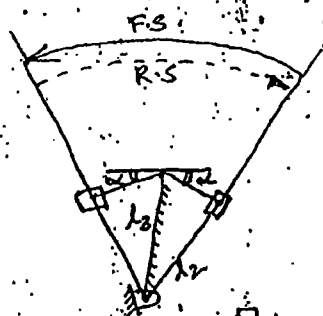
(4)



REVERSE stroke

SLOTTED LEVER MECHANISM.

(5)

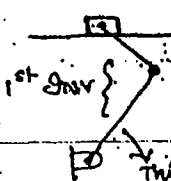


$$\frac{TFS}{TRS} = \frac{180 + 2\alpha}{180 - 2\alpha}$$

$$\alpha = \sin^{-1} l_2 / l_3$$

l_3 = fixed link

(6)



→ combination of 3rd inversion and the offset slider crank (1st) inversion.

crank & slotted lever type

this is in this mechanism a combination of 2nd & 1st

Q1

FOURTH INVERSION

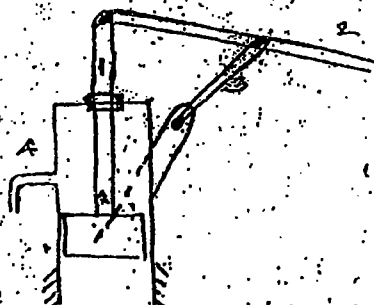
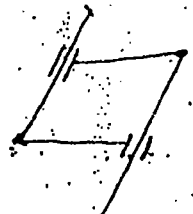


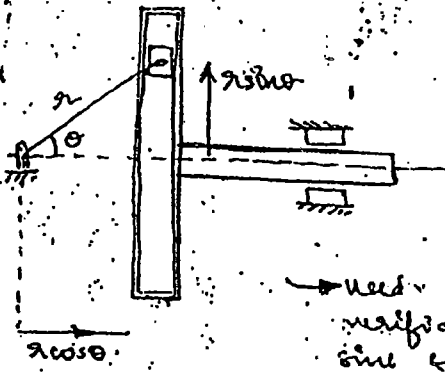
Fig: Application of 4th inversion in a hand-pump.

DOUBLE SLIDER CRANK MECHANISM for a 4-bar

- (1) 2 sliding pairs, 2 turning pairs & a CRANK
↳ called scotch yoke mechanism

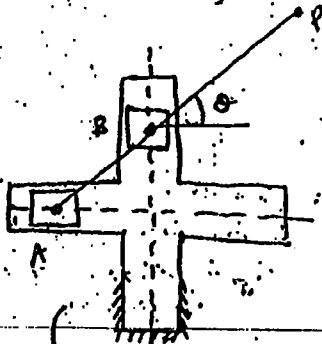


To generate a harmonic motion mechanically



Need for the verification of sine & cosine for in older times.

(2) ELLIPTICAL TRAMMEL



An ellipse is generated

$$x_p = BP \cos \theta$$

$$y_p = AP \sin \theta$$

$$\left(\frac{x_p}{BP}\right)^2 + \left(\frac{y_p}{AP}\right)^2 = 1$$

If $AP = BP \rightarrow$ Circle
is generated

(B) OLDHAM'S COUPLING

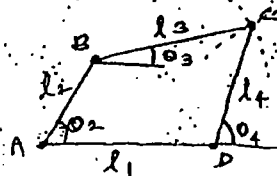
It's another inversion of double slider double crank 4-bar mechanism.

- used to join non-parallel, & mis-aligned shafts.
- used for small mis-alignments, low speed applications.

POSITION, VEL, ACCEL ANALYSIS

POSITION ANALYSIS

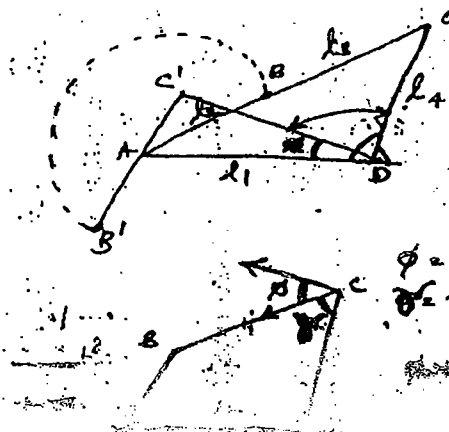
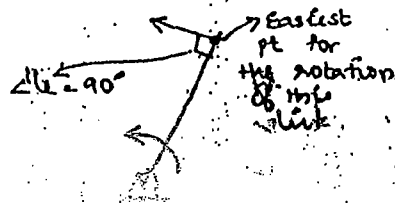
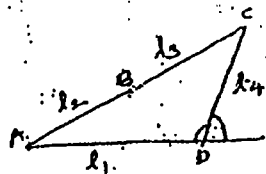
4-bar mechanism:-



To locate the pt C

$$\begin{cases} x_c = l_2 \cos \theta_2 + l_3 \cos \theta_3 = l_1 + l_4 \cos \theta_4 \\ y_c = l_2 \sin \theta_2 + l_3 \sin \theta_3 = l_4 \sin \theta_4 \end{cases}$$

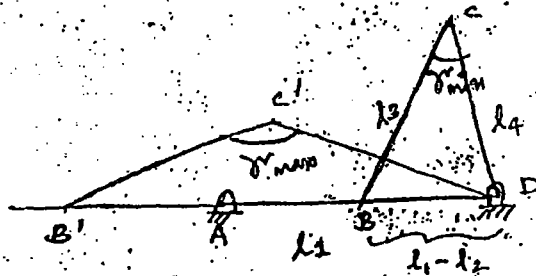
Extreme position: (for the rocker or lever)



$\phi = \alpha_2 \angle l_2$
 θ_2 transmits $\angle l_2$

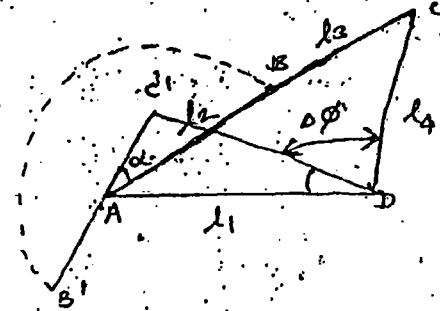
Transmission angle = $90^\circ \pm 30^\circ$ best < 6

when l_1 is aligned with l_2 : we get
min transmission < 6.



This is also a quick
return mechanism

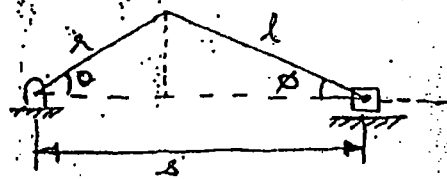
$$\frac{180 + \alpha}{180 - \alpha} = QRR$$



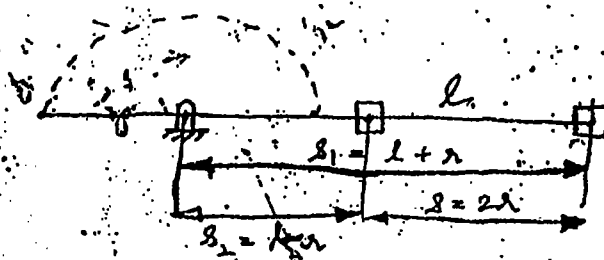
$\alpha \rightarrow$ Angle that indicates the angular
displacement at the i/p crank.

$$x = r \cos \alpha + l \cos \phi$$

$$r \sin \alpha = l \sin \phi$$



Extreme positions :-

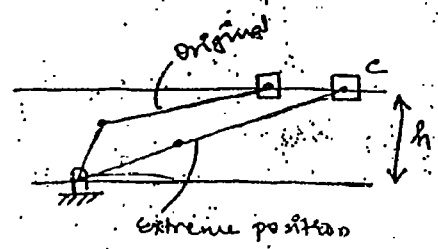
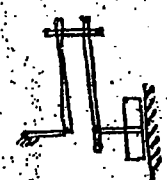
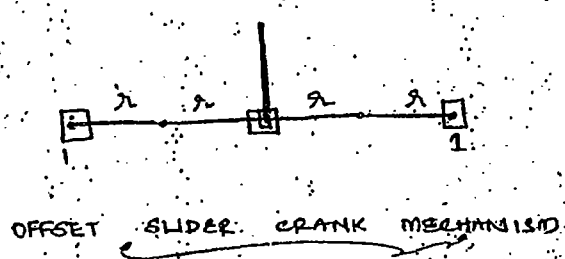


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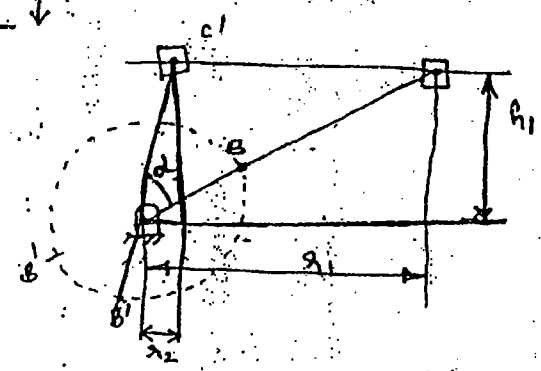
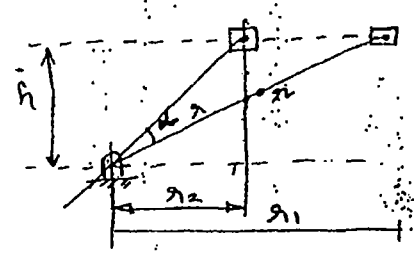
with
crank and guide

Q. What is the length of stroke if the length of connecting rod is equal to length of crank?

- a) r b) $2r$ c) $3r$ ☒ d) $4r$



Forward stroke
 $F.S \rightarrow 180 + \alpha$
 $B.S \rightarrow 180 - \alpha$

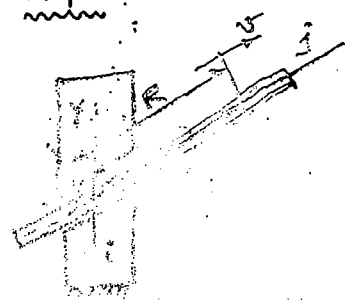


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Oct 23/2010

VELOCITY AND ACCELERATION ANALYSIS

SLIDING PAIRS



$$v = \frac{d}{dt} r = r \omega$$

$$a = \frac{dv}{dt} = r \alpha$$

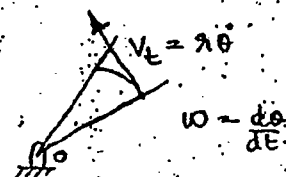
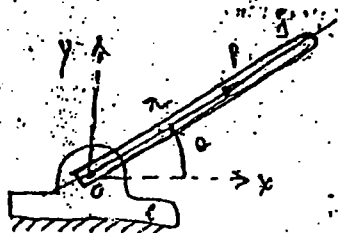
- If position vector inc
i.e. posn vector radially outwards } +ve vel

If position vector dec
i.e. r radially inwards } -ve vel.

- If link PQ is STATIONARY $\rightarrow$ the vel is
ABSOLUTE vel

If link PQ is MOVING $\rightarrow$ the vel is
RELATIVE vel.

TURNING
PAIR



vel $\rightarrow$ Always tangential
wrt O.

Accelerations:

(i) - Tangential accel

- parallel to tangential vel
- $\perp$ to position vector

$$f^t = r\alpha$$

$$\alpha = \frac{d\omega}{dt} = \ddot{\theta}$$

(ii) Centripetal accel

- always acts towards the centre of rotation
- responsible for tilt or change in direction of vel.

$$f^c = r\omega^2$$

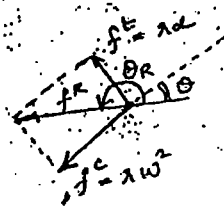


(60)

(8)

- If ω and α have the same sign i.e. its magnitude inc with resp to time, they are in the same direction.
- If ω and α have same sign & its acceleration
- If ω and α have diff sign: then it is deceleration.

RESULTANT acceleration

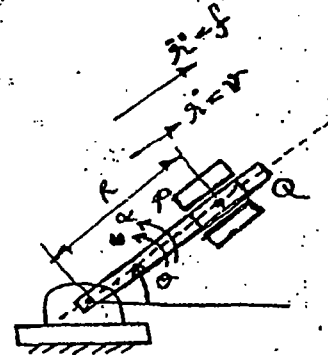


f^R - resultant accn

θ_R - angle the resultant makes.

TURN - SLIDE PAIR

combination of turning and sliding.



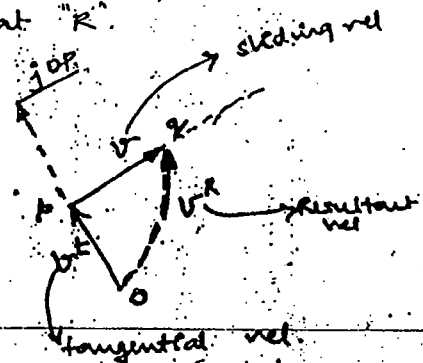
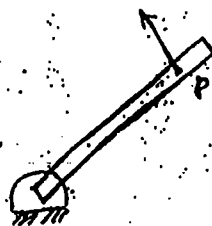
"Q" has some relative motion w.r.t "P".

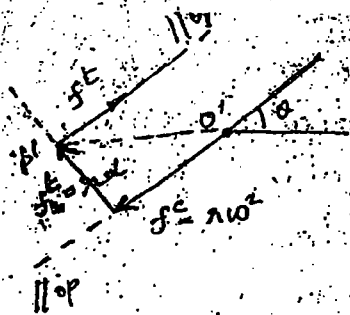
Instantaneously P & Q coincide and hence they are called COINCIDENT POINTS.

Instantaneous ^{ang} vel = ω .

Instantaneous ang accn = α .

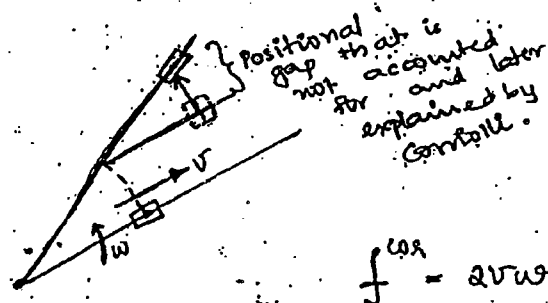
Block is instantaneously at "R".





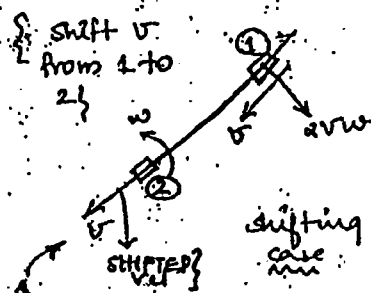
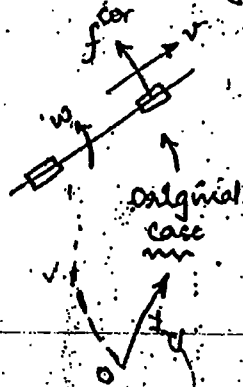
CORRIOLLI'S COMPONENT

Whenever axis of the sliding pair undergoes rotation $\rightarrow$ Coriolis component of acceleration



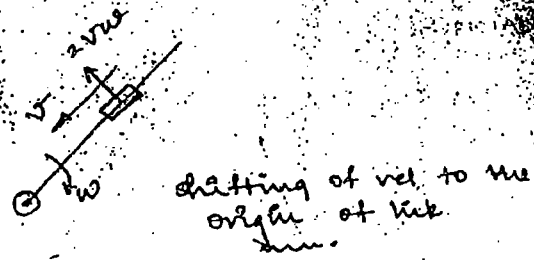
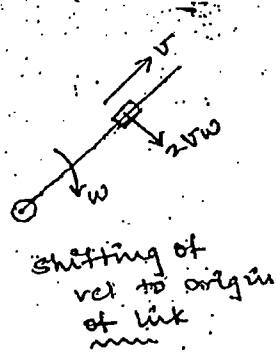
By rotating the vel vector in the direction of angular vel by $90^\circ \rightarrow$ To get direction of Coriolis's acceleration

- vel vector has to be drawn from the centre of origin of the link

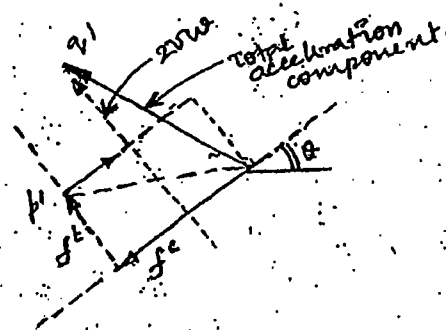


Here the vel has to be shifted from 1 to the pt of origin of the link i.e. (2)

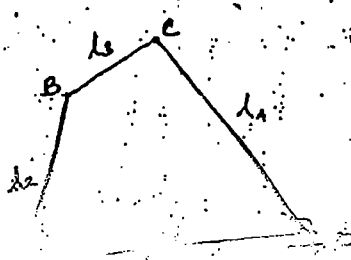
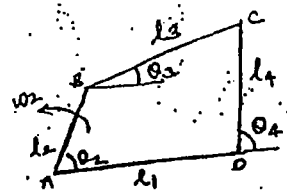
(61) (9)



using Coriolis for getting total acceleration

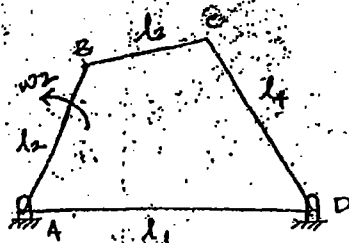


VELOCITY ANALYSIS OF A FOUR BAR MECHANISM



A & B are the reference pts (with absolute zero velocity) for the entire velocities.

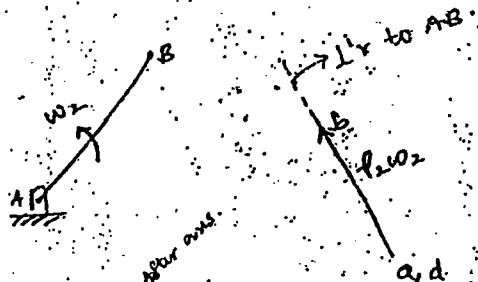
ANALYSIS:



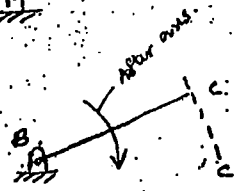
$$w_3 = ?$$

$$w_4 = ?$$

For AB:



For BC:

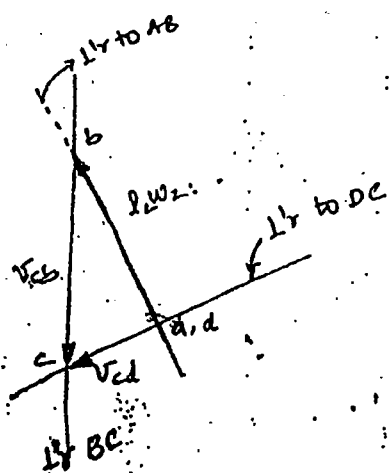


Not possible to arrive at a conclusion

For CD:



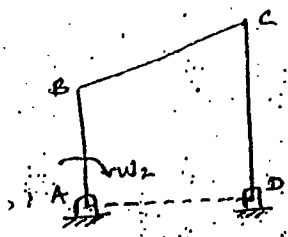
vel. of c wrt. , l'r to DC.



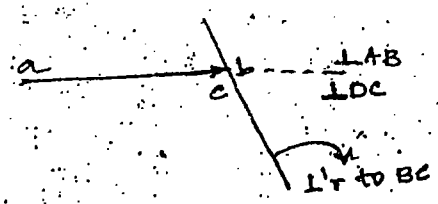
$$v_{cb} = l_3 w_3 \Rightarrow w_3 = \frac{v_{cb}}{l_3}$$

$$v_{cd} = l_4 w_4 \Rightarrow w_4 = \frac{v_{cd}}{l_4}$$

CASE 1: AB and CD are parallel to each other



$V_{cb} = 0$
 $\omega_3 = 0$
 $V_{cd} = V_{ba}$
 $I_A \omega_2 = I_D \omega_3$
 $V_b = V_c$



b and c coincides

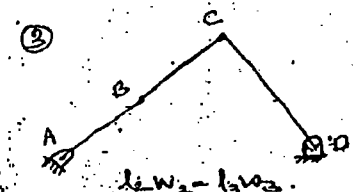
CASE 1(a):



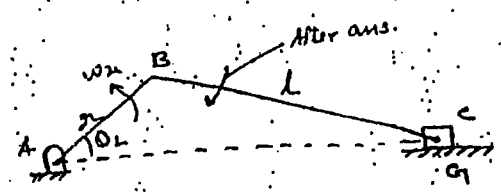
Here i/p and o/p will have net in opp directions.

Here also $AB \parallel CD$

Rotate the fig. (to get diff configuration)



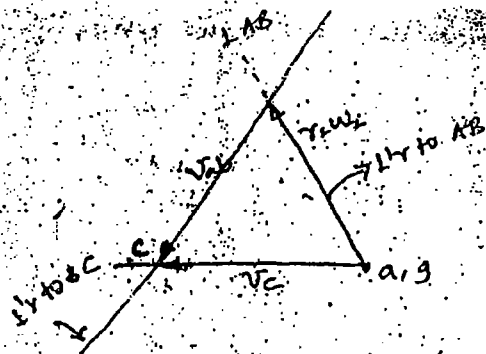
GLIDER - CRANK MECHANISM



$\omega_3 = ?$
 $V_A = ?$

Fixed pts $\rightarrow$ A and G.

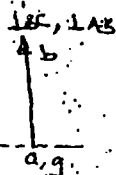
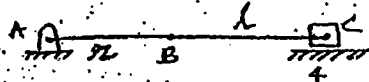
$\{ AB \text{ is crank.}$
 $V_{c/w/b} \rightarrow \text{Ltr to BC etc...}$



$$v_{cb} = l\omega_3 \Rightarrow \omega_3 = \frac{v_{cb}}{l} = \frac{v_c}{l}$$

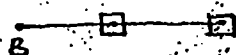
$$v_{slider} = v_c = g\vec{c} \text{ or } \vec{ac}$$

Ques: 180° configuration:



$v = 0$ for piston
So, DEAD centre position

$$v_c = 0$$

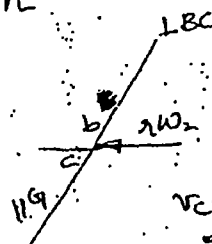
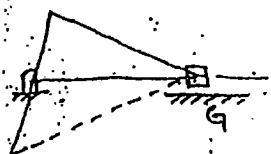


$$v_c = -v_{a,b}$$

$$l\omega_3 = -\lambda\omega_2$$

$$\omega_3 = -\frac{\lambda\omega_2}{l} = -\frac{\omega_2}{\lambda}$$

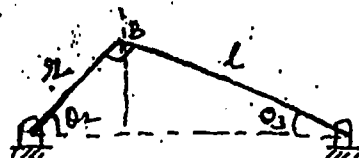
$$v_{slider} = v_{tip \text{ of crank}}$$



$$v_c = v_b = \lambda\omega_2$$

$$v_{slider} = v_{tip \text{ of the crank}}$$

Ques:



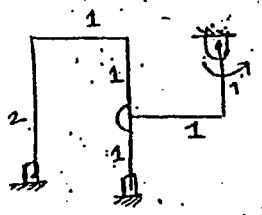
$$\theta_3 = 90 - \theta_2$$

$$l \sin \theta_2 = l \sin \theta_3$$

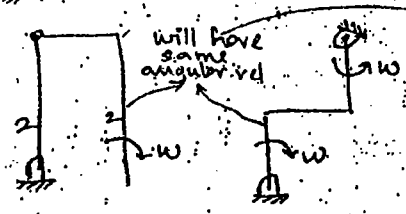
$$= l \cos \theta_2$$

$$\tan \theta_2 = l/r$$

Ques :



Separate the figure into 2 so that it becomes more simpler...

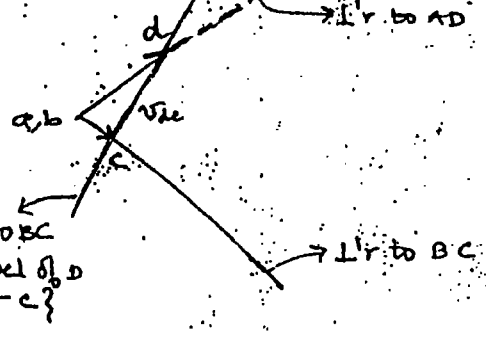
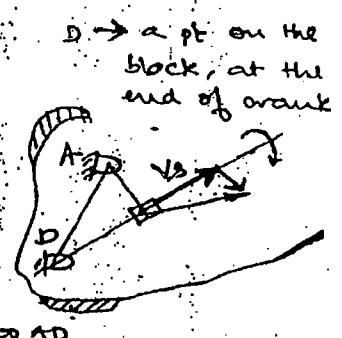
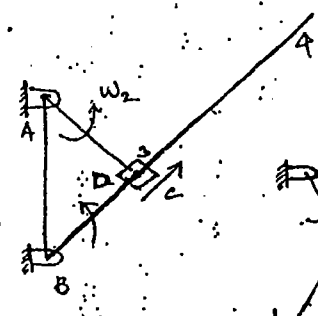


These 2 links are linked. They are together (adjacent) so angular vel on one is same as that on another.

2-5
5:40-1

NOV 13' 2010

VELOCITY ANALYSIS FOR INVERSIONS OF SLIDER CRANK



vel of D wrt C is $\parallel$ to BC.

C is a pt on link BC, At the moment C & D coincides, D slides along the link?

$$V_{da} = AD \cdot \omega_2$$

$$\omega_4 =$$

$$V_{cb} = \omega_4 \times BC$$

$$\omega_4 = \frac{V_{cb}}{BC}$$

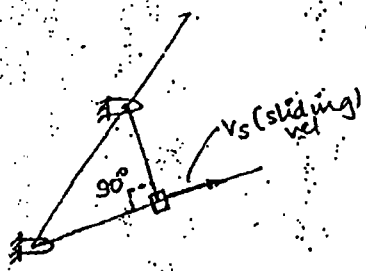
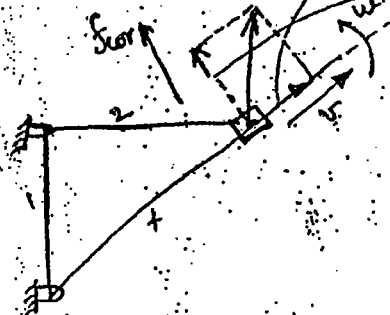
$$V_{sliding} = cd$$

$$f_{cor} = 2vw = 2V_{sliding} \times \omega_4$$

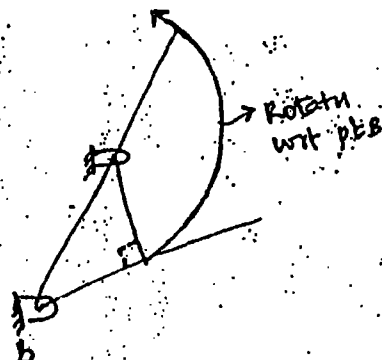
Coriolis's component of acen

Horizontal component

Vertical component (tangential)



V_s is the sliding vel..
Here in this case,
 V_s exists,,
but tangential component
is zero.
Hence the Coriolis's
acen is zero in this
case when the angle
included is 90°

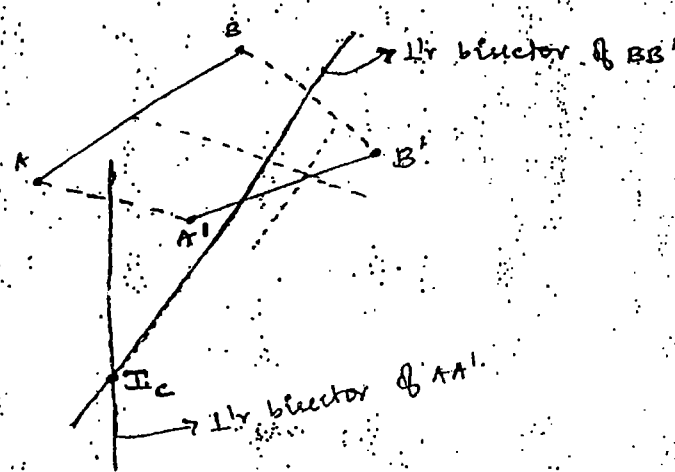


$\omega_4 =$ vel. of the link
when there is a
rotation with point 'B'.

ANALYSIS USING INSTANTANEOUS CENTRES

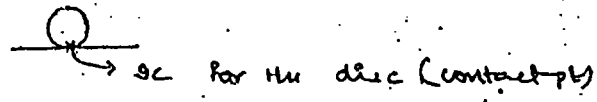
General rigid body motion $\rightarrow$ Translation + Rotation

For a particular pt the motion of the link can be pure rotation $\Rightarrow$ this pt is called instantaneous centre.



$\rightarrow$ For n rigid bodies in motion, there are nC_2 instantaneous centres.
Eg: for a 4 bar mechanism
4 links $\Rightarrow$ no. of IC's $= nC_2 = \frac{4 \times (4-1)}{2} = 6$

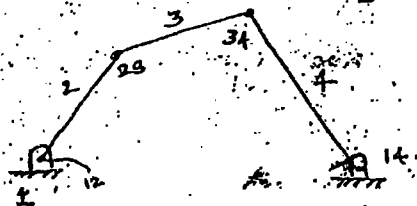
$\rightarrow$ For a disc rotating w/o slipping, IC will lie at the contact pt.



$\rightarrow$ If the disc is slipping, IC will lie on the axis of rotation.

the IC will lie on the axis of rotation.
Common link (ARNOLD / KENNEDY'S)

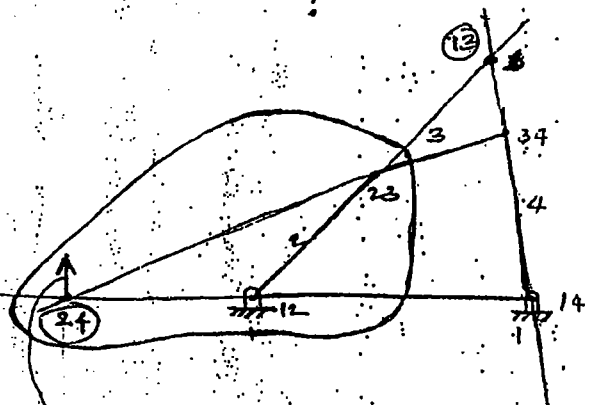
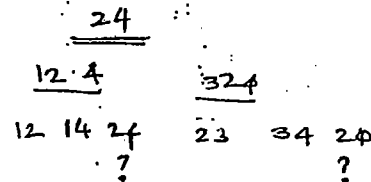
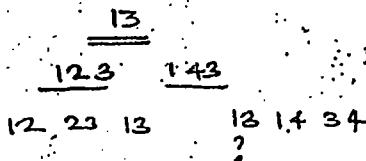
VEL. MECHANISM. 4-BAR MECHANISM USING INSTANTANEOUS CENTRES



when 2 pairs are joined directly, the joint itself is an instantaneous centre.

IC for 1-3:

1-3 not directly connected, so use the Kennedy's theorem or three centre theorem.



Input known $\Rightarrow$ output = ?

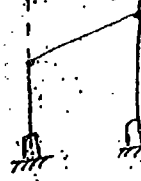
$$V_{24} = (12-24) \omega_2 = (14-24) \omega_4$$

If the body is a part of link 2

If the body is a part of link 4

$$V_{23} = (12-23) \omega_2 = (13-23) \omega_3$$

Ex:



Ans: $\omega_3 = 0$

Instantaneous centre at infinity

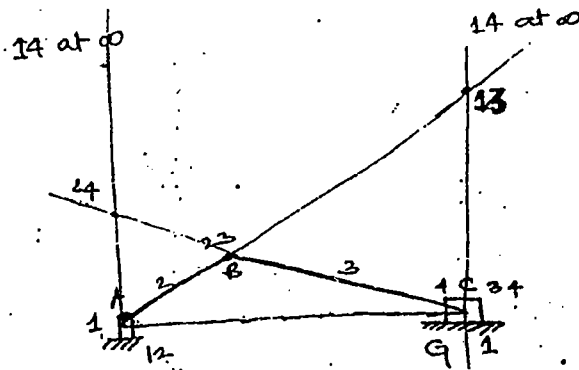
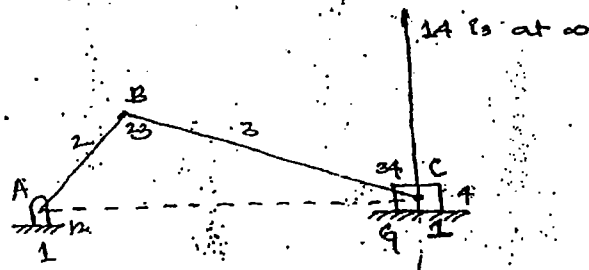
$$\text{so } V_{23} = (I_2 - I_3) \omega_2$$

$$V_{23} = (I_3 - I_2) \omega_3$$

If this dist $\rightarrow \infty$
 ω_3 has to be zero

INSTANTANEOUS CENTRE METHOD FOR A SLIDER

CRANK MECHANISM



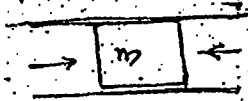
$$V_{24} = (I_2 - I_4) \omega_2 = V_4 \text{ (sliding vel)}$$

$$V_{23} = (I_2 - I_3) \omega_2 = (I_3 - I_2) \omega_3$$

$$V_{34} = (I_3 - I_4) \omega_3 = V_4$$

DYNAMIC ANALYSIS

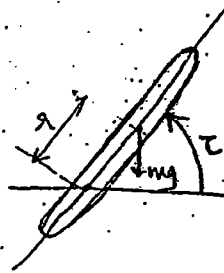
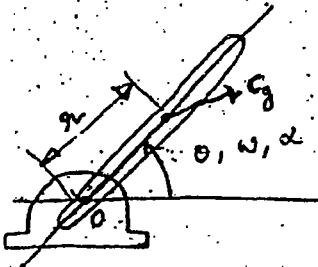
SLIDING PAIR :-



$$\sum F = ma$$

Net force along the axis of reciprocation.

TURNING PAIR :-



$$M_o = T - mgr \cos \theta = I_o \alpha$$

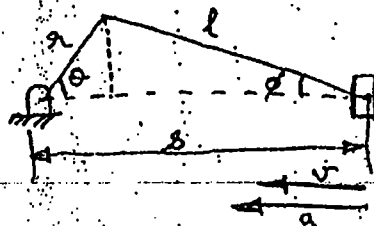
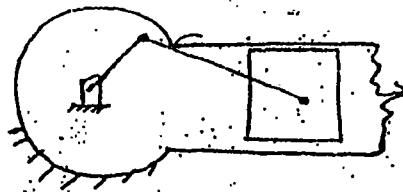
$$\alpha = \ddot{\theta} \quad \{\text{angle } \theta\}$$

$$I_o \ddot{\theta} + mgr \cos \theta = T \quad \text{Torque}$$

DYN ANALYSIS OF SLIDER CRANK BASED m/c

3 turning pairs
1 sliding pair

Slider crank mechanism based m/c



$$s = r \cos \theta + l \cos \phi$$

$$r \sin \theta = l \sin \phi, \quad \omega = \dot{\theta}$$

$$v = r \omega (\sin \theta + \frac{\sin 2\theta}{2n})$$

$$a = r \omega (\cos \theta + \frac{\cos 2\theta}{n})$$

$$\left. \begin{array}{l} v \\ a \end{array} \right\} n = \frac{l}{r} \text{ ratio}$$

Net force on the piston = PISTON EFFORT.

Pressure on the connecting rod = Thrust

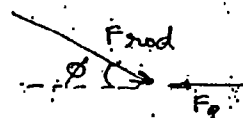
Thrust leads to turning moment.

Piston effort consists of:

$$F_p = F_g - ma$$

$m = \text{mass of}$

store due to gas

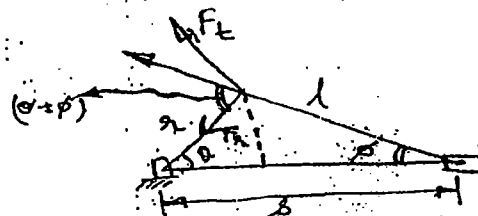


$$F_p = F_g - ma$$

$$F_{rod} = F_p / \cos \phi$$

$F_t = \text{Tangential force on the crank}$

($F_t = \text{CRANK EFFORT}$)



→ we have ignored the mass of the connecting rod.

$$F_t = F_{rod} \sin(\theta + \phi)$$

$$F_t = F_{rod} \times \cos(\theta + \phi)$$

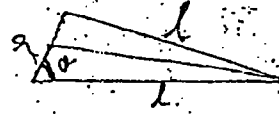
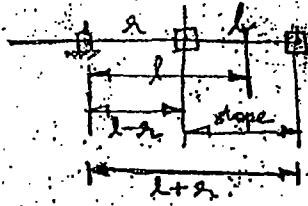
$$F_t = \frac{F_p \sin(\theta + \phi)}{\cos \phi}$$

$$\text{Turning moment, TM} = F_t \times r$$

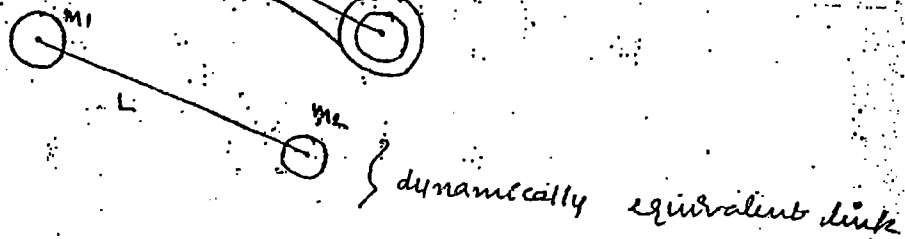
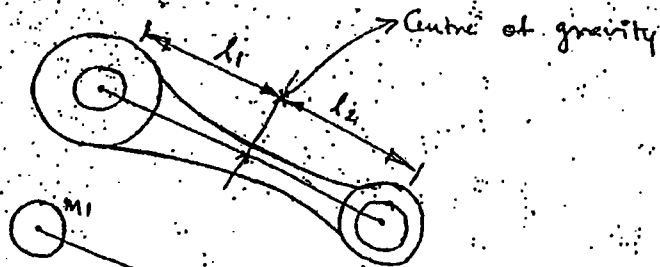
$$\text{TM} = \frac{r F_p \sin(\theta + \phi)}{\cos \phi}$$

Crank effort } of IC engine
turning moment

eg



DYNAMICALLY EQUIVALENT LINK (DEL)



conditions for DEL:-

- 1) $m_1 + m_2 = m$
- 2) $m_1 l_1 = m_2 l_2$
- 3) $I_{eq}(cg) = m_1 l_1 + m_2 l_2$

3 eqns cant be satisfied

↓ so we use the first 2 eqns
but then we mention that the moment of inertia are not the same.

so to account for the diff in MOI b/w the actual link & the DEL { error of moment of inertia }, we introduce a compensation for this error.

pg 5 [exo]

1 (c) $d = (L-1)3 - J \cdot 2$ $\left. \begin{matrix} L=8 \\ J=9 \end{matrix} \right\}$
 $= 21 - 18 = 3 //$

2 (b) $AB = 2.65 \sqrt{240^2 - 100^2}$

It's hence a Grashoff's linkage.

Also It's a crank rocker mechanism
 (link adj to shortest link is fixed)

3 (b) S_p link and d_p link are parallel.

$V_A = V_B$

$OA \times \omega_2 = OB \times \omega_4$

$60 \times 8 = 160 \times \omega_4$

$\omega_4 = \frac{60 \times 8}{160} = 3 \text{ rad/s}$

$\frac{60 \times 8}{160}$

4 (b) since planar $\rightarrow$ no gravity term in eqn
 (mg sine) = 0.

$\alpha = 0 \rightarrow$ no inertial torque.

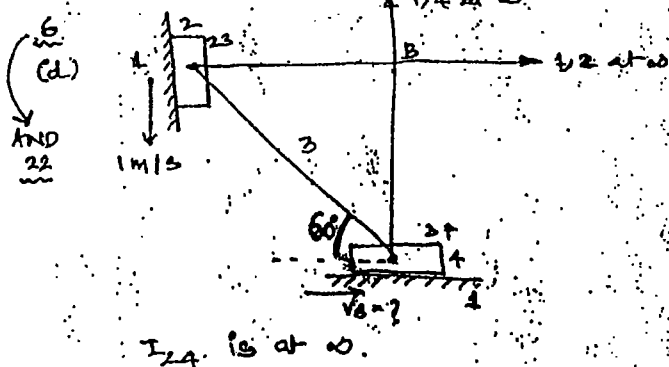
Axis of rotation & centre of mass coincides,
 hence no centrifugal force

along AB 30 N acts,
 At O2 also 30 N should act.

2x12
 24

5 (c) $I_{\text{rotor}} = 2V\omega$
 $= 2 \times 12 \times \frac{2\pi \times 60}{60} \omega$

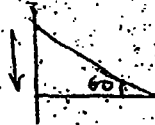
$\left\{ \begin{matrix} \omega = \frac{2\pi \times 60}{60} \\ \omega = 2\pi \times 1 \end{matrix} \right\}$



$\frac{24}{12 \times 4}$
 $\frac{12 \times 14 \times 24}{12 \times 14 \times 24}$

I_{24} is at ω .

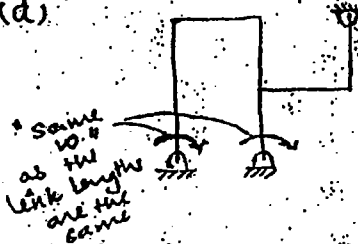
22. (1) same fig as 6



7 (c) R-2

8 (c) P-2 ? Allow 1, 5 are constrained for revolute pairs?

9 (d)



OD = 100 mm

$\omega = 2$

$v = r\omega = 200 \text{ mm/s}$

Centripetal accn = 400 mm/s^2

10 (c)

11 (c)

(Ans)

Vel polygon is a scaled Δ 's.

{2' link b/w crank and connecting rod = 90° }

Use pythagoras

12 (c)

$v_{BA} = ?$

$v_{BA} = v_B - v_A$

$v_B = OB \times \omega$, $v_A = OA \times \omega$

$v_{BA} = (r_B \times \omega) - (r_A \times \omega) = (r_B - r_A) \omega$

13 (d)

ω uniform
so only centripetal accn.

$f_{BA} = f_B - f_A = (r_B - r_A) \omega^2$

14 (b)

Higher pair - class II

If only lower pairs are allowed, the min no. of links is 4.

{ If higher is allowed, then the no. of links is 3.

15(b)

Only when axis of the slider undergoes rotation
so that the 2nd or 3rd members of slider
crank.

⇒ SHAPER

16(d)

Hooke's / universal → large misalignment
Oldham's → small misalignment

18()

$$V_B = EB \times \omega_3 = 0.5 \text{ m/s}$$

$$\omega_3 = -$$

$$V_C = EC \times \omega_3$$

19(b)

Pure rolling → at the centre of disc → (1)

Pure sliding → at centre of surface → (2)

Both rolling & sliding → at the line joining
(1) & (2).

→ So common N^l at the contact pt.

20(c)

4 links, 3 turning pairs → 12 dof

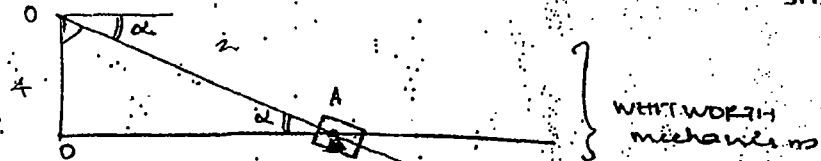
$$12 \text{ dof} - (5 \times 2) = 2 //$$

21(b)

LINK fixed is SHORTEST →

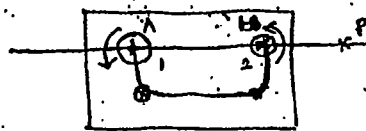
double CRANK
mechanism.

above



$$\alpha = \sin^{-1} \left(\frac{OO'}{OA} \right)$$

23(c)



$$AP(\omega_1) = BP(\omega_2)$$

$$r_1 \omega_1 = r_2 \omega_2$$

$$\omega_2 = \frac{r_1}{r_2} \omega_1 = 2\omega_1$$

Try

2 pulleys connected to a belt drive.

24 () x coord, $x_B = p$
 y coord, $y_B = p \tan \theta$

vel in x dir = $\frac{dp}{dt} = 0$

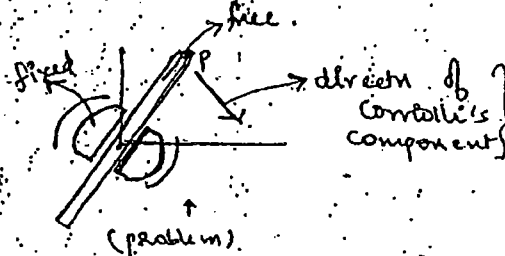
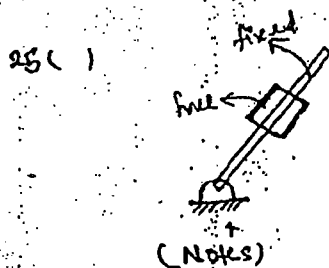
vel in y dir = $\frac{d(p \tan \theta)}{dt} = p \sec^2 \theta \cdot \dot{\theta}$
 $(v_y) = \omega p \sec^2 \theta //$

$a_x = 0$

$a_y = \frac{d(v_y)}{dt} = -2 \cos^3 \theta (-\sin \theta) p \cdot \dot{\theta} \cdot \dot{\theta}$
 $= 2 p \omega^2 \frac{\sin \theta}{\cos^3 \theta} //$

$\dot{\theta} = \frac{d\theta}{dt} = \omega$

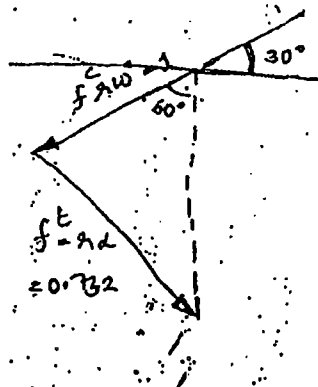
Note: If $v_x \neq 0$,
 then find resultant of v_x and v_y



No radial accel

v is -ve, ω is +ve

$f_{control}$
 $= 2 v \omega$
 $= 1 m/s^2$



26 (d)

Mechanical = $\frac{1}{vel ratio}$
 adv

$vel ratio = \frac{vel output member}{vel of input member}$

$\therefore$ Mech adv = ∞

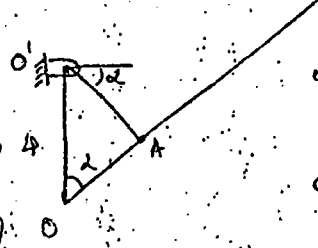
(69) (17)

27(c)
28(b)

21(b)

3rd
Inversion

Crank & slotted lever
mechanism



$$\alpha = \sin^{-1} \left(\frac{O'A}{O'O} \right)$$

$$30^\circ$$

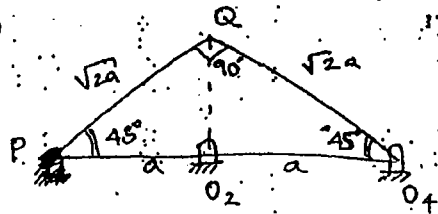
$$QRR = \frac{180 + 2\alpha}{180 - 2\alpha} = 2.7$$

$$\frac{d\theta}{dt} = \omega$$

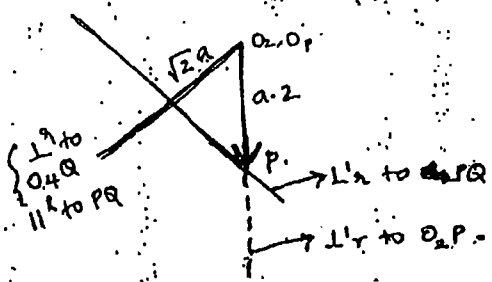
29(d)

30(a) At least one of the links must be a crank
↳ so Grashoff's or special Grashoff's.

31(c)



Relative vel method :-



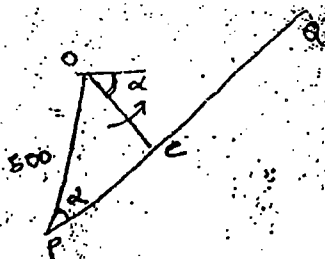
$$V_{QP} = V_{PQ} = \sqrt{2}a \cdot \omega_3$$

$$\omega_3 = \frac{\sqrt{2}a}{\sqrt{2}a} = 1 \text{ rad/sec}$$

$$V_{PQ} = \sqrt{2}a$$

$$V_{PQ} = V_{QP} = \sqrt{2}a$$

5



$$QPR = \frac{180 + 2x}{180 - 2x} = 2$$

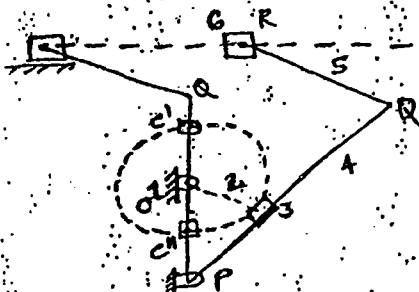
Q. 300

$$\sin \alpha = \frac{OC}{OP} \therefore \sin 30^\circ = \frac{1}{2}$$

$$OC = OP/2 = 250 \text{ mm} //$$

54 ط

max speed is when the angle is 90°
 i.e. when $\frac{y}{p}$ link is $\perp$ to line of stroke
 slider ~~max~~ will have max vel.
 { this can happen in forward and
 return stroke }



$$w_1 p c' = w_2 o c'$$

$$w_4(7.50) = 2.143 \cdot 2.10$$

$$W_4 = \frac{2 \times 210}{760} = \frac{2}{3} \text{ rev per sec.}$$

draining return stroke:

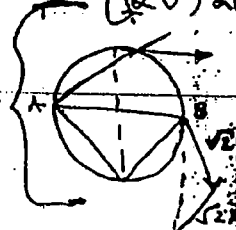
$$w_4 PC'' = w_2 OC''$$

$$\omega_1 = \omega_2$$

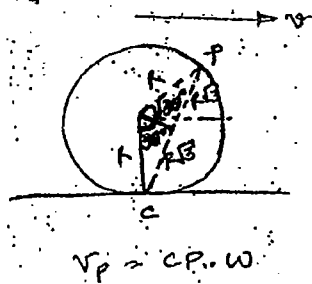
so return stroke w and forward stroke w are different.

25 (b) Ans c 1

end: $\phi_{\text{diag}}(2V)$ always



36()



$$v = R\omega$$

$$\omega = v/R$$

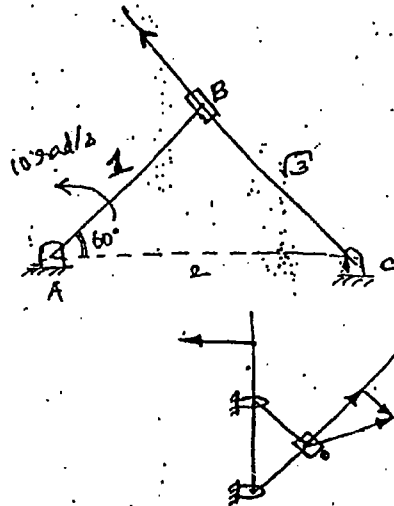
$$CP = 2R \frac{\sqrt{3}}{2} = \sqrt{3}R$$

$$v_p = \sqrt{3}R \omega = \sqrt{3}R \frac{v}{R} = \sqrt{3}v$$

Ans:

Ques: For the config shown, the angular vel of link AB is 10 rad/s counter clockwise. The magnitude of the relative sliding vel in m/s of slider B w.r.t link AC is...?

Ans:



$$\text{slider vel} = AB \cdot \omega_2$$

Pg 11 [Additional objectives]

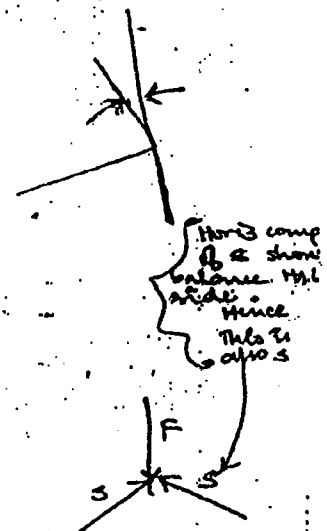
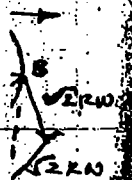
$$1(d) P = 3 \cos \alpha$$

$$S \sin \alpha + S \sin \alpha = F$$

$$F = 2S \sin \alpha$$

$$= \frac{2P}{\cos \alpha} \sin \alpha = 2P \tan \alpha$$

1. How the J. always



2. (b) $v_B = v_C$

$AB \omega_2 = DC \omega_4$

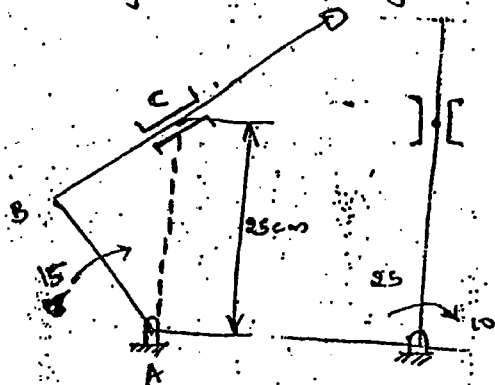
$CD = AB + 30$

$AB (5) = (AB + 30) 2$

$AB = 20 \text{ cm}$

3. (a) Rolling w/o slipping $\rightarrow$ contact pt. is A

4. (d)



$v_B = v_{ba}$

$a, c \rightarrow b$

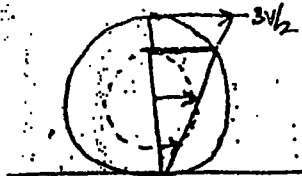
$BC \omega_3 = AB \omega_2$

$\omega_3 = \frac{AB}{BC} \omega_2 = \frac{15}{25} \times 10 = 6 \text{ rad/s}$

5. (c) Rolling vel = Radius $\times$ (Relative angular vel)
 $= r(\omega_1 + \omega_2)$

6. (c)

7. (c)



$\omega = v/r$

$r\omega = v$

8. 13(b) 1 $\rightarrow$ No 2 pts have same vel.

15. () $v_{BA} = v_B - v_A = r_b \omega - r_a \omega$
 $= (r_b - r_a) \omega$

$100 - 80 = 800 \omega$

(71)

(11)

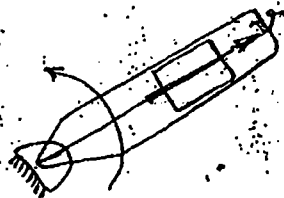
$$\frac{60000}{600} = 200 \text{ rad/sec} //$$

$$V_B = OB \omega = 140$$

$$OB = \frac{140}{\omega} = 700 \text{ mm}$$

$$\dot{\theta} \text{ (rad/sec)} = 2 \times 700 = 1400 \text{ mm/sec} \quad (\dot{\theta} \text{ (rad/sec)} = 2 \cdot OB)$$

8 (a)



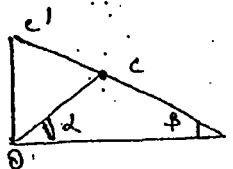
Uniform rotation } No tangential
 { piston reciprocating &
 cylinder rotating @ 80
 Coriolis accel is
 present.

9 (a)

They have to form a kinematic pair first.
 otherwise there is no question of constraint

10 (a)

11 (b)



$$vel = OC'(\omega)$$

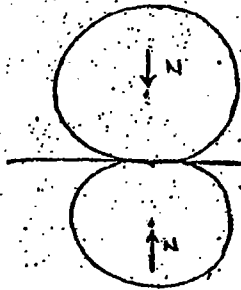
vel)

12 (d)

14 (d)

GEARS

- Toothed wheel to transmit power from one shaft to another.



FRICTION DISC

- Used only for slow power transmission

- Torque transmitted is low.

$$T = \mu N \times r$$

- By traditional methods involute is easier to generate than cycloid but with the advent of NC m/c's both are equally easy.

- Involute gears: Contact b/n convex regions
Cycloidal gears: Contact b/n convex & concave regions

- Contact stress is more for 2 convex regions
As contact stress $\uparrow$, wear $\uparrow$

- Hence higher contact stress for involute, hence higher wear for involute

- ϕ = pressure angle.

ϕ for involute $\downarrow$

$$r_b = r_p \cos \phi \text{ (remains constant)}$$

- For cycloidal, pressure angle keeps on varying inducing variable transmission loads. (disadv)

- In case of cycloidal gears, ω are and hence variation in centre of the vel ratio (disadv)
Hence they are no longer conjugate as
- Interference: contact b/w 2 non-conjug regions.
- Below base circle, involute generation is not possible. So we use circular arc, this region comes into contact with the tip of the other tooth (which is a conjugate part)
Circular arc: It is not a conjugate region
Region which is not involute can be removed by UNDERCUTTING.
- Undercutting reduces the strength of the gear
- If $\phi \uparrow$, T_{min} (min no. of teeth) can be reduced
- Composite teeth:-
Whenever a non-involute exists, make a cycloidal.

Hypocycloidal at bottom
Involute at middle
Epicycloid at top

} composite teeth profile.

- Arc of contact = $\frac{\text{path of contact}}{\cos \phi}$
 - No. of pairs of teeth in contact = contact ratio
- CONTACT RATIO:-
is the ratio of arc of contact to circular pitch
- Recommended value = 1.4 for contact ratio
(1.2 to 1.35)

- For 20% of time load is shared by the
prev pair of contact teeth.

For 60% of time load is taken by the
present pair of contact teeth.

For the last 20% of the time the load is
taken by the next pair of teeth.

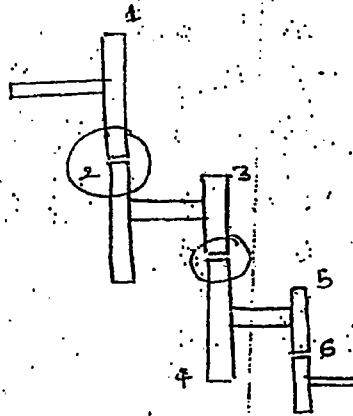
- Mfn contact ratio is desired.

COMPOUND GEAR TRAINS

(73)

Gear trains with one or more compound wheels:

- If 2 or more gears are keyed on to the same shaft $\rightarrow$ it's called compound.



$$VR = \frac{N_1}{N_4} = \frac{N_1}{N_2} \cdot \frac{N_2}{N_3} \cdot \frac{N_3}{N_4}$$

$$= -\frac{T_2}{T_1} \cdot 1 \cdot -\frac{T_4}{T_3}$$

$$VR = \left(\frac{T_2}{T_1}\right) \left(\frac{T_4}{T_3}\right)$$

$$\frac{N_1}{N_6} = \left(\frac{T_2}{T_1}\right) \left(\frac{T_4}{T_3}\right) \left(\frac{T_6}{T_5}\right) \left. \begin{array}{l} 3 \text{ stages} \\ \text{so} \\ -ve \dots \end{array} \right\}$$

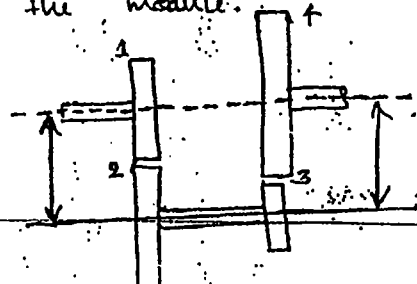
Even no. of stages $\rightarrow$ +ve vel ratio
 Odd no. of stages $\rightarrow$ -ve vel ratio

- With lesser no. of gears, we get larger VR's in case of compound G.T's

RIVETED G.T's

In a compound G.T., if input shaft and output shafts are coaxial, it's called a riveted G.T.

- In a SGT, all gears should have a common module
- In a CGT, 1 & 2, 3 & 4 etc should have same module {i.e. meshed gears should have same module}
- High speed gear has tangential force smaller the module.



$$r_1 + r_2 = r_3 + r_4$$

$$\frac{m_1(T_1 + T_2)}{2} = \frac{m_2(T_3 + T_4)}{2}$$

$$VR = \left(\frac{T_2}{T_1}\right) \left(\frac{T_4}{T_3}\right)$$

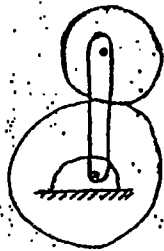
In a compound G.T. meshed gears should have the same module.
In Planetary too the gears can have different modules.

CYCLOIDAL GEAR TRAINS

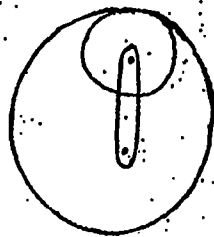
In any of the gear trains (simple, compound or planetary), if one of the axes is floating, it is called a Cycloidal G.T.

- So, any type of G.T. can be converted to a cycloidal type if one of the axes is floating / moving (not fixed).

EPICYCLOIDAL GEAR TRAIN :-



HYPOCYCLOIDAL GEAR TRAIN :-



Epicyclic & Hypocycloidal gear trains have been together called as the CYCLOIDAL G.T's.



$$\frac{N_{ga}}{N_{pa}} = -\frac{T_p}{T_s}$$

$$N_s \approx N_a + N_{sa} \rightarrow N_{sa} \approx N_s - N_a$$

$$N_p = N_a + N_{pa} \rightarrow N_{pa} = N_p - N_a$$

$$\frac{N_g - N_a}{N_p - N_a} = \frac{T_p}{T_s}$$

$$\left. \begin{array}{l} N_{Ra} \\ N_{Rr} = N_R - N_a \end{array} \right\}$$

For one more element $\frac{1}{2}$

$$\frac{N_p - N_a}{N_p + N_a} = \frac{T_R}{T_P}$$

Spring gear

$$\frac{N_{Pa}}{N_{Ra}} = \frac{T_R}{T_P}$$

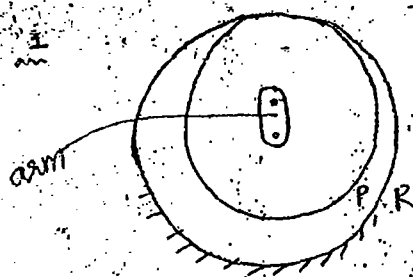
TABULAR Method:

S No.	Action	Arm	$\frac{S}{(T_s)}$	$\frac{P}{(T_p)}$	$\frac{R}{(T_r)}$
1)	Arm sliding H to S	0	+1	$-\frac{T_s}{T_p}$	$-\frac{T_s}{T_r}$
2)	Rel vel of wheel (unknown) = x	0	+x	$-\frac{T_s x}{T_p}$	$-\frac{T_s x}{T_r}$
} so specifying the rel vel of all the wheels in terms of unknown rel wheel {.					
3)	+4 to arm	y	x+4	$-\frac{T_s}{T_p} (x+4)$	$-\frac{T_s}{T_r} (x+4)$

{ so specifying the set of all the wheels in terms of unknown wheel }.

so at the end we get the absolute vel.

pg 23 { Numericals - GEAR TRAINS }



another view of the same arrangement.

$$N_1 = 0$$

$$T_1 = 104$$

$$T_2 = 96$$

$$N_2 = ?$$

$$N_3 = 60$$

N_p can be got directly, so we take the vel. rel to arm,

unknown So $N_p = N_a$ is taken.

$$\frac{N_p - N_a}{N_R - N_a} = \frac{T_R}{T_P}$$

$\begin{matrix} \nearrow 80 \\ \searrow 104 \end{matrix}$
 $\begin{matrix} \nearrow 96 \\ \searrow 60 \end{matrix}$

$$N_p = \text{---} \quad \{ N_p \text{ is the only unknown?}$$

$$\text{so } \frac{N_2 - N_3}{N_1 - N_3} = \frac{T_1}{T_2}$$

$$N_5 = 100, N_R = 0$$

$$T_5 = 20, T_P = 30, T_R = 80$$

$$R_S + 2R_P = R_R \quad \left\{ \begin{array}{l} \text{rad of} \\ \text{alg gear} \end{array} \right.$$

$$\text{so, } T_5 + 2T_P = T_R$$

$$\frac{N_5 - N_a}{N_R - N_a} = \frac{T_R}{T_5} \quad \text{--- (1)}$$

OR

we have $\frac{N_S - N_A}{N_P - N_A} = -\frac{T_P}{T_S}$ & $\frac{N_P - N_A}{N_P - N_A} = \frac{T_P}{T_P}$

xply the 2 eqns,

we get $\frac{N_S - N_A}{N_P - N_A} = -\frac{T_P}{T_S}$ } same as eqn

so $\frac{N_S - N_A}{N_P - N_A} = -\frac{T_P}{T_S}$

$\frac{100 - N_A}{0 - N_A} = -\frac{80}{20}$

$N_A = 20 \text{ rpm}$

3. ω/p & ω/p shafts are collinear $\rightarrow$ so RIVETED

Overall gear ratio = 12

first stage gear ratio = $g_1 = 4$

second " " " = $g_2 = 3$

1st module, $m_1 = 3 \text{ mm}$

2nd module, $m_2 = 4 \text{ mm}$

1st stage $\left\{ \begin{array}{l} z_1 = 16 \\ m = 3 \text{ mm} \\ g_1 = 4 \end{array} \right.$ $\left\{ \begin{array}{l} z_3 = 15 \\ m = 4 \\ g = ? \end{array} \right.$ 2nd stage

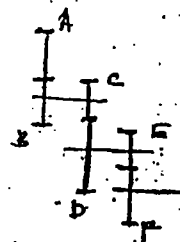
$$z_3 + z_4 = \frac{m_1 z_1}{2} + \frac{m_2 z_2}{2}$$

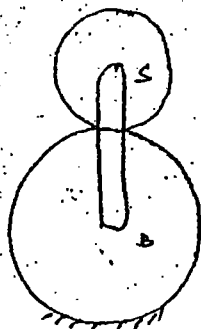
$$= \frac{3}{2} [15 + 45] = 90$$

POSSIBLE extension
 $m_1(z_1 + z_2) = m_2(z_3 + z_4)$
 so find module of the 2nd stage

5 b)
 6 b)

7 $\frac{N_F}{N_A} = \left(-\frac{T_A}{T_B} \right) \left(-\frac{T_C}{T_D} \right) \left(-\frac{T_E}{T_F} \right) = -\frac{T_A T_C T_E}{T_B T_D T_F} //$





$$N_B = 0$$

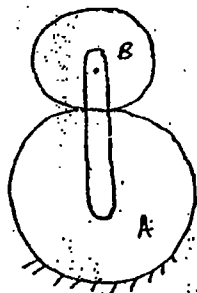
$$N_A = 1$$

$$N_S = ?$$

$$\frac{N_S - N_A}{N_B - N_A} = -\frac{T_B}{T_S}$$

$$\frac{N_S - 1}{0 - 1} = \frac{-100}{25}$$

$$N_S = 5 //$$



As wheel A is fixed,

$$N_A = 0$$

$$T_A = 40$$

$$T_B = 20$$

$$N_B = ?$$

$$N_B = ?$$

$$\frac{N_B - N_A}{N_A - N_A} = -\frac{T_A}{T_B}$$

$$\frac{N_B - 0}{0 - 0} = \frac{-40}{20}$$

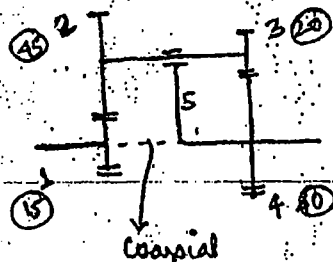
$$N_B = 9 //$$

10 (a)

die, inc

13.

As a riveted gear train

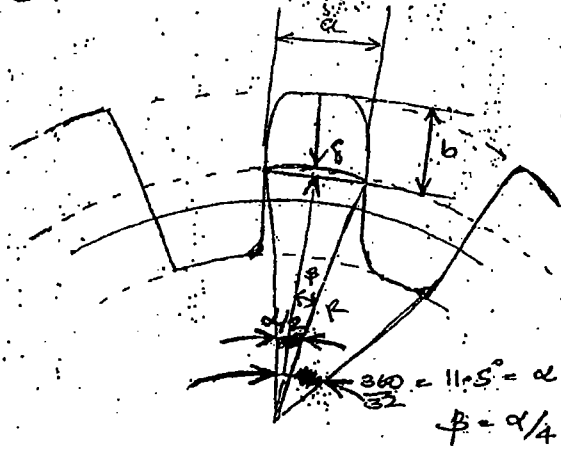


with 5, its made a cycloidal Gear train.

$$\frac{\omega_1 - \omega_5}{\omega_4 - \omega_5} = \frac{T_2}{T_1} \times \frac{T_4}{T_3} = \frac{45}{15} \times \frac{40}{20} = 6$$

15 (a) P-2, R-1

16



$$b = m + s$$

$$a =$$

$$\left. \begin{array}{l} 32 = T \\ 4 = m \end{array} \right\}$$

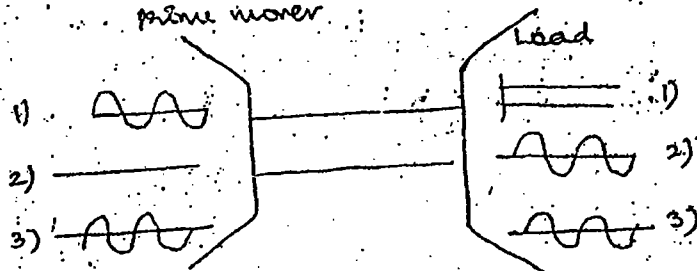
$$a = 2R \sin \beta$$

$$s = R - R \cos \beta$$

FLYWHEELS

- Reservoir of energy to decrease the fluctuations
- Useful in IC engines.
- Even out the fluctuations.

prime mover



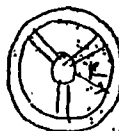
Case 1: fluctuating i/p & uniform o/p

Case 2: uniform i/p & fluctuating o/p

Case 3: fluctuating i/p & fluctuating o/p



$$I = \frac{MR^2}{2}$$



$$I = MR^2$$

$$\sigma_c = \frac{Jv^2}{R} = PR\omega^2$$

centrifugal stress.

ω_1 - max

ω_2 - min

$$\omega = \frac{\omega_1 + \omega_2}{2}$$

(Mean)

$$E_1 = \frac{1}{2} I \omega_1^2$$

$$E_2 = \frac{1}{2} I \omega_2^2$$

$$\Delta E = E_1 - E_2 = \frac{1}{2} I (\omega_1^2 - \omega_2^2)$$

$$\Delta E = \frac{1}{2} I (\omega_1 + \omega_2) (\omega_1 - \omega_2)$$

$$[\Delta E = I \omega \Delta \omega]$$

inclined

$$\Delta E = I \omega^2 \frac{\Delta \omega}{\omega} = I \omega^2 C_s \rightarrow \text{coeff of fluct of speed}$$

$$\Delta \omega = \omega_1 - \omega_2$$

$$C_s = \frac{\Delta \omega}{\omega} = \frac{\omega_1 - \omega_2}{\omega} = \frac{2(\omega_1 - \omega_2)}{\omega_1 + \omega_2}$$

$$\Delta E = I \omega^2 C_s$$

$$\Delta E = 2 E C_s$$

$$\frac{\Delta E}{E} = 2 C_s$$

$$[C_e = 2 C_s \rightarrow \text{coeff of fluctuation of energy}]$$

slp

slp

pg 18 [NUMERICALS - FLYWHEELS]

$$I \quad 80 \text{ kW at } 300 \text{ rpm} \rightarrow \omega = \frac{2\pi \times 300}{60}$$

$$\Delta E = 0.4 E \text{ per cycle}$$

$$C_s = 0.02$$

$$\sigma_c = 6 \times 10^6 \text{ N/m}^2, \quad \rho = 7500 \text{ kg/m}^3$$

$$d = ?$$

$$I = ?$$

$$\sigma_c = \rho R \omega^2$$

$$R = \frac{\sigma_c}{\rho \omega^2}$$

$$\text{diam} = 2R$$

$$\Delta E = I \omega^2 C_s \Rightarrow I = \frac{\Delta E}{\omega^2 C_s}$$

$$\text{Rate} = 80000 \text{ J/sec}$$

$$300 \text{ rpm} \rightarrow 5 \text{ rev/sec}$$

$$0.2 \text{ sec/rev}$$

$$1 \text{ cycle} = 2\pi \text{ rev} = 0.4 \text{ sec}$$

$$E \text{ per cycle} = 80000 \times 0.4 = 32000 \text{ J}$$

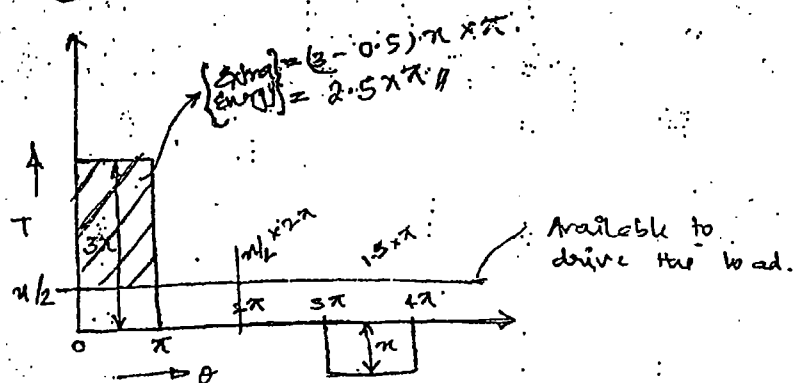
$$\Delta E = 0.9 \times (E \text{ per cycle})$$

$$\therefore \Delta E = 0.9 \times (32000)$$

$$I = \frac{\Delta E}{\omega^2 C_s} //$$

QUESTN CHANGES to make calculation easier:

Q: 20 kW at 240 rpm
 TM $\rightarrow$ turning moment
 TM is 3 times that of compression strokes
 Avg speed of 240 rpm



20 kW @ 240 rpm

$$C_s = 0.01$$

$$I = ?$$

work done / cycle = { Net Area under the }
 or { Turning moment }
 energy developed per cycle { diagram }

$$= 3\pi n - \pi n = 2\pi n$$

$$= \text{Mean torque} \times 4\pi = 2\pi n$$

$$(T_m)$$

$$T_m = \frac{2\pi n}{4\pi} = n/2 //$$

$$\Delta E = 2.5 \pi \text{ Joules}$$

$$= 2.5 \times \frac{5000}{\pi} \pi \text{ J.}$$

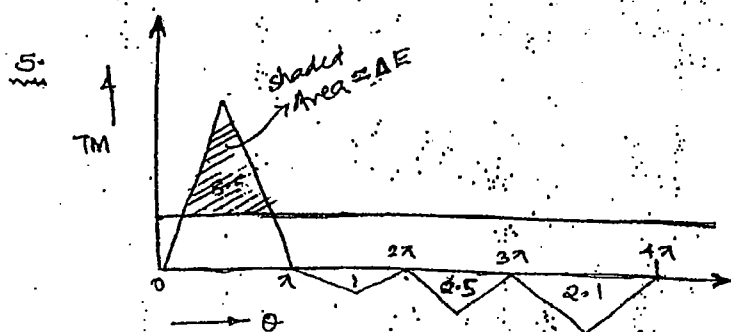
$$I = \frac{\Delta E}{\omega^2 C_s}$$

$$20 \text{ kW @ } 2400 \text{ rpm} \rightarrow 48 \text{ Hz} \rightarrow 0.25 \text{ rpm}$$

$$1 \text{ cycle} = 2 \pi \text{ rad} = 0.5 \text{ sec.}$$

$$2 \pi \text{ rad} = 10000 \text{ J}$$

$$\eta = \frac{5000}{\pi}$$



$$1 \text{ cm}^2 = 1400 \text{ J}$$

$$q_e = (10.2) \text{ rpm}$$

$$N = 100 \text{ rpm}$$

$$C_s = 0.04$$

$$I = ?$$

$$I = \frac{2 \Delta E}{\omega_1^2 - \omega_2^2}$$

$$\Delta E = -$$

$$\text{Energy done per cycle (WDPC)} = 8.5 - 0.5 - 0.8 - 2.1$$

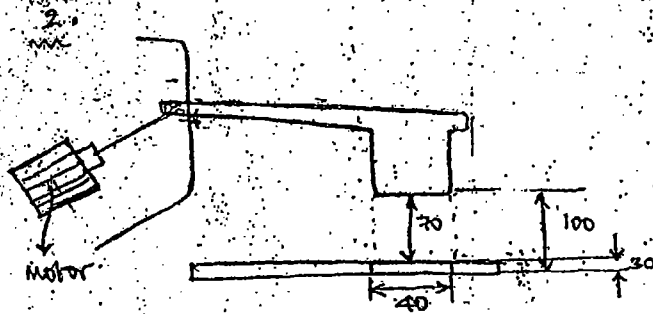
$$= 5.1 \text{ cm}^2/\text{r}$$

$$T_m = \frac{\text{WDPC}}{4\pi} = \frac{5.1}{4\pi}$$

$$1 \text{ cm} \times 1 \text{ cm} = 1400 \text{ J}$$

$$1.2 \text{ V} \quad 1400 \text{ Nm}$$

$$\Delta E = \text{cm}^2 \times 1400 = \text{J}$$



Punch comes down & goes up $\rightarrow$ one cycle

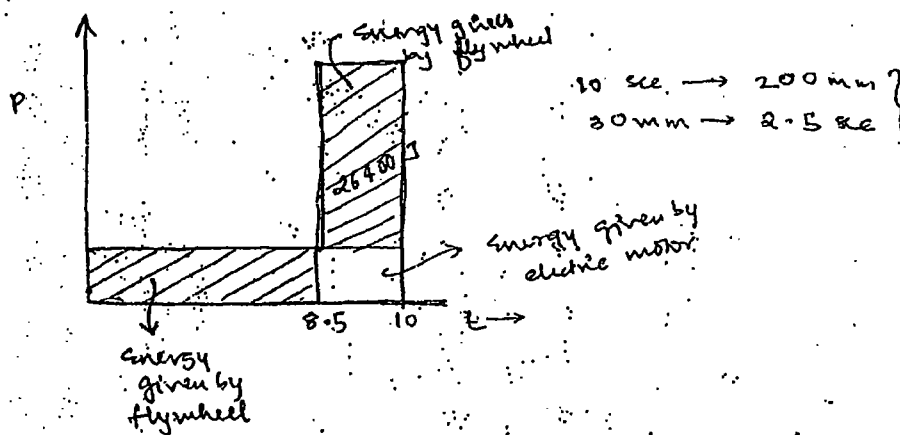
$$\text{Energy per hole} = \pi d t$$

$$= 26400 \text{ J}$$

$$\text{Energy per sec} = \frac{26400}{10} = 2640 \text{ J/s}$$

$$= 2640 \text{ W}$$

$$= 2.64 \text{ kW}$$



The 2 shaded areas must be equal.

$$\Delta E = 8.5 (2640)$$

$$26400 - (15) (2640)$$

$$C_s = 0.03$$

$$V = 25 \text{ m/s} = \text{KW}$$

$$\underline{\underline{M}}$$

$$\Delta E = I \omega^2 C_s$$

$$= M R^2 \omega^2 C_s$$

$$= M V^2 C_s$$

$$\Delta E = m v^2 c_s$$

$$m = \frac{\Delta E}{v^2 c_s}$$

Out of 10 secs, 8.5 secs energy is given by the flywheel.

3.

~~over energy~~

600 hls / hr

1 hls in 6 secs $\rightarrow$ Total cycle time $\} = 6 \text{ secs}$

1.5 secs $\downarrow$ NO WORK $\quad \quad \quad$ 4.5 secs $\downarrow$ 10000 Nm req. $\quad \quad \quad \} 1.5 + 4.5 = 6 //$

$2000 \times 4.5 =$ stored in flywheel.

Last 1.5 s $\rightarrow$ punching operation.

3000 J given by motor during 1.5 secs $\}$
7000 to be given by flywheel $\}$

so...
7000 &
9000
are both
incorrect

Quesn: whether to take 7000 or 9000 for the flywheel.

$\frac{10}{6} \times 4.5 =$ energy stored in flywheel.

In last 1.5 secs $\rightarrow$ $\begin{cases} 7.5 \text{ given by} \\ \text{flywheel.} \end{cases}$

$\begin{cases} 1.5 \text{ given by the} \\ \text{motor} \end{cases}$

whatever is req is delivered by the motor in 6 secs...

6.

Diag gives T.M req to drive the load
net area = energy for operation.

Q11 (Addtl Question)

300 kW at 90 rpm.

Cycle time = $\frac{4}{3}$ sec

$$EPC = \frac{4}{3} \times 300 = 400 \text{ kJ}$$

$$\text{out of this } \Delta E = 0.6 \times 400 = 240 \text{ kJ}$$

Energy per cycle //

Ans: (a)

Q11

Ans: d)

Mech Energy $\rightarrow$ Mech speed.

Q12

(d) Negative during major portion.

Approaching dead centre, deceleration.

Q12 (d) Net area $\times$ scale factor } 3rd option.

Q13

$T_1 = T_2$ and $A_1 = A_2$ } correct ans

NORMS

Q14 (a)

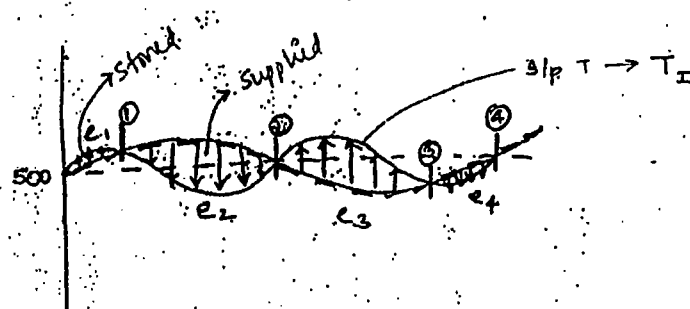
Area of $\Delta =$ Area of rect.

Q7

Load torque $= T_L = 500 + 50 \sin \theta$

(C)

input torque $= T_I = 500 + 50 \sin \theta$



$e_1 \rightarrow$ 3/p more, energy stored

$e_2 \rightarrow$ 3/p more, flywheel energy delivered

$e_3 \rightarrow$ 3/p more, energy stored

ex → flywheel gives out energy

$$T_L = 500 + 50 \sin \theta$$

$$T_E = 500 + 50 \sin 2\theta$$

$$\Delta T = T_{in} - T_{out}$$

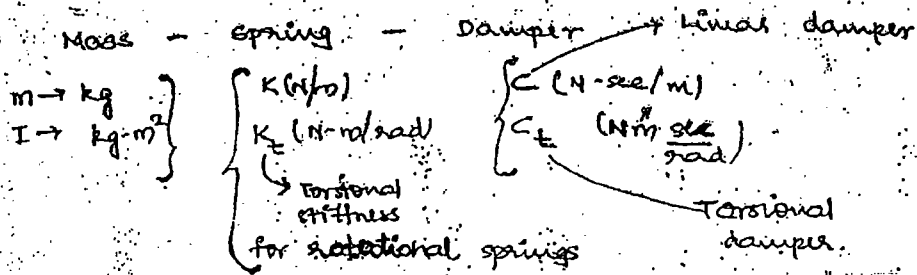
$$\int \Delta T d\theta$$

$$\rightarrow \Delta T = I \alpha$$

At pts ①, ②, ③, ④ value of torque of m/c and the engine will be equal.

{The starting and final pt can be taken as a single point.}

MECHANICAL VIBRATIONS



EQUILIBRIUM METHOD

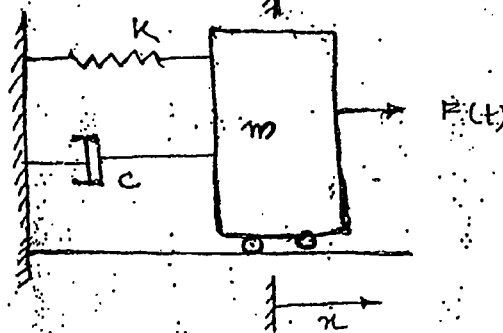
{ Newton-Euler method :- }

$$F = m \cdot a$$

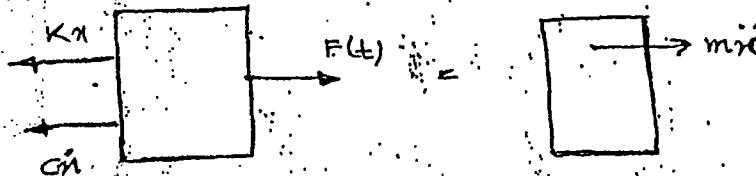
$$M = I \alpha$$

$\ddot{x} \rightarrow \text{accn}$
 $\dot{x} \rightarrow \text{vel}$
 $x \rightarrow \text{disp}$

{ (SEP) STATIC EQUILIB POSITION }



Freebody diag :



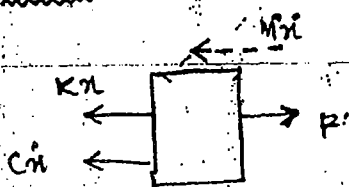
$$-Kx - C\dot{x} + F(t) = m\ddot{x}$$

General form

$$m\ddot{x} + C\dot{x} + Kx = F(t)$$

External force

mathematical eqn
- representation of vibration



• SPECIFIC EQN

{w/o any external force}

$$m\ddot{x} + c\dot{x} + Kx = 0 \quad \text{Free vibration}$$

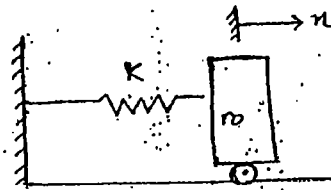
↳ Damped vibration

$$\rightarrow m\ddot{x} + Kx = 0$$

↳ Undamped vibration

Undamped free vibrations {No Friction}

UNDAMPED FREE VIBRATIONS



$$m\ddot{x} + Kx = 0$$

$$x(0) = x_0$$

$$\dot{x}(0) = v_0$$

Initial conditions

std form: $\ddot{x} + \frac{K}{m}x = 0$

$$\ddot{x} + \omega_n^2 x = 0$$

$$\left\{ \omega_n^2 = \frac{K}{m} \right\}$$

$$D^2 + \omega_n^2 = 0$$

$$D = \pm i\omega_n$$

$$\frac{K}{m} = \omega_n^2$$

$$x(t) = A \cos \omega_n t + B \sin \omega_n t$$

$$x(0) = A = x_0$$

$$\dot{x}(0) = B\omega_n = v_0$$

$$\therefore A = x_0$$

$$B = \frac{v_0}{\omega_n}$$

$$x(t) = x_0 \cos \omega_n t + \frac{v_0}{\omega_n} \sin \omega_n t$$

$$A = X \cos \phi$$

$$B = X \sin \phi$$

$$X = \sqrt{A^2 + B^2} = \sqrt{x_0^2 + \left(\frac{v_0}{\omega_n}\right)^2}$$

$$\phi = \tan^{-1} B/A = \tan^{-1} \frac{v_0}{x_0 \omega_n}$$

$$x(t) = X \cos(\omega_n t - \phi)$$

ω_n = natural freq of system

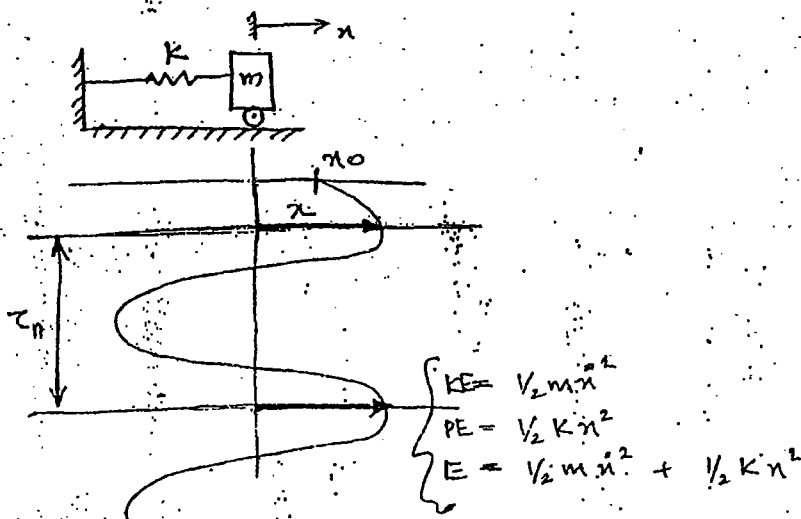
freq of system depends on K and m

$$T_n = \frac{2\pi}{\omega_n} = 2\pi \sqrt{m/K}$$

↳ Natural time period

T_n : time period

↳ depends on m and K only.



$$E = KE + PE = \text{const}$$

$\begin{matrix} 0 & PE_{\max} \\ KE_{\max} & 0 \end{matrix}$

$$\text{So } \frac{dE}{dt} = 0$$

$$\text{So } E = \frac{1}{2} m \dot{x}^2 + \frac{1}{2} K x^2$$

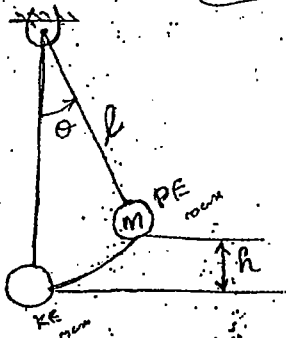
$$\text{So } \frac{dE}{dt} = \frac{1}{2} m 2\dot{x}\ddot{x} + \frac{1}{2} K 2x\dot{x} = 0$$

$$(m\ddot{x} + Kx)\dot{x} = 0$$

$\dot{x}$ cannot be equal to 0

$$\text{Hence } m\ddot{x} + Kx = 0$$

SIMPLE PENDULUM



$$K = \frac{1}{2} m l^2 \dot{\theta}^2$$

$$P = mgh$$

$$P = mgl (1 - \cos \theta)$$

$$\{h = l(1 - \cos \theta)\}$$

1.25 + 1.58 + 1.43 = 4.26

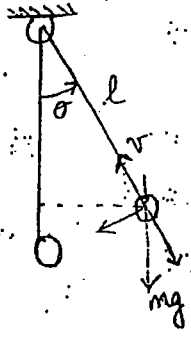
$$E = \frac{1}{2} m l^2 \dot{\theta}^2 + mgl (1 - \cos \theta)$$

$$\frac{dE}{dt} = \frac{1}{2} m l^2 2 \dot{\theta} \ddot{\theta} + mgl (\sin \theta) \dot{\theta} = 0$$

$$(ml^2) \ddot{\theta} + mgl \sin \theta = 0$$

$$\omega_n = \sqrt{\frac{K_{eq}}{m_{eq}}} = \sqrt{\frac{mgl}{ml^2}} = \sqrt{g/l}$$

- ENERGY METHOD:



Mean position $\rightarrow$ Net force $= 0$
Any other position $\rightarrow$ $mg \sin \theta$

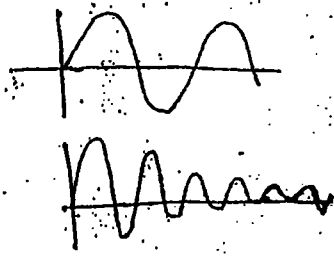
$$ml^2 \ddot{\theta} + mgl \sin \theta = 0$$

$$\therefore \text{As } -mgl \sin \theta = I \ddot{\theta} = ml^2 \ddot{\theta}$$

DAMPED FREE VIBRATION

$\tau_d > \tau_0 \rightarrow$ time for natural oscillations
 $\rightarrow$ time for damped oscillations

$$\omega_d < \omega_0$$



- If system is disturbed, it comes back to the original system: critical damping.

$$C_c = 2\sqrt{km}$$

- $C < C_c \rightarrow$ Under damped system.
- $C = C_c$ with critical damping system.
- $C > C_c \rightarrow$ Over damped system.

$$\zeta = \frac{C}{C_c} = \frac{C}{2\sqrt{km}}$$

DAMPED FREE VIBRATIONS

$$m\ddot{x} + C\dot{x} + Kx = 0.$$

$$x(0) = x_0, \quad \dot{x}(0) = v_0.$$

$$\ddot{x} + \frac{C}{m}\dot{x} + \frac{K}{m}x = 0.$$

$$\frac{K}{m} = \omega_n^2$$

$$\frac{C}{m} = \frac{C}{2\sqrt{K}} \frac{2}{\sqrt{m}} \frac{\sqrt{K}}{\sqrt{m}} = \frac{2C}{2\sqrt{Km}} \sqrt{\frac{K}{m}} = 2\zeta\omega_n$$

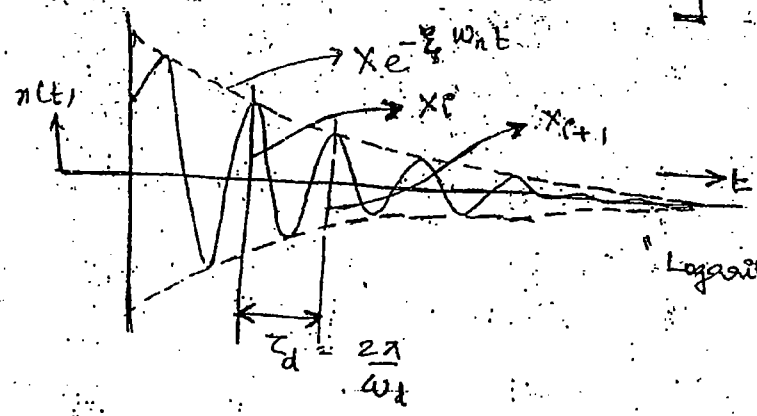
$$\ddot{x} + 2\zeta\omega_n\dot{x} + \omega_n^2x = 0$$

$$x(t) = e^{-\zeta\omega_n t} [A \cos \omega_d t + B \sin \omega_d t]$$

$$\omega_d = \sqrt{1 - \zeta^2} \omega_n$$

K. 10
g.

$$[x(t) = X e^{-\zeta \omega_n t} \cos(\omega_d t - \phi)]$$



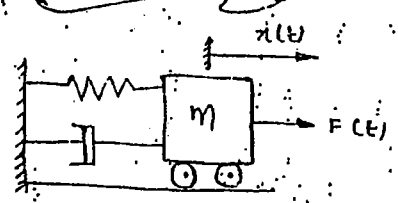
δ = logarithmic decrease or log-dec.

* $\rightarrow \delta = \ln \frac{X_i}{X_{i+1}}$ } logarithmic decrement.

$$\delta = \frac{1}{n} \ln \frac{X_i}{X_{i+n}}$$

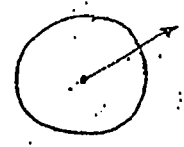
* $\rightarrow \zeta = \frac{\delta}{\sqrt{4\pi^2 + \delta^2}}$

FORCED VIBRATIONS:



$$m\ddot{x} + c\dot{x} + kx = F(t)$$

$\swarrow F \cos \omega t$
 $\searrow F \sin \omega t$



$m = 10 \text{ kg}$
 $e = 1 \text{ cm}$

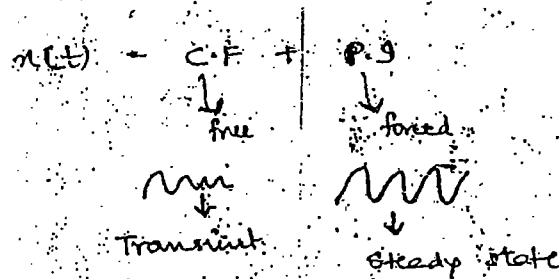
$N = 60,000 \text{ rpm}$

$\omega = \frac{2\pi N}{60} \approx 4000 \pi \text{ rad/sec}$

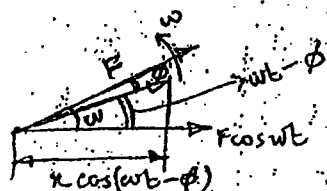
Force due to vibration

$$m e \omega^2 = 10 \times 10^{-2} \times (4000\pi)^2$$

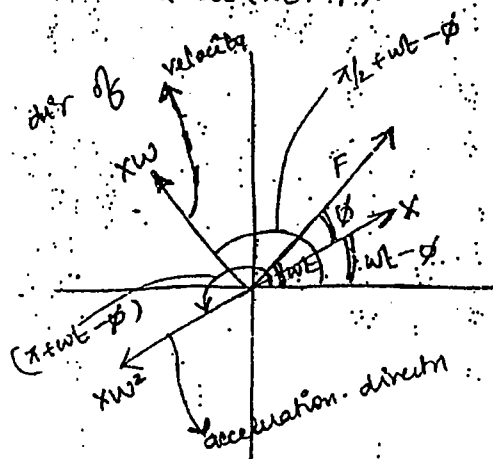
$$= 16000000 \text{ N}$$



$$x(t) = X \cos(\omega t - \phi)$$



For force at an angle $\omega t - \phi$, the magnitude of horiz component is $X \cos(\omega t - \phi)$.



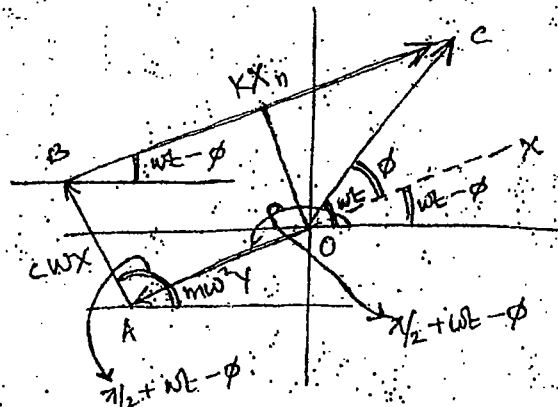
$$x = X \cos(\omega t - \phi)$$

$$\dot{x} = -X\omega \sin(\omega t - \phi) = X\omega \cos(\pi/2 + \omega t - \phi)$$

$$\ddot{x} = -X\omega^2 \cos(\omega t - \phi) = X\omega^2 \cos(\pi + \omega t - \phi)$$

$$m\ddot{x} + c\dot{x} + kx = F \cos(\omega t - \phi)$$

$$m\omega^2 X \cos(\pi + \omega t - \phi) + c\omega X \cos(\pi/2 + \omega t - \phi) + kX \cos(\omega t - \phi) = F \cos \omega t$$



Rotation vector diag that represents forced vibration
All vectors rotate with an angular vel ω

$\triangle ODC \rightarrow$ right angled $\triangle$

$$DC^2 + OD^2 = OC^2$$

$$(BC - BD)^2 + (OD)^2 = OC^2$$

$$BD = AD,$$

$$(KX - m\omega^2 X)^2 + (c\omega X)^2 = F^2$$

$$X = \frac{F}{\sqrt{(K - m\omega^2)^2 + (c\omega)^2}}$$

$$\phi = \tan^{-1} \left(\frac{c\omega}{K - m\omega^2} \right)$$

$$K - m\omega^2 = K \left(1 - \frac{m}{K} \omega^2 \right)$$

$$= K \left(1 - \left(\frac{\omega}{\omega_n} \right)^2 \right)$$

$$c\omega = K \left[\frac{c\omega}{K} \right]$$

$$= K \left[\frac{c\omega}{m\omega_n^2} \right] = K \left[\frac{c}{m} \left(\frac{\omega}{\omega_n} \right)^2 \right]$$

$$\frac{F}{k} = X_{st}$$

$$X = F/k$$

$$\sqrt{\left(1 - \left(\frac{\omega}{\omega_n}\right)^2\right)^2 + \left(2\zeta \frac{\omega}{\omega_n}\right)^2}$$

$$X = \frac{X_{st}}{\sqrt{\left(1 - \left(\frac{\omega}{\omega_n}\right)^2\right)^2 + \left(2\zeta \frac{\omega}{\omega_n}\right)^2}}$$

$$\sqrt{\left(1 - \left(\frac{\omega}{\omega_n}\right)^2\right)^2 + \left(2\zeta \frac{\omega}{\omega_n}\right)^2}$$

non dimensionalised eqns

$$\frac{X}{X_{st}} = \frac{1}{\sqrt{\left(1 - \left(\frac{\omega}{\omega_n}\right)^2\right)^2 + \left(2\zeta \frac{\omega}{\omega_n}\right)^2}}$$

X static

$$MF = \frac{1}{\sqrt{\left(1 - \left(\frac{\omega}{\omega_n}\right)^2\right)^2 + \left(2\zeta \frac{\omega}{\omega_n}\right)^2}}$$

$$\sqrt{\left(1 - \left(\frac{\omega}{\omega_n}\right)^2\right)^2 + \left(2\zeta \frac{\omega}{\omega_n}\right)^2}$$

$$\rightarrow X = X_{st} (MF)$$

Magnification factor

Freq response fn (FRF)

- Freq ratio & damping ratio decides the amplitude of the vibration.

$$\text{Also, } \phi = \tan^{-1} \frac{c\omega}{k - m\omega^2} = \tan^{-1} \frac{2\zeta \omega/\omega_n}{1 - (\omega/\omega_n)^2}$$

- MF is a fn of freq ratio and damping ratio.

$$MF = \frac{1}{\sqrt{\left(1 - \left(\frac{\omega}{\omega_n}\right)^2\right)^2 + \left(2\zeta \frac{\omega}{\omega_n}\right)^2}}$$

small force leading to large amplitude, $MF = \infty$ } Resonance

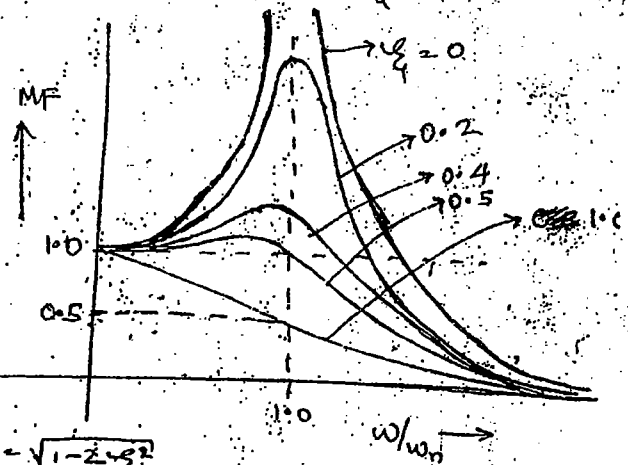
$\rightarrow$ For undamped system, $\zeta = 0$, $MF = \infty$

When there is damping

$$\frac{\omega}{\omega_n} = 1$$

$$MF = \frac{1}{2\zeta}$$

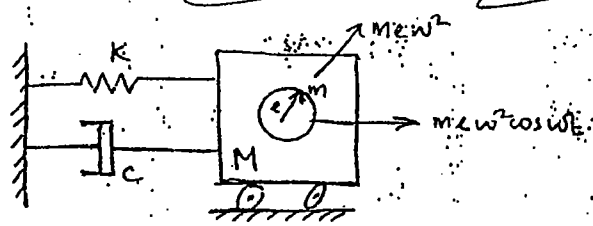
$\zeta = \text{finite value}$
 $\zeta \neq 0$



MF is max at: $\frac{\omega}{\omega_n} = \sqrt{1 - 2\zeta^2}$

$\phi = 0^\circ$	$\omega/\omega_n < 1$
$\phi = 90^\circ$	$\omega/\omega_n = 1$
$\phi = 180^\circ$	$\omega/\omega_n > 1$

ROTATING OR RECIPROCATING UNBALANCES



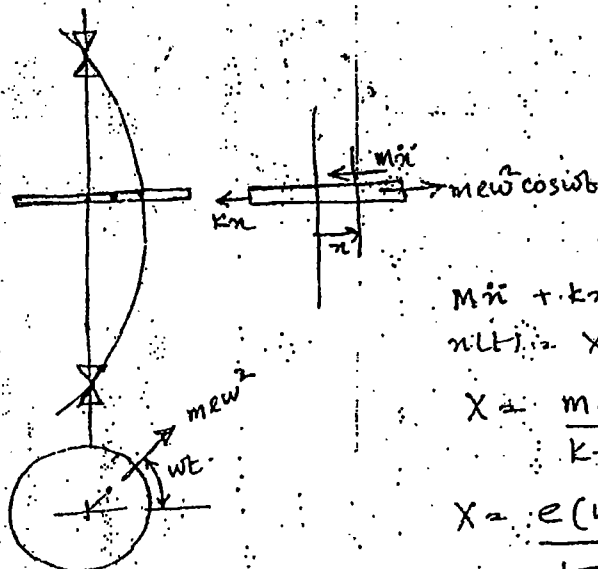
$$M\ddot{x} + C\dot{x} + Kx = m_e w^2 \cos wt$$

$$x(t) = X \cos(\omega t - \phi)$$

$$X = \frac{m_e w^2}{\sqrt{(K - M\omega^2)^2 + (C\omega)^2}}$$

WHIRLING OF SHAFTS / CRITICAL SPEEDS

- shaft rotates about its own axis
- shaft which is deformed rotates about the unbend axis $\rightarrow$ it called WHIRLING



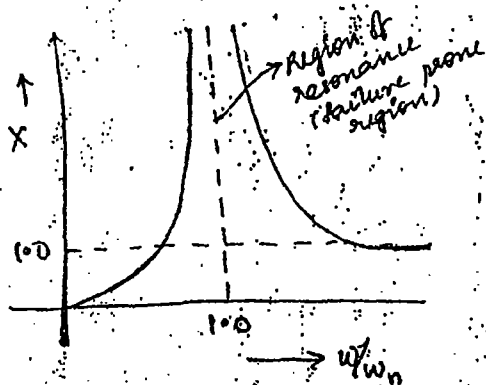
$$m\ddot{x} + kx = m\omega^2 x \cos \omega t$$

$$x(t) = X \cos(\omega t - \phi)$$

$$X = \frac{m\omega^2 r}{k - m\omega^2}$$

$$X = \frac{e(\omega/\omega_n)^2}{1 - (\omega/\omega_n)^2}$$

X = whirling radius or amplitude of whirling.

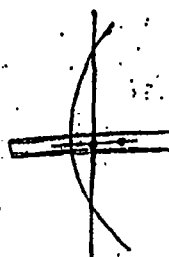


Critical speed $\omega = \omega_n = \sqrt{\frac{k}{m}}$

$$X = \frac{e(\omega/\omega_n)^2}{1 - (\omega/\omega_n)^2}$$

when $\omega/\omega_n = \infty$

$$X = \frac{e(\infty)^2}{1 - (\infty)^2} = -e$$



Shaft bends off to the direction of force

(84)

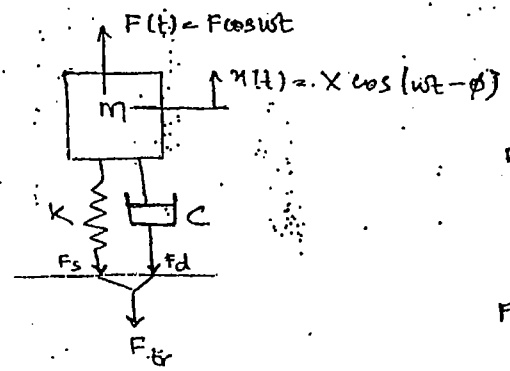
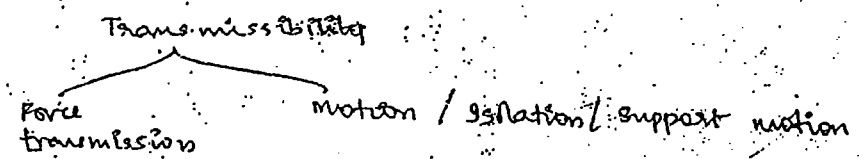
At high speeds it deflects backwards and it rotates around its centre of gravity.

If it cant bend $\rightarrow$ it will undergo failure.

It brings back the CG on its centre of rotation $\rightarrow$ so escapes the failure.

$\rightarrow$ coasting up and coasting down procedure to avoid shaft failure while crossing the resonance region / critical speed region.

TRANSMISSIBILITY



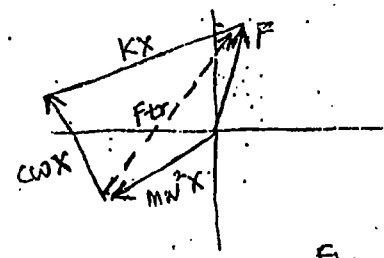
$$F_{tr} = F_s + F_d$$

$\downarrow$ spring force $\downarrow$ damping force

$$F_{tr} = \sqrt{(KX)^2 + (C\omega X)^2}$$

$$F_{tr} = X \sqrt{K^2 + (C\omega)^2}$$

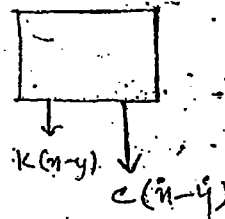
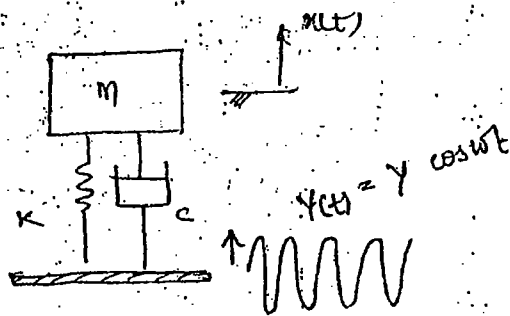
$$F_{tr} = \frac{\sqrt{K^2 + (C\omega)^2}}{\sqrt{(K - M\omega^2)^2 + (C\omega)^2}} F$$



$$\frac{F_{tr}}{F} = \frac{\sqrt{K^2 + (C\omega)^2}}{\sqrt{(K - M\omega^2)^2 + (C\omega)^2}}$$

Transmissibility ratio

$$\left\{ \frac{F_{tr}}{F} = \frac{\sqrt{1 + (2\xi \frac{\omega}{\omega_n})^2}}{\sqrt{(1 - (\frac{\omega}{\omega_n})^2)^2 + (2\xi \frac{\omega}{\omega_n})^2}} \right.$$



Both ends in motion
 so rel disp for spring.
 rel vel for damper.

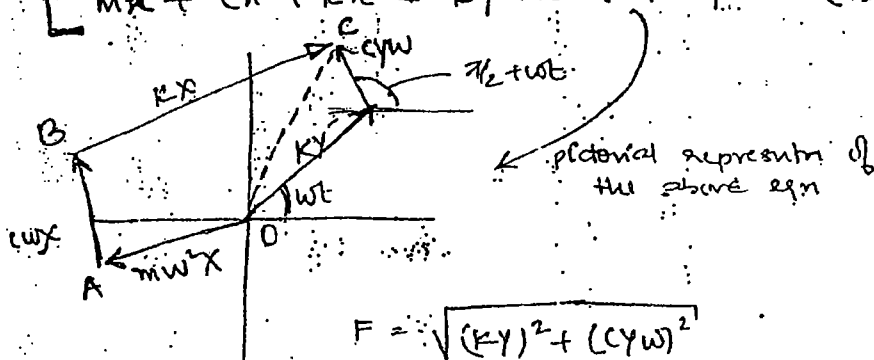
$$m\ddot{x} + c(\dot{x}-\dot{y}) + k(x-y) = 0$$

$$m\ddot{x} + c\dot{x} + kx = ky + c\dot{y}$$

$$y(t) = y \cos \omega t$$

$$\dot{y}(t) = -y\omega \sin \omega t = y\omega \cos(\pi/2 + \omega t)$$

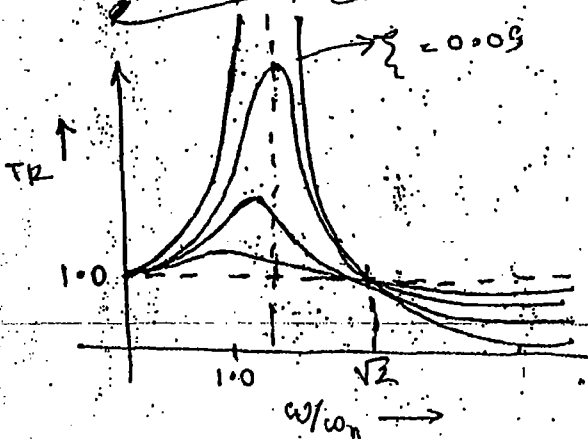
$$[m\ddot{x} + c\dot{x} + kx = ky \cos \omega t + cy\omega \cos(\pi/2 + \omega t)]$$



$$F = \sqrt{(ky)^2 + (cy\omega)^2}$$

$$x = \frac{\sqrt{(ky)^2 + (cy\omega)^2}}{\sqrt{(k - m\omega^2)^2 + (c\omega)^2}}$$

variation of transmissibility ratio



$\frac{w}{w_n} \approx 0$, TR small

TR is max at $\frac{w}{w_n} = 1$

$$0 \leq \frac{w}{w_n} \leq 1$$

$$\left. \begin{array}{l} \frac{TR \uparrow}{TR \downarrow} \quad \frac{w/w_n \uparrow}{\xi \uparrow} \end{array} \right\}$$

$$1 \leq \frac{w}{w_n} \leq \sqrt{2} \quad \left. \begin{array}{l} \frac{TR \downarrow}{TR \downarrow} \quad \frac{w/w_n \uparrow}{\xi \uparrow} \end{array} \right\} \rightarrow TR > 1$$

$$\left. \begin{array}{l} \frac{TR \downarrow}{TR \downarrow} \quad \frac{w/w_n \uparrow}{\xi \uparrow} \end{array} \right\}$$

At $\frac{w}{w_n} = \sqrt{2} \rightarrow TR = 1$ for all ξ

$$\frac{w}{w_n} > \sqrt{2}$$

$$TR < 1$$

$$TR \downarrow \quad w/w_n \uparrow$$

$$TR \uparrow \quad \xi \uparrow$$

$\frac{w}{w_n} \geq 4$ Transmissibility is less, so better isolation

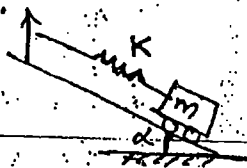
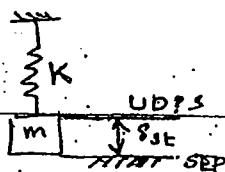
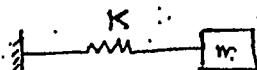
$\frac{w}{w_n} \approx 10$, Good isolation.

STANDARD PROBLEMS

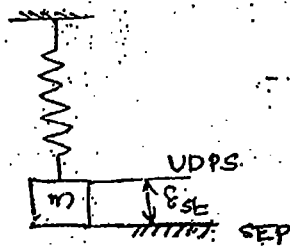
Q1) A spring of stiffness K is cut into $1/2$ & joined in parallel. What is stiffness?

Each piece will have stiffness of $2K$.
Then joining in parallel, resultant stiffness = $4K$

Q2)



By making static Equib position as the initial position, no need to bother about the wt mg , or the deflection or initial elongation.



SEP : static Equib position as reference

UDPS : Undeformed position of spring

δ_{SE} : static deflection due to self weight

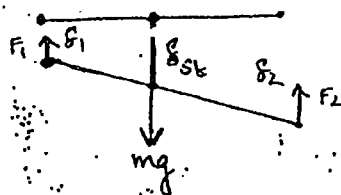
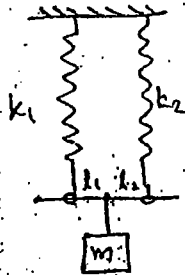
$$\delta_{SE} = \frac{mg}{k}$$

$$\omega_n = \sqrt{\frac{k}{m}} = \sqrt{\frac{g}{\delta_{SE}}}$$

$$\omega_n = \sqrt{g/\delta_{SE}}$$

$$m\ddot{u} + k u = 0$$

Q3)



$$\delta_{SE} = \delta_1 + \frac{(\delta_2 - \delta_1) l_1}{l_1 + l_2}$$

$$\omega_n = \sqrt{g/\delta_{SE}}$$

$$K_{eq} = \frac{mg}{\delta_{SE}}$$

Condition for springs being parallel:

$$\delta_1 = \delta_2$$

$$k_1 l_1 = k_2 l_2$$

otherwise they are not parallel,
it's a specific configuration

the
at
elongate

reference
this

in due

(Q4)



$$F_1 = mg$$

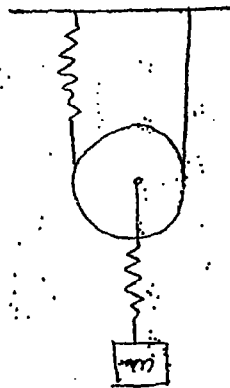
$$F_2 = 2mg$$

$$s_1 = \frac{F_1}{K_1}, \quad s_2 = \frac{F_2}{K_1}$$

$$s_{st} = s_1 + 2s_2$$

$$\omega_n = \sqrt{g/s_{st}}$$

(Q5)



$$F_1 = mg$$

$$F_2 = T = \frac{mg}{2}$$

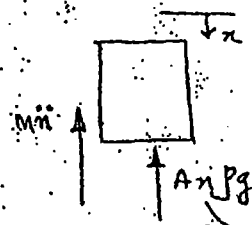
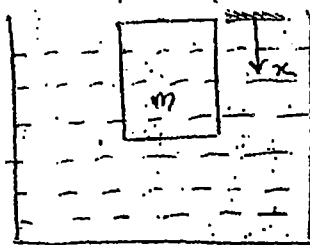
$$s_1 = \frac{mg}{K_1}$$

$$s_2 = \frac{mg}{2K_2}$$

$$s_{st} = s_1 + \frac{s_2}{2}$$

$$\omega_n = \sqrt{g/s_{st}}$$

(Q6) Tub filled with fluid. Block oscillates in fluid. What are the forces acting on the block.



Additional buoyancy force

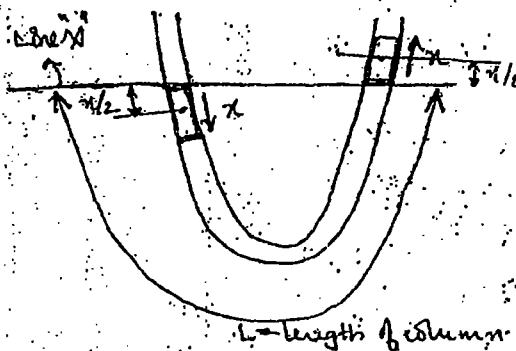
$$m\ddot{x} + A_n \rho g x = 0$$

$$\omega_n = \sqrt{\frac{A_n \rho g}{m}}$$

Q7) U-tube manometer, expose one limb to high pr, release it. Then the column oscillates. Find the freq.

A = Area of c.s

L = length of the column below the line "X"



$$PE = A \Delta p g L = \frac{1}{2} (2 A \Delta p g) L x^2$$

$\downarrow$
 K_{eq}

$$KE = \frac{1}{2} \frac{(A L \rho) \dot{x}^2}{m_{eq}}$$

$$E = ?$$

$$\frac{dE}{dx} = V$$

$$m_{eq} \ddot{x} + K_{eq} x = 0$$

$$A \rho L \ddot{x} + 2 A \rho g L x = 0$$

$$\omega_n = \sqrt{\frac{2 A \rho g L}{A \rho L}} = \sqrt{2g/L}$$

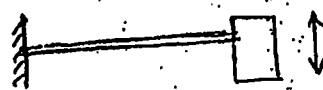
Q8) STRUCTURAL elements as Springs

Mass oscillates along the axis of the structural member $\rightarrow$ Axial vibration.



$$K = \frac{EA}{L} \text{ \& } K = F/\delta$$

Vibration $\perp$ to axis $\rightarrow$ Transverse/bending vibration.



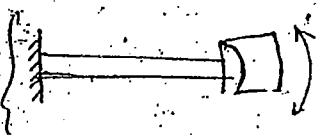
$$\delta = \frac{FL^3}{3EI}$$

$$K = F/\delta$$

$$\therefore K = \frac{3EI}{L^3}$$

$$\omega_n = \sqrt{\frac{K}{I}}$$

Case:
TORSIONAL
VIBRATION

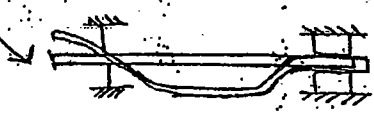


$$\frac{T}{J} = \frac{G\theta}{L}$$

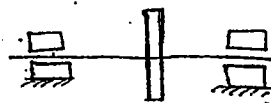
$$K_t = \frac{T}{\theta} = \frac{GJ}{L}$$

Case: SHAFT & ^{LONG} BEARING conditions.
shaft bearing gives a simply supported end condition.

Long bearings gives a fixed end condition.



$$\left\{ \begin{aligned} \delta &= \frac{FL^3}{48EI} \\ K &= \frac{48EI}{L^3} \end{aligned} \right.$$

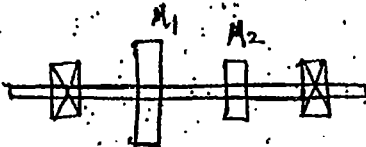


$$\left\{ \begin{aligned} \delta &= \frac{FL^3}{192EI} \\ K &= \frac{192EI}{L^3} \end{aligned} \right.$$

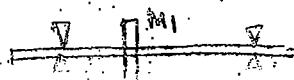
$$W_n = \sqrt{\frac{K}{m}}$$

Q9)

2 degree freedom system $\rightarrow$ 2 free masses.
{lowest freq is called fundamental freq}
 $\downarrow$
Use Dunkerley's empirical method to estimate the lowest of the 2 natural freq's.



$$\frac{1}{W_n^2} = \frac{1}{W_{n1}^2} + \frac{1}{W_{n2}^2} \quad \cdot \quad f_n = \frac{W_n}{2\pi} \rightarrow \text{in Hz}$$

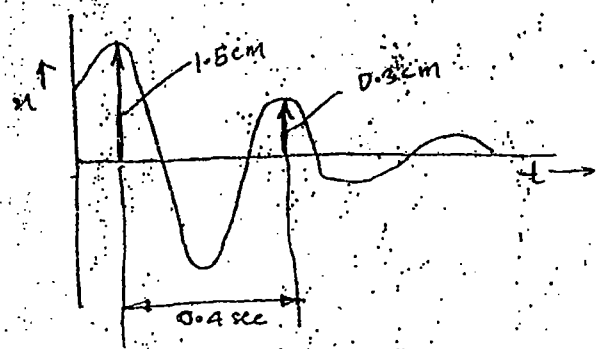


$$\frac{1}{f_n^2} = \frac{1}{f_{n1}^2} + \frac{1}{f_{n2}^2}$$

LINKED
LIVES
TYPE

spring-mass damper system

The free vibration curve for a fixed mass system is as shown in the figure.
If the mass of system = 1 kg, determine
(a) spring constant
(b) damping constant



$x_1 = 1.5 \text{ cm}, x_2 = 0.3 \text{ cm}$

$\delta = \ln \frac{x_1}{x_2}$

$\xi = \frac{\delta}{\sqrt{4\pi^2 + \delta^2}}$

If $\delta < 0.7$, only
then $\xi \approx \frac{\delta}{2\pi}$ approx

$\tau_d = 0.4 \text{ secs}$

$\omega_d = \frac{2\pi}{\tau_d}$

$\omega_n = \frac{\omega_d}{\sqrt{1-\xi^2}}$

$\frac{K}{m} = \omega_n^2$
So, $K = m\omega_n^2$ } Ans (a)
Stiffness calculated.

(b) $C = ?$

$\frac{C}{2m} = \xi$

$C_c = 2\sqrt{Km}$

$C = \xi C_c$

$C = 2\xi\sqrt{Km}$

ser. no.
ed. no.

(11) (29)
Q11) In a spring-mass damper system, the mass completes 5 oscillations in 0.5 sec. During these 5 oscillations, the amplitude decays to 5% of the original value. If the stiffness of the spring is 1 kN/m, determine the mass and the spring constant.

Ans:

$$\tau_d = 0.5 \text{ sec for } 5 \text{ oscillations} = \frac{0.5}{5} = 0.1 \text{ sec}$$

$$\omega_d = \frac{2\pi}{\tau_d} = 20\pi \text{ rad/sec}$$

$$\text{Original amp} = X_0$$

$$\text{After 5 oscillations, Amp} = X_5$$

$$X_0 - X_5 = 0.05$$

$$\delta = \frac{1}{5} \ln \frac{X_0}{X_5} = \frac{1}{5} \ln \frac{X_0}{0.05 X_0}$$

$$= \frac{1}{5} \ln 20$$

$$\zeta = \frac{\delta}{\sqrt{4\pi^2 + \delta^2}}, \quad \omega_n = \frac{\omega_d}{\sqrt{1 - \zeta^2}}$$

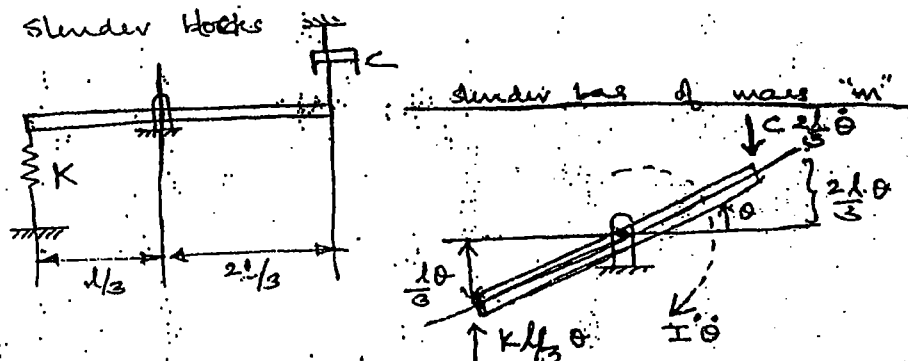
$$K = 1000 \text{ N/m}$$

$$M = \frac{K}{\omega_n^2}$$

$$C = 2\zeta \sqrt{KM}$$

Q12)

Slender Hooks

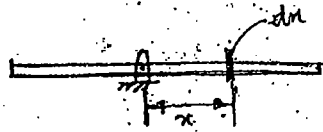


$$I \ddot{\theta} + \frac{C 2l}{3} \dot{\theta} + \frac{K l}{3} \theta = 0$$

$$I \ddot{\theta} + \frac{4}{9} C l^2 \dot{\theta} + \frac{K l^2}{9} \theta = 0$$

To find M.O.I.

(a) About the centre



$$\rho = \frac{m}{l}$$

$$\int_{-l/2}^{l/2} x^2 \rho dx = \rho \frac{x^3}{3} \Big|_{-l/2}^{l/2} = \frac{\rho l^3}{12} = \frac{m l^2}{12}$$

(b)

About another pt,

$$\rho \frac{x^3}{3} \Big|_{-l/2}^{2l/3} = \rho l^3 \frac{1}{9} = \frac{m l^2}{9}$$

$$I \ddot{\theta} + C \frac{2l}{3} \dot{\theta} \frac{2l}{3} + K \frac{l}{3} \theta \frac{l}{3} = 0$$

$$\left(\frac{m l^2}{9} \ddot{\theta} \right) + \left(\frac{4 C l^2}{9} \dot{\theta} \right) + \left(\frac{K l^2}{9} \theta \right) = 0$$

$$M_{eq} = \frac{m l^2}{9}$$

$$\text{So } \omega_n = \sqrt{\frac{K_{eq}}{M_{eq}}} = \sqrt{\frac{K}{m}}$$

Now, DAMPING ratio for the system :-

$$\zeta = \frac{C_{eq}}{2 \sqrt{K_{eq} M_{eq}}} = \frac{4 C l^2 / 9}{2 \sqrt{\frac{K l^2}{9} \frac{m l^2}{9}}} = \frac{2 C}{\sqrt{K m}}$$

Understanding and as normally you get ζ as $\frac{c}{2 \sqrt{K m}}$

pg 8 [Prev Gate Objectives]

01 In the mean position, Net force is ZERO
Ans: (A)

02 ω closer to 1 will have the max vibration
So, Ans: (C) $\rightarrow R$ with 128

03 Here,
 $C = C_c = 2\sqrt{km}$
 $K = \frac{3EI}{l^3}$

04 As a simply supported shaft,

So, $K = \frac{48EI}{l^3}$

$M = 20\text{ kg}$
 $l = 1\text{ m}$
 $25 \times 25\text{ mm}$
 $E = 200 \times 10^9\text{ N/m}^2$

$d = 20 \times 10^{-3}\text{ m}$

$l = 0.5\text{ m}$

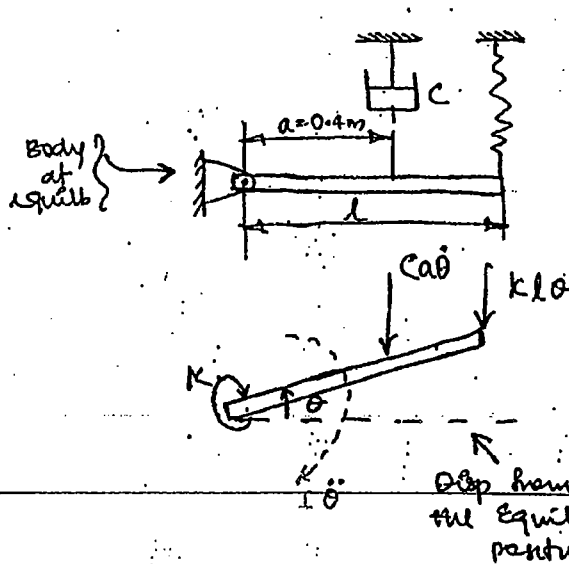
$E = 2.1 \times 10^{11}\text{ Pa}$

$N_c = ? \text{ Hz}$

$\omega_n = \sqrt{\frac{K}{m}} \text{ rad/sec}$

$N_c = \frac{\omega_n}{2\pi} \text{ Hz}$

05 (a) slender bar — mass is distributed uniformly



$I = \frac{ml^2}{3}$

$I\ddot{\theta} + ca\theta + Kl\theta + K\theta = 0$

$\left(\frac{ml}{3}\right)\ddot{\theta} + (ca^2)\theta + (K_0 + Kl^2)\theta = 0$

Body is rotational

Take moment
equilibrium about
axis of rotation

Dep from
the Equilib.
position

$$\omega_0 = \sqrt{\frac{K_{eq}}{m_{eq}}} = \sqrt{\frac{K_0 + K_1^2}{m l^2/3}} = 42.43 \text{ rad/sec.}$$

Ans: (a)

$$C_{eq} = C \alpha^2$$

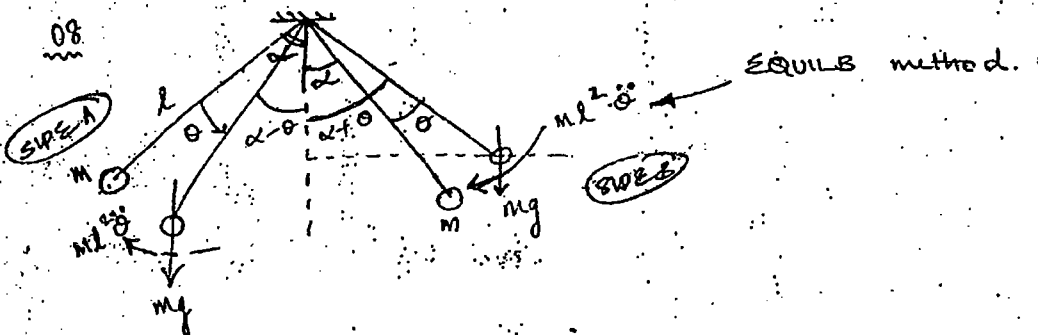
06

07 $K_1 \& K_2 \rightarrow$ in series $\left\{ \begin{array}{l} \text{they are in parallel} \\ \text{with } K_3 \end{array} \right.$

Ans: (b)

$$\sqrt{\frac{(K_1 K_2) + K_3}{J}}$$

08



$$m l^2 \ddot{\theta} + m l^2 \ddot{\alpha} + m g l \sin(\alpha + \theta) - m g l \sin(\alpha - \theta) = 0$$

$$2 m l^2 \ddot{\theta} + 2 m g l \cos \alpha \sin \theta = 0$$

{ for small angles, } $\sin \theta \approx \theta$

m_{eq}

K_{eq}

$$\omega_0 = \sqrt{\frac{K_{eq}}{m_{eq}}} = \sqrt{\frac{g \cos \alpha}{l}}$$

ENERGY method:

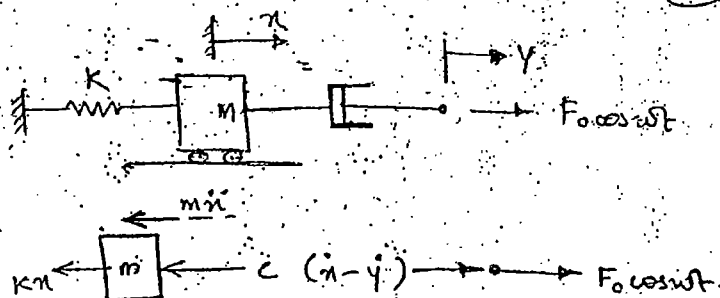
For side A, PE is -ve

For side B, PE is +ve

$$\text{So } (-mgl)_A = (mgh)_B \rightarrow \text{As it is rising}$$

and then differentiate with respect to time and equate to 0.

09



$$\left. \begin{aligned} M\ddot{x} + c(\dot{x} - \dot{y}) - Kx &= 0 \\ c(\dot{x} - \dot{y}) + F_0 \cos \omega t &= 0 \end{aligned} \right\}$$

It's a $1/2$ order dof system, can't exist on its own, so its coupled to another system. Hence it becomes a $1\frac{1}{2}$ dof system.

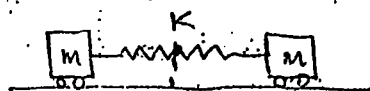
10 Natural freq. $= \frac{10000}{100} = 10 \text{ rad/sec}$
 in cycles/sec $= \frac{10}{2\pi} = \frac{5}{\pi} //$

12 Ans: (C) $10/\pi$ } same as Q 10

$-(0) = 0$

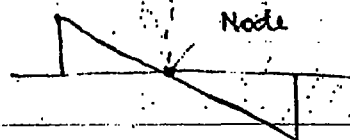
11: $\omega_n = 100 \text{ rad/sec}$
 $\zeta = 0.8$
 $\omega_d = ?$
 $\omega_d = \sqrt{1 - \zeta^2} \omega_n = 60 \text{ rad/sec}$
 Ans: (A)

13 It's a 2 degree freedom system.



It's a SEMI-DEFINITE system

$\omega_1 = 0$

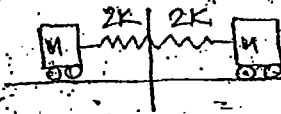


Node: pt with 0 displacement

For a 2-mass system, it can move freely in one direction in the condition that there is no tension in the string.

if the system consists of equal masses,

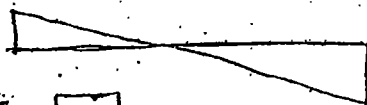
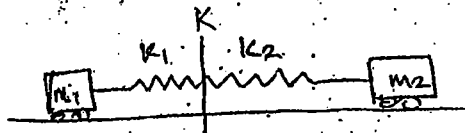
$$\omega_2 = \sqrt{\frac{2K}{m}}$$



divide the spring into 2 parts with
 $K_{\text{each}} = 2K$
 so,
 $\omega_2 = \sqrt{\frac{2K}{m}}$

if system consists of unequal masses,

Then the amplitudes are not equal,
 The node cent at the centre.



$$\sqrt{\frac{K_1}{m_1}} = \sqrt{\frac{K_2}{m_2}}$$

$$K = \frac{K_1 K_2}{K_1 + K_2}$$

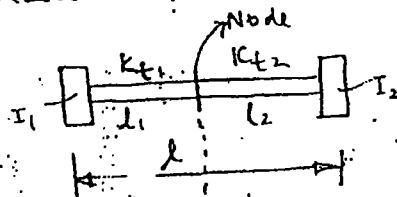
The freq's are 0, $\sqrt{\frac{K(m_1 + m_2)}{m_1 m_2}}$

In this eqn
 If you put
 $m_1 = m_2 = m$
 we get the
 equal mass
 case

EXTRA

NOTE:

Node is close to the heavy rotor.



$$K_t = \sqrt{\frac{GJ}{l}}$$



$$K_{t1} = \frac{GJ}{l_1}$$

$$K_{t2} = \frac{GJ}{l_2}$$

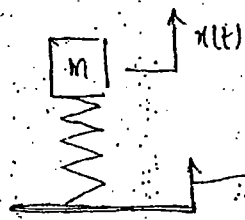
$$\sqrt{\frac{K_{t1}}{I_1}} = \sqrt{\frac{K_{t2}}{I_2}}$$

$$\Rightarrow I_1 l_1 = I_2 l_2$$

CASE 1:

divided
the spring
into 2 parts
with
sum of $2K$
So,
 $\omega_2 = \sqrt{\frac{2K}{m}}$

14



{ Motion allowed only vertically
The rollers are present to
ensure this vertical motion.

$$y(t) = 0.2 \sin 200\pi t \text{ mm}$$

$$Y = 0.2 \text{ mm}$$

$$\omega = 200\pi$$

$$C = 0$$

$$X \leq 0.01 \text{ mm}$$

$$m = 50 \text{ kg}$$

$$K = ?$$

$$\frac{X}{Y} = \sqrt{\frac{K^2}{(K^2 - m\omega^2)^2}}$$

$$\frac{X}{Y} = \frac{F}{K^2 - m\omega^2}$$

On Transmissibility
eqn, put $C = 0$

15

$$\frac{K_1 \times K_2}{(K_1 + K_2)} + K_3 = \left(\frac{3}{4} + 2 \right) \times 1000$$

$$\omega_n = \sqrt{\frac{11000}{4}}$$

Ans: (b) greater
(than)

16

$$\omega_d < \omega_n$$

17

stiffness becomes 4 times
deflection reduces to $1/4$

$$\text{So Ans: } \underline{D} \left\{ \frac{10}{4} = 2.5 \text{ mm} \right\}$$

18

$$\tau = 2\pi \sqrt{\frac{m}{K}} = 2 \text{ sec}$$

$$\tau' = 2\pi \sqrt{\frac{m+2}{K}} = 3 \text{ sec}$$

$$m = ? \text{ Ans: } \underline{(B)}$$

On Q15 they are
in SERIES
↑ Ans (A)
mm

Q19 is
(similar to Q15 on Pg 12)

19

(C) They are in parallel

20

(A) High damping reduces transmissibility

21

(C)

22

$$m = 250 \text{ kg}, K = 100,000 \text{ N/m}$$

23

$$F = 350 \text{ N @ } 2500 \text{ rpm}$$

$$\omega = \frac{2\pi N}{60}$$

$$\xi = 0.15$$

$$\text{Transmissibility} = \frac{\sqrt{k^2 + (c\omega)^2}}{\sqrt{(k - m\omega^2)^2 + (c\omega)^2}}$$

also
(TR)

if given (m, c, k, ω) use above formula.

if ω_n, ξ, ω is given use

$$\omega_n = \sqrt{k/m}$$

$$TR = \frac{1 + (2\xi\omega/\omega_n)^2}{\sqrt{(1 - (\omega/\omega_n)^2)^2 + (2\xi\omega/\omega_n)^2}}$$

Q8

$$m, k, c$$

$$c_c = 2\sqrt{km}$$

$$\delta = ?$$

$$\xi = \frac{c}{c_c}, \quad \delta = \frac{2\pi\xi}{\sqrt{1-\xi^2}}$$

pg 19 (last page)

Q1

(B)

Little less than undamped natural freq.

Q2

$$K/2 \text{ \& } K/2 \text{ in } \parallel^2$$

$$K \text{ \& } K \text{ in series}$$

Ans: (A)

$$K_1 = \frac{1}{\frac{1}{K} + \frac{1}{K}} = \frac{K}{2}$$

Q3

$$x(t) = Ce^{-\xi\omega_n t} \cos(\omega_d t - \phi)$$

$$e^{-\xi\omega_n t} (A \cos \omega_d t + B \sin \omega_d t)$$

$$x(0) = x \quad \dot{x}(0) = 0$$

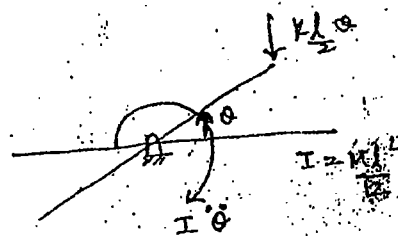
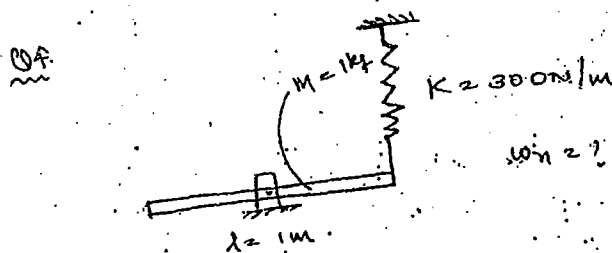
find x at $t = nT_d$

{do trial & error for faster solving}

check and compare exponents

$$-\xi\omega_n t; \quad = -\xi\omega_n nT_d$$

$$z = \sum_{n=1}^{\infty} \omega_n \frac{1}{\omega_d} = \frac{2\pi}{\omega_d} \sum_{n=1}^{\infty} \omega_n = \frac{2\pi}{\sqrt{1-\phi^2}} \sum_{n=1}^{\infty} \omega_n = \frac{2\pi}{\sqrt{1-\phi^2}} \cdot \frac{1}{2}$$



$$I_{\text{rod, about axis}} = \frac{Ml^2}{12}$$

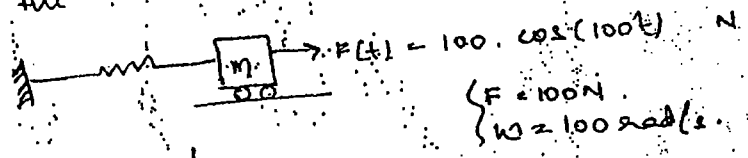
$$\frac{Ml^2}{12} \ddot{\theta} + \frac{Kl^2}{2} \theta = 0$$

$$\omega_n = \sqrt{\frac{Kl^2/4}{Ml^2/12}} = \sqrt{\frac{300 \times 12}{1 \times 4}} = \sqrt{900} = 30 \text{ rad/s}$$

Q5 2 springs in parallel

$$\sqrt{\frac{8000}{1.4}} = \sqrt{4000} = \frac{20\sqrt{10}}{2\pi} = 10 \text{ Hz}$$

2010 Gate : A mass m attached to a spring is subjected to a harmonic force as shown. The amplitude of the forced motion is 50 mm. The value of m is —.



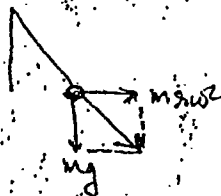
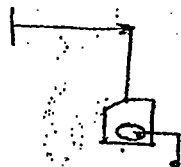
Ans: $K = 3000 \text{ N/m}$

$$X = \frac{F}{K - m\omega^2}$$

$$50 \times 10^{-3} = \frac{100}{3000 - m(100)^2}$$

Lift moves up $\rightarrow g-a$
 Lift moves down $\rightarrow g+a$ } Natural freq for pendulum in a LIFT,

Notes: Lift $\rightarrow$ Edg drum pack



$$\alpha = \tan^{-1} \frac{g\omega^2}{g}$$

$$g_e = \sqrt{g^2 + \omega^2 x^2}$$

$$\omega_n = \sqrt{\frac{g_e}{l}}$$

11 [CORRECTIONS - venow]
 competitive

1 (c) in G-P

2 (b)

3 (a)

stiffness is +ve
 then the system oscillates

$K \rightarrow -ve$ or 0, then it doesn't oscillate.
 UNSTABLE.

4 () $M = 100 \text{ kg}$ $m = 20 \text{ kg}$ $e = 0.5 \times 10^{-3} \text{ m}$

$k = 85000 \text{ N/m}$

$C = 0$ } damping negligible?

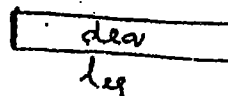
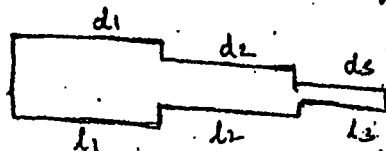
$\omega = 20 \pi \text{ rad/s}$

$$x = \frac{m\omega^2}{k - M\omega^2} \text{ when } C = 0$$

so find $x = ?$

5 (b)

stepped shaft can be replaced with an equivalent shaft



$$\frac{l_{eq}}{d_{eq}^4} = \frac{l_1}{d_1^4} + \frac{l_2}{d_2^4} + \frac{l_3}{d_3^4}$$

6(c)

$$e = 2 \text{ mm}$$

$$\omega_n = 10 \text{ rad/sec}$$

$$N = 300 \text{ rpm}, \quad \omega = \frac{2\pi N}{60} \text{ rad/sec}$$

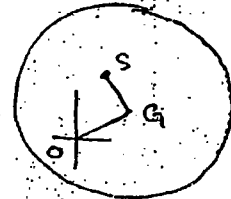
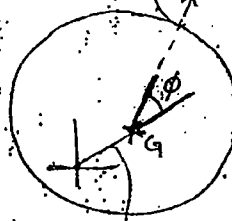
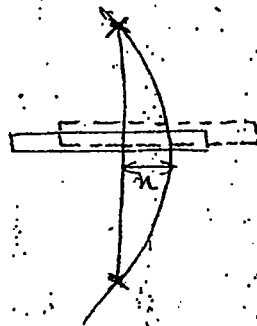
$$\therefore \omega = 10\pi \text{ rad/sec}$$

$$F = m\omega^2 e$$

$$\text{whirling rad} = X = \frac{m\omega^2 e}{k - m\omega^2} \quad (\text{no damping})$$

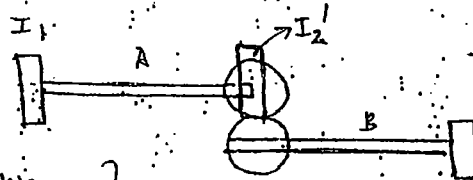
$$X = \frac{e\omega^2}{\omega_n^2 - \omega^2} \quad \left\{ \begin{array}{l} \text{sing by } m \end{array} \right.$$

7(c)



At critical speed, phase angle = 90°

8(c)



$$\frac{\omega_B}{\omega_A} = n$$

$$\frac{1}{2} I_2' \omega_A^2 = \frac{1}{2} I_2 \omega_B^2$$

$$I_2' = \left(\frac{\omega_B}{\omega_A} \right)^2 I_2 = n^2 I_2$$

9(c)

cantilever shaft

$$\omega_n = \omega_d \Rightarrow \text{critical speed}$$

critical speed depends on k & m and

NOT on eccentricity

so it remains same

10(c)

11(b)

12(a)

13(c)

Add equivalent spring mass $\Rightarrow$

$$N_c \propto \omega_n \sqrt{k/m}$$

$$N_c \propto \sqrt{k} \quad N_c \propto \frac{1}{\sqrt{m}} \quad k = \frac{56G}{d^3}$$

$$N_c \propto \sqrt{k} \quad N_c \propto \frac{1}{\sqrt{m}} \quad N_c \propto \frac{1}{\sqrt{d^3}}$$

19 (a) $\frac{\omega}{\omega_n} = 4, \quad \omega_n = \frac{1}{4} \times \omega$

26 () Bell crank lever is given angular dispⁿ θ

KE = $\frac{1}{2} K_1 (c\theta)^2 + \frac{1}{2} K_2 (b\theta)^2 + \frac{1}{2} K_3 (a\theta)^2$

KE = $\frac{1}{2} I \dot{\theta}^2 + \frac{1}{2} m_1 (c\dot{\theta})^2 + \frac{1}{2} m_2 (b\dot{\theta})^2$

$\downarrow$
vel = $c\dot{\theta}$
when its given
a disp θ

PE = $\frac{1}{2} \frac{(K_1 l^2 + K_2 b^2 + K_3 a^2)}{K_{eq}} \theta^2$

KE = $\frac{1}{2} \frac{(I + m_1 c^2 + m_2 b^2)}{M_{eq}} \dot{\theta}^2$

$\omega_n = \sqrt{\frac{K_{eq}}{M_{eq}}}$

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Q 21

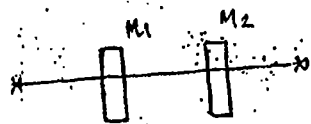
$m = 2.5 \text{ kg}$

$k = 8000 \text{ N/m}$

$\delta = \frac{1}{5} \ln \frac{x_0}{x_0/4} = \frac{1}{5} \ln 4$

$\delta = \frac{g}{\sqrt{4\pi^2 + g^2}} \quad \therefore C = 2.9 \sqrt{km}$

Q 22



$\begin{bmatrix} m_1 & 0 \\ 0 & m_2 \end{bmatrix} \quad \begin{bmatrix} k_{11} & k_{12} \\ k_{21} & k_{22} \end{bmatrix}$

Mass matrix

Stiffness matrix

$\begin{bmatrix} a_{11} & a_{12} \\ a_{21} & a_{22} \end{bmatrix}$

Flexibility matrix

Flexibility is reciprocal of stiffness.

$\omega_{n1} = \sqrt{\frac{k_{11}}{m_1}} \text{ or } \sqrt{\frac{1}{m_1 a_{11}}}$

$\frac{1}{\omega_{n1}^2} + \frac{1}{\omega_{n2}^2} + \frac{1}{\omega_{n3}^2}$

So $A = K^{-1}$

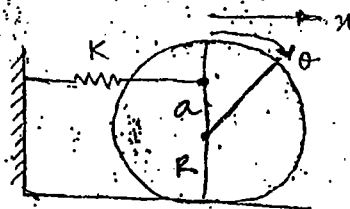
$\omega_{n2} = \sqrt{\frac{K_{22}}{M_2}} \quad \text{or} \quad \sqrt{\frac{1}{M_2 Q_{22}}}$

$\left[\frac{1}{\omega_n^2} \right] = \frac{1}{\omega_{n1}^2} + \frac{1}{\omega_{n2}^2}$

Similar to Q23

* Std problem: min

ENERGY method used



Condition for 1 dof

$x = R\theta$ } FOR NO slip
If there is a slip it becomes a 2 dof system

$KE = \frac{1}{2} m \dot{x}^2 + \frac{1}{2} \left(\frac{m R^2}{2} \right) \dot{\theta}^2$
 $= \frac{1}{2} m (R\dot{\theta})^2 + \frac{1}{2} \frac{m R^2}{2} \dot{\theta}^2$
 $= \frac{1}{2} \left(\frac{3}{2} m R^2 \right) \dot{\theta}^2$

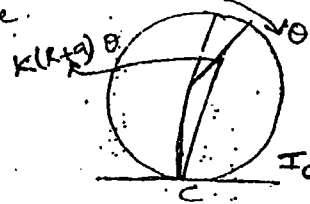
With disp x , it is relative to the disp x .
 Here occurs the angular disp θ .

So, $PE = \frac{1}{2} K (x + a\theta)^2$
 $PE = \frac{1}{2} K ((R+a)\theta)^2$ } $\omega_n = \sqrt{\frac{K(R+a)^2}{\frac{3}{2} m R^2}}$

EQUILB METHOD

Consider moment equals about instantaneous centre.

Q24



$I_C \ddot{\theta} + K(R+a)^2 \theta = 0$

Q24 } Bring down speed to (0.7×1400)
 2 methods } or find the critical speed
 } safe speed

$\omega \times R$

change critical speed

$$K = (192) \frac{EI}{l^3}$$

$$\omega_n \propto \sqrt{E}$$

$$\omega_n \propto \sqrt{I} \rightarrow \propto d^2$$

$$\omega_n \propto \frac{1}{l^{3/2}}$$

1 speed

$$K = (192) \frac{EI}{l^3}$$

3	→	1	1
48	→	16	4
	→	64	8

increase span → ↓ ω_n

decrease span → ↑ ω_n

Q45

$$M = 50 \text{ kg}$$

$$K = 35000 \text{ N/m}$$

$$C = C_c = 2\sqrt{KM}$$

$$f_d = C \omega_d \quad \left\{ \begin{array}{l} \omega_d \text{ is } \\ \text{given} \end{array} \right\}$$

prob 312

$$\left\{ \begin{array}{l} \delta_{am} = 8 \text{ cm} \\ \delta_{pm} = 6.10 \text{ cm} \end{array} \right.$$