## **Aircraft Weight and Balance Revision**

Tail Number:	Date: 12/7/2022						
Prepared by: Marvin L. Campbell A&P 2808375				Work Order No: verbal			
				Type Certificat Data No:	e		
Aircraft Make: Piper	Model: PA28-181		Serial No: 28-7890418		Time: 4937.8		
Registered Owner: Agri-Business Aviation, Inc.			Address: 1800 N. Market St. Sparta, IL 62286				
Maximum Weight 25	50	CG F	Range FWD 82	2.0	AFT 93.0		
As Received; Date of Previous Weight and Balance: 11/14/2022			Jseful Load: 1001.15	<b>EW:</b> 1548.85	<b>EWCG:</b> 87.6724	<b>Moment:</b> 135791.5	
Notes: Installation of Ele	ectronics International CGF	R-30 Combo M	lonitoring System				
				Weight	Arm	Moment	
Removed VSI, All engine Instr., Fuel gauges, Amp Guage, Vac gauge, & assoc. w				-7.2	59.9	-431.28	
Installed CGR-30P, CGR-30C, EDC, Vac.Transducer, & G-force Transducer.			3.9	59.9	233.61		
				0.00	0.00	0.00	
			-	0.00	0.00	0.00	
				0.00	0.00	0.00	
				0.00	0.00	0.00	
				0.00	0.00	0.00	
				0.00	0.00	0.00	
			7	0.00	0.00	0.00	
				0.00	0.00	0.00	
X As Calculated	Moment	135593.83	New Empt	y Weight CG	New U	Iseful Load	
As Weighed	Weight	1545.55	87.73		10	1004.45	
			Signature				
			Repair Age	ency 2808375 A	\&P		

## 6.7 WEIGHT AND BALANCE DETERMINATION FOR FLIGHT

- (a) Add the weight of all items to be loaded to the basic empty weight.
- (b) Use the Loading Graph (Figure 6-13) to determine the moment of all items to be carried in the airplane.
- (c) Add the moment of all items to be loaded to the basic empty weight moment.
- (d) Divide the total moment by the total weight to determine the C.G. location.
- (e) By using the figures of item (a) and item (d) (above), locate a point on the C.G. range and weight graph (Figure 6-15). If the point falls within the C.G. envelope, the loading meets the weight and balance requirements.

	Weight (Lbs)	Arm Aft Datum (Inches)	Moment (In-Lbs)
Basic Empty Weight			
Pilot and Front Passenger	340.0	80.5	27370
Passengers (Rear Seats)*	340.0	118.1	40154
Fuel (48 Gallon Maximum)		95.0	
Baggage*	F 3	142.8	
Total Loaded Airplane			

The center of gravity (C.G.) of this sample loading problem is at inches aft of the datum line. Locate this point ( ) on the C.G. range and weight graph. Since this point falls within the weight - C.G. envelope, this loading meets the weight and balance requirements.

IT IS THE RESPONSIBILITY OF THE PILOT AND AIRCRAFT OWNER TO INSURE THAT THE AIRPLANE IS LOADED PROPERLY.

\*Utility Category Operation - No baggage or rear passengers allowed.

## SAMPLE LOADING PROBLEM (NORMAL CATEGORY)

Figure 6-9

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	Weight (Lbs)	Arm Aft Datum (Inches)	Moment (In-Lbs)
Basic Empty Weight			100
Pilot and Front Passenger		80.5	10000000
Passengers (Rear Seats)*		118.1	
Fuel (48 Gallon Maximum)		95.0	a in the same
Baggage*		142.8	200
Total Loaded Airplane			

Totals must be within approved weight and C.G. limits. It is the responsibility of the airplane owner and the pilot to insure that the airplane is loaded properly. The Basic Empty Weight C.G. is noted on the Weight and Balance Data Form (Figure 6-5). If the airplane has been altered, refer to the Weight and Balance Record for this information.

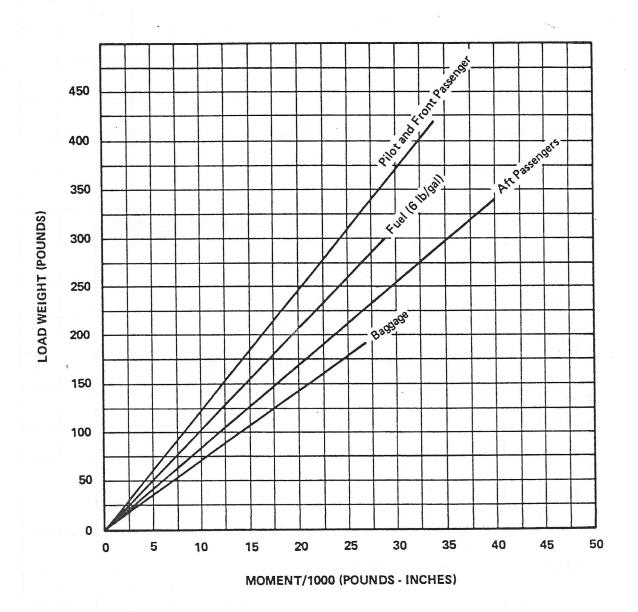
\*Utility Category Operation - No baggage or rear passengers allowed.

WEIGHT AND BALANCE LOADING FORM

Figure 6-11

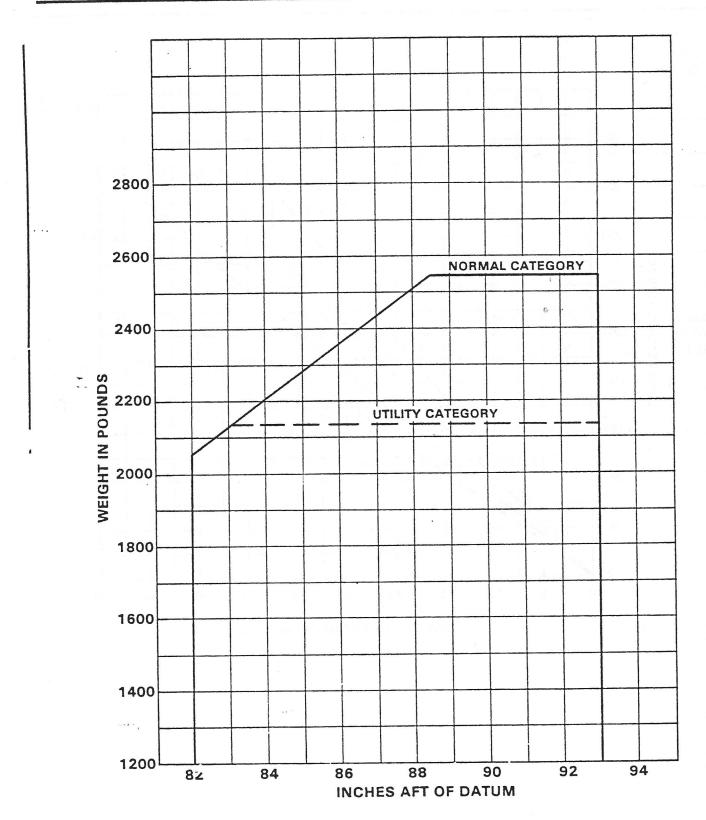
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LOADING GRAPH
Figure 6-13

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C. G. RANGE AND WEIGHT Figure 6-15

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**ISSUED: JUNE 18, 1976 REVISED: MAY 23, 1980**