

Full length article

# Shape optimisation of cold-formed steel columns with manufacturing constraints using the Hough transform

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## ABSTRACT

This paper introduces manufacturing constraints into a recently developed evolutionary algorithm for shape optimisation of CFS profiles. The algorithm is referred to as “self-shape optimisation” and uses Genetic Algorithm (GA) together with the Augmented Lagrangian (AL) method to avoid ill-conditioned problems. Simple manufacturing rules derived from the limitations of current cold-forming processes, i.e. a limited ability to form continuously curved surfaces without discrete bends, are described in the paper and incorporated into the algorithm. The Hough transform is used to detect straight lines and transform arbitrarily drawn cross-sections into manufacturable ones. Firstly, the algorithm is verified against a known optimisation problem and found to accurately converge to a manufacturable optimum solution. Secondly, the algorithm is applied to singly-symmetric CFS columns each of which is subject to an axial compressive load of 75kN and has a uniform wall thickness of 1.2 mm. The strength of the columns is evaluated by the Direct Strength Method (DSM) and all buckling modes are considered. Various column lengths (from 500 mm to 3000 mm) and numbers of roll-forming bends were investigated. The optimised cross-sections are presented and discussed.

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## 1. Introduction

Cold-formed steel (CFS) profiles are usually an attractive and cost-effective building structural solution relative to more “conventional” building materials, such as hot-rolled steel and concrete. They are thin-walled structural members with a high capacity-to-weight ratio [1] and can be pre-fabricated off-site and readily installed on-site.

The main advantage of CFS members lies in their manufacturing processes that allow the formation of almost any cross-sectional shape at room temperature. The profiles are typically formed by bending coils of thin steel sheets (up to 6 mm thick) with a finite number of rollers (roll-forming) or die blocks (brake-pressing). However, despite this flexibility, the use of CFS sections has been mainly restricted to Cee, Zed and Sigma cross-sectional shapes, with or without stiffeners, as shown in Fig. 1. As the cross-sectional shape controls the strength of CFS members, there is a real potential to eventually discover new optimised cross-sectional shapes tailored to specific applications, such as purlins, girts and studs for buildings, and uprights for storage racks for instance. Such discoveries will enhance the competitiveness of CFS

structures and are now facilitated by the development of a new structural design method, the Direct Strength Method (DSM) [2], which allows any cross-sectional shapes with the same degree of complexity to be designed.

This paper aims at incorporating manufacturing constraints for optimising the cross-sectional shape of CFS columns by minimising the cross-sectional area for a given design axial compressive load. Optimisation for specific applications is not considered in the present study, and will be investigated in the future. Simple roll-forming (or brake-pressing) manufacturing rules are defined and incorporated into the recently developed “self-shape optimisation” algorithm [3,4] as manufacturing constraints. The specificity of the algorithm is briefly described in the paper. The Hough transform, used to detect straight lines and transform non-manufacturable cross-sections into manufacturable ones, is explained herein. The algorithm is verified against a known optimisation problem, being the optimisation of the cross-sectional shape of a doubly-symmetric closed thin-walled profile for given second moments of area. The verified algorithm is then used to optimise simply-supported, singly-symmetric and free-to-warp open-section CFS columns. The optimised manufacturable cross-sections are presented and discussed in the paper. For comparison purposes, the algorithm is also run to obtain non-manufacturable cross-sections.

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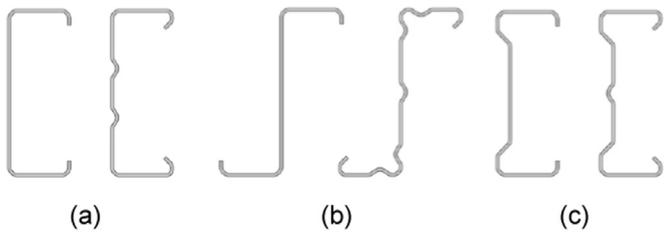


Fig. 1. Conventional CFS profiles with or without stiffeners, (a) Cee, (b) Z and (c)  $\Sigma$ -sections.

## 2. Background

### 2.1. General

One of the early studies on shape optimisation of CFS profiles is attributed to Liu et al. [5]. A knowledge-based global optimisation algorithm, aiming at optimising the capacity of CFS columns manufactured from coils of set width and thickness, was used. Leng et al. [6] optimised the cross-sectional shapes of CFS open columns using three different optimisation algorithms, namely gradient-based steepest descent method, Genetic Algorithm (GA) and Simulated Annealing (SA). Sections having a wall thickness of 1 mm and a perimeter of 280 mm were divided into 21 elements and optimised. “Open circular” and “S” cross-sections were found. Moharrami et al. [7] improved the study in [6] by introducing various types of boundary conditions into the algorithm. Gilbert et al. [3] proposed a GA-based Augmented Lagrangian (AL) constraint-handling shape optimisation method for CFS profiles. The accuracy of the algorithm was verified against an optimisation problem for which an analytical solution is known. Gilbert et al. [4] then applied the verified algorithm to optimise the cross-sectional shape of CFS simply-supported, singly-symmetric and open-section columns, subjected to a certain axial compressive load. A set of rules to accurately determine the local and distortional elastic buckling stresses from the Finite Strip signature curves was also developed.

Manufacturing constraints were first introduced into shape optimisation algorithms for CFS profiles by Leng et al. [8]. The authors introduced partial manufacturing and construction (geometric end uses) constraints using an SA algorithm. The constraints were implemented by defining (i) flat “horizontal” flanges, (ii) minimum dimensions for the “vertical” web, flanges and lips, (iii) allowance passage for utilities between lips and (iv) no overlapping elements in the cross-sections. The study was improved in Leng et al. [9] by introducing a limited number of rollers (representing the number of discrete bends between flat segments, see Section 2.3). These enhancements resulted in manufacturable cross-sections with improved capacities when compared to conventional Cee-sections of identical cross-sectional area. Leng et al. [10] also presented optimised cross-sectional shapes, i.e. singly-symmetric “Cee” and “Sigma” and anti-symmetric “S”, with both manufacturing and construction constraints. Franco et al. [11] proposed CFS shape grammar rules, with an “alphabet”, for shape optimisation of CFS profiles. Manufacturing constraints, with given stiffener sizes, were intrinsic to the shape grammar resulting in manufacturable cross-sections. Genetic Algorithm (GA) was used in [11] as a search algorithm.

### 2.2. Present shape optimisation algorithm

The algorithm referred to as “self-shape optimisation” and developed in [3,4] is used in the current study. The method rigorously explores the natural evolution process and the latent potential of GA in an innovative way. GA was initially developed by

Holland [12] and is an adaptive heuristic search algorithm that mimics the Darwin’s evolutionary survival of the fittest theory. It is less susceptible to be self-trapped into local optima, and is able to handle non-linear problems. The classical GA principles can be found in Adeli and Sarma [13].

GA is an unconstrained optimisation method, and constrained problems are transformed into unconstrained problems by using a fitness function  $f$  expressed as,

$$f = f(x) + \sum_{i=1}^n \alpha_i g_i(x) + \sum_{i=n+1}^k \beta_i h_i(x) \quad (1)$$

where  $f(x)$  is the objective function,  $x$  is the vector of design variables,  $g_i(x)$  and  $h_i(x)$  are the  $i$ th inequality and equality constraint violations ( $n$  inequality and  $k-n$  equality constraints), respectively, and  $\alpha_i$  and  $\beta_i$  are penalty factors. The algorithm aims at minimising  $f$ .

In theory, the penalty factors in Eq. (1) increase when the constraint violations decrease and convergence can be facilitated by increasing the penalty factors. Nevertheless, large values of the penalty factors lead to ill-conditioned problems or slow down the algorithm [14,15]. To avoid the problem of having penalty factors increasing to infinity, the Augmented Lagrangian (AL) constraint-handling method for GA proposed by Adeli and Cheng [14] is used in this research to solve the problem. The fitness function  $f$  is then re-arranged as,

$$f = f(x) + \frac{1}{2} \left\{ \sum_{i=1}^n \gamma_i \left[ \max[0, g_i(x) + \mu_i] \right]^2 + \sum_{i=n+1}^k \gamma_i \left[ h_i(x) + \mu_i \right]^2 \right\} \quad (2)$$

where  $\gamma_i$  and  $\mu_i$  are penalty function coefficients and real parameters associated with the  $i$ th inequality and equality constraints, respectively.  $\gamma_i$  and  $\mu_i$  are automatically updated at each GA generation but are kept to finite values [14].

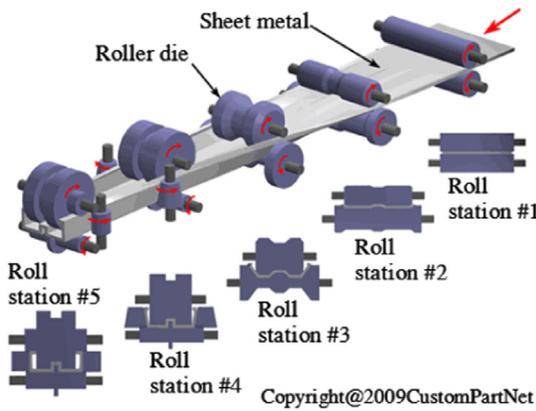
The main characteristics of the “self-shape optimisation” principle [3] are summarised below:

- The initial population in the GA is generated by arbitrarily drawing cross-sections using self-avoiding random walks in a defined design space. These random walks enable cross-sections to be generated without presumptions of their shapes.
- A floating-point type GA is used, implying that a cross-section is defined by floating-point numbers representing the coordinates of the points constituting the cross-section.
- Cross-over and mutation operators are performed in relation to the design space but not to the floating-point variables. The cross-over operator allows for the merging of two cross-sections to generate off-springs bearing similarity in cross-sectional shapes to the two parents. In the mutation operator, a part of the cross-section is deleted and redrawn.

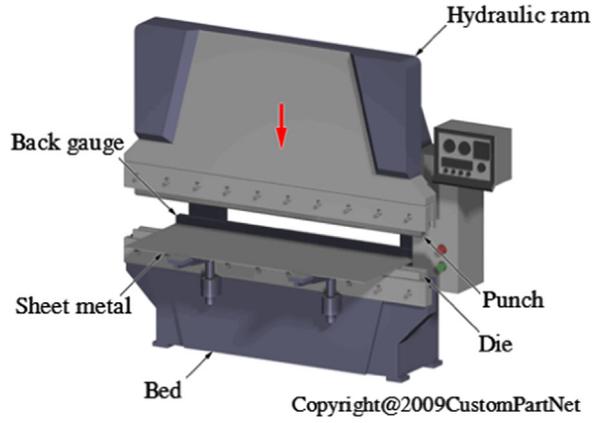
### 2.3. Manufacturing constraints

#### 2.3.1. Traditional manufacturing processes

CFS profiles are typically mass-produced by two main cold-forming processes, referred to as “roll-forming” and “brake-pressing”. Both processes involve bending a flat sheet of steel to a desired cross-sectional shape. In roll-forming operations, as shown in Fig. 2(a), the sheet is gradually rolled to a desired cross-sectional shape through successive rollers. This continuous manufacturing process allows long profiles to be manufactured. In brake-pressing operations, as shown in Fig. 2(b), the sheet is repetitively pressed between differently shaped brake punches and die blocks to bend it to the desired cross-sectional shape. Brake-pressing is limited in manufacturing long members. Both manufacturing processes can only bend the flat sheet of metal at discrete bending locations,



(a) roll-forming



(b) brake-pressing

Fig. 2. Cold-forming processes (Courtesy of CustomPartNet Inc.).

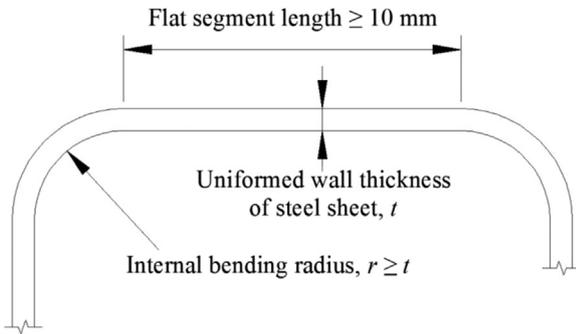


Fig. 3. Manufacturing rules.

leaving flat (straight) segments between bends. This limitation needs to be considered in the shape optimisation algorithms to obtain manufacturable cross-sections.

### 2.3.2. Simple manufacturing rules

Simple manufacturing rules have been defined herein based on the basic roll-forming constraints encountered by a European steel storage rack manufacturer. They consist of three main rules:

- (1) The minimum internal bending radius  $r$  to steel sheet thickness  $t$  ratio is 1.0 (Fig. 3);
- (2) The minimum length of a single flat segment is 10 mm (Fig. 3);
- (3) The number of flat segments per cross-section cannot exceed 20 (i.e. a maximum number of 19 bends per open cross-section and 10 flat segments per half cross-section).

In the present study, Rule (1) is neglected since it does not affect the basic shape of the optimised manufacturable cross-section, and only Rules (2) and (3) are considered. A nil internal bending radius (i.e. perfect bends) is assumed to simplify the algorithm. Actual bending radii can be added to the optimised cross-section prior to manufacture.

### 2.3.3. Hough transform

The Hough transform is used in this paper to detect straight lines, i.e. flat manufacturable segments, in the cross-section. This transform is commonly used in image processing to detect regular shapes, such as straight lines, circles and ellipses, from the discrete points forming the image [16].

The method consists of defining a “parametric space” in which

each straight line in the image is represented by its orientation angle  $\theta$ , with respect to the Cartesian  $x$ -axis, and its normal distance  $r$  to the origin, as shown in Fig. 4(a). If  $\theta$  is restricted to the interval  $[0^\circ; 180^\circ]$ , each straight line is represented by a unique coordinate  $(r, \theta)$  in the parametric space. An image point of coordinate  $(x_i, y_i)$  in the Cartesian  $x$ - $y$  space is transformed into a sinusoidal curve in the parametric  $r$ - $\theta$  space as

$$r = x_i \cos \theta + y_i \sin \theta \quad (3)$$

Sinusoidal curves having common intersecting points have collinear (aligned) points in the image. This is illustrated in Fig. 4 (b) with 4 points aligned on the line of coordinate  $(r=10 \text{ mm}, \theta=60^\circ)$  in the parametric space.

For image processing purposes, an array referred to as the accumulator array (or accumulator matrix), is created in the discretised parametric space. The columns of the array correspond to the increasing values of  $\theta$ , at  $\Delta\theta$  intervals, and the lines to increasing values of  $r$ , at  $\Delta r$  intervals. Aligned image points are detected as,

- Step 1: Set  $\theta=0^\circ$ .
- Step 2: For each image point (i) calculate its  $r$  value from Eq. (3) for the set value of  $\theta$ , (ii) calculate the closest discrete  $r$  value matching the lines of the accumulator array and (iii) add the point reference number to the corresponding cell in the accumulator array.
- Step 3: Set  $\theta=\theta+\Delta\theta$ . If  $\theta \geq 180^\circ$  go to Step 4, else go to Step 2.
- Step 4: All points sharing the same cell in the accumulator array are considered aligned.

The choice of the intervals  $\Delta\theta$  and  $\Delta r$  influences the ability and accuracy of the Hough transform in detecting straight lines. The smaller  $\Delta\theta$ , the more refined the search space. A larger value of  $\Delta r$  (wider corridor) represents a less stringent alignment tolerance, as illustrated in Fig. 5, where two values of  $\Delta r$  are shown. A larger  $\Delta r_1$  results in all the four points in Fig. 5 being aligned by the Hough transform. A smaller  $\Delta r_2$  results in only two points being aligned by the Hough transform.

### 2.3.4. Manufacturing constraints in the GA

The manufacturing constraints are introduced into the fitness function (see Eq. (1)) as an equality constraint  $h_{align}$ , expressed as,

$$h_{align} = \omega \left| \frac{nbAligned}{nbElement} - 1 \right| \quad (4)$$

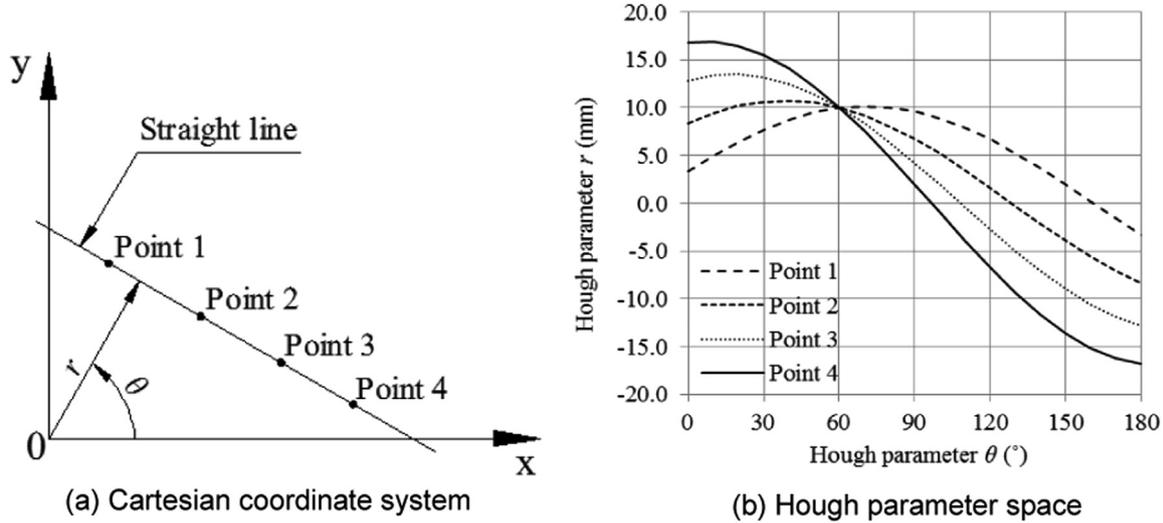


Fig. 4. Hough transform from Cartesian space to Hough parametric space.

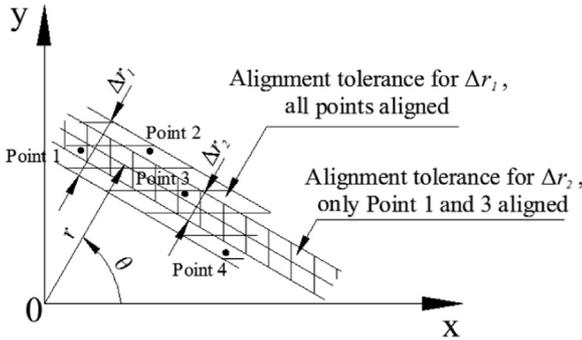


Fig. 5. Alignment tolerance for Hough transform.

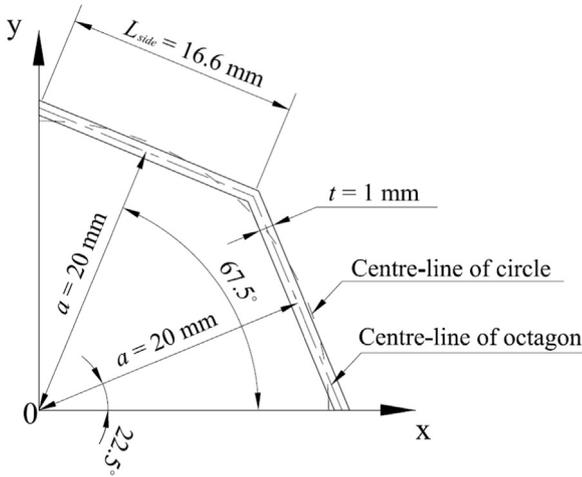


Fig. 6. Optimum octagon, only a quarter section shown.

Table 1  
AL penalty function coefficient  $\gamma_{align}$  and associated weight  $\omega$  selected for parametric study.

Combination	$\gamma_{align}$			
	0.01	0.10	1.00	
$\omega$	0.3	(1)	(4)	(7)
	0.5	(2)	(5)	(8)
	0.7	(3)	(6)	(9)

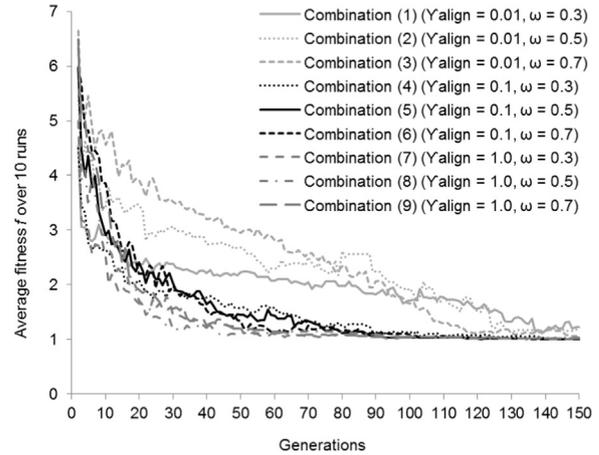


Fig. 7. Average fitness  $f$  for parametric study of AL penalty function coefficient  $\gamma_{align}$  and weight  $\omega$ .

where  $\omega$  is a weight associated with the constraint,  $nbElement$  is the total number of elements per half cross-section and  $nbAligned$  represents the number of aligned elements composed of the longest non-concurrent flat segments per half cross-section. The number of non-concurrent flat segments is taken to  $N_{max}$  if there are more than  $N_{max}$  non-concurrent flat segments per half cross-section or to the actual number of non-concurrent flat segments otherwise.  $N_{max}$  corresponds to a maximum number of flat segments per half cross-section set by the manufacturer, with  $N_{max}$  less than or equal to the maximum possible number of flat segments defined in Rule (3) outlined in Section 2.3.2. In the algorithm, a flat segment is determined from the Hough transform as consecutive aligned cross-sectional elements of total length equal to or greater than the minimum manufacturable length as defined in Rule (2) in Section 2.3.2. If the cross-section is made of less than  $N_{max}$  flat segments, the constraint is considered satisfied and  $h_{align} = 0$ .

### 3. Validation

#### 3.1. Optimisation problem

A similar optimisation problem to the one used in [3], for which an analytical solution exists, is used herein to verify the ability and

**Table 2**  
Average results over 10 runs for parametric study of  $\gamma_{align}$  and  $\omega$ .

Combination	Cross-sectional area			Second moment of area $I_x$		Second moment of area $I_y$		$nbAligned/nbElement$	
	$A_s$ (mm <sup>2</sup> )	Error <sup>a</sup> (%)	CoV (%)	Error <sup>b</sup> (%)	CoV (%)	Error <sup>b</sup> (%)	CoV (%)	Error (%)	CoV (%)
(1)	132.01	-0.4	0.5	-0.1	3.7	-0.2	2.1	-1.1	3.4
(2)	132.32	-0.2	0.6	0.5	2.3	-0.3	1.6	0.0	0.0
(3)	132.36	-0.1	0.6	0.5	2.0	0.4	0.8	0.0	0.0
(4)	132.13	-0.3	0.4	0.1	0.6	0.4	0.6	0.0	0.0
(5)	131.96	-0.4	0.2	0.1	0.2	0.1	0.3	0.0	0.0
(6)	132.20	-0.3	0.3	0.0	0.2	0.4	0.6	0.0	0.0
(7)	132.88	0.2	0.3	0.2	0.7	0.6	1.0	0.0	0.0
(8)	133.12	0.4	0.9	1.4	4.0	0.2	0.5	0.0	0.0
(9)	133.71	0.9	0.6	1.0	2.5	0.6	1.1	0.0	0.0

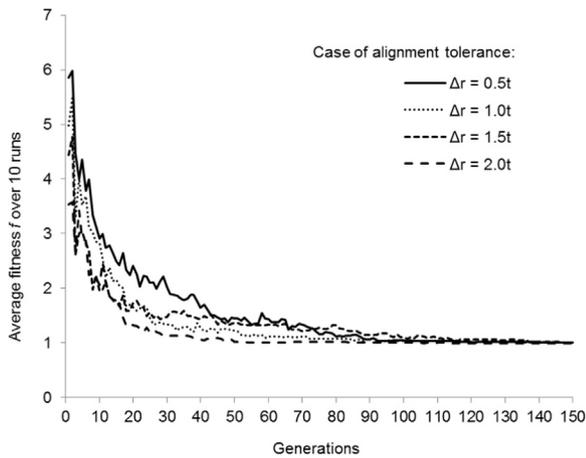
Note: the negative error represents the optimised results are less than the optimum octagon's

<sup>a</sup> Relative error when compared to  $A_o = 132.55$  mm<sup>2</sup> of the optimum octagon.

<sup>b</sup> Relative error when compared to  $I_{xt} = I_{yt}$  are 28,043.3 mm<sup>4</sup> of the optimum octagon.

**Table 3**  
Alignment tolerance  $\Delta r$  selected for parametric study.

$\Delta r$ (mm)	$\Delta r/L_{ele}$
0.5t	0.03
1.0t	0.06
1.5t	0.09
2.0t	0.12



**Fig. 8.** Average fitness  $f$  for parametric study of alignment tolerance  $\Delta r$ .

accuracy of the algorithm in optimising manufacturable cross-sectional columns. It consists of minimising the cross-sectional area  $A_s$  of a thin-walled bisymmetric closed cross-section for given

second moments of area,  $I_{xt}$  and  $I_{yt}$ , about the two axes of symmetry. Ragnedda and Serra [17] indicated that, when  $I_{xt}$  equals  $I_{yt}$ , the optimised cross-section is a circle and therefore a regular polygon of  $n$  sides if the cross-section is manufactured with  $n$  flat segments.

A regular octagon ( $n=8$ ) with apothem  $a$  (at mid-wall thickness) of 20 mm and wall thickness  $t$  of 1 mm is used herein to verify the algorithm. The cross-sectional area of the octagon  $A_o$  is 132.55 mm<sup>2</sup> and the length  $L_{side}$  of one side for the octagon is therefore 16.6 mm. Its second moments of area  $I_{xt} = I_{yt}$  are 28,043.3 mm<sup>4</sup>. As the problem is bisymmetric, only a quarter of the cross-section in Fig. 6 is optimised herein and the maximum number of flat segments  $N_{max}$  is therefore set to 2. The fitness function  $f$  derived from (Eqs. (1) and 4) is expressed as,

$$\text{Minimise } f = \frac{A_s}{A_o} + \alpha_x \max\left(0, 1 - \frac{I_x}{I_{xt}}\right) + \alpha_y \max\left(0, 1 - \frac{I_y}{I_{yt}}\right) + \alpha_{align} \omega \left| \frac{nbAligned}{nbElement} - 1 \right| \quad (5)$$

where  $I_x$  and  $I_y$  are the calculated second moments of area of the cross-section, and  $\alpha_x$ ,  $\alpha_y$  and  $\alpha_{align}$  are penalty factors associated with each constraint. In Eq. (5), the constraints on the given second moments of area are expressed as inequality constraints. In other words, the algorithm is not penalised if  $I_x \geq I_{xt}$  or  $I_y \geq I_{yt}$ . This was found to have significantly improved convergence.

In Fig. 6, the circle with the same second moment of area and wall thickness as the octagon is also shown for comparison. The cross-sectional area  $A_c$  of the circle is 130.31 mm<sup>2</sup>, i.e. 1.7% less than that of the manufacturable octagon.

The AL fitness function used in the algorithm and derived from (Eqs. (2) and 5) is given as,

**Table 4**  
Average results over 10 runs for parametric study of  $\Delta r$ .

Case of $\Delta r$	Cross-sectional area			Second moment of area $I_x$		Second moment of area $I_y$		$nbAligned/nbElement$	
	$A_s$ (mm <sup>2</sup> )	Error <sup>a</sup> (%)	CoV (%)	Error <sup>b</sup> (%)	CoV (%)	Error <sup>b</sup> (%)	CoV (%)	Error (%)	CoV (%)
0.5t	131.96	-0.4	0.2	0.1	0.2	0.1	0.3	0.0	0.0
1.0t	131.06	-1.1	0.1	-0.1	0.2	0.1	0.1	0.0	0.0
1.5t	131.23	-1.0	0.1	0.2	0.4	0.1	0.3	0.0	0.0
2.0t	130.43	-1.6	0.0	0.0	0.1	0.0	0.1	0.0	0.0

Note: the negative error represents the optimised results are less than the optimum octagon's.

<sup>a</sup> Relative error when compared to  $A_o = 132.55$  mm<sup>2</sup> of the optimum octagon.

<sup>b</sup> Relative error when compared to  $I_{xt} = I_{yt}$  are 28,043.3 mm<sup>4</sup> of the optimum octagon.

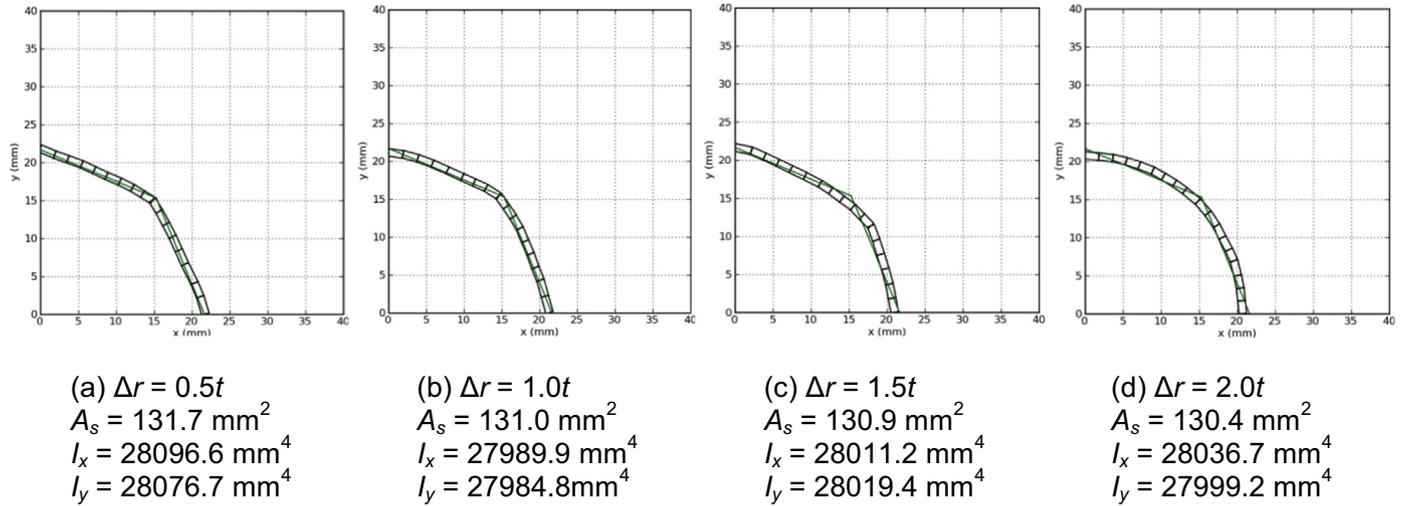


Fig. 9. Fittest optimised cross-sections at the last generation (150th) for all cases of  $\Delta r$  from (a) to (d).

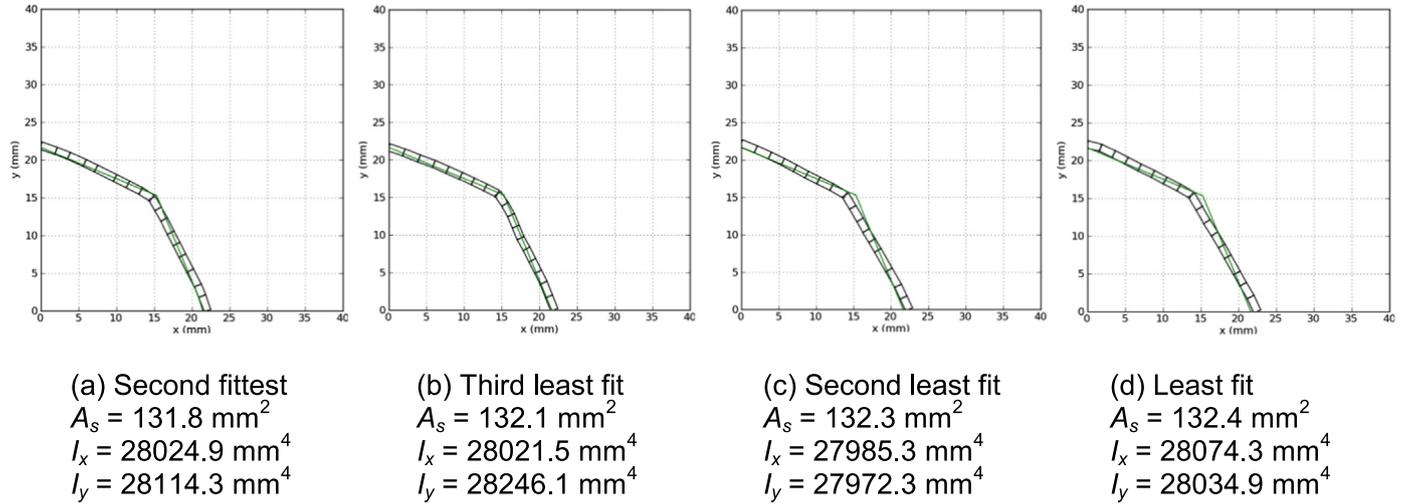


Fig. 10. Optimised cross-sections at the last generation (150th), for the second fittest (a), third fittest (b), second least fit (c) and least fit (d) cross-sections out of 10 runs for the case  $\Delta r = 0.5t$ .

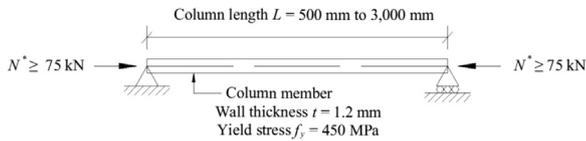


Fig. 11. Optimisation problem.

$$f = \frac{A_s}{A_o} + \frac{1}{2} \left\{ \gamma_x \left[ \max(0, 1 - \frac{I_x}{I_{xt}} + \mu_x) \right]^2 + \gamma_y \left[ \max(0, 1 - \frac{I_y}{I_{yt}} + \mu_y) \right]^2 + \gamma_{align} \left[ \omega \left| \frac{nbAligned}{nbElement} - 1 \right| + \mu_{align} \right]^2 \right\} \quad (6)$$

where  $\gamma_x$ ,  $\gamma_y$  and  $\gamma_{align}$  are the AL penalty function coefficients,  $\mu_x$ ,  $\mu_y$  and  $\mu_{align}$  are the real parameters associated with each penalty function coefficient. Gilbert et al. [3] investigated the appropriate initial values for  $\gamma_x$  and  $\gamma_y$ , and their recommendation of  $\gamma_x = \gamma_y = 2.0$  is used herein. For initial values of  $\gamma_x$  and  $\gamma_y$  less than 0.5,

the algorithm tends to select cross-sections with a small number of elements as the fittest ones. The appropriate initial value of  $\gamma_{align}$ , associated with  $\omega$ , is investigated in Section 3.3.1. Initial values of  $\mu_x = \mu_y = \mu_{align} = 0$  are used, as recommended by Belegundu and Arorat [18].

### 3.2. Other parameters used

An AL penalty increasing constant  $\beta$  of 1.05 and a convergence rate  $\rho$  of 1.5, as recommended in [3], are used. A value of  $\beta$  greater than 1.5 forces the algorithm to converge prematurely [3]. The design space is set to 40 mm  $\times$  40 mm [3] and the maximum number of generations to 150 per run. Ten runs are performed to verify the robustness of the algorithm. The number of individuals is set to 700 per generation and the cross-sections are drawn with short elements of nominal length of 2 mm (see [3] for more details). The probabilities of cross-over and mutation operations in the GA are equal to 80% and 1%, respectively, as used in [3].

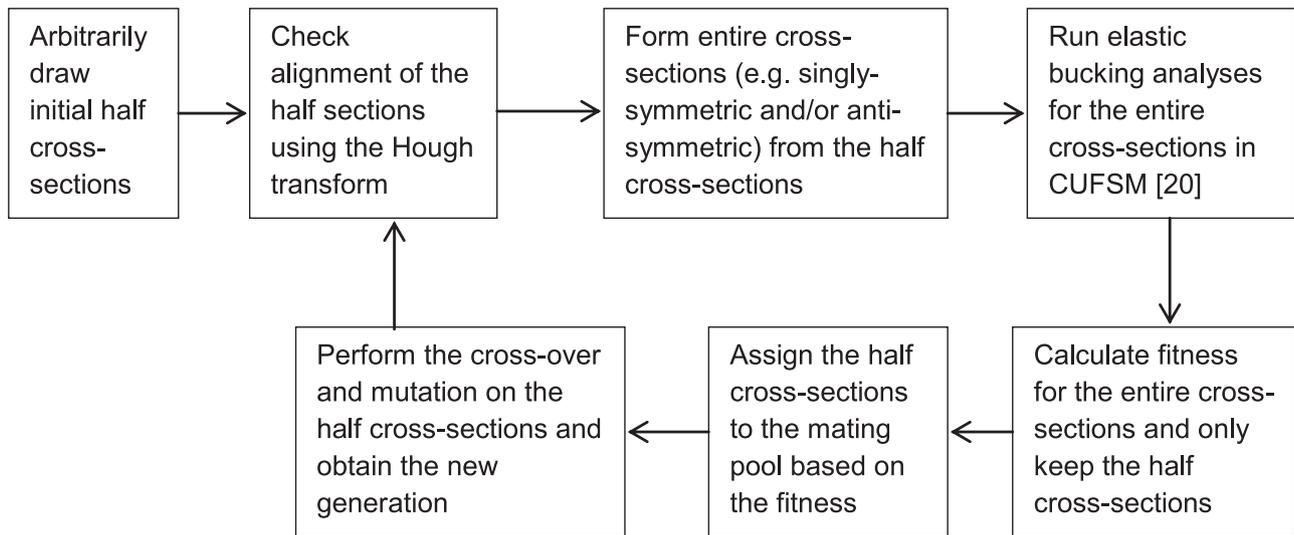
### 3.3. Parametric studies and results

#### 3.3.1. AL penalty function coefficient $\gamma_{align}$ and associated weight $\omega$

In order to determine the appropriate initial value for  $\gamma_{align}$  and constant value for weight  $\omega$ , nine different combinations are

**Table 5**  
Alignment tolerance  $\Delta r$  proposed for various  $N_{max}$  and column length ( $t=1.2$  mm).

500 mm Column				1500 mm Column				3000 mm Column			
$N_{max}$	$\Delta r$ (mm)	$L_{max}$ (mm)	$\Delta r/L_{max}$	$N_{max}$	$\Delta r$ (mm)	$L_{max}$ (mm)	$\Delta r/L_{max}$	$N_{max}$	$\Delta r$ (mm)	$L_{max}$ (mm)	$\Delta r/L_{max}$
3	1.5t	40	0.045	3	2.5t	50	0.060	3	4.0t	65	0.074
4	1.3t	32	0.049	4	2.5t	44	0.068	4	3.7t	60	0.074
5	1.0t	32	0.038	5	2.0t	36	0.067	5	3.5t	46	0.091
6	1.0t	32	0.038	6	1.5t	36	0.050	6	3.5t	46	0.091
–	–	–	–	7	1.5t	36	0.050	7	3.0t	40	0.090
–	–	–	–	–	–	–	–	8	2.5t	36	0.083



**Fig. 12.** Simplified flowchart of the algorithm.

investigated, as shown in Table 1. The value of  $\gamma_{align}$  and  $\omega$  are selected in the intervals [0, 1]. The alignment tolerance and step of orientation angle in the Hough transform are set to  $\Delta r=0.5t$  and  $\Delta\theta=0.5^\circ$ , respectively, in this section.

Fig. 7 illustrates the average fitness function  $f$  given in Eq. (5) over 10 runs, with  $\alpha=10$ , for all studied combinations of  $\gamma_{align}$  and  $\omega$ . The convergence rate improves as  $\gamma_{align}$  increases from 0.01 to 1.0. The algorithm tends not to converge when  $\gamma_{align}$  is too small (equal to 0.01), as the weight of the alignment penalty function in the AL fitness function in Eq. (6) is too small when compared to the objective function. The algorithm, on the other hand, is able to converge to a solution for the remaining studied combinations (4) to (9).

Table 2 compares the average results over 10 runs to the known optimum solution for all studied combinations. A negative sign in Table 2 means that the optimised results are less than the optimum solution. Combination (1), i.e.  $\gamma_{align}=0.01$  and  $\omega=0.3$ , experiences incomplete alignment with a relative alignment error of  $-1.1\%$ , while all the remaining combinations allow complete alignment. The average second moments of area are similar to the optimum ones for all combinations within a maximum average error of 1.4% (combination (8)). Yet, the coefficient of variation (CoV) on the second moments of area is minimum for  $\gamma_{align}=0.1$  (combination (4) to (6)). The minimum average cross-sectional area of  $131.96\text{ mm}^2$  is found for combination (5) and is 0.4% less than the one of the optimum octagon, therefore implying this combination represents a more optimal solution. Combination (5), i.e.  $\gamma_{align}=0.1$  and  $\omega=0.5$ , is therefore chosen for further analysis in this study.

### 3.3.2. Alignment tolerance $\Delta r$

Table 3 shows the values of  $\Delta r$  and corresponding alignment tolerance ratios  $\Delta r/L_{side}$  analysed to find the appropriate value of the alignment tolerance to ensure convergence and accuracy of the algorithm.  $\Delta\theta$  is kept constant at  $0.5^\circ$ .

Similar to Fig. 7, Fig. 8 plots the average fitness function  $f$  over 10 runs for all studied cases of  $\Delta r$ . The algorithm always converges to a solution. The larger the alignment tolerance  $\Delta r$ , the faster the convergence rate is, with  $\Delta r=2.0t$  converging at the 50th generation approximately. The algorithm converges in about 100 generations for  $\Delta r=0.5t$ .

Table 4 summarises the average results for all studied cases of  $\Delta r$ . The algorithm always satisfies the alignment criteria and converges to consistent solutions with small CoVs (within 0.5%) for all cases. In the case  $\Delta r=2.0t$ , the average relative errors on the cross-sectional area and second moments of area about the two axes of symmetry are  $-1.6\%$  (CoV=0.0%) and  $0.0\%$  (CoV=0.1%), respectively. This average cross-sectional area ( $130.43\text{ mm}^2$ ) is closer to (less than 0.1%) the absolute optimum circle cross-sectional area than the targeted octagon. To illustrate, Fig. 9 plots the fittest optimised cross-sections, with the wall thickness of 1 mm shown, at the 150th (final) generation for all cases of  $\Delta r$ . The larger the alignment tolerance is, the closer the cross-section to the absolute optimum circle is. A “circle” like shape is mainly observed for the case  $\Delta r=2.0t$ , while an “octagon” like shape is mainly observed for the other cases. Fig. 10 shows the second, third, second least and least fit optimised cross-sections, out of ten runs, for the stringiest alignment tolerance,  $\Delta r=0.5t$ . The fittest cross-section for this case is given in Fig. 9(a). All cross-sections

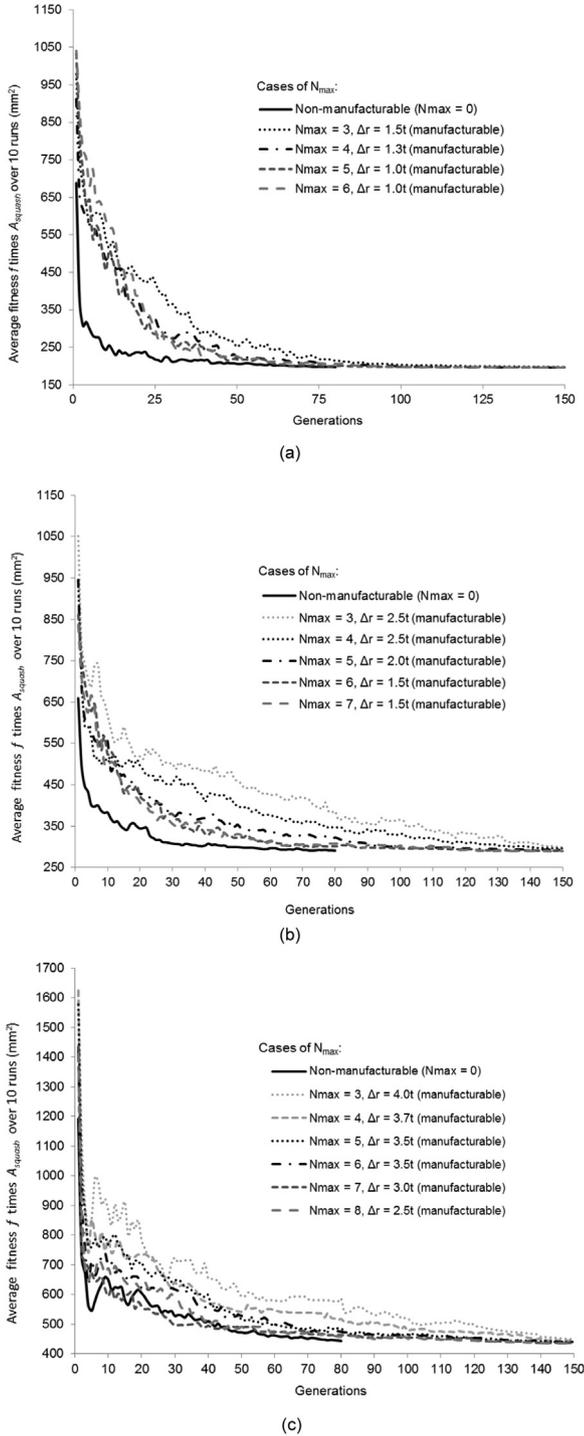


Fig. 13. Average fitness  $f$  for the (a) 500 mm, (b) 1500 mm and (c) 3000 mm long columns.

can converge to a consistent “octagon” shape outlining the robustness of the algorithm.

Based on the above results and as a simple rule, an alignment tolerance ratio  $\Delta r/L_{side}$  greater than 0.03 but no more than 0.1 is used in the Hough transform through this research, as a compromise between accuracy and convergence rate. Alignment tolerance ratios  $\Delta r/L_{side}$  less than 0.03 would allow more optimal shapes, i.e. closer to the octagon, but will require a large number of generations to be analysed and may cause convergence issues due to a stringent alignment tolerance.

## 4. Optimisation of CFS columns

### 4.1. Optimisation problem

The validated algorithm is applied to minimise the cross-sectional area  $A_s$  of manufacturable CFS columns subjected to an axial compressive force  $N^*$  of 75 kN. The columns are simply-supported, symmetric and free to warp, with a uniform wall thickness  $t$  of 1.2 mm. Column lengths of 500 mm (short), 1500 mm (intermediate) and 3000 mm (long) are investigated. The yield stress  $f_y$  of the column is 450 MPa, the Young’s modulus  $E$  is 200 GPa and the shear modulus  $G$  is 80 GPa. The optimisation problem is illustrated in Fig. 11. The algorithm is also run without considering the manufacturing constraints for comparison purposes.

The constrained optimisation problem is transformed into an unconstrained problem suitable for the GA and involves minimising the fitness function  $f$ ,

$$f = \frac{A_s}{A_{squash}} + \alpha \max\left(0, \frac{N^*}{N_c} - 1\right) + \alpha_{align} \omega \left| \frac{nbAligned}{nbElement} - 1 \right| \quad (7)$$

where the first term in the equation represents the objective.  $A_{squash}$  is the squash area, defined as the lower bound cross-sectional area of the profile:

$$A_{squash} = \frac{N^*}{f_y} \quad (8)$$

The second term in Eq. (7) represents the constraint on the axial capacity and is expressed as an inequality constraint. In other words, the cross-section is not penalised if its capacity exceeds the targeted capacity of 75 kN.  $N_c$  is the nominal member capacity in compression, calculated from the Australian standard AS4600 [19] (see Section 3.1), and  $\alpha$  is the penalty factor associated with the capacity constraint. The last term represents the manufacturing constraint and is expressed as an equality constraint (see Section 2.3.4).  $N_{max}$  is the maximum possible number of flat segments per half cross-section set by the manufacturer, but is less than the upper limit of flat segments defined in Rule (3) in Section 2.3.2. In the algorithm, a flat segment is determined from the Hough transform as consecutive aligned cross-sectional elements of a total length equal to or greater than the minimum manufacturable length (10 mm), defined in Rule (2) in Section 2.3.2. If the half cross-section is made of less than  $N_{max}$  flat segments, the constraint is considered satisfied.

The AL method for GA proposed by Adeli and Cheng [14] is also used herein to handle the axial capacity and manufacturing constraints. The fitness function  $f$  is then expressed as,

$$f = \frac{A_s}{A_{squash}} + \frac{1}{2} \left\{ \gamma \left[ \max\left(0, \frac{N^*}{N_c} - 1 + \mu\right) \right]^2 + \gamma_{align} \left[ \omega \left| \frac{nbAligned}{nbElement} - 1 \right| + \mu_{align} \right]^2 \right\} \quad (9)$$

where  $\gamma$  and  $\gamma_{align}$  are the penalty function coefficients, and  $\mu$  and  $\mu_{align}$  are the real parameters associated with each penalty function coefficient. From the parametric study in Section 3 and [3], initial values of  $\gamma=2.0$ ,  $\gamma_{align}=0.1$ ,  $\mu=\mu_{align}=0$  and  $\omega=0.5$  are used. An AL penalty increasing constant  $\beta$  of 1.05 and a convergence rate  $\rho$  of 1.5 are also used [3].

### 4.2. Column capacity

#### 4.2.1. Design method for CFS columns

Similar to [4], the DSM [2] as detailed in Section 7 of the Australian Standard AS4600 [19] is used in this research to

**Table 6**  
Average results over 10 runs for the 500 mm long columns.

$N_{max}$	Cross-sectional area		Nominal member capacity			Ultimate compressive stress	Alignment	
	$A_s$ (mm <sup>2</sup> )	CoV (%)	$N_c$ (kN)	Error <sup>b</sup> (%)	CoV (%)	$N_d/A_s$ (MPa)	Error (%)	CoV (%)
3	196.8	0.64	74.99	0.06	0.08	381.0	0.00	0.00
4	195.4	0.33	75.01	0.04	0.05	383.9	0.00	0.00
5	196.1	0.63	74.98	0.05	0.07	382.4	0.00	0.00
6	195.6	0.74	75.00	0.03	0.04	383.4	0.00	0.00
$\infty^a$	194.7	0.48	75.00	0.16	0.19	385.2	–	–

<sup>a</sup> Algorithm ran without manufacturing constraints (non-manufacturable cross-section).

<sup>b</sup> Absolute error when compared to  $N^* = 75$  k N.

**Table 7**  
Average results over 10 runs for the 1500 mm long columns.

$N_{max}$	Cross-sectional area		Nominal member capacity			Ultimate compressive stress	Alignment	
	$A_s$ (mm <sup>2</sup> )	CoV (%)	$N_c$ (kN)	Error <sup>b</sup> (%)	CoV (%)	$N_d/A_s$ (MPa)	Error (%)	CoV (%)
3	297.0	0.91	75.30	0.40	0.49	253.6	0.0	0.00
4	292.1	0.56	74.97	0.23	0.29	256.7	0.0	0.00
5	290.4	0.27	74.99	0.12	0.13	258.5	0.0	0.00
6	289.6	0.35	75.07	0.06	0.06	259.0	0.0	0.00
7	289.1	0.31	75.01	0.04	0.05	259.4	0.0	0.00
$\infty^a$	288.1	0.26	74.97	0.16	0.20	260.2	–	–

<sup>a</sup> Algorithm ran without manufacturing constraints (non-manufacturable cross-section).

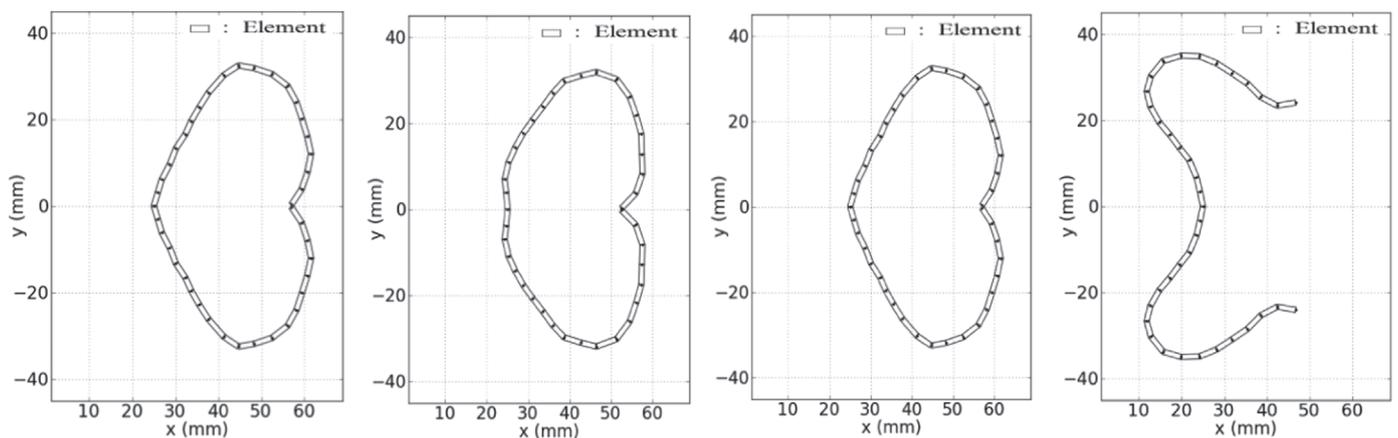
<sup>b</sup> Absolute error when compared to  $N^* = 75$  k N.

**Table 8**  
Average results over 10 runs for the 3000 mm long columns.

$N_{max}$	Cross-sectional area		Nominal member capacity			Ultimate compressive stress	Alignment	
	$A_s$ (mm <sup>2</sup> )	CoV (%)	$N_c$ (kN)	Error <sup>b</sup> (%)	CoV (%)	$N_d/A_s$ (MPa)	Error (%)	CoV (%)
3	444.7	0.64	75.37	0.50	0.35	169.5	0.0	0.00
4	440.3	0.32	75.30	0.40	0.32	171.0	0.0	0.00
5	436.5	0.20	75.15	0.20	0.30	172.2	0.0	0.00
6	435.8	0.23	75.10	0.13	0.34	172.3	0.0	0.00
7	434.3	0.19	75.03	0.04	0.16	172.8	0.0	0.00
8	434.5	0.29	75.02	0.02	0.08	172.7	0.0	0.00
$\infty^a$	435.3	0.35	75.16	0.21	0.54	172.7	–	–

<sup>a</sup> Algorithm ran without manufacturing constraints (non-manufacturable cross-section).

<sup>b</sup> Absolute error when compared to  $N^* = 75$  k N.



(a)  $A_s = 193.8$  mm<sup>2</sup>  
 $N_c = 75.16$  kN  
 $N_d/A_s = 387.8$  MPa

(b)  $A_s = 193.7$  mm<sup>2</sup>  
 $N_c = 75.07$  kN  
 $N_d/A_s = 387.6$  MPa

(c)  $A_s = 195.8$  mm<sup>2</sup>  
 $N_c = 74.90$  kN  
 $N_d/A_s = 382.5$  MPa

(d)  $A_s = 196.9$  mm<sup>2</sup>  
 $N_c = 74.81$  kN  
 $N_d/A_s = 379.9$  MPa

**Fig. 14.** Optimised cross-sections for the 500 mm long columns and the non-manufacturable case, (a, b) fittest cross-sections and (c, d) least fit cross-sections.

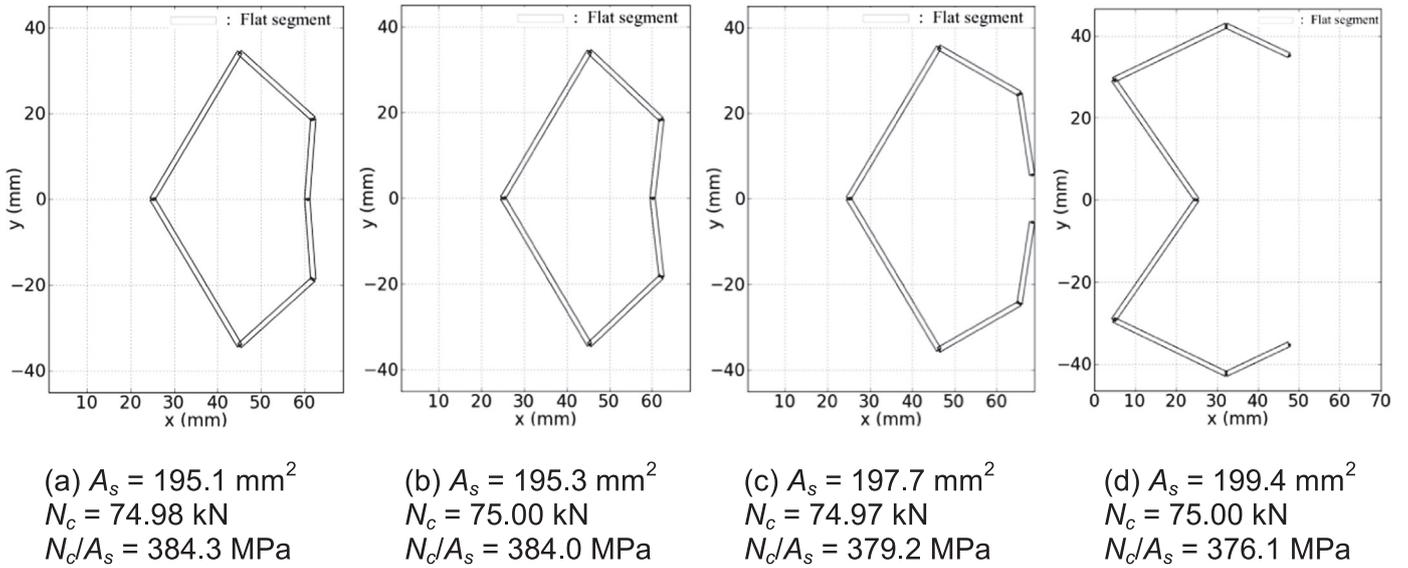


Fig. 15. Optimised cross-sections for the 500 mm long columns and  $N_{max}=3$ , (a, b) fittest cross-sections and (c, d) least fit cross-sections.

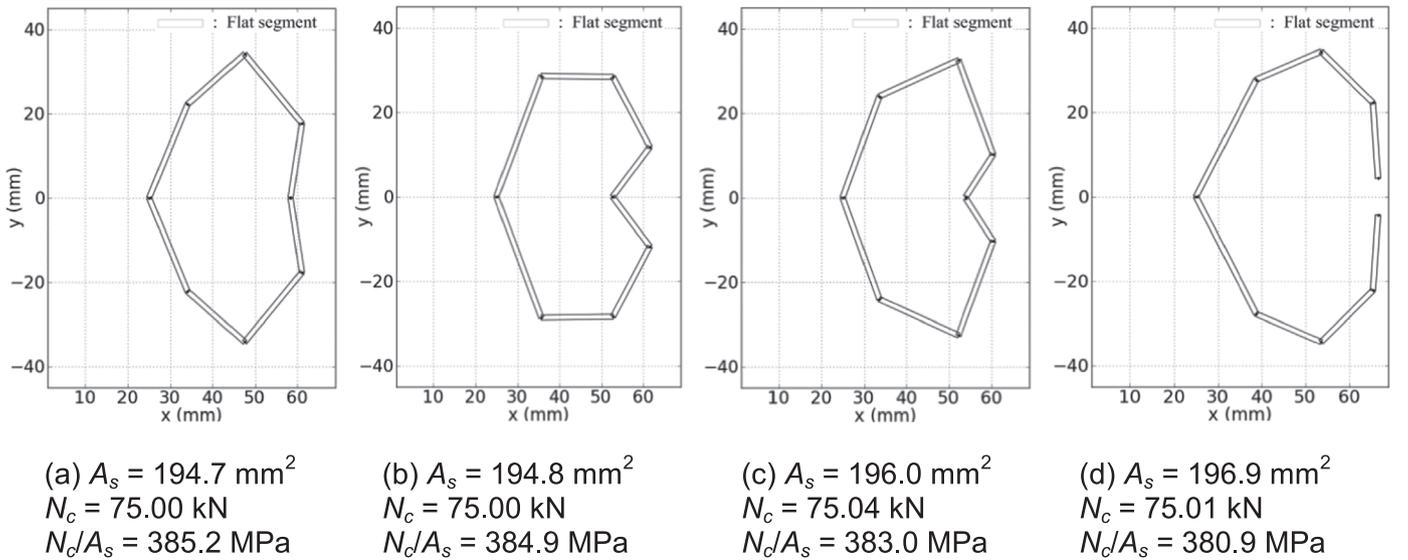


Fig. 16. Optimised cross-sections for the 500 mm long columns and  $N_{max}=4$ , (a, b) fittest cross-sections and (c, d) least fit cross-sections.

calculate the nominal member capacity in compression,  $N_c$  of the cross-sections given as,

$$N_c = \min(N_{ce}, N_{cl}, N_{cd}) \quad (10)$$

where  $N_{ce}$ ,  $N_{cl}$  and  $N_{cd}$  are the nominal axial compressive capacities for global, local and distortional buckling, respectively (see [19] for more details).

#### 4.2.2. Determination of the elastic buckling stresses

In this study, the elastic global buckling stress  $f_{oc}$  of the cross-sections is estimated from Timoshenko's buckling theory, as given in Clause 3.4.3 for singly-symmetric open cross-section of the Australian Standard AS4600 [19]. The rules developed in [4] to automatically estimate the elastic local and distortional buckling stresses  $f_{ol}$  and  $f_{od}$ , respectively, from the Finite Strip Method (FSM) [20] and the constrained Finite Strip Method (cFSM) [21] signature curves, are used. The open source software CUFSM [22] is used to perform the Finite Strip analyses.

#### 4.3. Other parameters

To study the influence of the maximum number of discrete bends on the optimised cross-sectional shape, various maximum numbers of flat segments  $N_{max}$  per half cross-section are investigated. The number of investigated flat segments  $N_{max}$  considered for each column length is given in Table 5.  $\Delta r$  is adjusted to the value of  $N_{max}$  and column lengths according to the results obtained from Section 3.3.2. Specifically, the maximum length of the cross-sectional elements  $L_{max}$  for the optimised solutions was first estimated for each combination of  $N_{max}$  and column length by pre-running the algorithm with a large alignment tolerance.  $\Delta r$  was then chosen for each case to satisfy  $\Delta r/L_{max}$  within the interval [0.03, 0.1]. The value of  $\Delta r$  for each studied case is given in Table 5. The step of orientation angle  $\Delta\theta$  is set to  $1^\circ$  for all cases.

In view of the parameters of the GA, the design space is set to  $100 \text{ mm} \times 100 \text{ mm}$  [3] and the maximum number of generations per run to 150 for manufacturable cross-sections and 80 [4] for non-manufacturable ones. 10 runs are performed. The number of individuals per generation is set to 500 and the cross-sections are

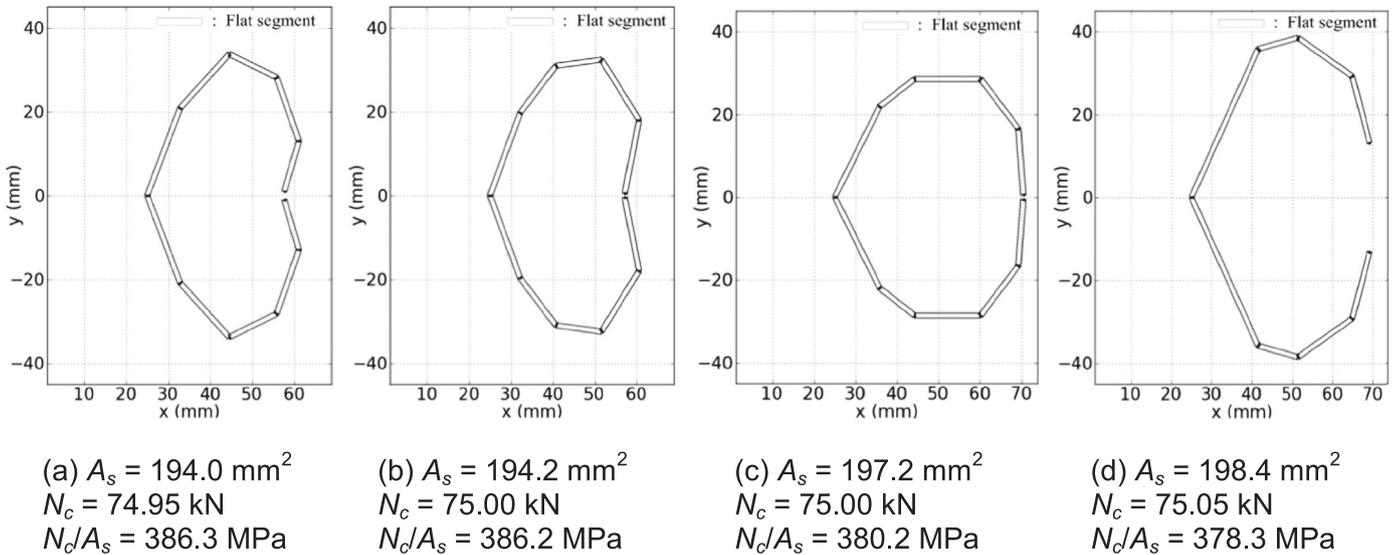


Fig. 17. Optimised cross-sections for the 500 mm long columns and  $N_{max}=5$ , (a, b) fittest cross-sections and (c, d) least fit cross-sections.

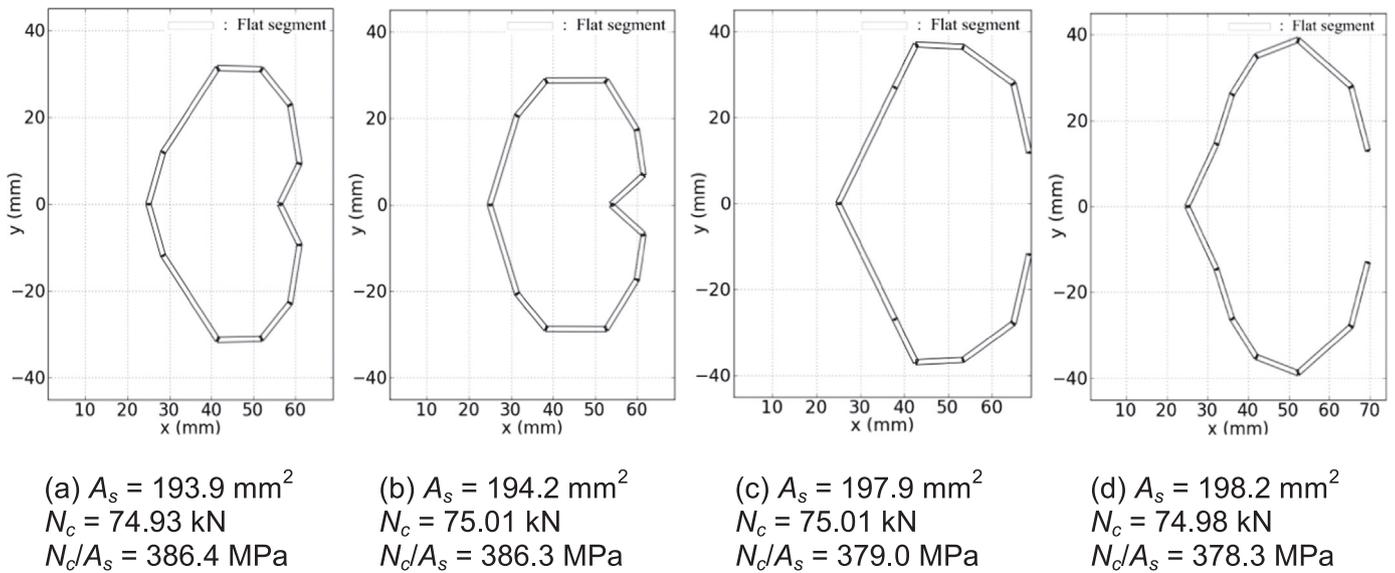


Fig. 18. Optimised cross-sections for the 500 mm long columns and  $N_{max}=6$ , (a, b) fittest cross-sections and (c, d) least fit cross-sections.

arbitrarily drawn with short elements of nominal length of 4 mm [3]. The cross-over and mutation operators for singly-symmetric cross-sections detailed in [4] apply. The probabilities of the cross-over and mutation operators in the GA are equal to 80% and 1%, respectively [3].

To improve computation time when elements are aligned, i.e. forming a flat segment, only the coordinates of the flat segments are entered into CUFISM [22]. However, for accuracy of the Finite Strip analysis [10], flat segments between 10 mm and 15 mm in length are divided into two segments of equal length, and flat segments longer than 15 mm are divided into three segments of equal length.

A simplified flowchart is presented in Fig. 12 showing that half cross-sections are used in the GA and entire cross-sections are only formed for elastic buckling analyses in CUFISM [22]. The detailed flowchart of the algorithm is shown in [3]. While singly-symmetric and/or anti-symmetric sections can be readily incorporated into the algorithm, only singly-symmetric columns are optimised in this study.

#### 4.4. Results and discussions

##### 4.4.1. Convergence

Fig. 13 shows the average fitness functions  $f$  given in Eq. (7) times the squash area  $A_{squash}$  over 10 runs for 500 mm (Fig. 13(a)), 1500 mm (Fig. 13(b)) and 3000 mm (Fig. 13(c)) long columns. In Fig. 13, the penalty factor in Eq. (7) is set to  $\alpha=10$ . The algorithm always converges to an optimised solution. Typically, the higher  $N_{max}$ , the fastest the convergence is, likely because the length of the flat segments of the optimised cross-sections increases when  $N_{max}$  decreases. The algorithm always converges the fastest for the non-manufacturable cases. The algorithm approaches an optimised solution at about the 100th generation for all manufacturable cases of the 500 mm long columns (Fig. 13(a)),  $N_{max}=5$  to 7 of the 1500 mm long columns (Fig. 13(b)) and  $N_{max}=5$  to 8 of the 3000 mm long columns (Fig. 13(c)). An optimum solution is approached after the 140th generation for the remaining cases. For the non-manufacturable cases, 60 generations are sufficient to approach an optimised solution for all column lengths.

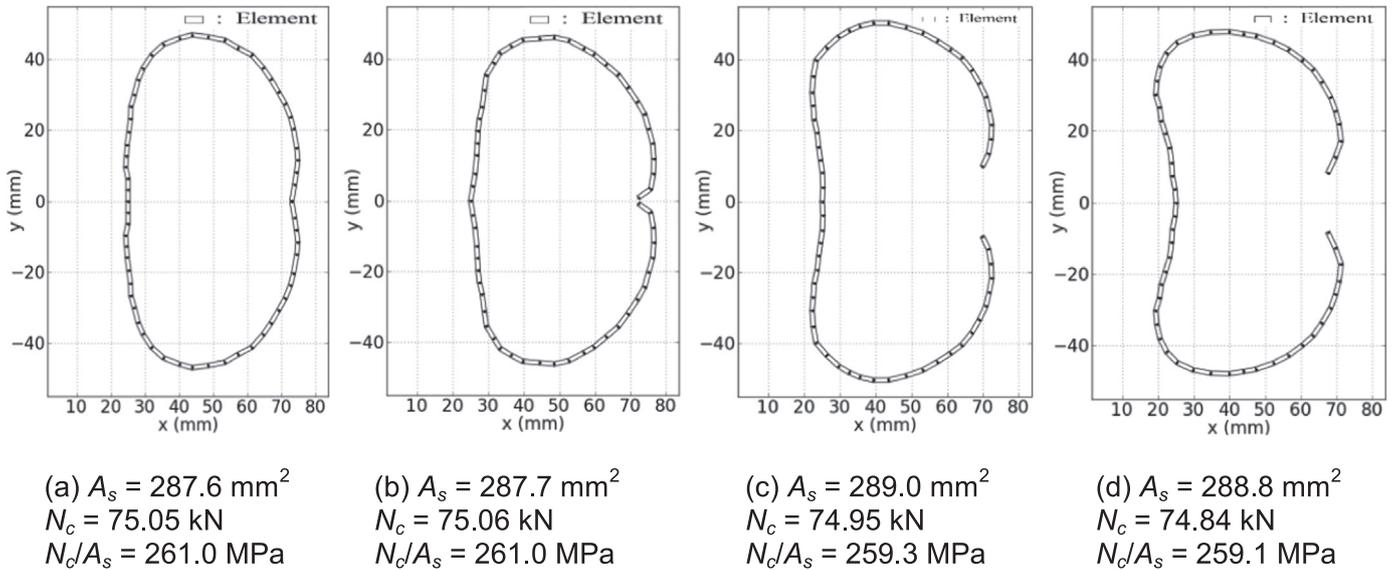


Fig. 19. Optimised cross-sections for the 1500 mm long columns and the non-manufacturable case, (a, b) fittest cross-sections and (c, d) least fit cross-sections.

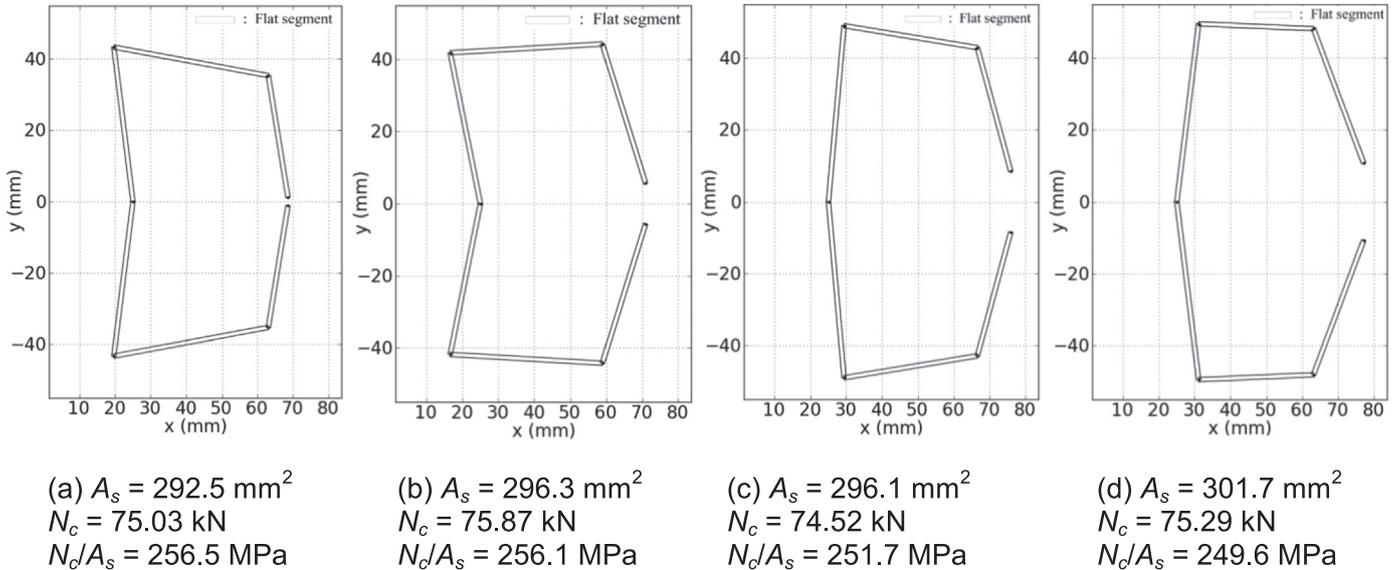


Fig. 20. Optimised cross-sections for the 1500 mm long columns and  $N_{max}=3$ , (a, b) fittest cross-sections and (c, d) least fit cross-sections.

The computation time is highly dependent on the number of cross-sectional elements analysed in CUFSM [22] (written in MATLAB), to perform the Finite Strip analyses. The greater the number of elements, the longer the computation time is. The main algorithm is written in Python and does not use parallel processing, while the Finite Strip analyses operated in CUFSM does. Up to 8 computer cores are selected per MATLAB analysis in a high performance computer cluster consisting of a mixture of SGI Altix XE and SGI® Rackable™ C2114-4TY14 servers for this purpose. The Finite Strip analyses take about 80% of the computation time and the main algorithm takes the remaining 20%. It takes on average 20, 35 and 65 min to optimise one generation for the manufacturable columns and 30, 50 and 75 min for the non-manufacturable ones for the 500 mm, 1500 mm and 3000 mm long columns, respectively.

In this study, a strict maximum of 75,000 solutions are evaluated (150 generations  $\times$  500 cross-sections) per run with the algorithm often converging in less than 50,000 solutions. This number is of the same order of magnitude as the 40,000 solutions evaluated in [6,23].

#### 4.4.2. Average results

Table 6 to Table 8 summarise the average results over 10 runs for the 500 mm, 1500 mm and 3000 mm long columns, respectively. When the algorithm is run with manufacturing constraints, the algorithm always satisfies the alignment criteria and converges to consistent solutions with small CoVs on the cross-sectional area (maximum of 0.91% when  $N_{max}=3$  in Table 7). This further confirms the robustness of the algorithm. The average nominal member capacity  $N_c$  is always close to the targeted axial compressive capacity of 75 kN, with the largest average absolute error equal to 0.50% for the 3000 mm long columns when  $N_{max}=3$  (see Table 8). For all cases, except for  $N_{max}=7$  and 8 of the 3000 mm long columns, the average ultimate compressive stress (nominal axial capacity-to-area ratio) is always lower than the same of the non-manufacturable solutions, as the latter are more optimum. However, for  $N_{max}=7$  and 8 of the 3000 mm long columns (Table 8), the average ultimate compressive stress is up to 0.6% higher than the same of the non-manufacturable cross-sections. This outcome is due to the extremely close cross-sectional shapes between the manufacturable and non-manufacturable cases and the

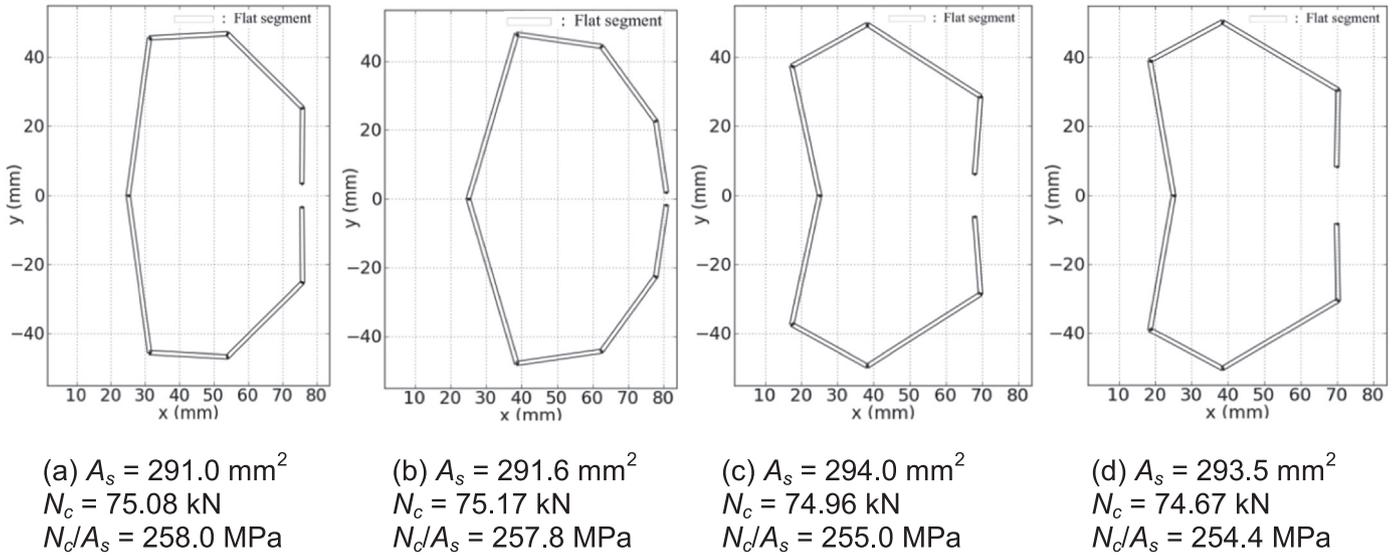


Fig. 21. Optimised cross-sections for the 1500 mm long columns and  $N_{max}=4$ , (a, b) fittest cross-sections and (c, d) least fit cross-sections.

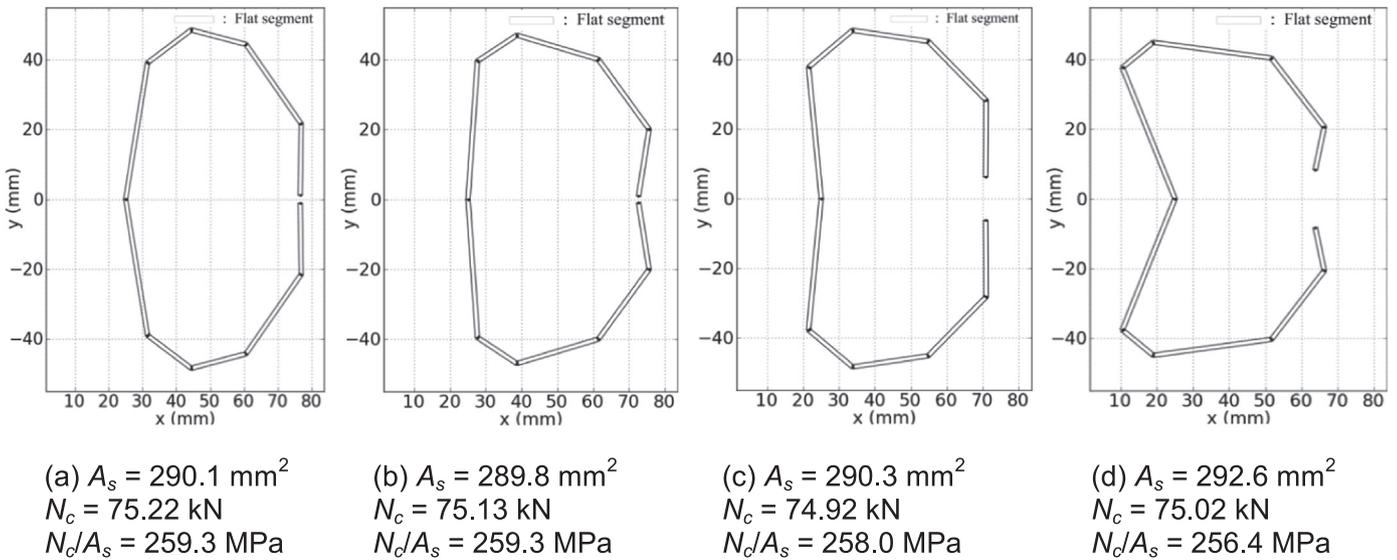


Fig. 22. Optimised cross-sections for the 1500 mm long columns and  $N_{max}=5$ , (a, b) fittest cross-sections and (c, d) least fit cross-sections.

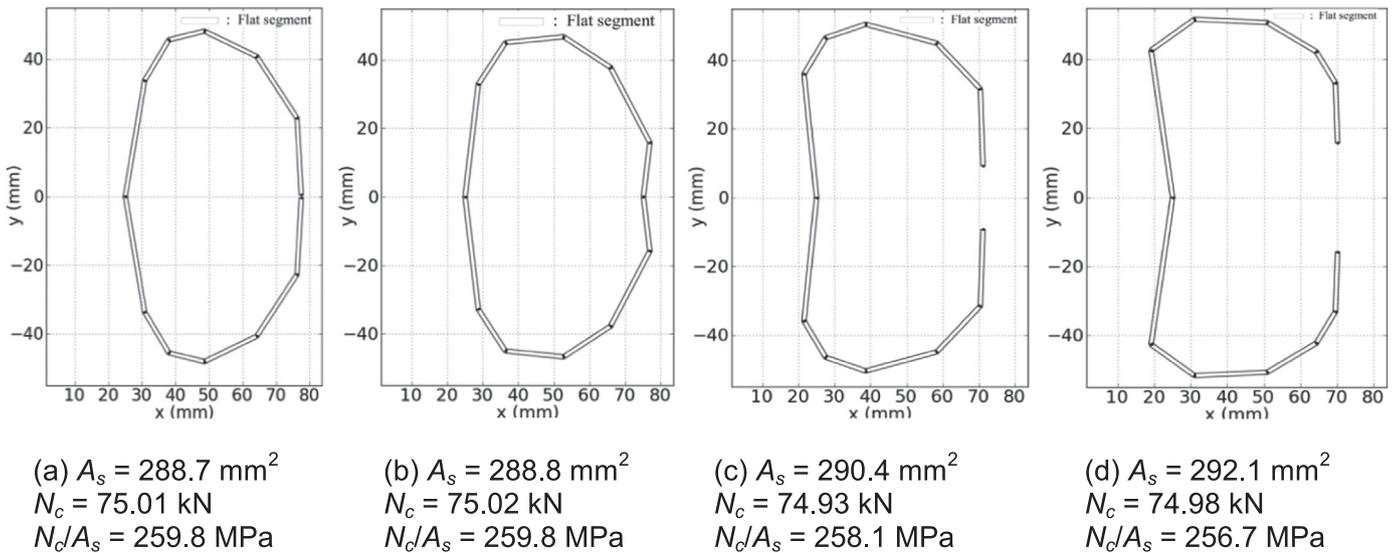


Fig. 23. Optimised cross-sections for the 1500 mm long columns and  $N_{max}=6$ , (a, b) fittest cross-sections and (c, d) least fit cross-sections.

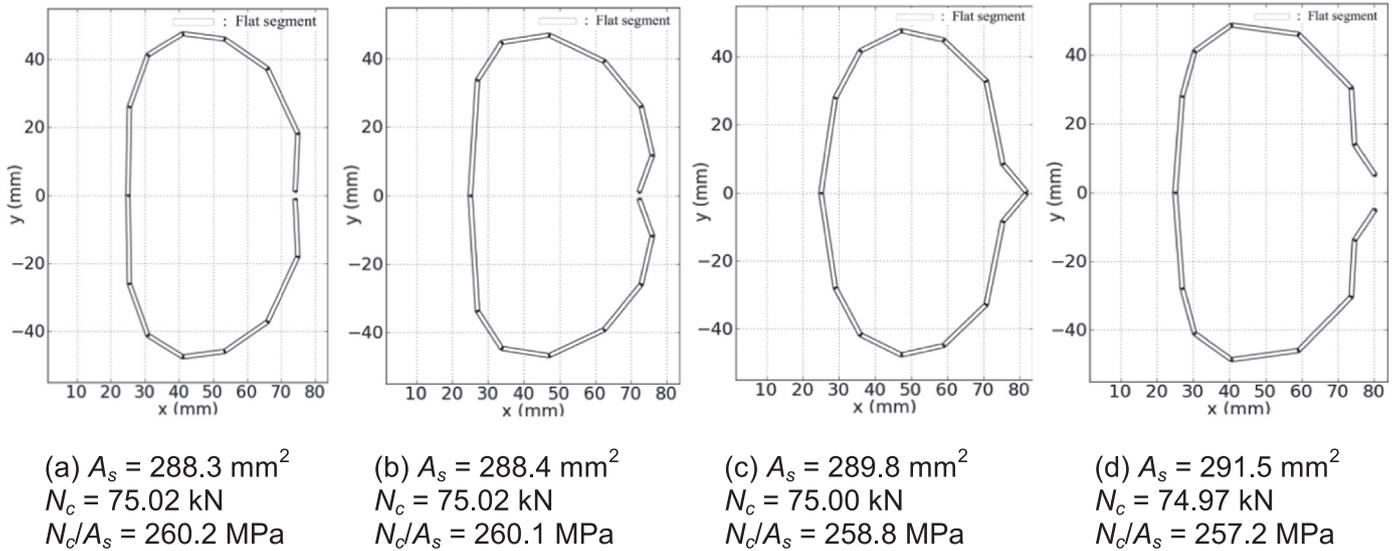


Fig. 24. Optimised cross-sections for the 1500 mm long columns and  $N_{max}=7$ , (a, b) fittest cross-sections and (c, d) least fit cross-sections.

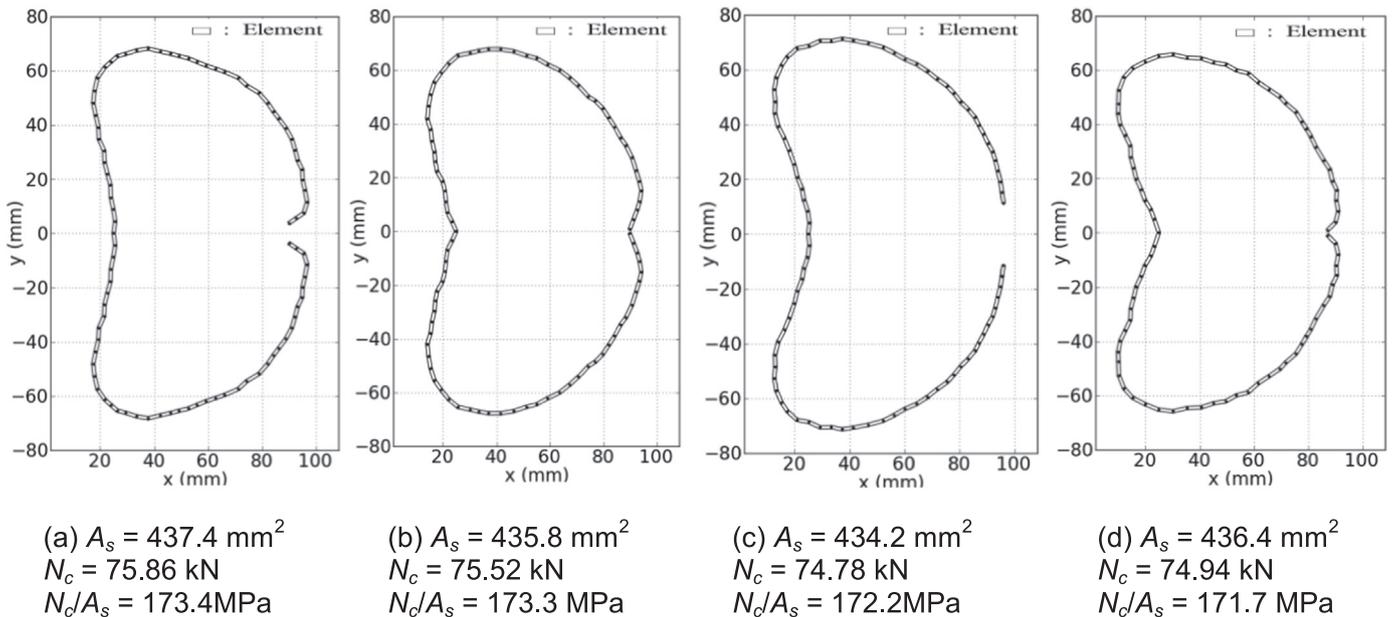


Fig. 25. Optimised cross-sections for the 3000 mm long columns and the non-manufacturable case, (a, b) fittest cross-sections and (c, d) least fit cross-sections.

large number of elements for the non-manufacturable cross-sections that limits the ability of the algorithm in forming perfectly curved cross-sections. The lowest average ultimate compressive stress of the manufacturable cases, found for  $N_{max}=3$ , is 1.1%, 2.5% and 1.9% lower than the same of the non-manufacturable case for the 500 mm (Table 6), 1500 mm (Table 7) and 3000 mm (Table 8) long columns, respectively. This validates the finding in [8] that introducing manufacturing constraints into shape optimisation algorithms marginally reduces the performance of the sections.

#### 4.4.3. Cross-sectional shapes

Figs. 14–18 illustrate the optimised non-manufacturable (Fig. 14) and manufacturable (Figs. 15–18) cross-sections for the 500 mm long columns. Based on the value of the ultimate compressive stress, the two fittest and two least fit cross-sections are shown in each figure. The two fittest cross-sections for all cases typically converge to “bean” cross-sectional shapes (subscripts (a–b) in Fig. 14 to Fig. 18), while the least fit ones mostly converge to open “Cee” sections (Fig. 13(c), Fig. 16(d), Fig. 17(d) and Fig. 18(c,

d)). Fig. 17(c) shows a closed “Cee” section, Fig. 14(d) and Fig. 13 (d) a “Sigma” cross-sectional shape, and Fig. 14(c) and Fig. 16(c) a “bean” cross-sectional shape. The fittest non-manufacturable cross-section in Fig. 14(a) has the largest ultimate compressive stress, 387.8 MPa, of all cross-sections. Its cross-sectional depth and width are 64.9 mm and 36.5 mm, respectively, i.e. a depth-to-width ratio of 1.78. The case  $N_{max}=3$  (Fig. 13(a)) has the lowest ultimate compressive stress of all fittest manufacturable cross-sections, which is only 0.9% lower than that of the fittest non-manufacturable solution (Fig. 14(a)). Its cross-sectional depth and width are 5.2% and 1.4%, respectively, greater than the fittest non-manufacturable solution.

Similar to Figs. 14–18, Figs. 19–24 show the fittest and least fit optimised cross-sections for the 1500 mm long columns. The fittest cross-sections converge to “bean” (Fig. 19(a, b), Fig. 20(a, b), Fig. 22(b), Fig. 23(b) and Fig. 24(a, b)) and closed “Cee” (Fig. 21(a, b), Fig. 22(a) and Fig. 23(a)) cross-sectional shapes, while the least fit ones converge to “Sigma” cross-sectional shapes (subscripts (c, d) in Figs. 19 and 21 to Fig. 23) and nearly closed “Cee” (Fig. 20(c, d)).

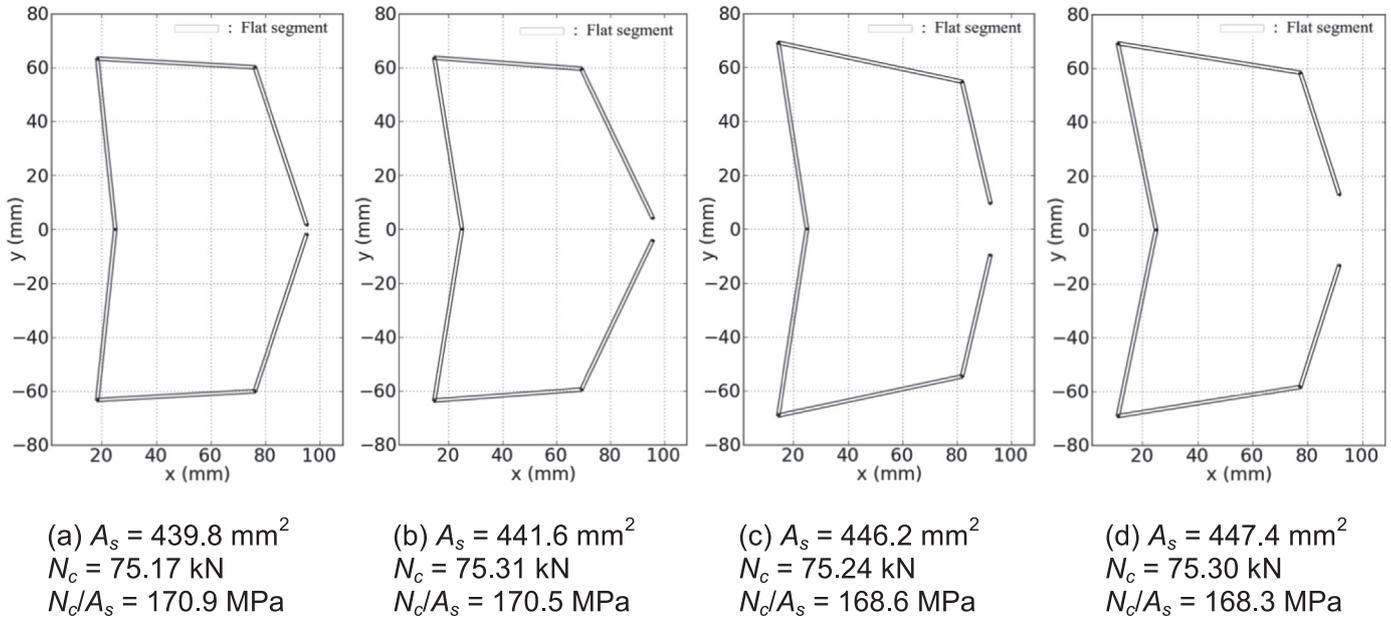


Fig. 26. Optimised cross-sections for the 3000 mm long columns and  $N_{max}=3$ , (a, b) fittest cross-sections and (c, d) least fit cross-sections.

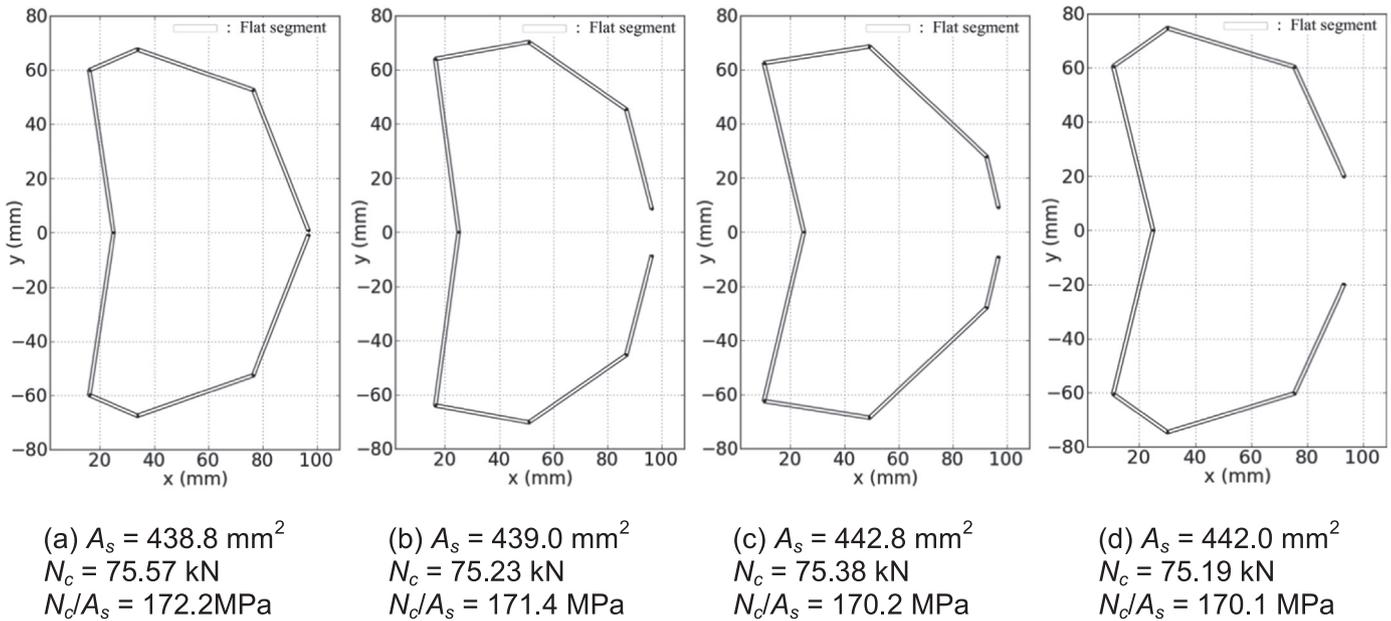


Fig. 27. Optimised cross-sections for the 3000 mm long columns and  $N_{max}=4$ , (a, b) fittest cross-sections and (c, d) least fit cross-sections.

For  $N_{max}=7$ , Fig. 24(c, d) show closed (or nearly closed) “Cee” sections with lip stiffeners.

Similar conclusions to the ones drawn for the 500 mm long columns apply: (i) the fittest non-manufacturable cross-section (Fig. 19(a)) has the largest ultimate compressive stress (261.0 MPa) of all studied cases, and (ii) the fittest manufacturable cross-section with the lowest ultimate compressive stress ( $N_{max}=3$  in Fig. 20(a)) performs similarly to the fittest non-manufacturable solution, with only 2.5% difference in ultimate compressive stresses between the two solutions. The cross-sectional depth and width of the former manufacturable cross-section are 7.5% and 2.8%, respectively, lower than that of the fittest non-manufacturable solution.

Figs. 25–31 show the two fittest and two least fit optimised non-manufacturable and manufacturable cross-sections for the 3000 mm long columns. All cross-sections converge to “bean” cross-sectional shapes, with the least fit cross-sections usually

having a more open cross-sectional shape than the fittest ones. The large number of elements forming the non-manufacturable cross-sections in Fig. 25 results in not perfectly curved shapes, and the largest ultimate compressive stress (173.6 MPa) is found for the manufacturable case  $N_{max}=6$  in Fig. 29(a). However, this value is only 0.11% greater than the ultimate compressive stress of the fittest non-manufacturable case in Fig. 25(a). The cross-sectional depth and width of the fittest cross-section are 134.1 mm and 82.6 mm, respectively, which represents a cross-section that is 1.8% less deep and 4.2% wider than the fittest non-manufacturable solution. The ultimate compressive stresses of the fittest cross-sections for all studied cases are close to each other and no more than 1.8% apart.

#### 4.4.4. Improvement in capacity

The nominal member capacities in compression of the optimised manufacturable solutions in Table 6 through Table 8 are

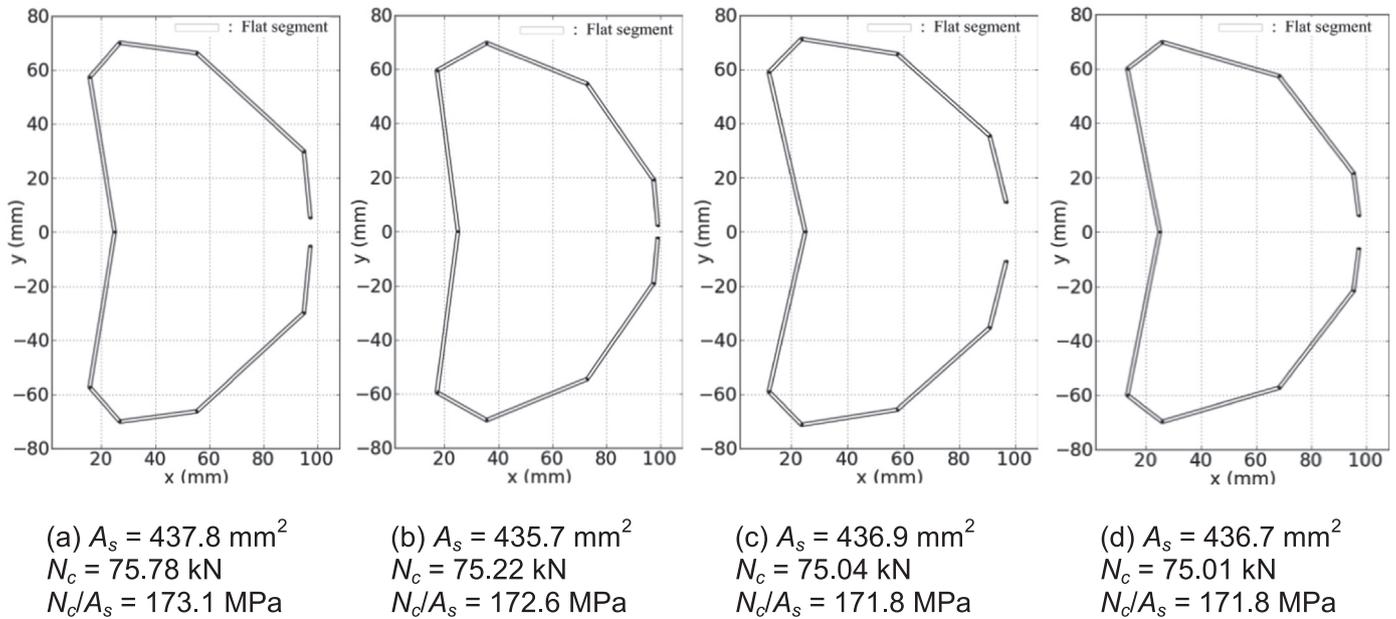


Fig. 28. Optimised cross-sections for the 3000 mm long columns and  $N_{max}=5$ , (a, b) fittest cross-sections and (c, d) least fit cross-sections.

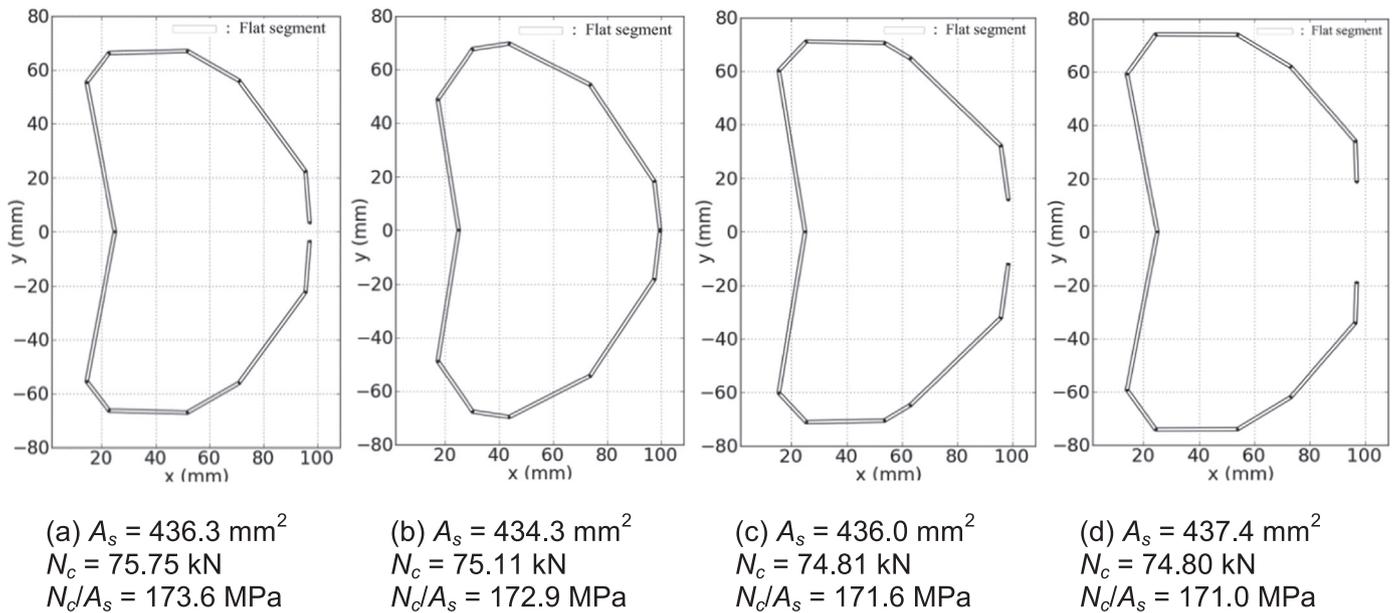


Fig. 29. Optimised cross-sections for the 3000 mm long columns and  $N_{max}=6$ , (a, b) fittest cross-sections and (c, d) least fit cross-sections.

compared to conventional lipped channel cross-sections that have similar aspect ratios to the sections manufactured in Australia by Bluescope Lysaght [24] and the same cross-sectional area of the optimised sections. Table 9 summarises the dimensions and capacity of the conventional lipped channels used for each investigated column length. The conventional sections in Table 9 satisfy the geometric limitations for pre-qualified DSM compression members given in Table 7.1.1 of Australian standard AS4600 [19]. As noted in Table 9, the nominal member capacities of the optimised solutions are significantly larger than the corresponding conventional ones. The improvement ranges from 30% for the 500 mm long columns to 151% for the 3000 mm long columns.

## 5. Conclusions

This paper has defined a set of simple manufacturing rules and incorporated them into the previously developed "self-shape

optimisation" algorithm for CFS profiles using the Hough transform. The objective of the fitness function is to minimise cross-sectional area subjected to a targeted axial member compressive capacity and manufacturing constraints. The ability and accuracy of the algorithm in optimising manufacturable thin-walled cross-sections have been verified against a known optimisation problem. The algorithm was used to shape-optimize manufacturable simply-supported, singly-symmetric and free to warp CFS open-section columns for all buckling modes. Short, intermediate and long columns, with various numbers of manufacturable flat segments, were investigated. The main conclusions are summarised below:

- The robustness of the algorithm is demonstrated by the consistency of the optimised solutions over 10 runs.
- The Hough transform accurately allows the formation of manufacturable CFS cross-sections, and the algorithm always

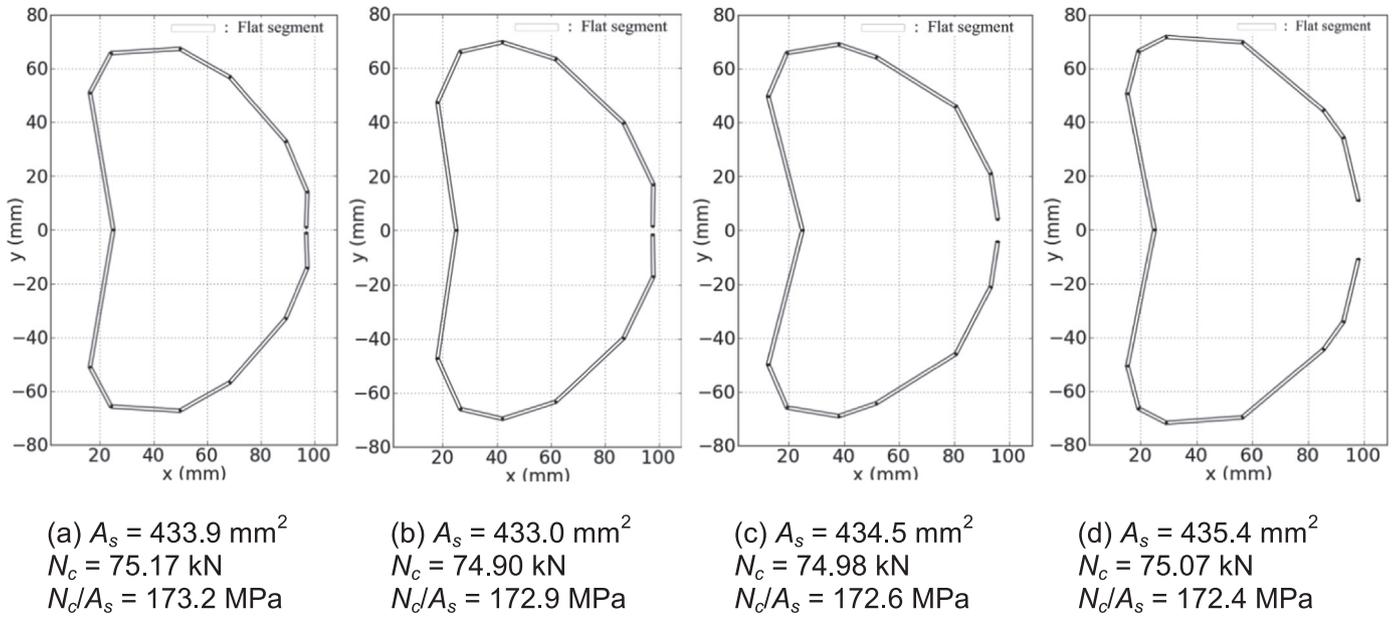


Fig. 30. Optimised cross-sections for the 3000 mm long columns and  $N_{max}=7$ , (a, b) fittest cross-sections and (c, d) least fit cross-sections.

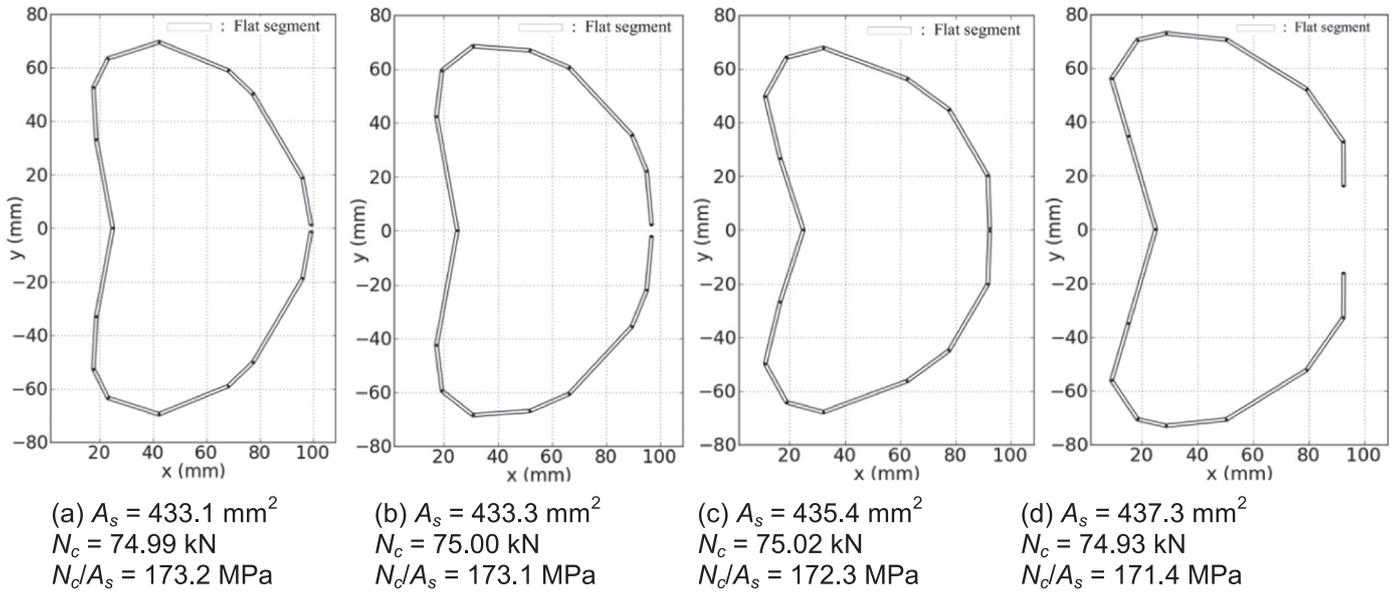


Fig. 31. Optimised cross-sections for the 3000 mm long columns and  $N_{max}=8$ , (a, b) fittest cross-sections and (c, d) least fit cross-sections.

Table 9  
 Results of conventional lipped channel cross-sections.

Column length (mm)	Conventional lipped channel section							Improvement in capacity <sup>b</sup> (%)
	Depth (mm)	Width (mm)	Lip (mm)	Thick-ness (mm)	Cross-section area <sup>a</sup> (mm <sup>2</sup> )	Nominal member capacity (kN)	Ultimate compressive stress (MPa)	
500	68.6	42.3	10.8	1.2	196.0	57.7	294.4	30
1500	118.6	55.3	13.1	1.2	290.3	39.3	135.4	91
3000	195.2	75.2	16.5	1.2	435.3	29.9	68.7	151

<sup>a</sup> Equal to the average optimised cross-sectional area of manufacturable cases.  
<sup>b</sup> Compared to the average optimised capacity of manufacturable cases.

converges to optimised solutions. The convergence rate for manufacturable cross-sections is slower than for non-manufacturable ones.

- However, the computation time per generation to shape-optimize short to long manufacturable columns is approximately

26–66% faster than for non-manufacturable ones, because in CUFSM it takes more time to analyse non-manufacturable cross-sections composed of a large number of short elements than to analyse manufacturable ones with a small number of flat segments.

- Introducing manufacturing constraints into shape optimisation algorithms was found to marginally affect the performance of the resulting sections, with the average ultimate compressive stress of the manufacturable columns being within 1.1% of that of the non-manufacturable ones.
- The manufacturable cross-sectional shapes were usually found to be similar. “Bean” and closed “Cee” cross-sectional shapes without local stiffeners were mainly found to be the fittest, and likely represent optimum manufacturable or non-manufacturable cross-sectional shapes. Typically, local buckling is prevented by shaping these rounded optimum cross-sections rather than forming local stiffeners in the algorithm.
- The optimised singly-symmetric manufacturable cross-sections have a capacity 30–151% higher than that of the conventional lipped channel sections.

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