

AVIONICS MAINTENANCE RECORD

AIRCRAFT SERIAL AND REGISTRATION_

T24002091

N246VB

D5520-2-13



INSTRUCTIONS

This record should be completed and maintained in accordance with FAA Federal Aviation Regulations:
91.411 Altimeter system and altitude reporting equipment tests and inspections,
91.413 ATC transponder tests and inspections,

and

43.9, 43.11 and 91.215, and/or other prevailing government regulations.

NOTE: Record required tests in the back of this record

Record Service Letters and Airworthiness Directives in the back of this record as well as the Airframe Maintenance Record.

Copyright © 1996

DATE	TOTAL TIME IN SERVICE HOURS 10THS	DESCRIPTION OF THE WORK PERFORMED	AUTHORIZED SIGN CERTIFICATE T & NUMBER
	Cessna	Van Bortel Aircraft Inc. 4912 S. Collins St., Arlington, TX 76018 FAA CRS VNFR171	
	Repaired and rep	6VB SN: T24002091 Tach: 14.5 DATE 03/28/2016 W.O. blaced wiring from UZ006 and UZ007 Transient Surge Suppressors in Circuit Breakers above was repaired and inspected in accordance with current regulations and its approved for return to service with respect to the work above, pertinent during station under this work order.	or Panel as required. OF THE FEDERAL AVIATION
	Repaired and repai	laced wiring from UZ006 and UZ007 Transient Surge Suppressors in Circuit Breakers ABOVE WAS REPAIRED AND INSPECTED IN ACCORDANCE WITH CURRENT REGULATIONS OF THE WORK ABOVE, PERTINENT D	or Panel as required. OF THE FEDERAL AVIATION



IN:	TAL TIME SERVICE		/ 11 - 1	IONICS
	OLITTIOL	DESCRIPTION OF THE WORK PERF	ALTIMETER SCALE CO	PRRECTION CARD
	RS 10THS		AIRCRAFT N#:_/ ALTIMETER S/N:_	1246 VB
		The following altimeter and altitude reporting equipmen	SCALE CORR	
		as required by 14CFR 91.411, IAW 14CFR Part 43, App	Add algebraically at room temp.	Add algebraically at room temp
		- Part (a) Static System //4/22	-1000 -16	14000 +20
		Part (b) Altimeters Pilot's 606-74A Copilot'sSta	0 +10	16000 +20
			500 + 20	18000 + 36
		sn011-0088710snsn	2000 7 40	20000 +56
	ALCOHOLD TO A	Tested to 25,000 feet	1500 + 10	22000 170
	A 140 LA	Part (c) Integrated System Test 1/14/22 (da		25000 +40
		Totalin Villan	3000 +20	30000
		DUNKIRK AVIONICS LLC. FAA CRS# DKZR372	4000 +15	35000 —
THE REAL PROPERTY.			6000 +20	40000
		ATC Transponder test and inspections as required by 14	8000 +26	45000
		14CFR Part 43, Appendix F, have been performed.	10000 + 20	50000
		Transponder Model 411-53 S/N	12000 +20	
		Transponder Model S/N Technician W/C DUNKIRK AVIONICS LLC. FAA CRS# DKZR372	SCALE CORRECTIONS SHOWN ON THIS CARD ARE THE ONLY THOSE SCALE ERRORS DETERMINED BY THE UTO AN ACCURATE MERCURIAL BAROMETER AT THE FOR ACCEPTANCE WILL BE USED IN DETERMING SCARD. THE COMPLETED SCALE CORRECTION CARD WHEN IT IS NOT INSTALLED IN THE AIRCRAFT.	SE OF A MASTER ALTIMETER TRACEABLE TIME THE ALTIMETER IS CALIBRATED ALE CORRECTIONS FOR ENTRY ON THIS
UNIE 1			FAA CRS# DKZR372X	FORM 106

НО		REFERENCE ALTITUED IN FT.	ALTIMETER READS	REFERENCE ALTITUED IN FT.	ALTIMETER READS	
	THOROUGHBRED	-1000	-1000	14000	13990	
-	AVIATION MAINTENANCE	0	0	16000	15980	
	05/550/2024	500	500	18000	17970	
	SYSTEM TEST DATE: 06/FEB/2024	1000	1000	20000	19940	
		1500	1500	22000	21900	
	AIRCRAFT REG #: N246VB	2000	2000	25000	24850	
		3000	3000	30000	***************************************	
200	AIRCRAFT MAKE/MODEL: TEXTRON T	4000	4000	35000	***************************************	
		6000	6000	40000	***************************************	
	I HEREBY CERTIFY THAT THE ALTIMETER	8000	8000	45000	***************************************	
	BEEN PERFORMED AND MEET SPECIFICAT	10000	10000	50000		
200	25.000	12000	12000			
2	PFD ADC TESTED TO 25,000	Tested By: (My Manual)	,	N#: N246VB	Date: 06/FEB/2024	
	I HEREBY CERTIFY THAT THE ATC TRANSPC COMPLY WITH FAR 43, APPENDIX F.		4316 H Lexingt	red Aviation Maintenance, Inc angar Drive on, KY 40510 RS # R81R199Y		

4117846

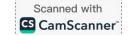
CERT #:

#1 TRANSPONDER MAKE: GARMIN

Form TAM021-03/22

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CS CamScanner

The following altimeter and altitude reporting equipments as required by 14CFR 91.411, IAW 14CFR Part 43, A performed. Part (a) Static System 1/14/22	The state of the s
Part (b) Altimeters Pilot's OL-74A Copilot's	Standby E 5 I-1000
SN 011-6088710	S/N 9200-3210C
Tested to 2500 feet Part (c) Integrated System Test 1/14/22 Technician DUNKIRKAVIONICS LLC. FAA CRS# DKZR.	W/O# 6853
	N 89130353
	10# 6853



PFD ADC ALTIMETER SCALE CORRECTON CARD

REFERENCE ALTITUED IN FT.	ALTIMETER READS	REFERENCE ALTITUED IN FT.	ALTIMETER READS		
-1000	-1000	14000	13990		
0	0	16000	15980		
500	500	18000	17970		
1000	1000	20000	19940		
1500 1500		22000	21900		
2000	2000	25000	24850		
3000	3000	30000			
4000	4000	4000 35000	4000 35000	4000 35000	
6000	6000 40000				
8000 8000 45000		45000			
10000	10000	50000			
12000	12000				
Tested By: (My Manual)	N	#: N246VB	Date: 06/FEB/2024		



Thoroughbred Aviation Maintenance, Inc. 4316 Hangar Drive Lexington, KY 40510 FAA CRS # R81R199Y



STBY ADC ALTIMETER SCALE CORRECTON CARD

REFERENCE ALTITUED IN FT.	ALTIMETER READS	REFERENCE ALTITUED IN FT.	ALTIMETER READS
-1000	-1000	14000	14000
0	0	16000	16000
500	500	18000	17980
1000	1000	20000	19960
1500	1500	22000	21900
2000	2000	25000	24850
3000	3000	30000	
4000	4000	35000	
6000	6000	40000	
8000	8000	45000	
10000	10000	50000	
12000	12000		
ested By: Aw Innale	N	#: N246VB	Date: 06/FEB/2024



Thoroughbred Aviation Maintenance, Inc. 4316 Hangar Drive Lexington, KY 40510 FAA CRS # R81R199Y





DATE	HO HO	DESCRIPTION OF THE WORK PERFORMED	AUTHORIZED SIGNATURE, CERTIFICATE TYPE & NUMBER
		THOROUGHBRED THOROUGHBRED THOROUGHBRED CRS#R81R199Y THOROUGHBRED	
	A A	RCRAFT REG #: N246VB AIRCRAFT SERIAL #: T24002091 RCRAFT MAKE/MODEL: TEXTRON T240 HEREBY CERTIFY THAT THE ALTIMETER, ENCODER, AND STATIC SYSTEM TESTS REQUIRED BY FAR 91.411 HAVE SEEN PERFORMED AND MEET SPECIFICATIONS SET FORTH IN FAR 43 APPENDIX E.	
	#:	PFD ADC TESTED TO 25,000 FEET #1 ENCODER TESTED TO 25,000 FEET STBY ADC TESTED TO 25,000 FEET HEREBY CERTIFY THAT THE ATC TRANSPONDER TEST REQUIRED BY FAR 91.413 WAS PERFORMED AND FOUND TO OMPLY WITH FAR 43, APPENDIX F. L TRANSPONDER MAKE: GARMIN MODEL: GTX 33 SERIAL #: 89130353	
		STED BY: My hamby CERT #: 4117846	

Phone: 8174687788

FAA Airworthiness Directives Compliance Record

Company:

Model: T240

Aircraft Registration No: N246VB

Category: Airframe

Position:

ATP Revision: 11/14/2019

Manufacturer: Cessna Aircraft Company

P/N:

S/N: T24002091

Facility Issue Number Cert No. / Type Effective Date Authorized By Method of Amendment # Signed Compliance Next Due Description Recur Complied D: --Hrs: --Hrs: --

Category: Engine Position: ATP Revision: 11/14/2019

Manufacturer: Teledyne Continental

P/N:

Model: TSIO-550-C

S/N: 1031118

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
93-10-02 08/12/1993	TO PREVENT AN ENGINE FAILURE DUE TO A MISSING CYLINDER VALVE RETAINER KEY		N/A BY S/N	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
96-12-22 07/31/1996	TO PREVENT LOSS OF ENGINE OIL CAUSED BY LOOSE OR SEPARATED OIL FILTER ADAPTERS, WHICH COULD RESULT IN ENGINE, CONTD.	 Hrs: C:	N/A, PART IS NOT INSTALLED	Yes	D: Hrs: N/A C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
99-09-17 L 04/22/1999	Superseded by 99-19-01	 Hrs: C:	SUPERSEDED	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park

Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
99-19-01 09/30/1999	To prevent crankshaft failure due to crankshaft cheek cracks, which could result in total engine power loss, contd.	 Hrs: C:	N/A BY S/N	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
2000-08-51 E 04/28/2000	Superseded by 2000-23-21	 Hrs: C:	SUPERSEDED	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
2000-23-21	To prevent crankshaft connecting rod journal fracture, which could result in total engine power, contd.	 Hrs: C:	N/A BY S/N	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
00-00-01 01/22/2001	Important for Cessna Oil Filter Adapter Assemblies listed in AD 96-12-22	 Hrs: C:	SEE AD 96-12-22	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
2004-08-10 05/05/2004	To prevent loss of engine power due to cracks in the cylinder head & possible engine failure caused,contd.	 Hrs: C:	N/A BY CYL. P/N	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
2007-16-10 08/23/2007	To prevent the turbine rotor from separating from the shaft of the turbocharger due to a machining defect in the,contd.	 Hrs: C:	N/A BY S/N'S OF TURBOS	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
2009-24-52 E 11/18/2009	Superseded by 2010-11-04	 Hrs: C:	SUPERSEDED	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
2010-11-04 06/16/2010	To prevent excessive hydraulic lifter wear, which can result in loss of engine power & loss of control of the airplane	 Hrs: C:	N/A BY ENGINE S/N	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
2011-13-03 07/13/2011	To prevent seizure of the turbocharger turbine, which could result in damage to the engine, & smoke in the,contd.	 Hrs: C:	N/A, HOUSINGS MARKED "JT"	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park

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To prevent an in-flight engine shutdown due to a failed fuel servo diaphragm, and damage to the airplane To prevent failure of the cylinder assemblies, which could lead to failure of the engine, in-flight, contd. To prevent failure of the oil cooler cross the engine, in-flight, contd. To prevent failure of the oil cooler cross fitting and engine, IFSD, and loss of the airplane To prevent failure of the oil cooler cross thris: PCW DURING PRODUCTION, INSTALLATION IS N/A, SERVO NOT INSTALLED N/A, SERVO NOT INSTALLED N/A, SERVO NOT INSTALLED N/A, SERVO NOT INSTALLATION IS NO D: Van Bortel Aircraft, VNFR171L/CRS Van Bortel Aircraft, VNFR171L/CRS PRODUCTION, INSTALLATION IS Paul Park	mendment #	Description	Complied		Recur	Next Due	Facility Cert No. / Type Authorized By Signed
109/15/2016 assemblies, which could lead to failure of the engine, in-flight, contd. To prevent failure of the oil cooler cross fitting and engine, IFSD, and loss of the airplane PCW DURING PRODUCTION, INSTALLATION IS PCW DURING PRODUCTION, INSTALLATION IS Paul Park PRODUCTION, INSTALLATION IS Paul Park PRODUCTION PRODUCTIO		To prevent an in-flight engine shutdown due to a failed fuel servo diaphragm, and	Hrs:		No	Hrs:	
fitting and engine, IFSD, and loss of the airplane Hrs: PRODUCTION, INSTALLATION IS C: Paul Park Paul Paul Paul Paul Paul Paul Paul Paul		assemblies, which could lead to failure of	C	N/A BY CYL. P/N	No		
WITH CSB15-2	1/18/2016	fitting and engine, IFSD, and loss of the	the state of the s	PRODUCTION, INSTALLATION IS IN COMPLIANCE		Hrs:	Deat Deat

Category: Propeller Position:

Manufacturer: McCauley

P/N:

Model: D3A34C447

S/N: 101010

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
2005-14-11 08/17/2005	To prevent blade failure that could result in separation of a propeller blade and loss of control of the airplane		N/A, NO REPAIR BY SO. CAL. PROP. SERVICE	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park

ATP Revision: 11/14/2019 Position: Category: Turbocharger Systems

Manufacturer: Hartzell Engine Tech

P/N: 646677

Model: (P/N) 466304-0003

S/N: H-SBL00087 H-SBL00086

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
		A CONTRACT OF THE STATE OF THE		-	D:	
		Hrs:			Hrs:	- Market - 11 11 10 20 1 9
		C:			C: -	

ATP Revision: 11/14/2019 Position: Category: Magnetos

P/N: 10-500556-101 Manufacturer: Teledyne Continental

11/15/2019 7:22 AM

CATP

S/N: D15GA092 D15GA069

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
73-07-04 10/11/1973	Superseded by 94-01-03	 Hrs: C:	SUPERSEDED	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
74-26-09 12/24/1974	S-20,-200,-1200 SERIES MAGNETOS	 Hrs: C:	N/A BY S/N'S	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
78-09-07 R3 01/17/1983	Superseded by 96-12-07	 Hrs: C:	SUPERSEDED	Yes	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
82-20-01 06/14/1983	TO PREVENT FAILURE OF IMPULSE COUPLING DUE TO IMPROPERLY HEAT TREATED (SOFT) FLYWEIGHTS RESULTING IN ENGINE, CONTD.	 Hrs: C:	N/A BY S/N'S	No	D: ` Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
94-06-09 05/20/1994	TO PREVENT INJURY OR DEATH TO GROUND PERSONNEL DUE TO A NON-GROUNDED MAGNETO	 Hrs: C:	N/A BY S/N'S	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
94-01-03 R2 06/28/1995	TO PREVENT MAGNETO FAILURE AND SUBSEQUENT ENGINE FAILURE	 Hrs: C:	N/A BY S/N'S	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
96-12-07 07/18/1996	Superseded by 2005-12-06	 Hrs: C:	SUPERSEDED	Yes	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
2005-12-06 07/19/2005	To prevent failure of the magneto impulse coupling assembly and possible engine failure	 Hrs: C:	N/A BY ENGINE MFR.	Yes	D: - Hrs: N/A C: -	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
					ATD Davidson	*****

Category: Ignition Systems

Position:

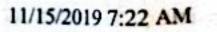
ATP Revision: 11/14/2019

Manufacturer: ACS Products Company

P/N:

Model: IGNITION SWITCHES

S/N:





Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
93-05-06 04/29/1993	TO PREVENT FAILURE OF IGNITION SWITCHES	 Hrs; C:	INSPECTION PER ACS SB92-01, DUE EACH 2000 HRS	Yes	D: Hrs: 2000 HRS. TT C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park

CATP

11/15/2019 7:22 AM



FAA Airworthiness Directives Compliance Record

Company:

Manufacturer: Cessna Aircraft Company

Category: Airframe

Model: T240

Position:

P/N:

S/N: T24002091

Aircraft Registration No: N246VB

ATP Revision: 01/22/2019

Facility Issue Number Cert No. / Type Effective Date Authorized By Method of Amendment # Signed Compliance Next Due Description Complied Recur D: --Hrs: --Hrs: --ATP Revision: 01/22/2019 Position:

gory: Engine

Manufacturer: Teledyne Continental

Model: TSIO-550-C

P/N:

S/N: 1031118

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
93-10-02 08/12/1993	TO PREVENT AN ENGINE FAILURE DUE TO A MISSING CYLINDER VALVE RETAINER KEY	 Hrs: C:	N/A BY S/N	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
96-12-22 07/31/1996	TO PREVENT LOSS OF ENGINE OIL CAUSED BY LOOSE OR SEPARATED OIL FILTER ADAPTERS, WHICH COULD RESULT IN ENGINE, CONTD.	 Hrs: C:	N/A, PART IS NOT INSTALLED	Yes	D: Hrs: N/A C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
19-17 L 04/22/1999	Superseded by 99-19-01	 Hrs: C:	SUPERSEDED	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park

ssue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
99-19-01 9/30/1999	To prevent crankshaft failure due to crankshaft cheek cracks, which could result in total engine power loss, contd.	-	N/A BY S/N	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
000-08-51 E 4/28/2000	Superseded by 2000-23-21	 Hrs: C:	SUPERSEDED	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
000-23-21 2/12/2000	To prevent crankshaft connecting rod journal fracture, which could result in total engine power,contd.	 Hrs: C:	N/A BY S/N	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
00-00-01 1/22/2001	Important for Cessna Oil Filter Adapter Assemblies listed in AD 96-12-22	 Hrs: C:	SEE AD 96-12-22	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
004-08-10 5/05/2004	To prevent loss of engine power due to cracks in the cylinder head & possible engine failure caused, contd.	 Hrs: C:	N/A BY CYL. P/N	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
007-16-10 8/23/2007	To prevent the turbine rotor from separating from the shaft of the turbocharger due to a machining defect in the,contd.	 Hrs: C:	N/A BY S/N'S OF TURBOS	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
009-24-52 E 1/18/2009	Superseded by 2010-11-04	 Hrs: C:	SUPERSEDED	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
010-11-04 6/16/2010	To prevent excessive hydraulic lifter wear, which can result in loss of engine power & loss of control of the airplane	 Hrs: C:	N/A BY ENGINE S/N	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
011-13-03 7/13/2011	To prevent seizure of the turbocharger turbine, which could result in damage to the engine, & smoke in the,contd.	 Hrs: C:	N/A, HOUSINGS MARKED "JT"	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park

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ssue Number ffective Date mendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
012-03-06 C 2/24/2012	To prevent an in-flight engine shutdown due to a failed fuel servo diaphragm, and		NI/A D	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
016-16-12 9/15/2016		 Hrs: C:	N/A BY CYL. P/N	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
016-21-04 1/18/2016	To prevent failure of the oil cooler cross fitting and engine, IFSD, and loss of the airplane	 Hrs: C:	PCW DURING PRODUCTION, INSTALLATION IS IN COMPLIANCE WITH CSB15-2	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park

Category: Propeller Position:

Manufacturer: McCauley

S/N: 101010 Model: D3A34C447

tive Date Aarendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
2005-14-11 08/17/2005	To prevent blade failure that could result in separation of a propeller blade and loss of control of the airplane	 Hrs: C:	N/A, NO REPAIR BY SO. CAL. PROP. SERVICE	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park

ATP Revision: 01/22/2019 Position: Category: Turbocharger Systems

P/N:

P/N: 646677 Manufacturer: Hartzell Engine Tech

S/N: H-SBL00087 H-SBL00086 Model: (P/N) 466304-0003

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
Que la constant de la	Cardinary Community	 Hrs: C:		-	D: Hrs: C:	

ATP Revision: 01/22/2019 Position: Category: Magnetos

P/N: 10-500556-101 Manufacturer: Teledyne Continental

01/22/2019 12:21 PM CATP

S/N: D15GA092 D15GA069

Superseded by 94-01-03	Complied	Compliance	Recur	Next Due	Authorized By Signed
	Hrs: C:	SUPERSEDED	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
S-20,-200,-1200 SERIES MAGNETOS	 Hrs: C:	N/A BY S/N'S	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
Superseded by 96-12-07	 Hrs: C:	SUPERSEDED	Yes	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
	Carried and the second of the	N/A BY S/N'S	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
TO PREVENT INJURY OR DEATH TO GROUND PERSONNEL DUE TO A NON- GROUNDED MAGNETO	 Hrs: C:	N/A BY S/N'S	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
	 Hrs: C:	N/A BY S/N'S	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
	 Hrs: C:	SUPERSEDED	Yes	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
	 Hrs: C:	N/A BY ENGINE MFR.	Yes	D: Hrs: N/A C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
	TO PREVENT FAILURE OF IMPULSE COUPLING DUE TO IMPROPERLY HEAT TREATED (SOFT) FLYWEIGHTS RESULTING IN ENGINE, CONTD. TO PREVENT INJURY OR DEATH TO GROUND PERSONNEL DUE TO A NON-GROUNDED MAGNETO TO PREVENT MAGNETO FAILURE AND SUBSEQUENT ENGINE FAILURE Superseded by 2005-12-06 To prevent failure of the magneto impulse coupling assembly and possible engine	Superseded by 96-12-07 C: TO PREVENT FAILURE OF IMPULSE COUPLING DUE TO IMPROPERLY HEAT TREATED (SOFT) FLYWEIGHTS RESULTING IN ENGINE, CONTD. TO PREVENT INJURY OR DEATH TO GROUND PERSONNEL DUE TO A NON-Hrs: C: TO PREVENT MAGNETO FAILURE AND SUBSEQUENT ENGINE FAILURE Hrs: C: Superseded by 2005-12-06 To prevent failure of the magneto impulse coupling assembly and possible engine failure Fig C:	Superseded by 96-12-07 TO PREVENT FAILURE OF IMPULSE COUPLING DUE TO IMPROPERLY HEAT TREATED (SOFT) FLY WEIGHTS RESULTING IN ENGINE, CONTD. TO PREVENT INJURY OR DEATH TO GROUND PERSONNEL DUE TO A NON-Hrs: GROUNDED MAGNETO TO PREVENT MAGNETO FAILURE AND SUBSEQUENT ENGINE FAILURE Hrs: C: Superseded by 2005-12-06 To prevent failure of the magneto impulse coupling assembly and possible engine failure N/A BY S/N'S SUPERSEDED N/A BY S/N'S N/A BY ENGINE MFR.	Superseded by 96-12-07 TO PREVENT FAILURE OF IMPULSE COUPLING DUE TO IMPROPERLY HEAT TREATED (SOFT) FLYWEIGHTS RESULTING IN ENGINE, CONTD. TO PREVENT INJURY OR DEATH TO GROUND PERSONNEL DUE TO A NON- GROUNDED MAGNETO TO PREVENT MAGNETO FAILURE AND SUBSEQUENT ENGINE FAILURE Firs: C: Superseded by 2005-12-06 To prevent failure of the magneto impulse coupling assembly and possible engine failure SUPERSEDED N/A BY S/N'S No Hrs: C: C: To prevent failure of the magneto impulse coupling assembly and possible engine failure	Superseded by 96-12-07

P/N:

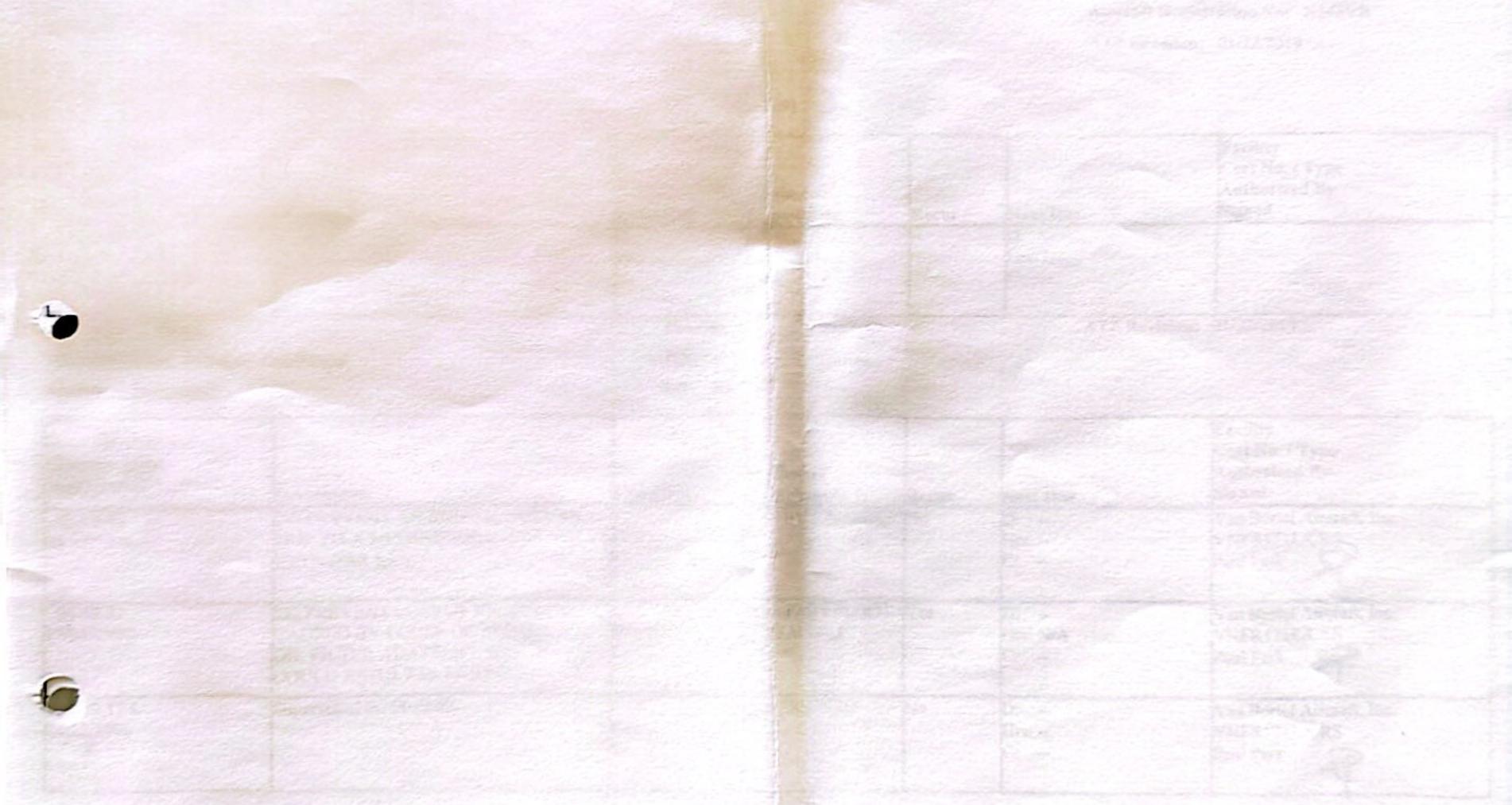
S/N:

Manufacturer: ACS Products Company

Model: IGNITION SWITCHES

01/22/2019 12:21 PM

Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed	
93-05-06 04/29/1993	TO PREVENT FAILURE OF IGNITION SWITCHES	 Hrs: C:	INSPECTION PER ACS SB92-01, DUE EACH 2000 HRS.	Yes	D: Hrs: 2000 HRS. TT C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park	



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CATP

engine

Company Make/Model Chautauqua Aircraft Sales, Inc Continental – TSIO-550-C

Tail # Serial #

N246VB

Location

1031118 Front Biweekly Number TSMOH 2024-22 TT:1131.1

Total Time

1131.1

Total Cycles

AD# Effective	Subject	Method of Compliance	ss	R	Date & Hours @ Comp	Next Comp @ Hrs/Date	Auth Sig & Cert No
12-03-06 02/24/2012	TO PREVENT AN IN-FLIGHT SHUTDOWN DUE TO A FAILED FUEL SERVO DIAPHRAGM	N/A Not Installed			12/28/2021	N/A	CRS# IC1R213K
16-16-12 09/15/2016	ECI CLASS 71 OR CLASS 76 CYLINDER ASSEMBLY FAILURES/	N/A by Cylinder P/N			12/28/2021	N/A	CRS# IC1R213K
16-21-04 11/18/2016	OIL COOLER CROSS FITTING, NIPPLE, AND BUSHING INTEGRITY/	PCW @ Cessna MFG.			12/28/2021	N/A	CRS# IC1R213K
20-16-11 09/21/2020	CROSS-FLOW CYLINDER ASSEMBLIES INTEGRITY/	Found PCW - Modification per MSB18-08A / MSB18-08B .			12/28/2021	N/A	CRS# IC1R213K
23-05-16 03/15/2023	IMPROPER INSTALLATION OF COUNTERWEIGHT RETAINING RINGS/	N/A Engine and component S/N unaffected			11/13/2024	N/A	CRS# IC1R213K
23-09-09 07/17/2023	FAILURES OF SPOT-WELDED, MULTI-SEGMENT V-BAND COUPLINGS/	N/A Not installed		x	11/13/2024	N/A	CRS# IC1R213K

Prepared by

Date 11/83/24

^{*} Indicates possible recurring inspections Tdata, Inc (Tdata.com)

airframe

Company

Chautauqua Aircraft Sales, Inc

Make/Model

Cessna - T240

Tail# Serial # N246VB T24002091

Biweekly Number TACH

2024-22 1131.1

A/C Cert. Date

Total Time

1131.1

AD# Effective	Subject	Method of Compliance	ss	R	Date & Hours @ Comp	Next Comp @ Hrs/Date	Auth Sig & Cert No
10-26-53 SS by 2010-26-54	TO PREVENT FAILURE OF THE WING DUE TO DISBONDING OF THE WING SKIN FROM THE WING SPAR/	SUPERCEDED	×		12/28/2021		CRS# IC1R213K
10-26-54 01/10/2011	TO INCREASE THE SERIAL NUMBERS IN THE APPLICABILITY SECTION AND TO CORRECT A SERIAL NUMBER IN THE APPLICABILITY SECTION/	N/A by A/C model / serial number.			12/28/2021		CRS# IC1R213K
11-03-04 03/14/2011	TO DETECT AND CORRECT DAMAGE IN THE RUDDER HINGES AND THE RUDDER HINGE BRACKETS/	N/A by A/C model / serial number.		x	12/28/2021		CRS# IC1R213K
13-11-10 07/26/2013	POH/AFM - MAXIMUM BRAKING LIMITATIONS AND PROCEDURES/	N/A by A/C model / serial number.			12/28/2021		CRS# IC1R213K
23-09-09 07/17/2023	FAILURES OF SPOT-WELDED, MULTI-SEGMENT V-BAND COUPLINGS/	N/A not installed		x	11/13/2024	N/A	CRS# IC1R213K

^{*} Indicates possible recurring inspections Tdata, Inc (Tdata.com)

Prepared by

Company

CHAUTAUQUA AIRCRAFT SALES, INC

Make/Model

Cessna - T240

Tail#

N246VB

Serial #

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T24002091

Biweekly Number

2022-01

Tach/Hour Meter

Tach = 942.5

A/C Cert. Date

Total Time

942.5

Subject	Method of Compliance	R	Date & Hours @ Comp	Next Comp @ Hrs/Date	Auth Sig & Cert No
CORRECTION - DETECT AND CORRECT CRACKS IN THE ELEVATOR TORQUE TUBE ASSEMBLY/	N/A by A/C model / serial number.	x	12/28/2021		CRS# IC1R213K
TO PREVENT A SHORT CIRCUIT CONDITION AT THE DEICE HEATER CONNECTOR/	N/A by no STC installed.		12/28/2021		CRS# IC1R213K
PREVENT JAMMING IN THE AILERON AND ELEVATOR CONTROL SYSTEMS/	N/A by A/C model / serial number.	X	12/28/2021		CRS# IC1R213K
TO PREVENT JAMMING IN THE AILERON AND ELEVAT OR CONTROL SYSTEMS/	N/A by A/C model / serial number.	X	12/28/2021		CRS# IC1R213K
TO DETECT AND CORRECT DAMAGE IN THE RUDDER HINGES AND THE RUDDER HINGE BRACKETS/	SUPERCEDED	x	12/28/2021		CRS# IC1R213K
TO PREVENT FAILURE OF THE WING DUE TO DISBONDING OF THE WING SKIN FROM THE WING SPAR/	SUPERCEDED		12/28/2021		CRS# IC1R213K
TO INCREASE THE SERIAL NUMBERS IN THE APPLICABILITY SECTION AND TO CORRECT A SERIAL NUMBER IN THE APPLICABILITY SECTION/	N/A by A/C model / serial number.		12/28/2021		CRS# IC1R213K
	CORRECTION - DETECT AND CORRECT CRACKS IN THE ELEVATOR TORQUE TUBE ASSEMBLY/ TO PREVENT A SHORT CIRCUIT CONDITION AT THE DEICE HEATER CONNECTOR/ PREVENT JAMMING IN THE AILERON AND ELEVATOR CONTROL SYSTEMS/ TO PREVENT JAMMING IN THE AILERON AND ELEVAT OR CONTROL SYSTEMS/ TO DETECT AND CORRECT DAMAGE IN THE RUDDER HINGES AND THE RUDDER HINGE BRACKETS/ TO PREVENT FAILURE OF THE WING DUE TO DISBONDING OF THE WING SKIN FROM THE WING SPAR/ TO INCREASE THE SERIAL NUMBERS IN THE APPLICABILITY SECTION AND TO CORRECT A	CORRECTION - DETECT AND CORRECT CRACKS IN THE ELEVATOR TORQUE TUBE ASSEMBLY/ TO PREVENT A SHORT CIRCUIT CONDITION AT THE DEICE HEATER CONNECTOR/ PREVENT JAMMING IN THE AILERON AND ELEVATOR CONTROL SYSTEMS/ TO PREVENT JAMMING IN THE AILERON AND ELEVAT OR CONTROL SYSTEMS/ TO DETECT AND CORRECT DAMAGE IN THE RUDDER HINGES AND THE RUDDER HINGE BRACKETS/ TO PREVENT FAILURE OF THE WING DUE TO DISBONDING OF THE WING SKIN FROM THE WING SPAR/ TO INCREASE THE SERIAL NUMBERS IN THE APPLICABILITY SECTION AND TO CORRECT A	CORRECTION - DETECT AND CORRECT CRACKS IN THE ELEVATOR TORQUE TUBE ASSEMBLY/ TO PREVENT A SHORT CIRCUIT CONDITION AT THE DEICE HEATER CONNECTOR/ PREVENT JAMMING IN THE AILERON AND ELEVATOR CONTROL SYSTEMS/ TO PREVENT JAMMING IN THE AILERON AND ELEVAT OR CONTROL SYSTEMS/ TO DETECT AND CORRECT DAMAGE IN THE RUDDER HINGES AND THE RUDDER HINGE BRACKETS/ TO PREVENT FAILURE OF THE WING DUE TO DISBONDING OF THE WING SKIN FROM THE WING SPAR/ TO INCREASE THE SERIAL NUMBERS IN THE APPLICABILITY SECTION AND TO CORRECT A	Subject Method of Compliance R Hours @ Comp CORRECTION - DETECT AND CORRECT CRACKS IN THE ELEVATOR TORQUE TUBE ASSEMBLY/ TO PREVENT A SHORT CIRCUIT CONDITION AT THE DEICE HEATER CONNECTOR/ PREVENT JAMMING IN THE AILERON AND ELEVATOR CONTROL SYSTEMS/ TO PREVENT JAMMING IN THE AILERON AND ELEVATOR CONTROL SYSTEMS/ TO DETECT AND CORRECT DAMAGE IN THE RUDDER HINGE BRACKETS/ TO PREVENT FAILURE OF THE WING DUE TO DISBONDING OF THE WING SPAR/ TO INCREASE THE SERIAL NUMBERS IN THE APPLICABILITY SECTION AND TO CORRECT A Method of Complance N/A by A/C model / serial number. X 12/28/2021 X 12/28/2021 12/28/2021 12/28/2021	Subject Method of Compliance R Hours @ Comp Next Comp @ Hrs/Date CORRECTION - DETECT AND CORRECT CRACKS IN THE ELEVATOR TORQUE TUBE ASSEMBLY/ TO PREVENT A SHORT CIRCUIT CONDITION AT THE DEICE HEATER CONNECTOR/ PREVENT JAMMING IN THE AILERON AND ELEVATOR CONTROL SYSTEMS/ TO PREVENT JAMMING IN THE AILERON AND ELEVATOR CONTROL SYSTEMS/ TO PREVENT JAMMING IN THE AILERON AND ELEVATOR CONTROL SYSTEMS/ TO DETECT AND CORRECT DAMAGE IN THE RUDDER HINGE BRACKETS/ TO PREVENT FAILURE OF THE WING DUE TO DISBONDING OF THE WING SKIN FROM THE WING SHAP WING SPAR/ TO INCREASE THE SERIAL NUMBERS IN THE APPLICABILITY SECTION AND TO CORRECT A N/A by A/C model / serial number. 12/28/2021 12/28/2021 12/28/2021

^{*} Indicates possible recurring inspections Tdata, Inc (Tdata.com)

Prepared by Syn Call

Date 1/14/22

airframe

Company

CHAUTAUQUA AIRCRAFT SALES, INC

Make/Model

Cessna - T240

Tail#

N246VB

T24002091

Serial #

2022-01

Biweekly Number Tach/Hour Meter

Tach = 942.5

A/C Cert. Date

Total Time

942.5

AD# Effective	Subject	Method of Compliance	R	Date & Hours @ Comp	Next Comp @ Hrs/Date	Auth Sig & Cert No	
* 11-03-04 03/14/2011	TO DETECT AND CORRECT DAMAGE IN	N/A by A/C model / serial number.	x	12/28/2021		CRS# IC1R213K	
	AND THE RUDDER HINGE BRACKETS/					The	
13-11-10 POH/AFM - MAXIMUM 07/26/2013 BRAKING LIMITATIONS AND PROCEDURES/		N/A by A/C model / serial number.		12/28/2021	2/28/2021 CRS# IC1R21		

Indicates possible recurring inspections Tdata, Inc (Tdata.com)

Company CHAUTAUQUA AIRCRAFT SALES, INC

Make/Model Continental -- TSIO-550-C

Location Front
Tail # N246VB
Serial # 1031118
Biweekly Number 2022-01

TSMOH Tach= 942.5

AD # Effective	Subject	Method of Compliance	R	Date & Hours @ Comp	Next Comp @ Hrs/Date	Auth Sig & Cert No
93-10-02 05/17/1993	CYLINDER VALVE RETAINER/	N/A by S/N		12/28/2021	N/A	CRS# IC1R213K
99-19-01 09/30/1999	CRANKSHAFT FAILURE/	N/A by S/N		12/28/2021	N/A	CRS# IC1R213K
00-23-21 12/12/2000	FRACTURE OF THE CRANKSHAFT CONNECTING ROD JOURNAL/	N/A by S/N		12/28/2021	N/A	CRS# IC1R213K
04-08-10 05/05/2004	PREVENT LOSS OF ENGINE POWER/	N/A Not Installed		12/28/2021	N/A	CRS# IC1R213K
07-16-10	TO PREVENT THE	N/A by S/N		12/28/2021	N/A	CRS# IC1R213K
08/23/2007	SEPARATING FROM THE SHAFT OF THE TURBOCHARGER DUE TO A MACHINING DEFECT IN THE TURBOCHARGER COMPRESSOR/					Kn
10-11-04 06/16/2010	TO PREVENT EXCESSIVE HYDRAULIC LIFTER WEAR/	N/A by Serial Number		12/28/2021	N/A	CRS# IC1R213K
12-03-06 02/24/2012	TO PREVENT AN IN-FLIGHT SHUTDOWN DUE TO A FAILED FUEL SERVO DIAPHRAGM/	N/A Not Installed		12/28/2021	N/A	CRS# IC1R213K
16-16-12 09/15/2016	ECI CLASS 71 OR CLASS 76 CYLINDER ASSEMBLY FAILURES/	N/A by Cylinder P/N		12/28/2021	N/A	CRS# IC1R213K
16-21-04 11/18/2016	OIL COOLER CROSS FITTING, NIPPLE, AND BUSHING INTEGRITY/	PCW @ Cessna MFG.		12/28/2021	N/A	CRS# IC1R213K
20-16-11 09/21/2020	CROSS-FLOW CYLINDER ASSEMBLIES INTEGRITY/	Found PCW - Modification per MSB18-08A / MSB18-08B .		12/28/2021	N/A	CRS# IC1R213K

Prepared by 3 M C Date 1/14/22

^{*} Indicates possible recurring inspections Tdata, Inc (Tdata.com)

propeller

AIRCRAFT RECORDS - DO NOT DESTROY

Company

CHAUTAUQUA AIRCRAFT SALES, INC

Make/Model

McCauley - D3A34C447

Location

Front

Tail#

N246VB

Serial #

101010

Biweekly Number

2022-01

TSMOH

Tach= 942.5

Subject Effective MAINTENANCE BY SOUTHERN CALIFORNIA PROPELLER SERVICE/		Method of Compliance	R	Date & Hours @ Comp	Next Comp @ Hrs/Date	Auth Sig & Cert No
		N/A No repair done.		12/28/2021	N/A	CRS# IC1R213K

^{*} Indicates possible recurring inspections Tdata, Inc (Tdata.com)

Appliances

Company

CHAUTAUQUA AIRCRAFT SALES, INC

Tail#

N246VB

Biweekly Number

2022-01

AD # Subject Effective		Method of Compliance	R	Date & Hours @ Comp	Next Comp @ Hrs/Date	Auth Sig & Cert No	
* 93-05-06 04/29/1993	93-05-06 - ACS IGNITION SWITCHES/	Due @ 2000 Hrs TIS	x	12/28/2021	Tach 2,000	CRS# IC1R213K	
20-19-06 11/03/2020	IDLER GEAR BEARING UNAPPROVED VARIANT/	Found PCW - N/A by S/N.		12/28/2021	N/A	CRS# IC1R213K	

Prepared by Tyr Coll Date 4/14/22

^{*} Indicates possible recurring inspections Tdata, Inc (Tdata.com)



12/01/2020 to 0/14/2022 SERVICE INFORMATION

BY TEXTRON AVIATION (Search Criteria

4 Records Found.

Pub Number

Title Description

Model/Variant

Serial Number

T240

T24002091

Compliance

Publication Type

M

It is the responsibility of the owner/operator to ensure that a Service Bulletin is applicable to their airplane before performing an inspection or modification. Warranty adjudication will not be administered based on search results. If necessary, contact Customer Support for technical assistance: 800-429-5372.

Pub No.	Pub Name	Date	Compliance	Title Description	
SEL-24-11 Electrical Power - Transmittal of Continental Aer				Electrical Power - Transmittal of Continental Aerospa Technologies Critical Service Bulletin CSB19-06	
SEL-79-03 filter Lut	Oil - Transmittal of Continental Aerospace Techno	04/21/2021 lac @ 6:1	Mandatory	Oil - Transmittal of Continental Aerospace Technologies CSB07-1A Connecting Rod Piston Pin Bushing Inspection	
SEL-71-10	Power Plant - Transmittal Of Continental Motors No Affection	01/26/2021 c/g/ nuts	Mandatory	Power Plant - Transmittal Of Continental Motors SB18-04A Replacement Of Flanged Nut	
SEL-71-11	Power Plant-Transmittal Of Continental Motors MSB	12/10/2020	Mandatory	Power Plant-Transmittal Of Continental Motors MSB 18-08C Inspection And Maintenance Modification Of Cylinder Assemblies	

Phone: 8174687788

FAA Airworthiness Directives Compliance Record

Company:

Category: Airframe

Manufacturer: Cessna Aircraft Company

Model: T240

Position:

P/N:

S/N: T24002091

Aircraft Registration No: N246VB

ATP Revision: 12/17/2020

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
		-	四日 三丁		D:	
		Hrs: C:			Hrs: C:	

Category: Engine Position: . ATP Revision: 12/17/2020

Manufacturer: Teledyne Continental P/N:

Model: TSIO-550-C S/N: 1031118

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
93-10-02 08/12/1993	TO PREVENT AN ENGINE FAILURE DUE TO A MISSING CYLINDER VALVE RETAINER KEY		N/A BY S/N	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
96-12-22 07/31/1996	TO PREVENT LOSS OF ENGINE OIL CAUSED BY LOOSE OR SEPARATED OIL FILTER ADAPTERS, WHICH COULD RESULT IN ENGINE, CONTD.	 Hrs: C:	N/A, PART IS NOT INSTALLED	Yes	D: Hrs: N/A C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
99-09-17 L 04/22/1999	Superseded by 99-19-01	 Hrs: C:	SUPERSEDED	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park

Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
To prevent crankshaft failure due to crankshaft cheek cracks, which could result in total engine power loss, contd.	 Hrs: C:	N/A BY S/N	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
Superseded by 2000-23-21	 Hrs: C:	SUPERSEDED	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
To prevent crankshaft connecting rod journal fracture, which could result in total engine power,contd.	 Hrs: C:	N/A BY S/N	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
Important for Cessna Oil Filter Adapter Assemblies listed in AD 96-12-22	 Hrs: C:	SEE AD 96-12-22	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
To prevent loss of engine power due to cracks in the cylinder head & possible engine failure caused, contd.	 Hrs: C:	N/A BY CYL. P/N	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
To prevent the turbine rotor from separating from the shaft of the turbocharger due to a machining defect in the,contd.	 Hrs: C:	N/A BY S/N'S OF TURBOS	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
Superseded by 2010-11-04	 Hrs: C:	SUPERSEDED	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
To prevent excessive hydraulic lifter wear, which can result in loss of engine power & loss of control of the airplane	 Hrs: C:	N/A BY ENGINE S/N	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
To prevent seizure of the turbocharger turbine, which could result in damage to the engine, & smoke in the,contd.	 Hrs: C:	N/A, HOUSINGS MARKED "JT"	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
	To prevent crankshaft failure due to crankshaft cheek cracks, which could result in total engine power loss, contd. Superseded by 2000-23-21 To prevent crankshaft connecting rod journal fracture, which could result in total engine power, contd. Important for Cessna Oil Filter Adapter Assemblies listed in AD 96-12-22 To prevent loss of engine power due to cracks in the cylinder head & possible engine failure caused, contd. To prevent the turbine rotor from separating from the shaft of the turbocharger due to a machining defect in the, contd. Superseded by 2010-11-04 To prevent excessive hydraulic lifter wear, which can result in loss of engine power & loss of control of the airplane To prevent seizure of the turbocharger turbine, which could result in damage to the	To prevent crankshaft failure due to crankshaft cheek cracks, which could result in total engine power loss, contd. Superseded by 2000-23-21 Hrs: C: To prevent crankshaft connecting rod journal fracture, which could result in total engine power, contd. Important for Cessna Oil Filter Adapter Assemblies listed in AD 96-12-22 Hrs: C: To prevent loss of engine power due to cracks in the cylinder head & possible engine failure caused, contd. C: To prevent the turbine rotor from separating from the shaft of the turbocharger due to a machining defect in the, contd. Superseded by 2010-11-04 Hrs: C: To prevent excessive hydraulic lifter wear, which can result in loss of engine power & loss of control of the airplane To prevent seizure of the turbocharger turbine, which could result in damage to the Hrs:	Description To prevent crankshaft failure due to crankshaft cheek cracks, which could result in total engine power loss, contd. Superseded by 2000-23-21 To prevent crankshaft connecting rod journal fracture, which could result in total engine power, contd. Important for Cessna Oil Filter Adapter Assemblies listed in AD 96-12-22 To prevent loss of engine power due to cracks in the cylinder head & possible engine failure caused, contd. To prevent the turbine rotor from separating from the shaft of the turbocharger due to a machining defect in the, contd. Superseded by 2010-11-04 To prevent excessive hydraulic lifter wear, which can result in loss of engine power & loss of control of the airplane To prevent seizure of the turbocharger turbine, which could result in damage to the Tirs: N/A BY S/N SUPERSEDED N/A BY S/N'S OF TURBOS C: N/A BY ENGINE S/N N/A, HOUSINGS MARKED "JT"	To prevent crankshaft failure due to crankshaft cheek cracks, which could result in total engine power loss, contd. Superseded by 2000-23-21 To prevent crankshaft connecting rod journal fracture, which could result in total engine power, contd. Important for Cessna Oil Filter Adapter Assemblies listed in AD 96-12-22 To prevent loss of engine power due to cracks in the cylinder head & possible engine failure caused, contd. To prevent the turbine rotor from separating from the shaft of the turbocharger due to a machining defect in the, contd. Superseded by 2010-11-04 To prevent excessive hydraulic lifter wear, which can result in loss of engine power & loss of control of the airplane Compliance N/A BY S/N No NO NO NO NO NO NO NO NO N	Description Complied Compliance To prevent crankshaft failure due to crankshaft cheek cracks, which could result in total engine power loss, contd. Superseded by 2000-23-21 Hrs: C: Superseded by 2000-23-21 Fro prevent crankshaft connecting rod journal fracture, which could result in total engine power, contd. Important for Cessna Oil Filter Adapter Assemblies listed in AD 96-12-22 Hrs: C: To prevent loss of engine power due to cracks in the cylinder head & possible engine failure caused, contd. To prevent the turbine rotor from separating from the shaft of the turbocharger due to a machining defect in the, contd. Superseded by 2010-11-04 To prevent excessive hydraulic lifter wear, which can result in loss of engine power & loss of control of the airplane C: N/A BY S/N No D: Hrs: C: N/A BY CYL. P/N No D: Hrs: C: N/A BY S/N'S OF No D: Hrs: C: To prevent excessive hydraulic lifter wear, which can result in loss of engine power & loss of control of the airplane To prevent seizure of the turbocharger urbine, which could result in damage to the Hrs: N/A BY ENGINE No D: N/A BY S/N'S OF No D: Hrs: C: N/A BY S/N'S OF No D: Hrs: C: N/A BY ENGINE No D: Hrs: C: -

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CS CamScanner

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
2012-03-06 C 02/24/2012	To prevent an in-flight engine shutdown due		N/A, SERVO NOT INSTALLED	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
2016-16-12 09/15/2016	To prevent failure of the cylinder assemblies, which could lead to failure of the engine, in-flight, contd.	 Hrs: C:	N/A BY CYL. P/N	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
2016-21-04 11/18/2016	To prevent failure of the oil cooler cross fitting and engine, IFSD, and loss of the airplane	 Hrs: C:	PCW DURING PRODUCTION, INSTALLATION IS IN COMPLIANCE WITH CSB15-2	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
2020-16-11 09/21/2020		11/15/2019 Hrs: 263.6 C:	MODIFICATION PER MSB18-08A & IN COMPLIANCE WITH MSB18-08B	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/FAA CRS Paul Park

Position: Category: Propeller

Manufacturer: McCauley

S/N: 101010 Model: D3A34C447

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
2005-14-11 08/17/2005	To prevent blade failure that could result in separation of a propeller blade and loss of control of the airplane	Hrs: C:	N/A, NO REPAIR BY SO. CAL. PROP. SERVICE	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park

ATP Revision: 12/17/2020 Position: Category: Turbocharger Systems

P/N:

P/N: 646677 Manufacturer: Hartzell Engine Tech

S/N: H-SBL00087 H-SBL00086 Model: (P/N) 466304-0003

Issue Number Effective Date Amendment #			Method of			Facility Cert No. / Type Authorized By
	Description	Complied	Compliance	Recur	Next Due	Signed

ssue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
56.17.07 07/(2:1896		 Hrs: C:		-	D: Hrs: C:	

Category: Magnetos Position: ATP Revision: 12/17/2020

Manufacturer: Teledyne Continental P/N: 10-500556-101

Model: SC-20 SERIES S/N: D15GA092 D15GA069

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
73-07-04 10/11/1973	Superseded by 94-01-03	 Hrs: C:	SUPERSEDED	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
74-26-09 12/24/1974	S-20,-200,-1200 SERIES MAGNETOS	 Hrs: C:	N/A BY S/N'S	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
78-09-07 R3 01/17/1983	Superseded by 96-12-07	 Hrs: C:	SUPERSEDED	Yes	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
82-20-01 06/14/1983		 Hrs: C:	N/A BY S/N'S	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
94-06-09 05/20/1994	TO PREVENT INJURY OR DEATH TO GROUND PERSONNEL DUE TO A NON- GROUNDED MAGNETO	 Hrs: C:	N/A BY S/N'S	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
94-01-03 R2 06/28/1995	TO PREVENT MAGNETO FAILURE AND SUBSEQUENT ENGINE FAILURE	 Hrs: C:	N/A BY S/N'S	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
96-12-07 07/18/1996	Superseded by 2005-12-06	 Hrs: C:	SUPERSEDED	Yes	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park
2005-12-06 07/19/2005	To prevent failure of the magneto impulse coupling assembly and possible engine failure	 Hrs: C:	N/A BY ENGINE MFR.	Yes	D: Hrs: N/A C:	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park

Category: Ignition Systems

Position:

ATP Revision: 12/17/2020

Manufacturer: ACS Products Company

P/N:

Model: IGNITION SWITCHES

S/N:

Issue Number Effective Date Amendment #	Description	Complied	Method of	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
93-05-06 04/29/1993	TO PREVENT FAILURE OF IGNITION SWITCHES		INSPECTION PER ACS SB92-01, DUE EACH 2000 HRS.	Yes	D: Hrs: 2000 HRS. TT C: -	Van Bortel Aircraft, Inc. VNFR171L/CRS Paul Park

Category: Appliance

Position:

ATP Revision: 12/17/2020

Manufacturer: McCaulcy

P/N: C290D3-R/T43

Model: C290D3-[X]/T[X]

S/N: 150365

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
2020-19-06 11/03/2020	To Prevent Failure of the Idler Gear Bearin This Failure Could Result in Failure of the Governor, Loss of Propeller Pitch Control, Engine and Propeller Over Speed, Engine Oil Contamination, and Loss of Control of the Airplane	Hrs: C:	N/A BY S/N & AFFECTED BEARING HAS NOT BEEN INSTALLED.	No	D: Hrs: C:	Van Bortel Aircraft, Inc. VNFR171L/FAA CRS Paul Park