# LIMITATIONS Pilot & Plane



GIRLS LOVE TO FLY

## PA.I.F PERFORMANCE & LIMITATIONS

Atmosphere

Density Altitude & Weather

3

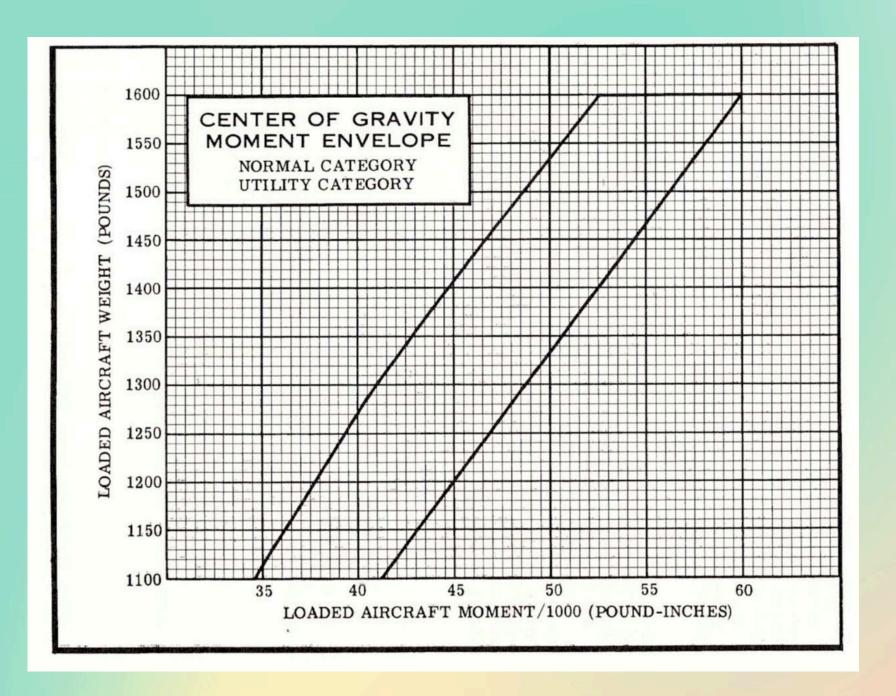
**Configuration**Flaps & Landing Gear

Pilot Technique
Level of experience &
comfort

4

Airport Environment
Terrain, obstacles, runway
surface





## WEIGHT & BALANCE

Forward vs Aft CG

WEIGHT X ARM = MOMENT

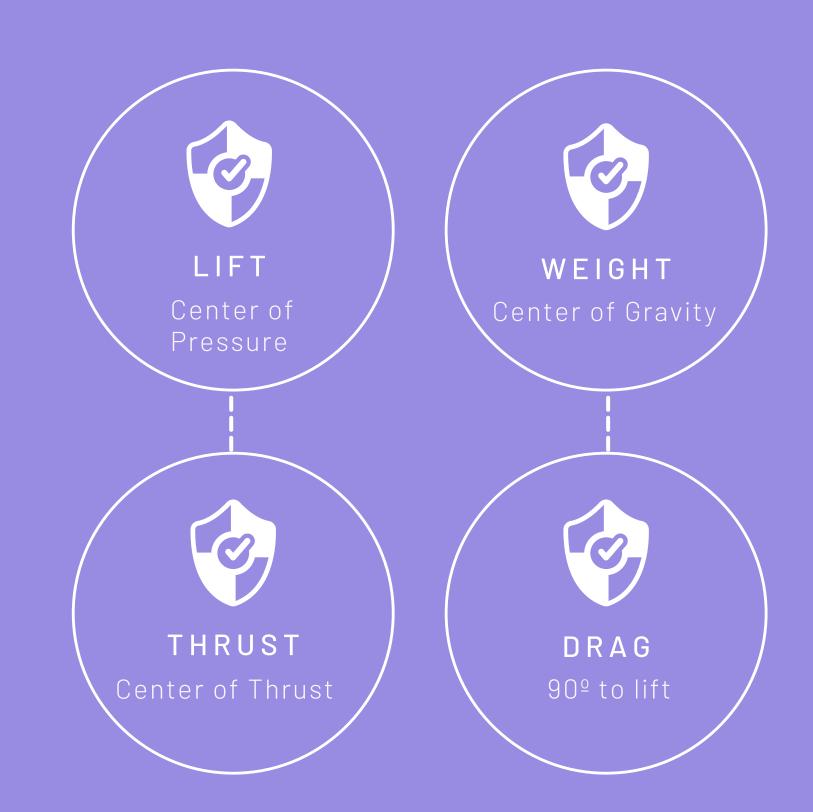
ITEM WEIGHT ARM MOMENT

### AERODYNAMICS

4 FORCES OF FLIGHT

Four forces work together to determine an aircraft's behavior.

Straight and level flight is the perfect example of what happens when all the flight forces are in equilibrium.



## Theories of LIFT

If a fluid flow speeds up, there is a simultaneous pressure drop.





For every action, there is an equal and opposite reaction.



Newton

## ANGLE OF ATTACK

The plane stalls when it exceeds the "critical angle of attack".





#### **Chord Line**

An imaginary line from leading edge to trailing edge of the camber.



Camber
Measure of the curvature of an airfoil.



#### Relative Wind

Direction of the wind in relation to the airfoil. Always parallels opposite of motion.



#### Va

Maneuvering speed

### CHARTS 91.103

### When in doubt, keep numbers conservative.

	TAI	KE-	OFF	DIST	ANC	<b>E</b>	LAPS RETRA	CTED HAR	D SURFACE	RUNWAY
GROSS WT. LBS.	IAS 50 FT. MPH	HEAD WIND KNOTS	AT SEA LEVEL & 59° F.		AT 2500 FT. & 50° F.		AT 5000 F	T. & 41° F.	AT 7500 FT. & 32° F.	
			GROUND RUN	TOTAL TO CLEAR 50 FT.OBS	GROUND RUN	TOTAL TO CLEAR 50 FT. OBS	GROUND RUN	TOTAL TO CLEAR 50 FT. OBS	GROUND RUN	TOTAL TO CLEAR 50 FT. OBS
1600	64	0 10 20	735 500 305	1385 1035 730	910 630 395	1660 1250 890	1115 780 505	1985 1510 1090	1360 970 640	2440 1875 1375

NOTES: 1. Increase the distances 10% for each 35°F. increase in temperature above standard for the particular altitude.

2. For operation on a dry, grass runway, increase distances (both "ground run" and "total to clear 50 ft. obstacle") by 7% of the "total to clear 50 ft. obstacle" figure.

	XAN	IMU	MF	TAS	E-01	F-CL	IMI	B D	ATA
GROSS WEIGHT LBS.	AT SEA	LEVEL &	59° F.	AT 5000 FT. & 41° F.			AT 10000 FT. & 23° F.		
	IAS, MPH	RATE OF CLIMB FT./MIN.	FUEL USED, GAL.		RATE OF CLIMB FT./MIN.	USED	IAS, MPH	RATE OF CLIMB FT./MIN.	FUEL USED FROM S.L., GAL.
1600	73	670	0.6	69	440	1.6	65	220	3.0

NOTES: 1. Flaps retracted, full throttle, mixture leaned to smooth operation above 5000 ft.

2. Fuel used includes warm-up and take-off allowances.

3. For hot weather, decrease rate of climb 15 ft./min. for each 10°F above standard day temperature for particular altitude.

-1	AND	ING	DISTA	NCE			LOWERED TO SURFACE RUNV		
	APPROACH SPEED, IAS, MPH	AT SEA LE	VEL & 59° F.	AT 2500 F	T. & 50° F.	AT 5000 F	T. & 41° F.	AT 7500 FT. & 32° F.	
GROSS WEIGHT LBS.		GROUND ROLL	TOTAL TO CLEAR 50 FT.OBS	GROUND ROLL	TOTAL TO CLEAR 50 FT, OBS	GROUND ROLL	TOTAL TO CLEAR 50 FT. OBS	GROUND ROLL	TOTAL TO CLEAR 50 FT. OBS
1600	58	445	1075	470	1135	495	1195	520	1255

NOTES: 1. Decrease the distances shown by 10% for each 4 knots of headwind.

2. Increase the distance by 10% for each 60°F, temperature increase above standard.

 For operation on a dry, grassy runway, increase distances (both "ground roll" and "total to clear 50 ft. obstacle") by 20% of the "total to clear 50 ft. obstacle" figure.

#### Figure 5-3.

#### CRUISE PERFORMANCE

WITH LEAN MIXTURE

	RPM	%внр			END	. HOURS	RANGE, MILES	
ALTITUDE			TAS MPH	GAL/HR.	STANDARD LONG RANG		SE STANDARD	LONG RANG
					22.5 GAL.	35 GAL.	22.5 GAL.	35 GAL.
2500	2750	92	121	7.0	3.2	5.0	390	605
	2700	87	119	6.6	3.4	5.3	410	635
	2600	77	114	5.8	3.9	6.1	445	690
	2500	68	108	5.1	4.4	6.9	445	740
	2400	60	103	4.6				790
	2300	53	96		4.9	7.7	505	
	2200	46		4.1	5.5	8.6	535	830
	2100	40	89 79	3.6 3.2	6. 2 7. 0	9.7	550 555	860 865
5000	2750	0.5						
5000		85	121	6.4	3.5	5.5	425	660
	2700	80	118	6.0	3.8	5.8	445	690
	2600	71	113	5.3	4.2	6.6	475	740
	2500	63	107	4.8	4.7	7.4	505	790
	2400	56	101	4.3	5.3	8.2	530	830
8	2300	49	93	3.8	5.9	9.2	550	860
- 1	2200	43	84	3.4	6.6	10.3	560	870
	2100	37	71	3.0	7.5	11.7	540	835
7500	2700	74	117	5.5	4.1	6.3	480	745
	2600	66	111	4.9	4.6	7.1	505	790
- 1	2500	58	105	4.4	5.1	7.9	535	830
- 1	2400	52	98	4.0	5.7	8.8	555	860
1	2300	45	89	3.6	6.3	9.8	560	875
1	2200	40	77	3.2	7.1	11.1	550	850
				0.12				
10,000	2700	68	116	5.1	4.4	6.8	510	790
	2600	61	109	4.6	4.9	7.6	535	830
- 1	2500	54	102	4.1	5.4	8.5	555	865
- 1	2400	48	93	3.7	6.1	9.4	565	880
- 1	2300	42	82	3.3	6.8	10.6	555	860
12,500	2650	60	110	4.5	5.0	7.8	550	855
,	2600	56	106	4.3	5.3	8.2	555	865
	2500	50	97	3.9	5.8	9.1	565	880
	2400	44	86	3.5	6.5	10.1	560	870
		1.0000000	10000				000	
			المرابع					

NOTES: 1. Maximum cruise is normally limited to 75% power.

 In the above calculations of endurance in hours and range in miles, no allowances were made for take - off or reserve.

Figure 5-4.

### PA.I.H HUMAN FACTORS

IMSAFE (AIM 8-1-1)

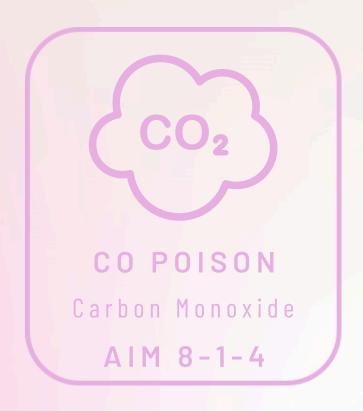
85% of general aviation accidents are caused by pilot error.

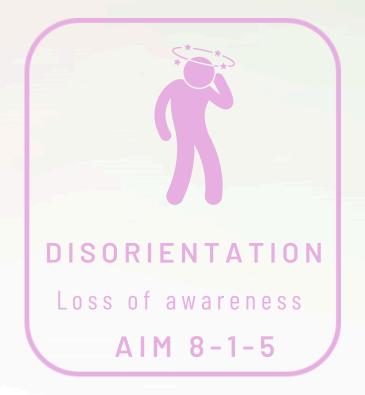
#### Human Factors include:

- Aeromedical
- Stress/Emotions
- Energy Levels
- Attitudes
- Confirmation Bias









# ALCOHOL & DRUGS 91.17

REQUIRED DUTY

Route Plan

REQUIRED DUT

Pre-Flight

REQUIRED DUTY

Check Wx

REQUIRED DUTY

Weight & Balance



### HAZARDOUS ATTITUDES

#### ANTI-AUTHORITY

Follow the rules, they are probably right.

#### **MACHO**

Taking chances is foolish.

## HULNERABILITY

It could happen to me,

#### **IMPULSIVITY**

Not so fast, think first.

#### RESIGNATION

I am not helpless. I can make a difference. Maneuvering Speed 4 Forces of Flight

## THANKAYOU

