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	٠,	199	For EAA Use

U S Department of Transportation Federal Aviation Administration

AF MAJOR REPAIR AND ALTERATION 15 199 (Airframe, Powerplant, Propeller, or Appliant DO NASH

For FAA Use Only Office Identification

INSTRUCTIONS; Print or type all entries. See FAR 43.9, FAR 43 Appendix B. and AC 43.9-1 (or subsequent revision thereof) for instructions

	osition of this form Th	nis report is required on 901 Federal Aviat	bylaw ((49 U.S.C. 1421). Fa	ilureto rep	ort can result in a c	ivil penalt	y not to exce	ed \$1,000
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Post Office Box 60627 Cerbficated Repair Station Manufacturer									
		or alteration made to	the un	it(s) identified in iter	n 4 above	and described on th	e reverse	or attachme	nts hereto
have l	been made in accord	lance with the require od correct to the best	ements	of Part 43 of the U					
Date				Signature of Auth	orized Indi	ividual	7	~ <i>1</i>	
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			7. Ap	proval for Relurn To	Service				
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BY	AA Designee	Repair Station		Person Approved b	y Transportess Group	t		-	
Date of App	proval or Rejection	Certificate or Designation No.		Signature of Author	orized Indi	vidual a J	Ho.	2	
4-11-97		529210 1.	A	5700		Thurston A. Tho	_		

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

	- -	•	
			-
I. Aircraft mod	ified to Colemill Enterpr	ises, Inc. "FOXSTA	R" per STC SA1762S(
dated 03/18/85 ar	d drawing #76000 revise	ed 03/18/85 with the	installation of two
Teledyne Contine	ntal I0550C Engines and	l two Hartzell HC-C4	YF-2E Propellers, and
Hartzell D4511P	Spinners, Nose Cowling	per Colemill P/N 74	000-1, and Winglets p
	00-50 and 66000-51 app	roved Flight Manual	Supplement dated
02/21/95 Installed	in P.O.H.		
II. Fuel flow ga	auge removed and Shadir	n fixel flow totalizer I	P/N 910522 installed in
	STC SA486GL and Repo		,
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	WEIGHT A	AND BALANCE	
Daving out list or	d weight and balance up	dated and installed in	iPOH
Equipment list at	a weight and darance ub	dated and instance in	
It has been determ	nined by Flight Test, that	t Auto Pilot Model K	ing KFC 200 previous
It has been deten	atible with the above ST	C and will produce n	o adverse effect on
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United States of America

Department of Transportation—federal Aviation Administration

pplemental Type Certificate

This cortificate, issued to

Colemill Enterprises

P.O. Box 60627

Cornelia Fort Airpark Nashville, TN 37206

cortifies that the change in the type design for the following product with the limitations and conditions

therefor as specified hereon meets the airworthiness requirements of Part 3 of the Civil

Regulations.

Original Product - Type Cortificate Number: 3A16

Make: Beech Aircraft Corporation

Model: 95-C55, D55, E55 & 58

Description of Trype Design Change:

Install two Teledyne Continental Model IO-550-C engines, Hartzell or McCauley propellers, nose cowling per Colemill P/N 74000-1, and winglets per Colemill P/N 66000-50 and -51, all per Colemill Enterprises' "Installation Instructions For Modification of Beachcraft Baron Model: C-55, D-55, E-55 & 58", dated September 14, 1984, or later FAA Approved Revisions.

Limitations and Conditions:

This approval should not be extended to other aircraft of this model on which other previously approved modifications are incorporated unless it is determined by the installer that the interrelationship between this change and any other previously

(See Page 2)

This cortificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, revoked, or a termination date is otherwise established by the Administratur of the Federal Aviation Administration.

Date of application: July 26, 1984

Sale reissued: March 22, 1985

Sale of issuance: March 18, 1985

Sale umended:

February 4, 1991 February 21, 1995

By direction of the Administrator

Roger D. Andersongnature)

Manager, Atlanta Aircraft

Certification Office

Any alteration of this certificate is punishable by a fine of not exceeding \$1,000, or imprisonment not exceeding 3 years, or both.

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United States of America

Department of Transportation—Federal Aviation Administration

Supplemental Type Certificate

(Continuation Sheet)

Number SA1762SO

Limitations and Conditions: (Continuation)

approved modifications will produce no adverse effect upon the airworthines of that airplane.

- 2. The following Colemill Enterprises' Airplane Flight Manual Supplements or later FAA approved revisions are required with this installation:
 - a. For Models 95-C55, D55, E55: "Pilot's Operating Handbook and Airplane Flight Manual Supplement for Beechcraft Baron", dated March 18, 1985.
 - b. For Models 58, & 58A: "Pilot's Operating Handbook and Airplane Flight Manual Supplement for Beechcraft Baron", dated March 18, 1985.

END

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United States of America

Bepartment of Transportation—Jederal Aviation Administration

Supplemental Type Certificate

Number SA486GL

This certificate, issued to

Shadin Company, Inc. 6950 Wayzata Boulevard Minneapolis, MN:55426

cortifies that the change in the type design for the following product with the limitations und conditions

therefor as specified hereon meets the aircrathiness requirements of Fart 3 of the Civil Air

Hegulations. See Type Certificate Data Sheet No. 3A16 for complete certification basis.

Original Product - Tripo Contificato Number

3A16

Beech -

Medd 95-C55, 95-C55A, D55, D55A, E55, E55A, 58, 58A, 56TC and A56TC

Inscription of Japo S. sign Change.

Installation of a Shadin Company Fuel Flow Indicating System in accordance with Shadin Company Report Number 4023, revised April 22, 1985, or other FAA approved revision.

Limitations and Genditions :

This approval should not be extended to other airplanes of this model that incorporate any other previously approved modification, unless it is determined that the interrelationship between this change and any other previously approved modification will introduce no adverse effect on the airworthiness of these airplanes.

This certificate and the supporting data which is the basis for approval shull remain in affect until surrendered, suspended, rowked, or a termination date is otherwise established by the Administrator of the

Tederal Swiation Administration

Interofapplication. August 29, 1980

Jule masuel

Inte of issuance

January 13, 1981

Sate assembled November 19, 1981, May 13, 1982 December 16, 1982, July 18, 1985

By derete a of the delimenistrate .

W. F. Horn (Signature) Manager, Chicago Aircraft Certification Office Central Region, ACE-115C

Am alteration of this certificate is punishable by a fine of not exceeding \$1,00%, or impresonment not exceeding 3 years, or both.

. This certificate may be transferred in accordance with FAR 21.47.

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NOTICE

Weight and balance of operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets, Identify with aircraft nationality and registration mark and date work completed,

Removed the following Collins Equipment, AMR-350 audio amp., GLS-350 Glideslope Rec., ANS-351 RNAV system, both VHF-251 VHF Comm systems, both VIR-351 NAV Recievers, a IND-350A NAV Indicator, both PWR-150 power converters, and both VHF Comm Antenna's. Installed the following equipment, a Bendix/King KMA-24 audio amp., a Bendix/King KNS -80 RNAV system, a Bendix/King KY-196A VHF Comm system, a Bendix/King KX-155 NAV/COMM system with Glideslope reciever, a Bendix/King KI-204 NAV Indicator, a Bendix/King KA-120 tuning adapter, replaced both VHF Comm antenna's and installed two AFT intercomm positions for the existing nat AA80 intercomm system. The afore mentioned equipment were installed in accordance with the following installation manuals.

Bendix/King KMA-24 P/N 006-0180-00 JAN. 19979 REV. 0 Installation manual Bendix/King KNS-80 P/N 006-0154-02 JULY 1979 REV. 2 Installation manual Bendix/King KY-196A P/N 006-00695-0002 NOV. 1991 REV. 2 Installation manual Bendix/King KX-155 P/N 006-0179-05 FEB. 1988 REV. 5 Installation manual Bendix/King KI-204 P/N 006-0137-04 March 1988 REV.4 Installation manual Bendix/King KA-120 P/N 006-0549-00 June 1982 REV. 0 Installation manual

All System's have been functionally checked in accordance with manufactures instructions and meets all specifications. The KNS-80 RNAV system is placarded "AREA NAVIGATION USE LIMITED TO VFR ONLY" PENDING IFR Approval. Electrical load checked and meets Beechcraft specifications. Compass accuracy checked good. See revised aircraft equipment list dated 3/16/94 for exact equipment locations and revised weight and balance. Log book entries completed this date. All details on work performed are driftle under work order 11-05-14242 at this repair station.

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WU.S. GOVERNMENT PRINTING OFFICE. 1992-769 ...:2/60157



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MAJOR REPAIR AND ALTERATION

Form Approved OMB No. 2120-0020 For FAA Use Only

Federal Aviation Administration

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. The appendix B and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. The appendix B and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. The appendix B and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. The appendix B and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. The appendix B and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. The appendix B and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. The appendix B are appendix B and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. The appendix B are appendix B and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. The appendix B are appendix B are appendix B and AC 43.9-1 (or subsequent revision thereof) for instructions are appendix B are appen

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NOTICE

Weight and balance of operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

Description of Work Accomplished
(If more space is required, attach additional sheets, Identify with aircraft nationality and registration mark and date work completed.)

Installed BFGoodrich 1000+/stormscope system consisting of # 78-8060-5900-8 display # 78-8051-9160-4 processor and # 78-8051-9200-8 antenna. System installed in accordance with 3M Stormscope installation manual #78-0299-7885-7 and AC 43.13.2A. Installed Bendix / King KT-76a Transponder with Terra AT-3000 blind encoder. Transponder and encoder were installed in accordance with Bendix / King manual # 006-0143-04 revision 4 dated March 1992. Terra encoder installed in accordance with manual # 1900-4099-00 revision B, ECO #804. g. Followed approiate guidelines of AC 43.13.2A and checked systems in accordance with FAR 91.411 and 91.413. Removed Sperry P-200 radar system. Installed Bendix / King RDS-82VP Radar System consisting of the following: # 066-3084-32 Antenna / 12" array and # 066-3084-323 Indicator. System installed in accordance with Bendix / King installation manual # 006-00955-0006 revision 6 dated March 1992. Removed existing 3" turn and bank indicator and installed 2' T&B. Relocated suction gauge to copilots . instrument panel. Installed King service bulletin KI-525A-1 adding heading synchro to HSI indicator. All work performed in accordance with AC 43.13.2A and all equipment manufactures installation instructions. All electrical loads checked on busses new wiring wired to in accordance with manufactures specifications. All wiring and electrical parts used meet approiate MIL Specifications. See revised aircraft equipment list dated 8/5/93 for current weight and balance and exact equipment locations. All work on file this repair station under work order 1105013360.

ระบบจากับ ของการเมาเมาเลง เราะบับการเรื The second of maken to terrible the second to the second t คอสสมใช้เกียกใช้สา**จะ**ทำอาโละสาร Steel beingelich periteil in ☐ Additional Sheets Are Attached

QU.S. GOVERNMENT PRINTING OFFICE: 1992-769-012/60157

FAA AIRCRAFT REGISTRY DATE: 1/-27-90 CAMERA NO. /

UNITED STATES OF AMERICA
DEPARTMENT OF TRANSPORTATION—FEDERAL AVIATION ADMINISTRATION

STANDARD AIRWORTHINESS CERTIFICATE

NATIONALITY AND REGISTRATION MARKS

MANUFACTURER AND MODEL

AIRCRAFT SERIAL NUMBER

4' · CATEGORY

N257MD

BEECH 58

TH-1286

NORMAL

5. AUTHORITY AND BASIS FOR ISSUANCE
This airworthiness certificate is issued pursuant to the Federal Aviation Act of 1958 and certifies that, as of the date of issuance, the aircraft to which issued has been inspected and found to conform to the type certificate therefor, to be in condition for safe operation, and has been shown to meet the requirements of the applicable comprehensive and detailed airworthiness code as provided by Annex 8 to the Convention on International Civil Aviation, except as noted herein.

NONE

TERMS AND CONDITIONS
 Unless sooner surrendered, suspended, airworthiness cht-ticate is effective as lor accordance with Parts 21, 43, and 91 of the Factor Conditions.

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Any atteration, reproduction, or misuse of this certificate may be punishable by a fine not exceeding \$1,000, or imprisonment not exceeding \$4 years, or both, THIS CERTIFICATE MUST BE DISPLAYED IN THE AIRCRAFT, IN ACCORDANCE WITH APPLICABLE FEDERAL AVIATION REGULATIONS.

FAA Form 8100-2 (8-82)

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1 Northstar installation manual PN GM295 REV:F 2 AC43.13elA Chapter 11 3 AC43.13e2A Chapter 1 4 AC43.13e2A Chapter 2 paragraphs 21, 22, 23, 24 and	27
5 AC43.13-2A Chapter 2 6 AC20-121 paragraph 6.2 appendix 1 paragraph 3.1, 2,	
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Form Approved Budget Bureau No. 04-R060.1 DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION FOR FAA USE ONLY MAJOR REPAIR AND ALTERATION OFFICE IDENTIFICATION 7.0.5 (Airframe, Powerplant, Propeller, or Appliance) INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. MODEL 58 NATIONALITY AND REGISTRATION MARK Beechcraft 1. AIRCRAFT SERIAL NO. TH-1286 N1832K ADDRESS (As shown on registration certificate) 1951 SW Winners Drive NAME (As shown on registration certificate) 2. OWNER Stuart, Florida Tomac, Inc. 3. FOR FAA USE ONLY 4, UNIT IDENTIFICATION 5. TYPE ALTER SERIAL NO. : REPAIR UNIT ATION X AIRFRAME Ne (As described in item 1 above) eccessossessessessesses RECEIVE POWERPLANT SO-GADO-8 APR 4 1983 PROPELLER APPLIANCE MANUFACTURER 6. CONFORMITY STATEMENT C. CERTIFICATE NO. B. KIND OF AGENCY A. AGENCY'S NAME AND ADDRESS U.S. CERTIFICATED MECHANIC Hangar One, Inc. P. O. Box 30100 Tampa, FL 33630 FOREIGN CERTIFICATED MECHANIC X- CERTIFICATED REPAIR STATION -~764-85 MANUFACTURER D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge. SIGNATURE OF AUTHORIZED INDIVIDUAL DATE March 24, 1983 Remeth C. Hostnick

7. APPROVAL FOR RETURN TO SERVICE

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is APPROVED REJECTED

CANADIAN DEPARTMENT

SIGNATURE OF AUTHORIZED INDIVIDUAL

FAA Form 337 (7-67)

DATE OF APPROVAL OR

BY

FAA FLT, STANDARDS INSPECTOR

FAA DESIGNEE

MANUFACTURER

REPAIR STATION

CERTIFICATE OR

DESIGNATION NO.

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Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

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FEDERAL AVIATION ADMINISTRATION

MAJOR REPAIR AND ALTERATION

Form Approved Budget Bureau No. 04-R060.1

FOR FAA USE ONLY

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NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

This is a supplement to the FAA Form 337 dated March 3, 1983. The purpose of this 337 is approval of the installed RNAV equipment for IFR and approach categories.

This aircraft was test flown on March 4, 1983 and the installed Collins ANS351 RNAV system was found to meet the requirements of AC 90.45A Appendix A Para 2 & 3.

Area navigation system is approved for IFR enroute, terminal and approach categories.

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DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION

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FAA Form 337 (7-67)

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NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record.

An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

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FAA Form 8100-2 (7-67) FORMERLY FAA FORM 1362