



Space Is Hard. Artemis Is Harder.

What we don't yet know how to do

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LiftOff Summer Institute 2026

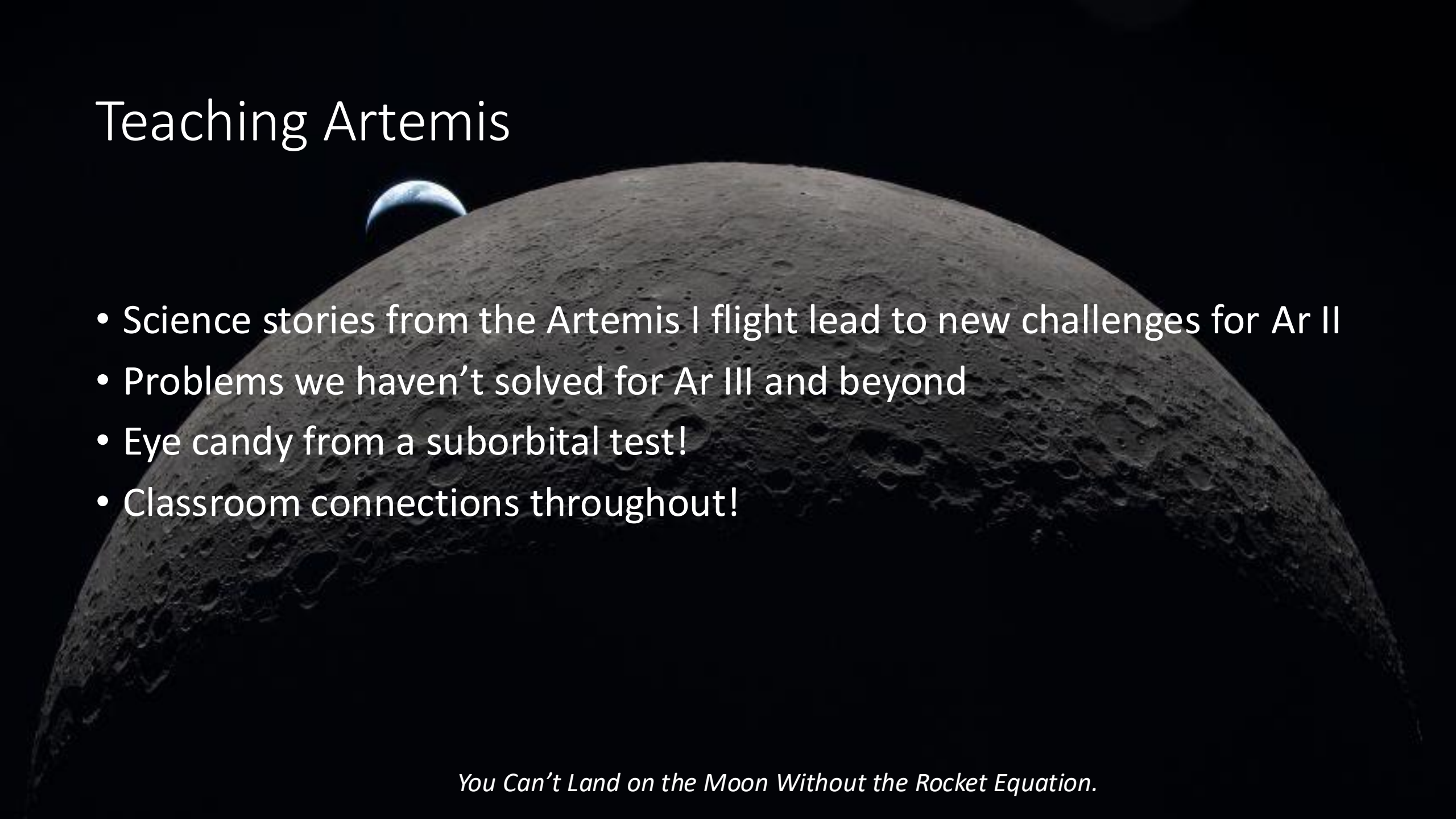
June 23, 2026

Teaching Artemis

Science Lessons from Launch to Splashdown

You Can't Land on the Moon Without $PV = nRT$.

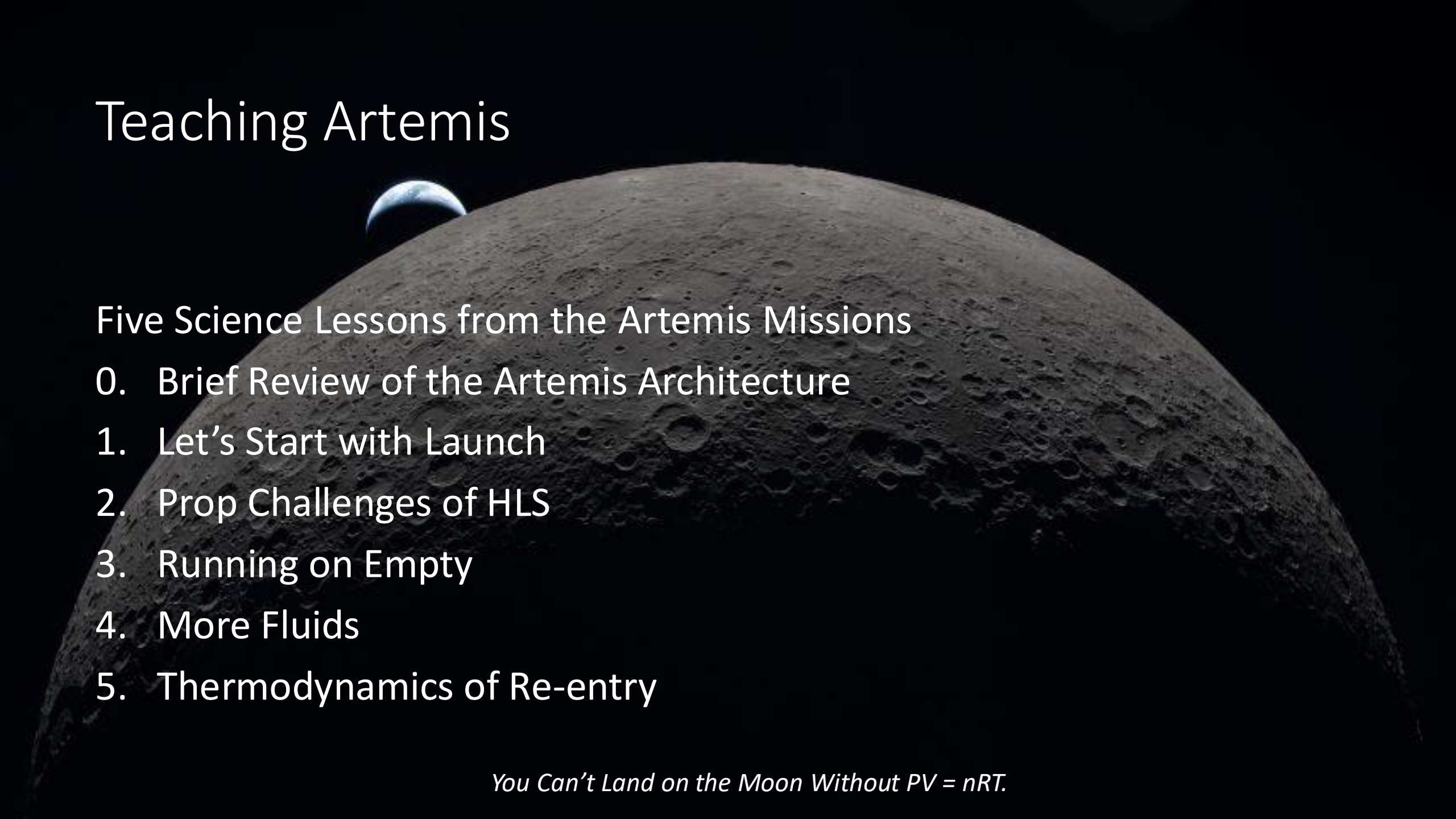
Teaching Artemis



- Science stories from the Artemis I flight lead to new challenges for Ar II
- Problems we haven't solved for Ar III and beyond
- Eye candy from a suborbital test!
- Classroom connections throughout!

You Can't Land on the Moon Without the Rocket Equation.

Teaching Artemis



Five Science Lessons from the Artemis Missions

0. Brief Review of the Artemis Architecture
1. Let's Start with Launch
2. Prop Challenges of HLS
3. Running on Empty
4. More Fluids
5. Thermodynamics of Re-entry

You Can't Land on the Moon Without $PV = nRT$.

0. A brief review of the Artemis Architecture

Oh no! NASA charts!



SPACE LAUNCH SYSTEM



NASA's Moon Rocket

ORION SPACECRAFT



Our Ride to the Moon

HUMAN LANDING SYSTEM

SPACE X STARSHIP



Artemis Astronauts' Lunar Landers

BLUE ORIGIN BLUE MOON



EXPLORATION GROUND SYSTEMS



Assembly, Launch, Recovery

EXPLORATION EVA SYSTEMS



Spacesuits for Moonwalks

HUMAN-CLASS DELIVERY LANDER

SPACE X STARSHIP



Lunar Freight Deliveries

BLUE ORIGIN MARK 2 CARGO



LUNAR TERRAIN VEHICLE



The Artemis Moon Buggy

PRESSURIZED ROVER



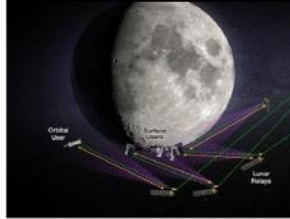
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The Moon's Mobile Network

NASA COMMUNICATIONS NETWORKS

NEAR EARTH NETWORK



Call Home from Space

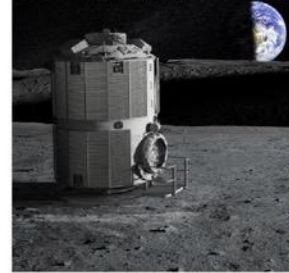
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INITIAL SURFACE HABITAT

Government Reference Concept



Home on the Lunar Surface

LUNAR SURFACE CARGO LANDER

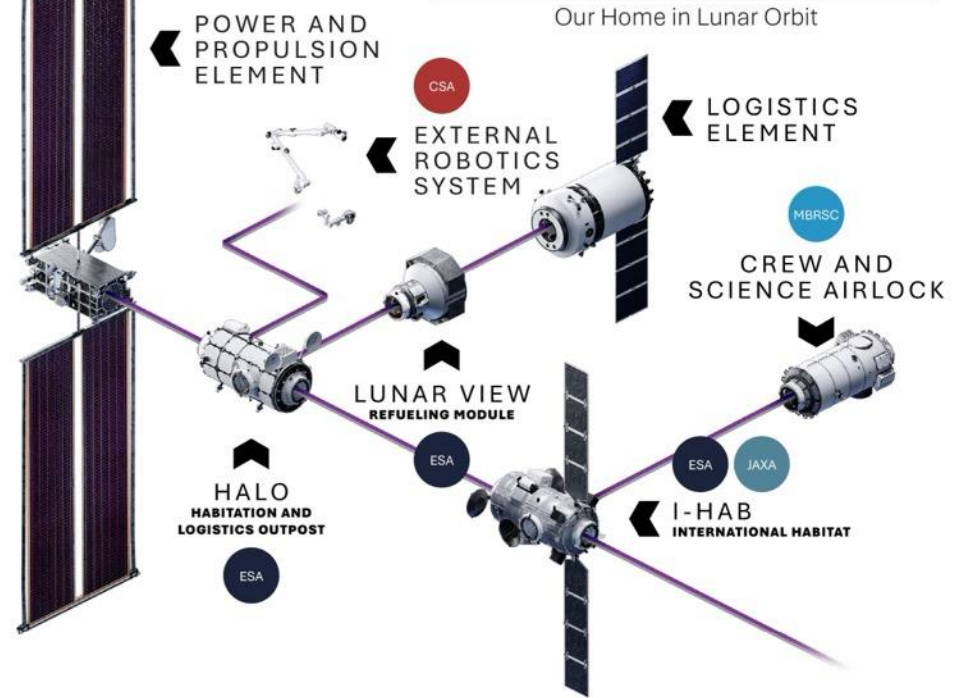
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Logistics, Delivered

GATEWAY

Our Home in Lunar Orbit



International Partners

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- ESA** European Space Agency
- JAXA** Japan Aerospace Exploration Agency
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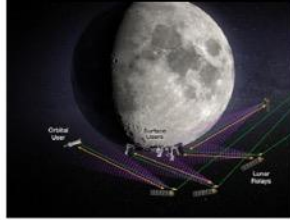
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BLUE ORIGIN MARK 2 CARGO



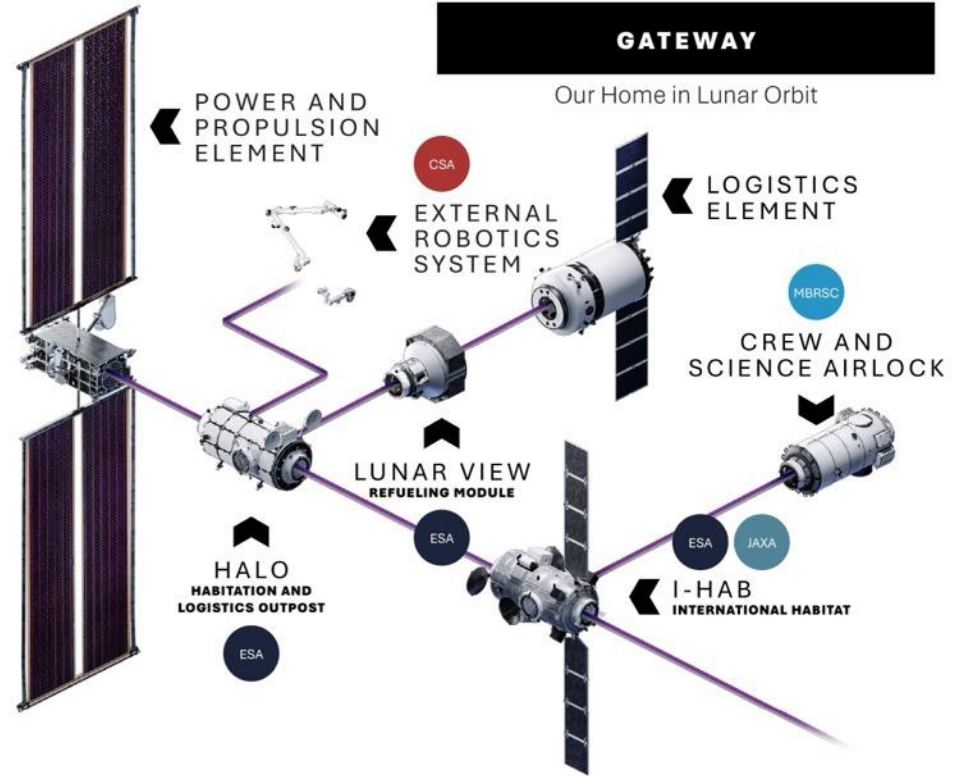
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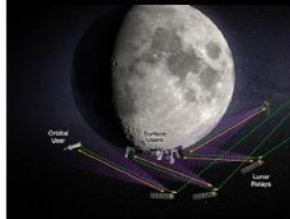
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Science and Tech Deliveries

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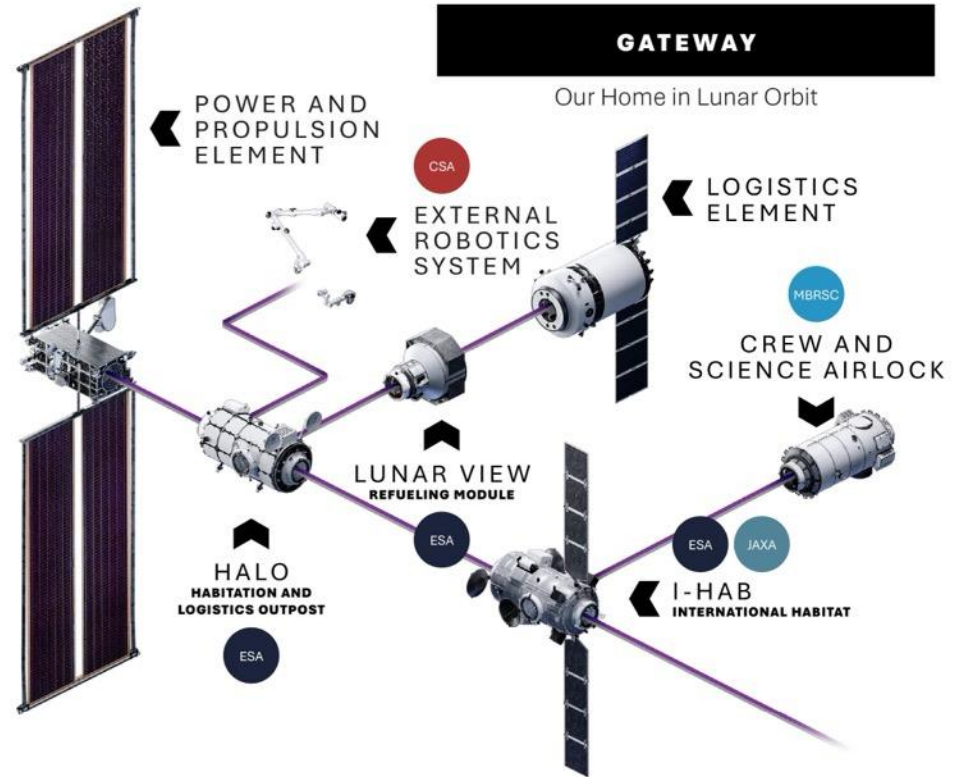
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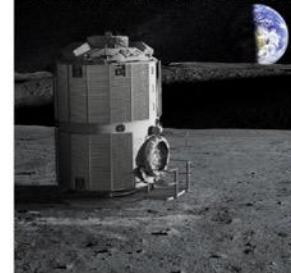
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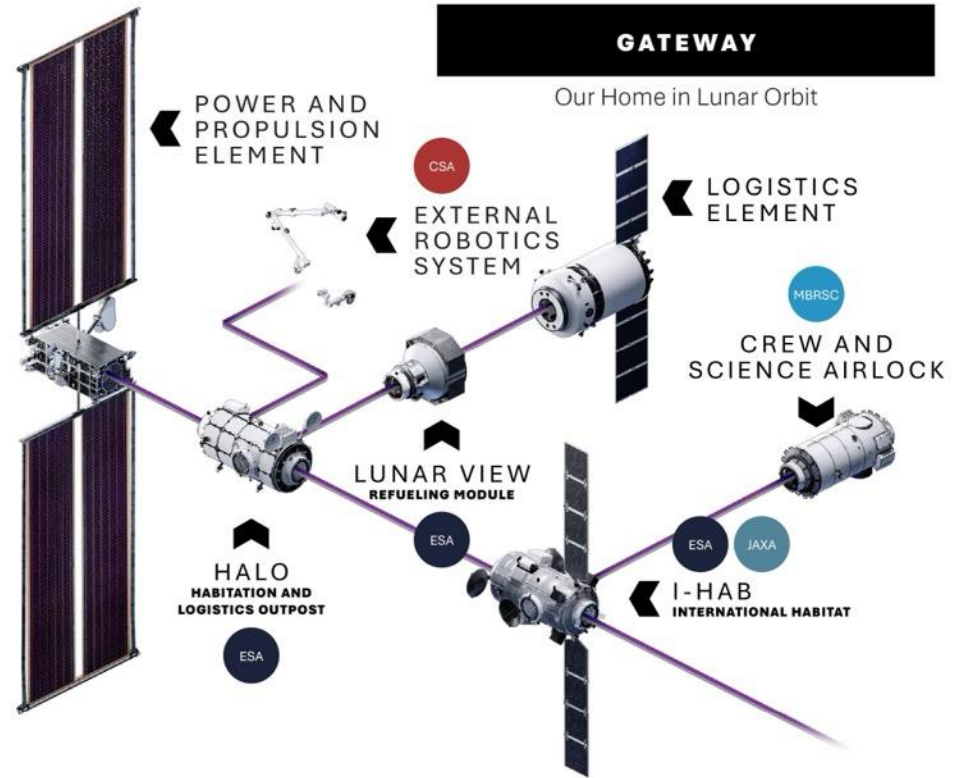
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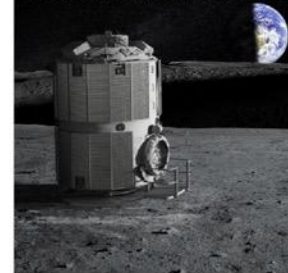
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Assembly, Launch, Recovery

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LUNAR TERRAIN VEHICLE



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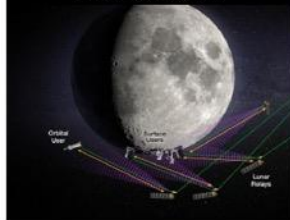
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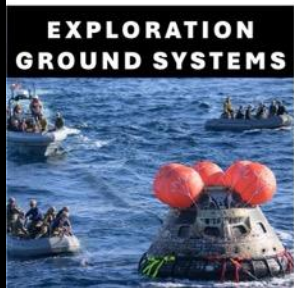


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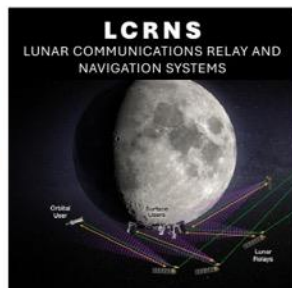
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Spacesuits for Moonwalks



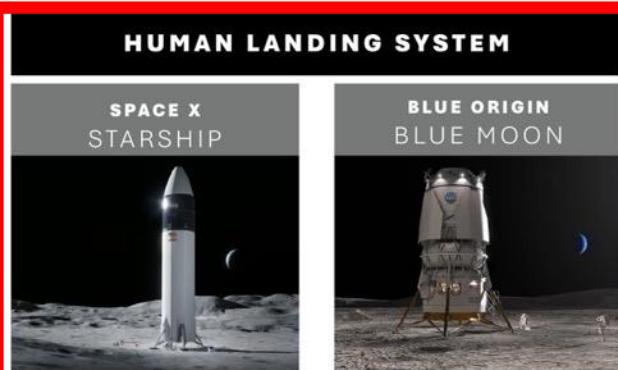
PRESSURIZED ROVER

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HUMAN LANDING SYSTEM

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BLUE ORIGIN BLUE MOON

Artemis Astronauts' Lunar Landers

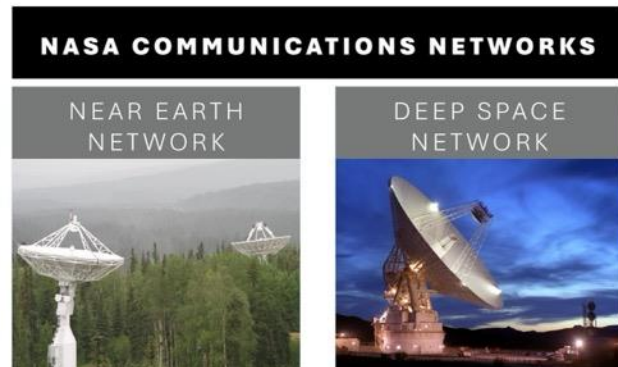


HUMAN-CLASS DELIVERY LANDER

SPACE X STARSHIP

BLUE ORIGIN MARK 2 CARGO

Lunar Freight Deliveries

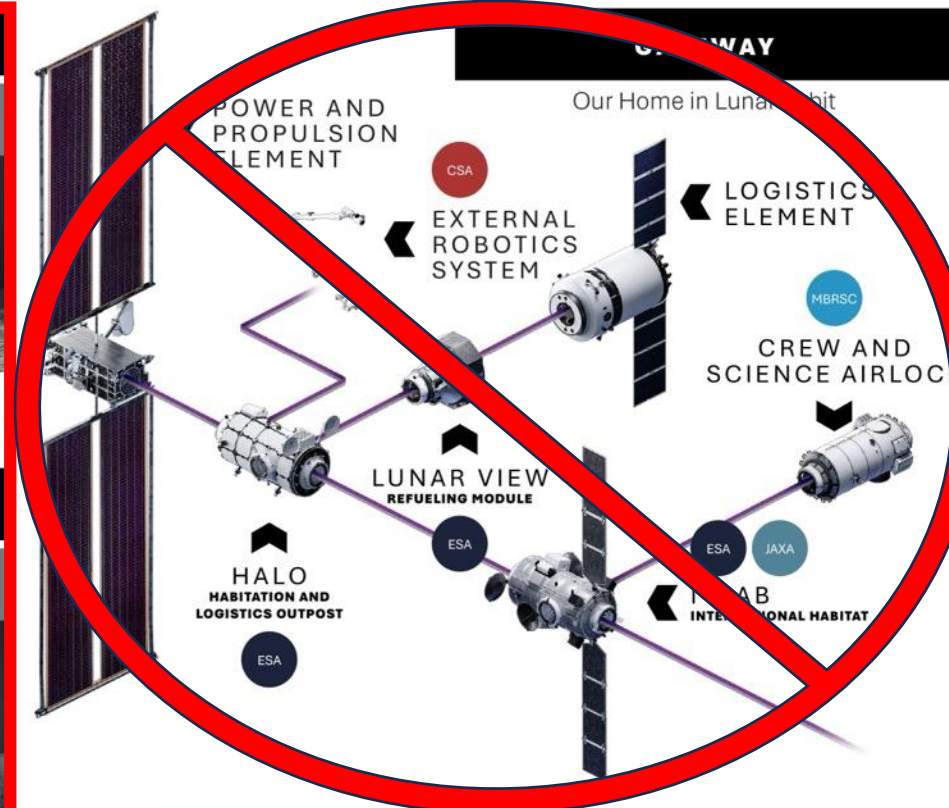


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Call Home from Space



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INITIAL SURFACE HABITAT

Government Reference Concept

Home on the Lunar Surface



LUNAR SURFACE CARGO LANDER

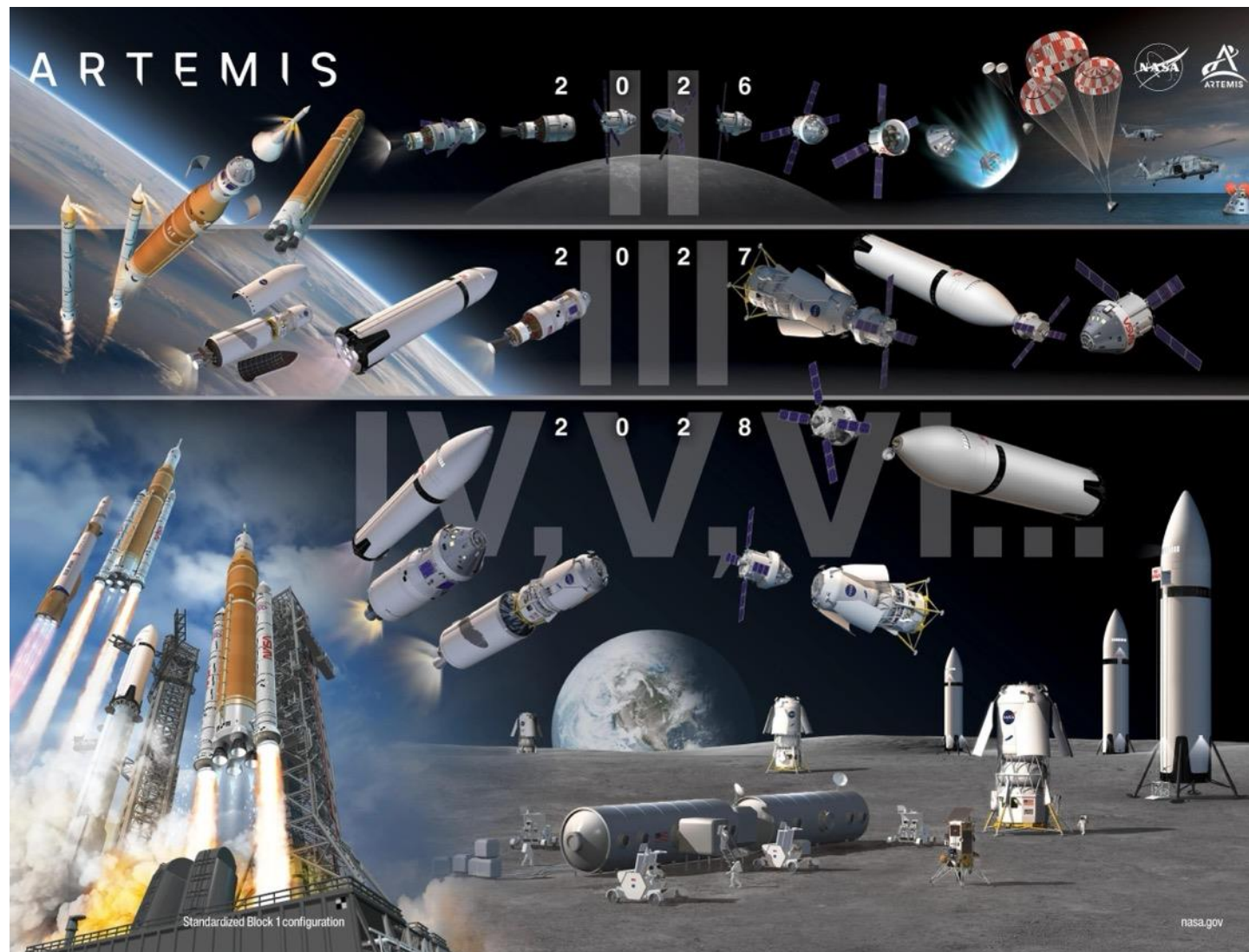
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Artemis Architecture Updated Yet Again

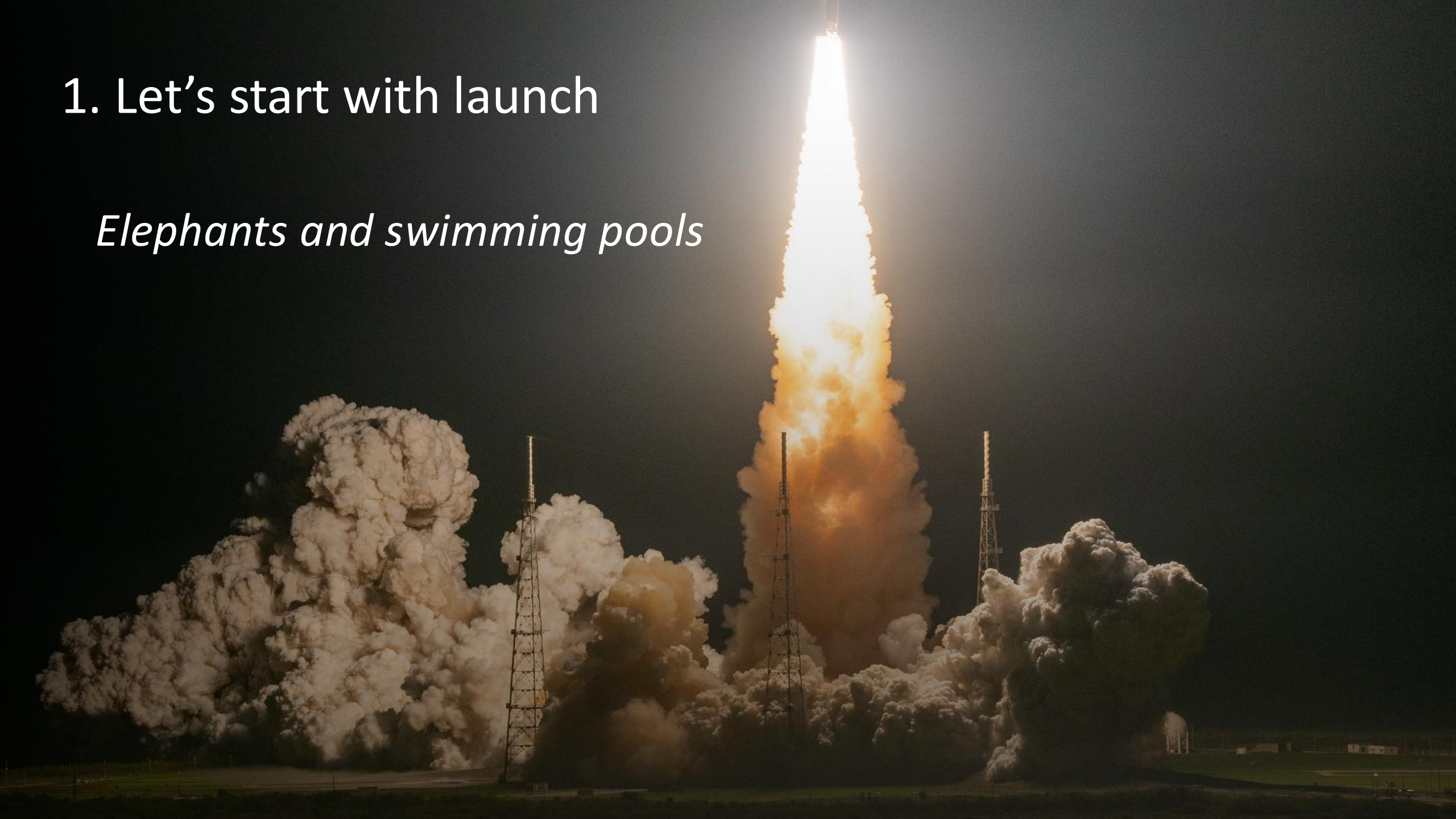


Artemis Architecture Updated Yet Again



1. Let's start with launch

Elephants and swimming pools

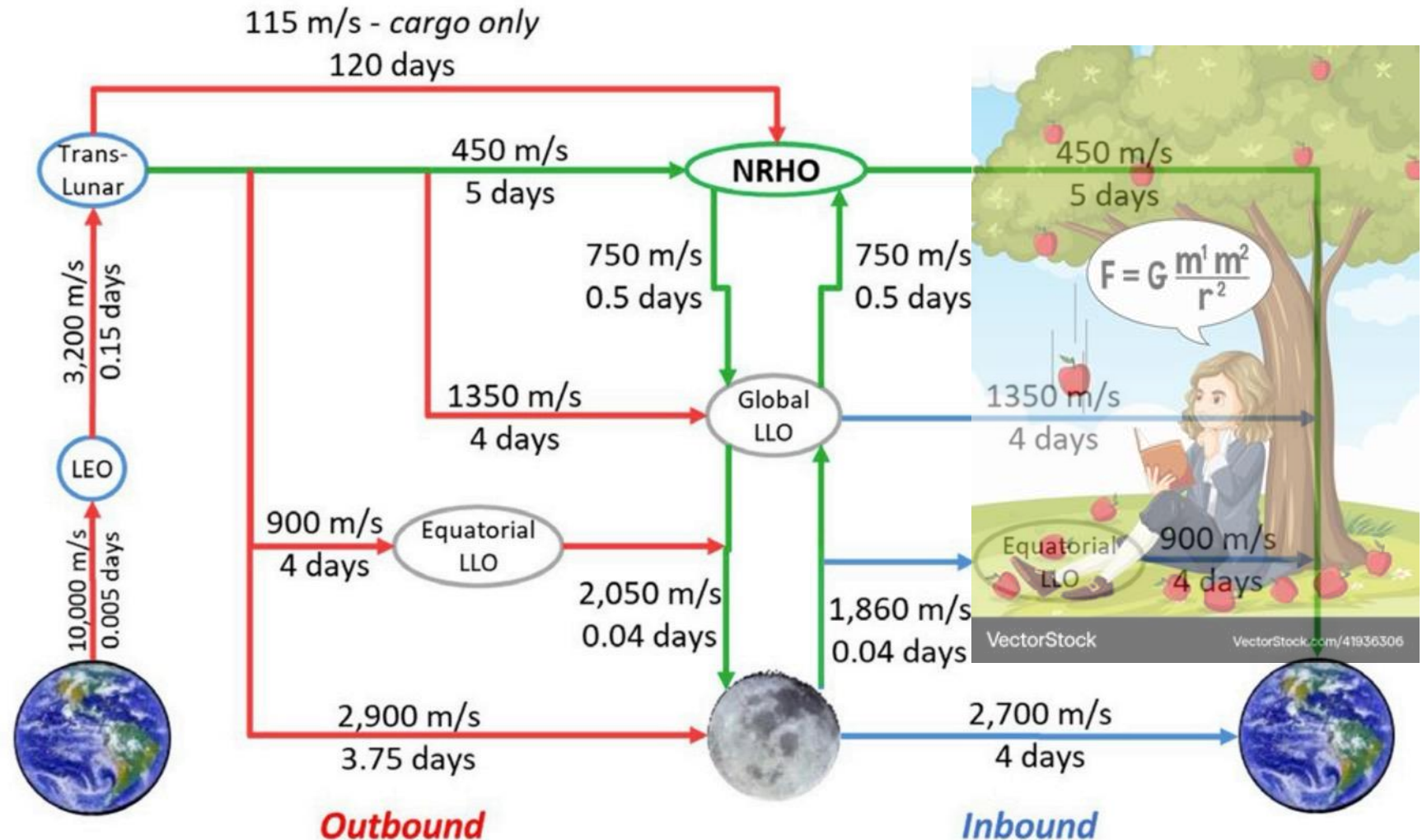


1. Let's start with launch

Δv is the Metric of Space

Energy requirements are set by the masses in the solar system via Newtonian gravity.

These energies are expressed in Δv : the required speed change a vehicle must achieve to climb and brake in gravity wells.



Δv is the Metric of Space

Example: Earth to LLO
to Lunar Surface:

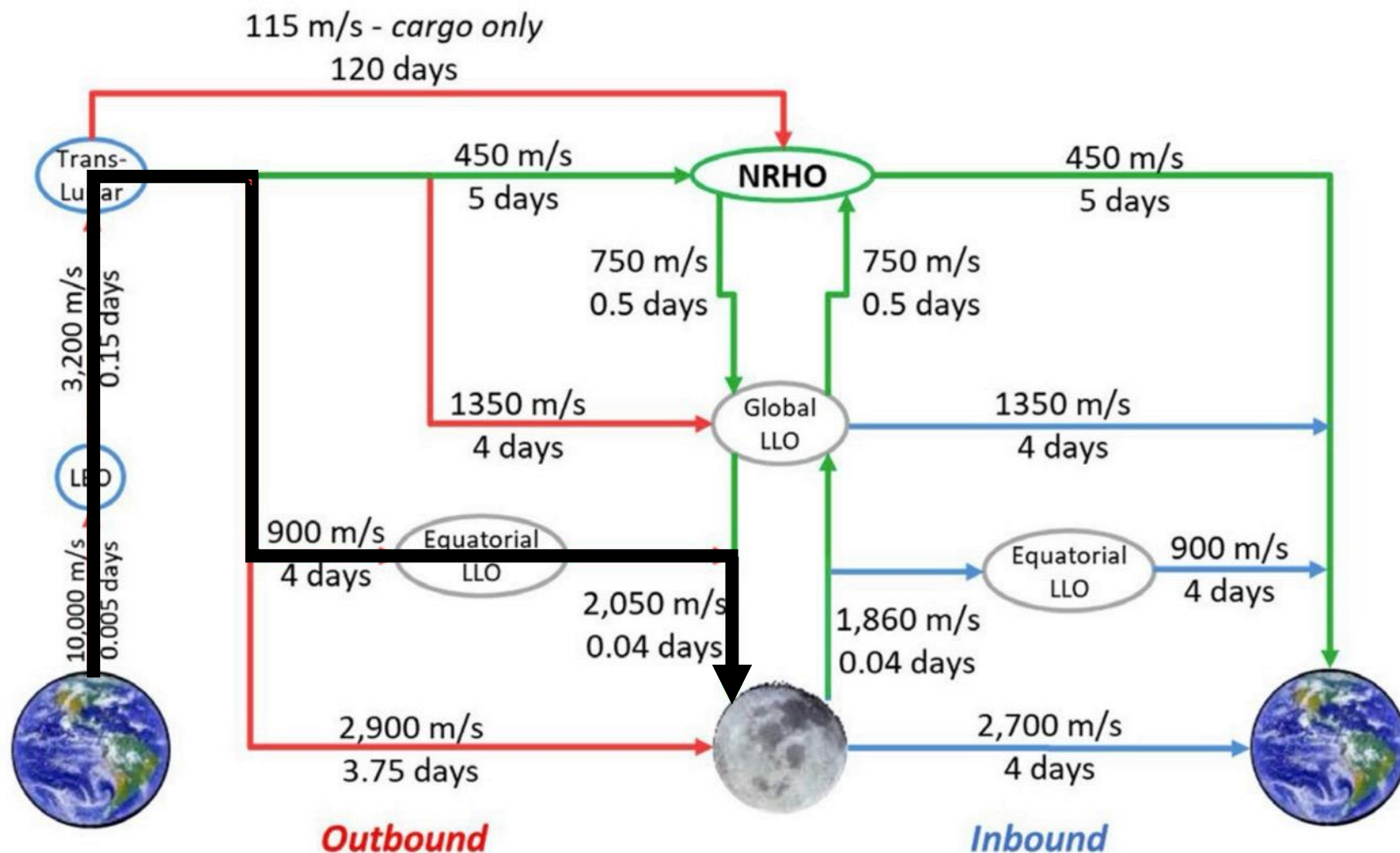
ES-LEO: 10 km/s

LEO-TLO: 3.2 km/s

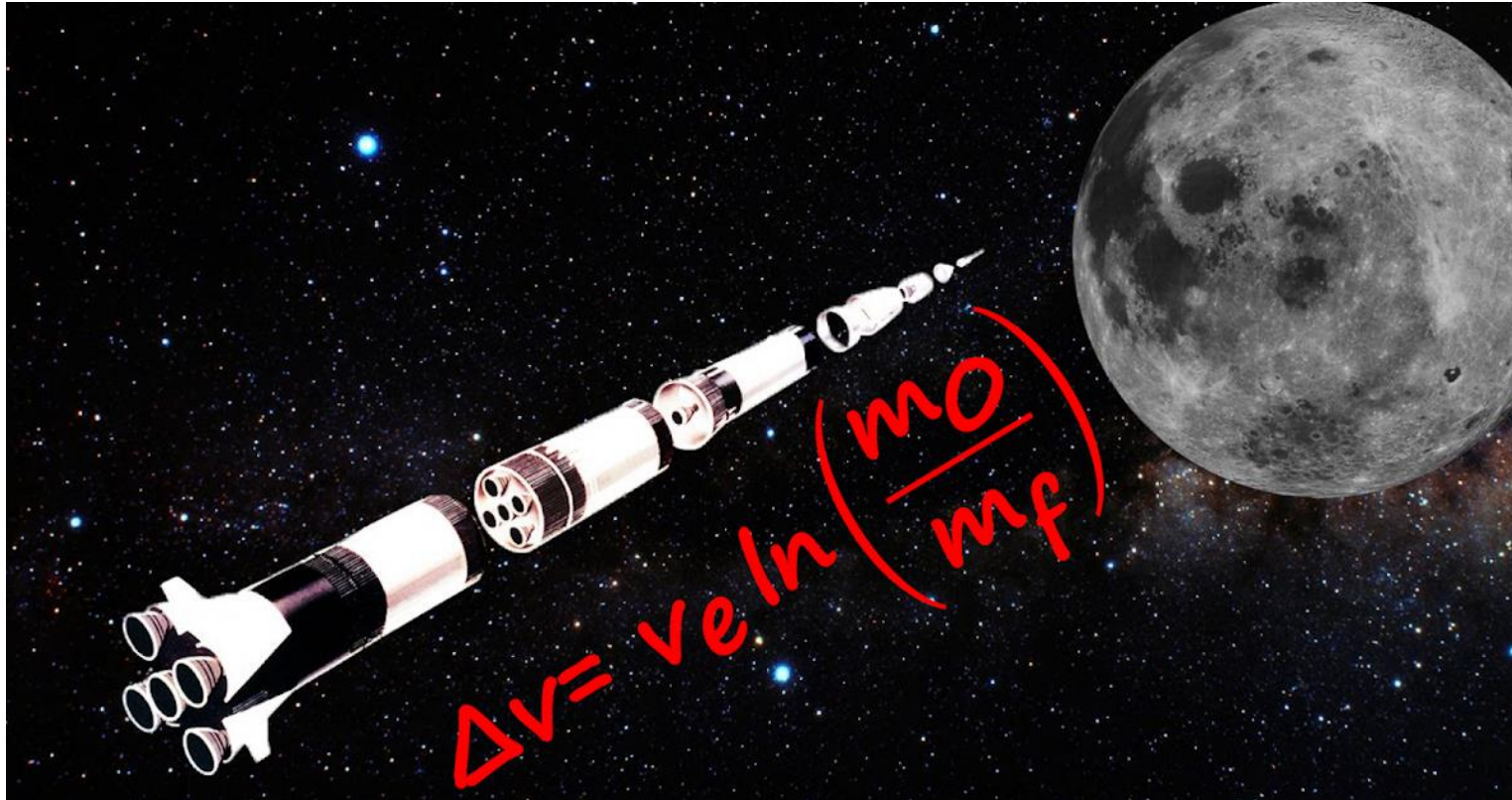
TLO-LLO: 0.9 km/s

LLO-LS: 2.05 km/s

Total: 16.15 km/s



Tyranny of the Rocket Equation



m_0 = initial mass

m_f = final mass

Δv = change in speed

v_e = exhaust speed

To achieve required Δv , there are only two dials: engine design (v_e) and vehicle mass.

Tyranny of the Rocket Equation



m_0 = initial mass

m_f = final mass

Δv = change in speed

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Saturn V solves the Tyranny by *staging masses*

Tyranny of the Rocket Equation



m_0 = initial mass

m_f = final mass

Δv = change in speed

v_e = exhaust speed

Starship HLS must carry all its mass back and forth. Starship solves Tyranny by brute force (mass)



What would you bring to the Moon?

The Artemis Packing List:

Pick **three things** you would take to the Moon. Estimate their mass in kg.

In spaceflight, mass is money, propellant, risk, and choice.



What would you bring to the Moon?

Now estimate the additional propellant required for your payload mass

Rocket Equation applied to Artemis IV:

One way trip: **90 kg propellant per kg payload mass**

Round trip: **165 kg propellant per kg payload mass**

How much propellant are you going to need for your mementos?

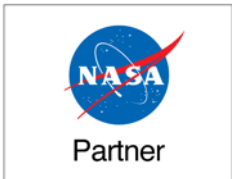


What would you bring to the Moon?

Moon-Bag Challenge Worksheet

Propellant shadow = mass x 90 | Average mission-cost share = mass x \$150,000

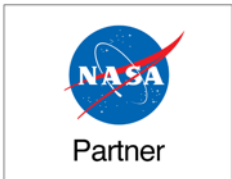
Item	Estimated mass	Propellant shadow	Average cost share
	kg	mass x 90	mass x \$150,000
	kg	mass x 90	mass x \$150,000
	kg	mass x 90	mass x \$150,000



Moon-Bag Quick Reference

A classroom estimate for Artemis-scale Moon-bound mass

Item	Typical mass	Propellant shadow	Average cost share
Photo, patch, letter, or small charm	0.02 kg	1.8 kg	\$3,000
Phone	0.2 kg	18 kg	\$30,000
Paperback book	0.3 kg	27 kg	\$45,000
Journal and pens	0.5 kg	45 kg	\$75,000
1 liter of water	1.0 kg	90 kg	\$150,000
Laptop	1.5 kg	135 kg	\$225,000
Camera kit	2.5 kg	225 kg	\$375,000
Guitar	2.0 kg	180 kg	\$300,000
10 kg comfort bag	10.0 kg	900 kg	\$1,500,000



ARTEMIS II

Official Flight Kit

OFK for ArII is 4.5 kg

ITEM	QTY
Shavings from SLS (Space Launch System) core stage	1
Mini U.S. flags	5
U.S. flags	3
Artemis II rubber stamp	1
Copy of JPL Ranger Program lunar photo negative	1
U.S flags	50
Lockheed Martin patches	17
U.S. flags	50
Exploration Ground System (EGS) program flags	40

ITEM	QTY
Extravehicular Activity and Human Surface Mobility Program (EHP) patches	15
Artemis embossing discs	4
America 250 flags	3
Artemis embroidered ribbons	5
Artemis II holographic seals	850
Artemis II pins	150
Artemis II patches	200
SD card ("Send Your Name to Space" campaign)	1
Unflown Apollo 18 mission flag	1

2. Propellant Challenges of HLS Starship HLS



Blue Moon Mk2



Apollo LEM



Cryogenic Propellant Transfer and Storage

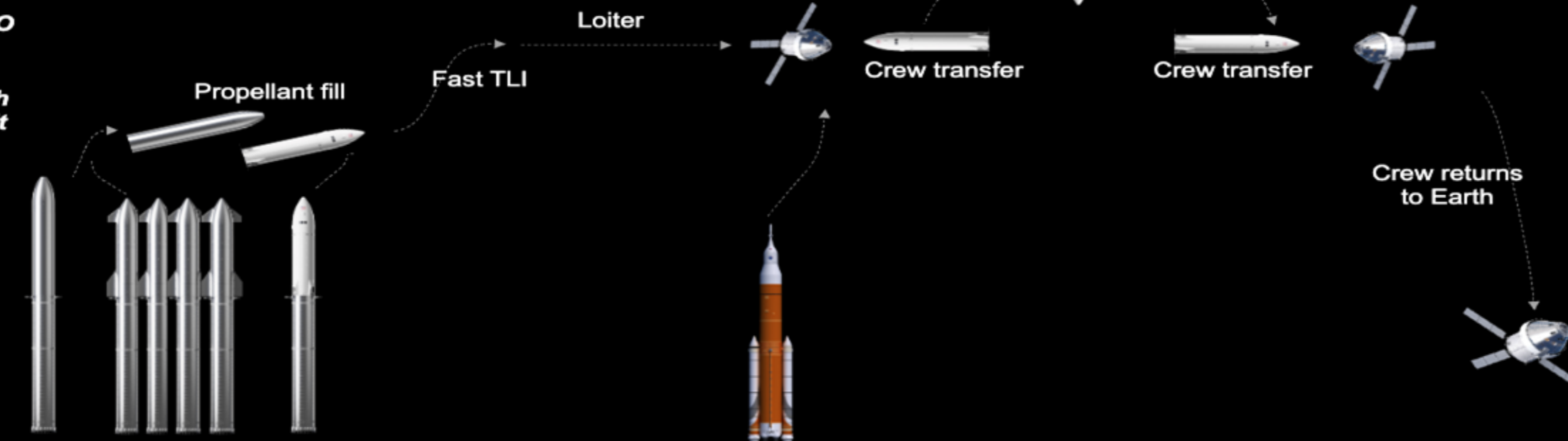
Artemis III Concept of Operations

Moon

NRHO

Earth Orbit

Earth



Propellant
aggregation

HLS Starship
launches

Extended loiter if
needed

Orion launch

Variable
Stay on the Moon

Crew returns to
Orion

Crew returns
to Earth

Crew transfer

Crew transfer

Fast TLI

Loiter

Propellant fill

Cryogenic Propellant Transfer and Storage

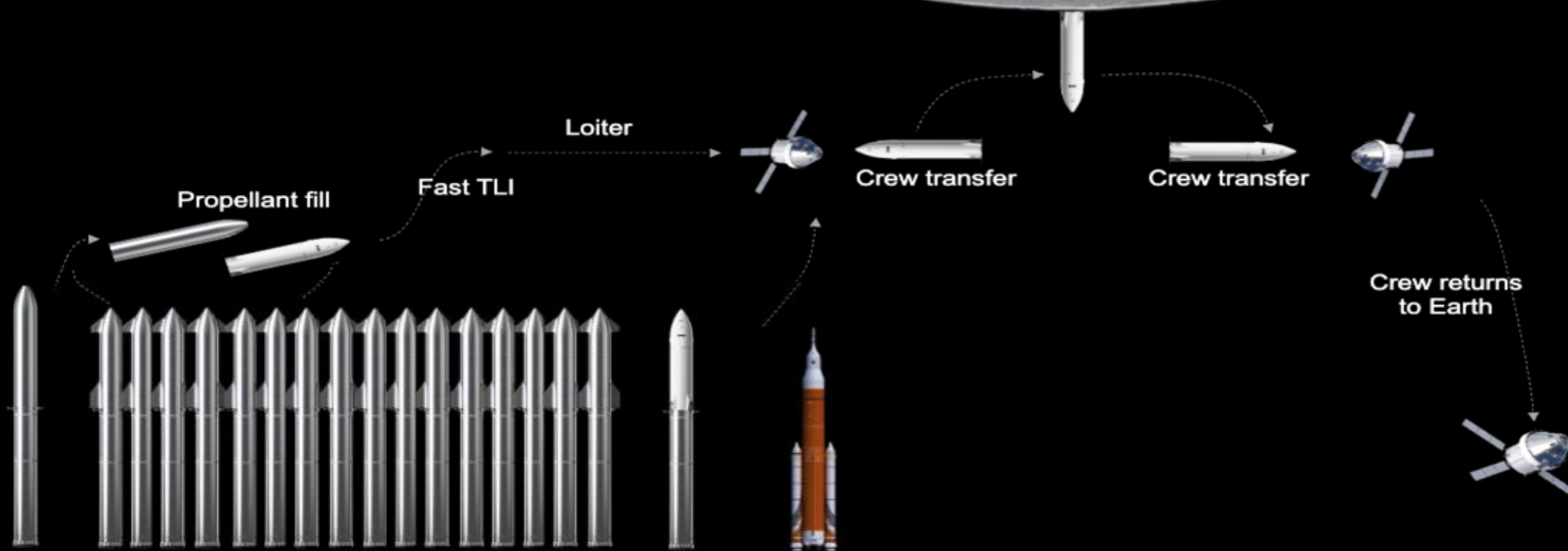
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Propellant aggregation

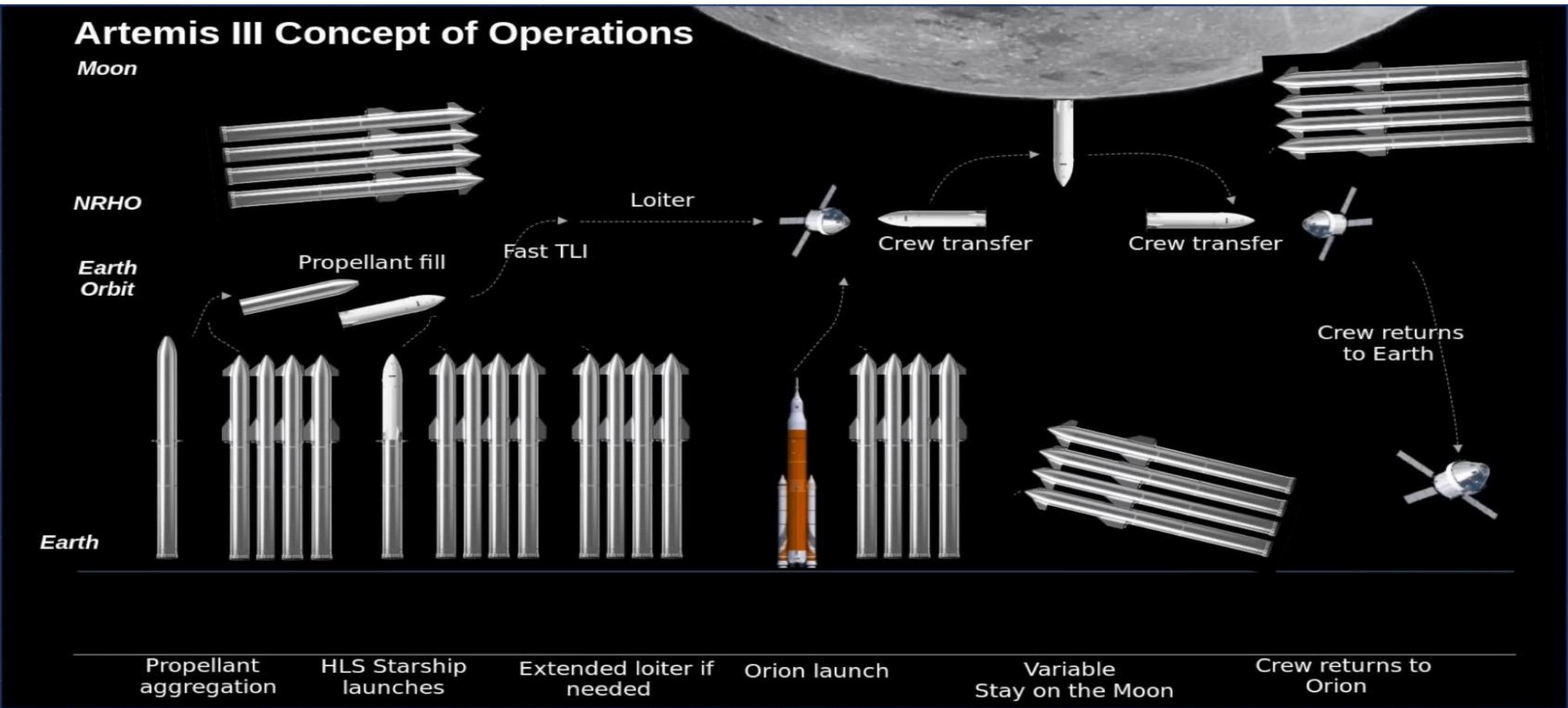
HLS Starship launches

Extended loiter if needed

Orion launch

Variable Stay on the Moon

Crew returns to Orion



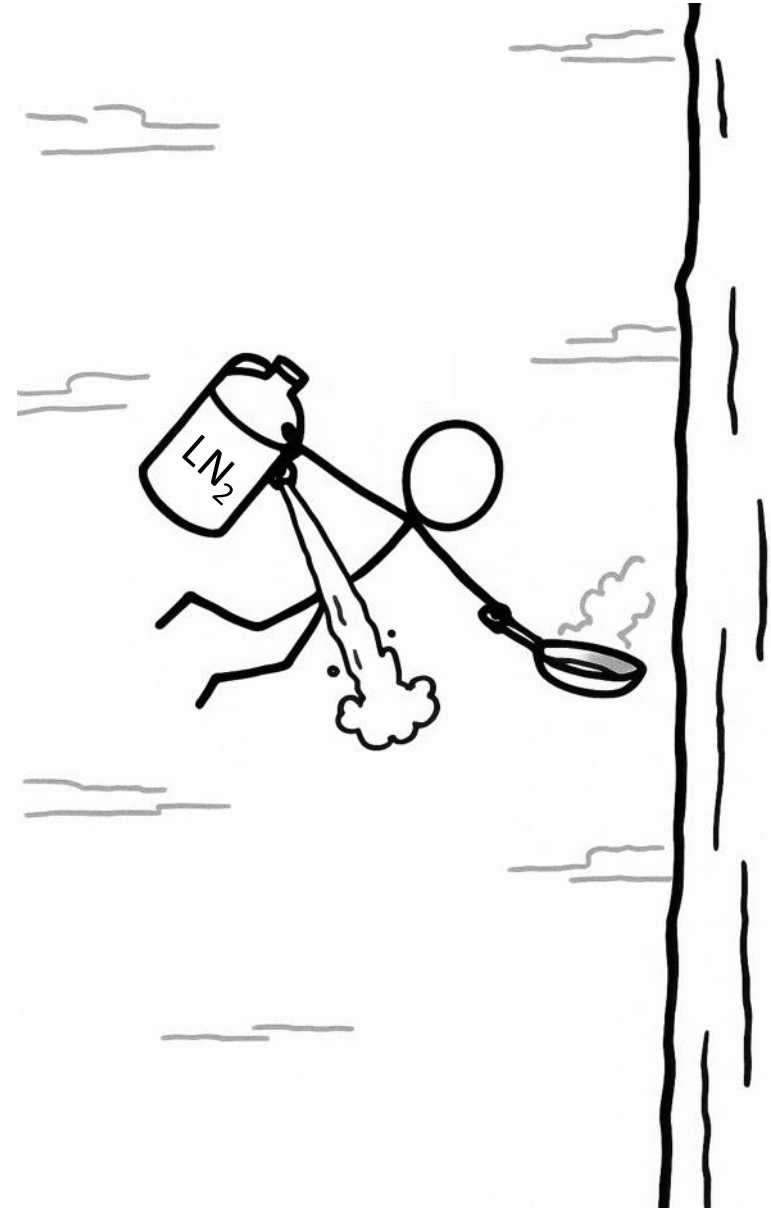
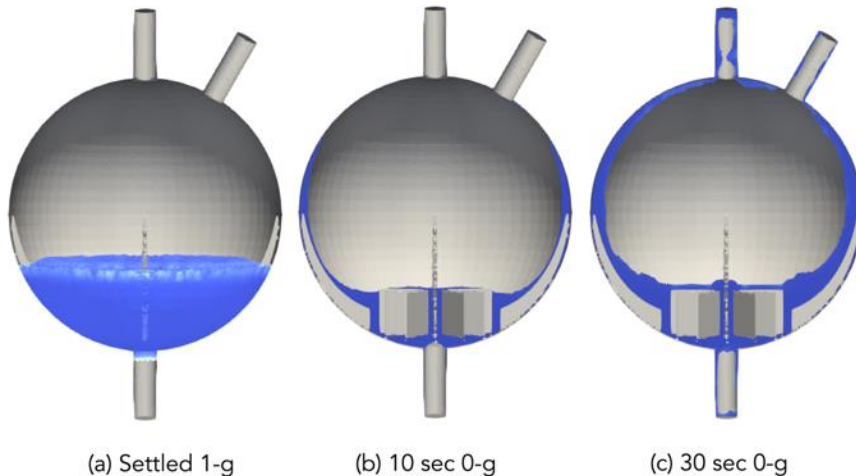
Why So Hard?

Cryogenic Challenges:

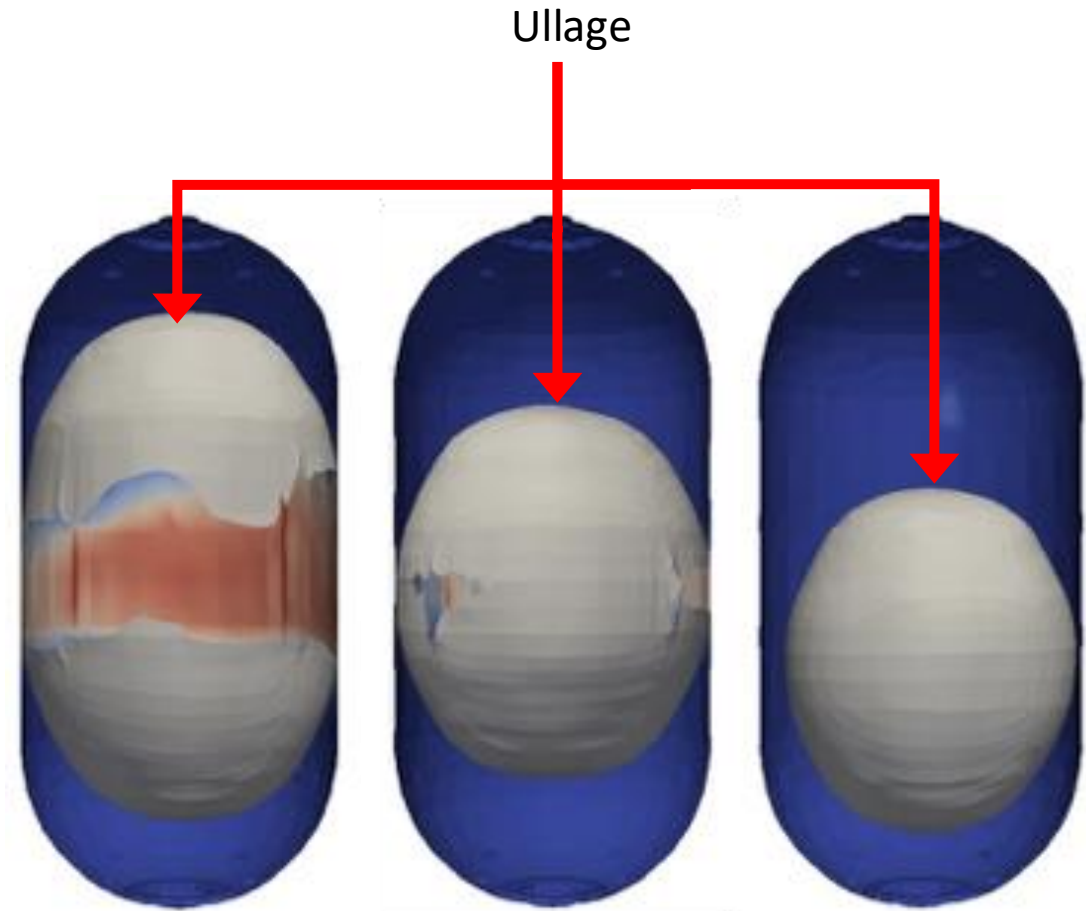
- Transfer lines cause propellant boil-off
- Storage in space causes propellant boil-off
- Boil-off causes autogenous pressurization
- Propellant becomes a frothy mixture of gasses and liquid

Microgravity Challenges

- Pressure must be vented: need *phase separation*
- Capillary adhesion causes propellant to cling to tank walls
- No reliable mass gauge



In-Space Propellant Transfer



Your Turn

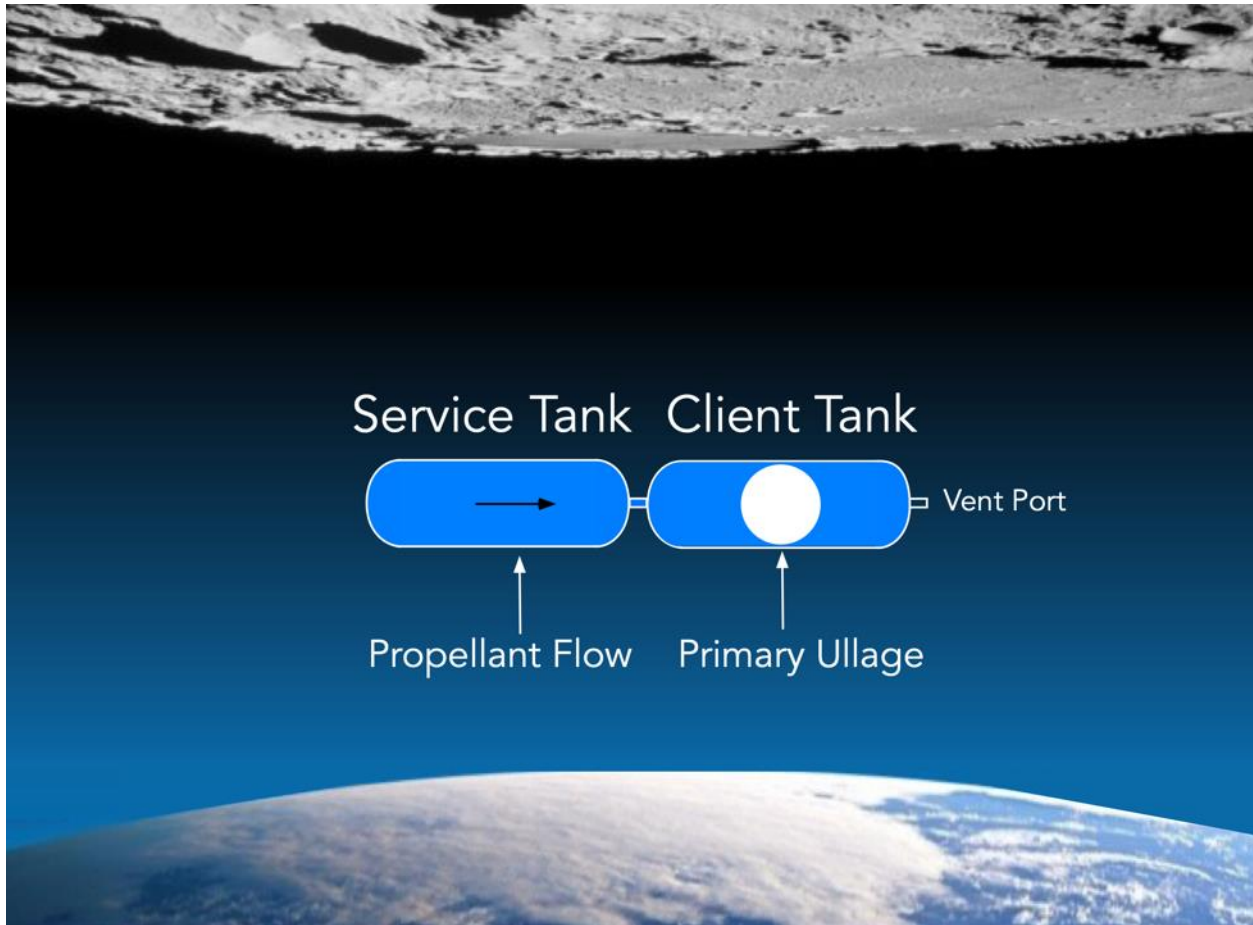
Transfer propellant from the service to the client tank:

1. Without allowing propellant to escape
2. Without moving the tanks





Microgravity Ullage Trapping

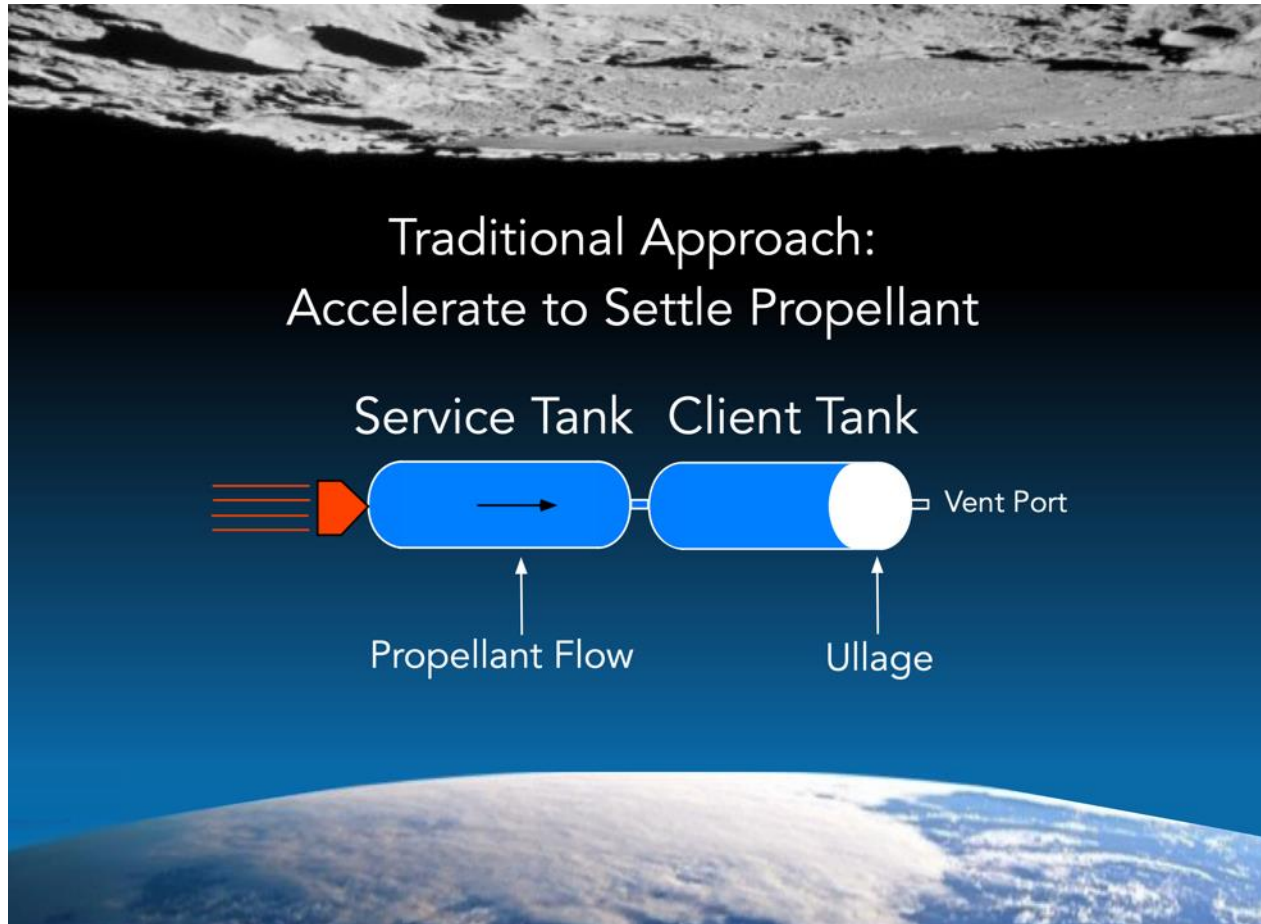


To transfer propellant from Service to Client tank, we need the ullage at the vent port.

In microgravity, the ullage drifts and has no preferential location in the tank.

Most methods of prop transfer involve moving the primary ullage.

Microgravity



Ullage Settling is the proposed (but untested) approach to *phase separation* for Artemis HLS vehicles.

Ullage Settling creates artificial gravity that separates the gas and liquid phases.

Drawback: complexity and risk

Microgravity Ullage Trapping



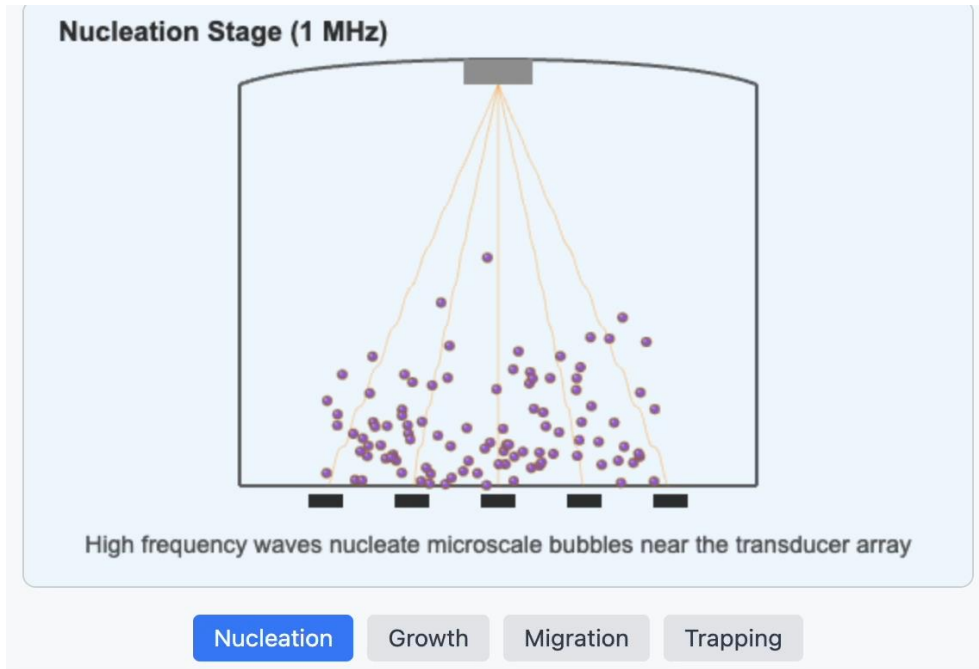
We propose desaturating the client tank to form a *secondary* ullage.

Desaturating: drop the client tank pressure by a small amount. Saturated pressurant (helium) comes out of solution.

Acoustic transducers evolve, migrate, and trap the secondary ullage near the vent port.

Bubble Trapping as Closed-Loop State Control

A phased acoustic array can create, move, and hold a secondary ullage near a vent port when the drive is adapted to the evolving bubble population.



Hidden State

- Bubble size distribution
- Void fraction
- Sheet location
- Local sound speed
- Trap stability

Sensor / Data Inputs

- Acoustic response
- Drive frequency, phase,
- Amplitude
- Temperature
- Vent / flow response

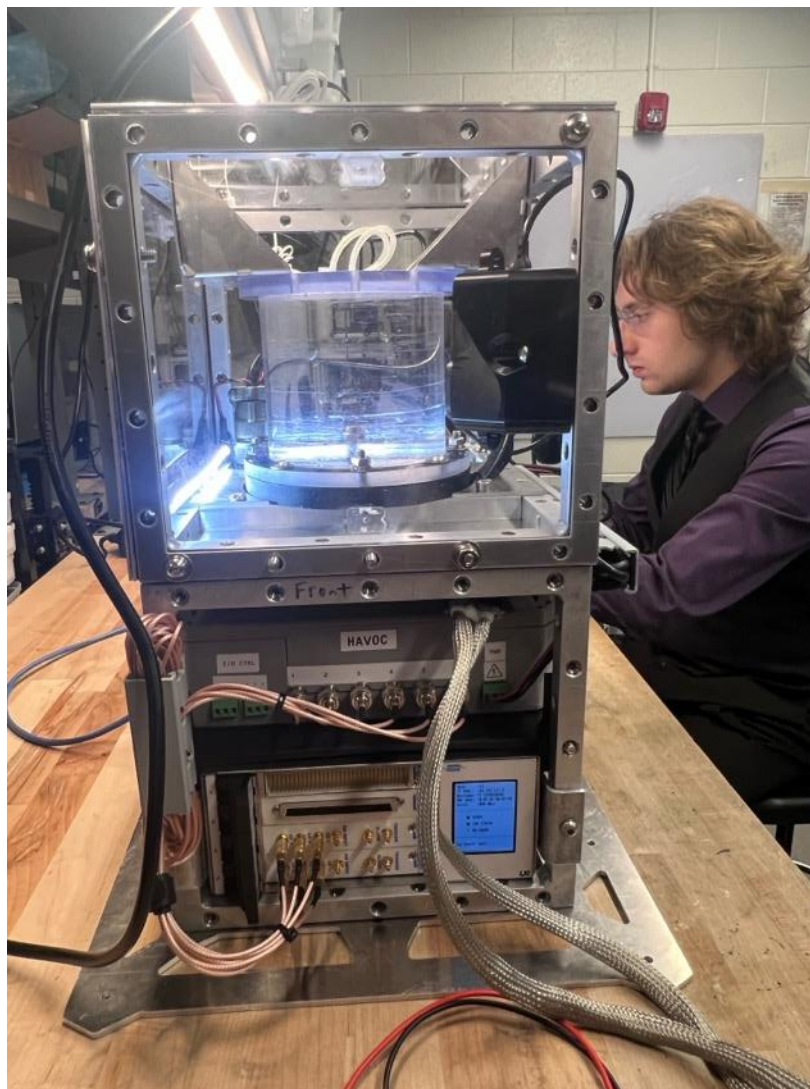
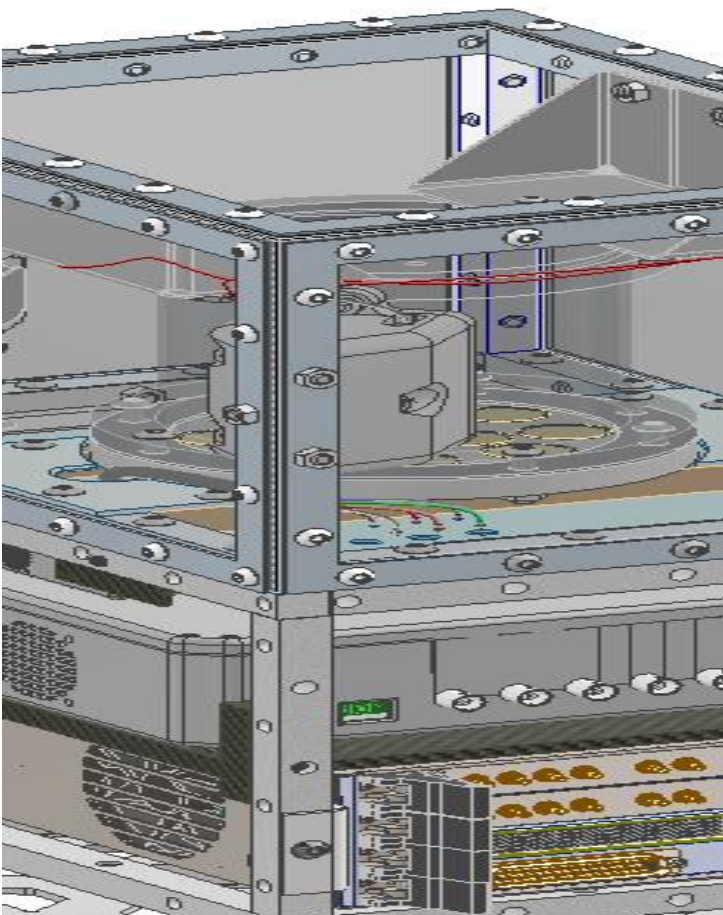
Estimator

- Bubble-sheet state
- Average bubble size
- Minnaert frequency band
- Trap confidence

Control Outputs

- Retuned frequency
- Phase pattern
- Amplitude envelope
- Migration wave
- Trapping wave

Suborbital Flight Experiment: MUTT

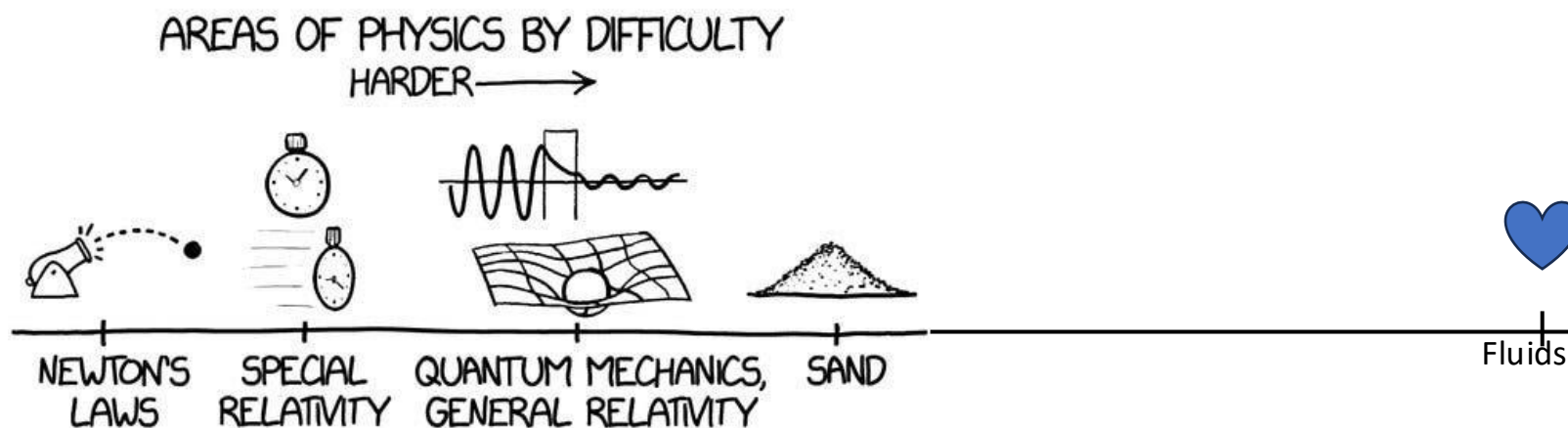
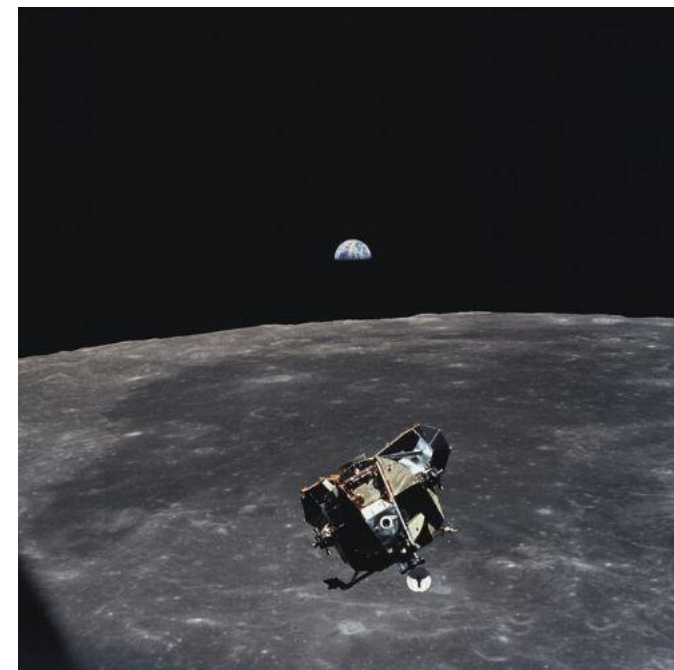


3. Running on Empty



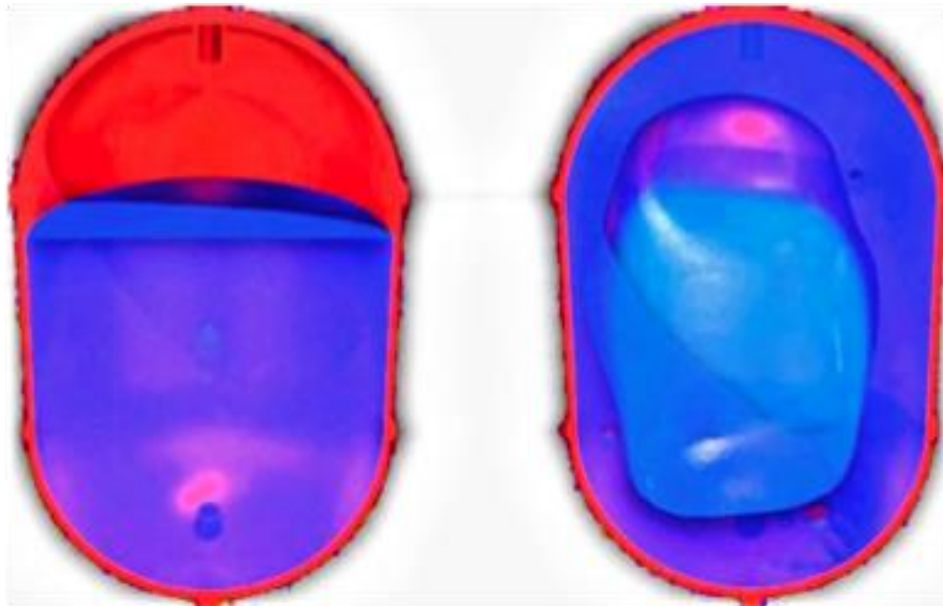
Suborbital Testing of Propellant Mass Gauging Technology

- 56 years post Apollo 11: still no reliable mass gauge
- No buoyancy in microgravity means no level sensors



Modal Propellant Gauging

Using sound to infer fluid distribution inside the tank



1-g

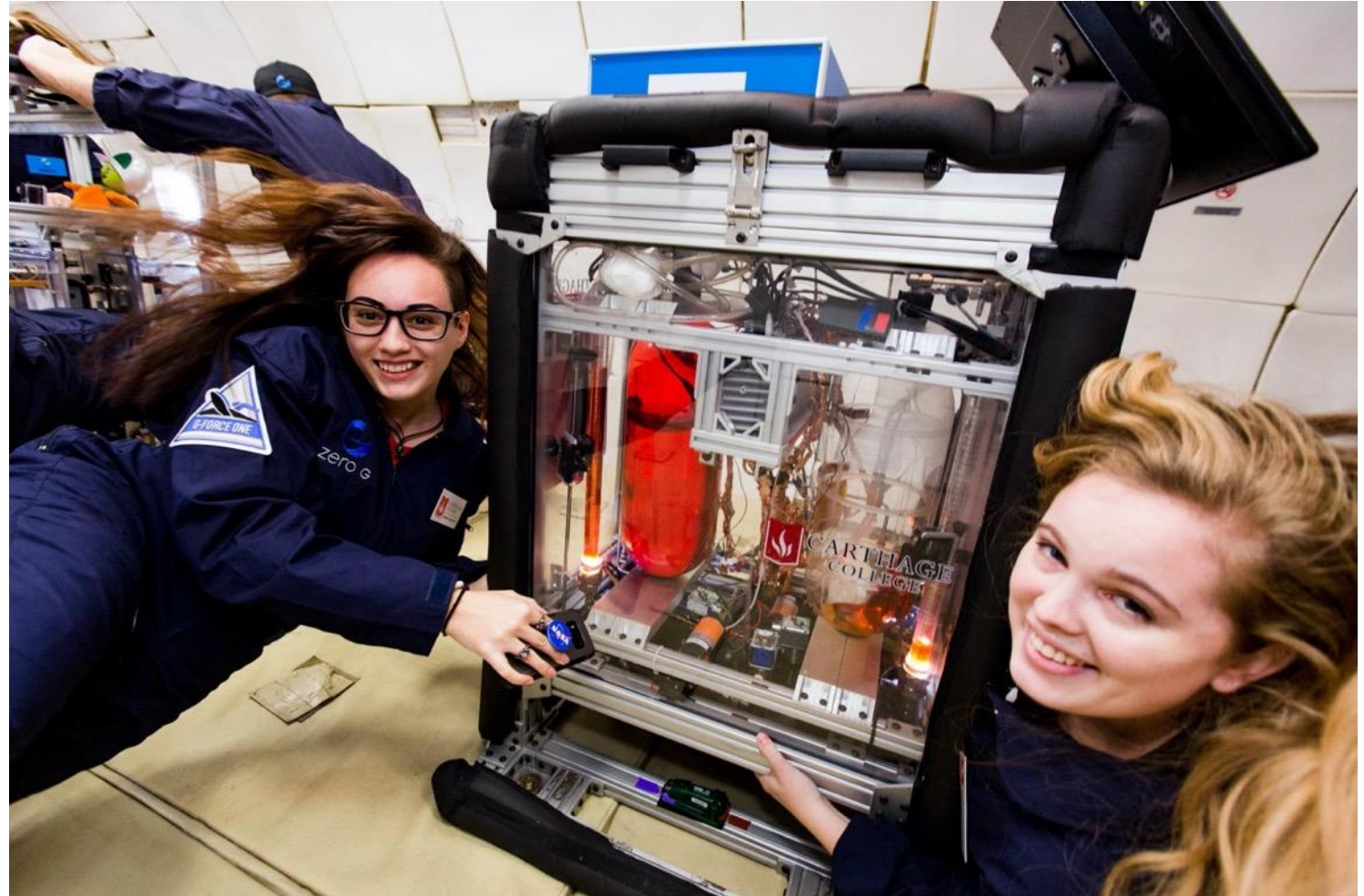
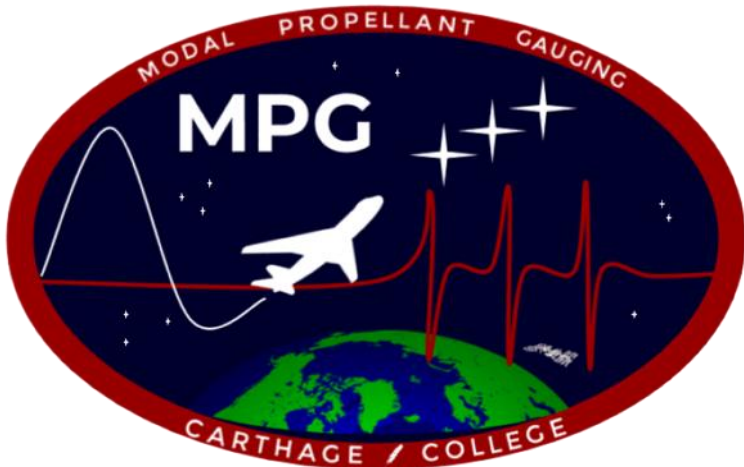
0-g

In 0-g, liquid mass sticks to the walls and **mass loads** the surface

Zero-g Complications

- Wetting
- Stratification
- Film boiling
- Slosh memory

MPG uses *Modal Response* of Tank to Estimate Contained Mass



08:07:08



CARTHAGE
COLLEGE

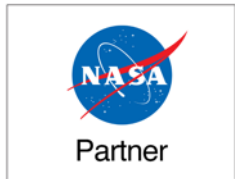


07:23



Altitude (km): 55

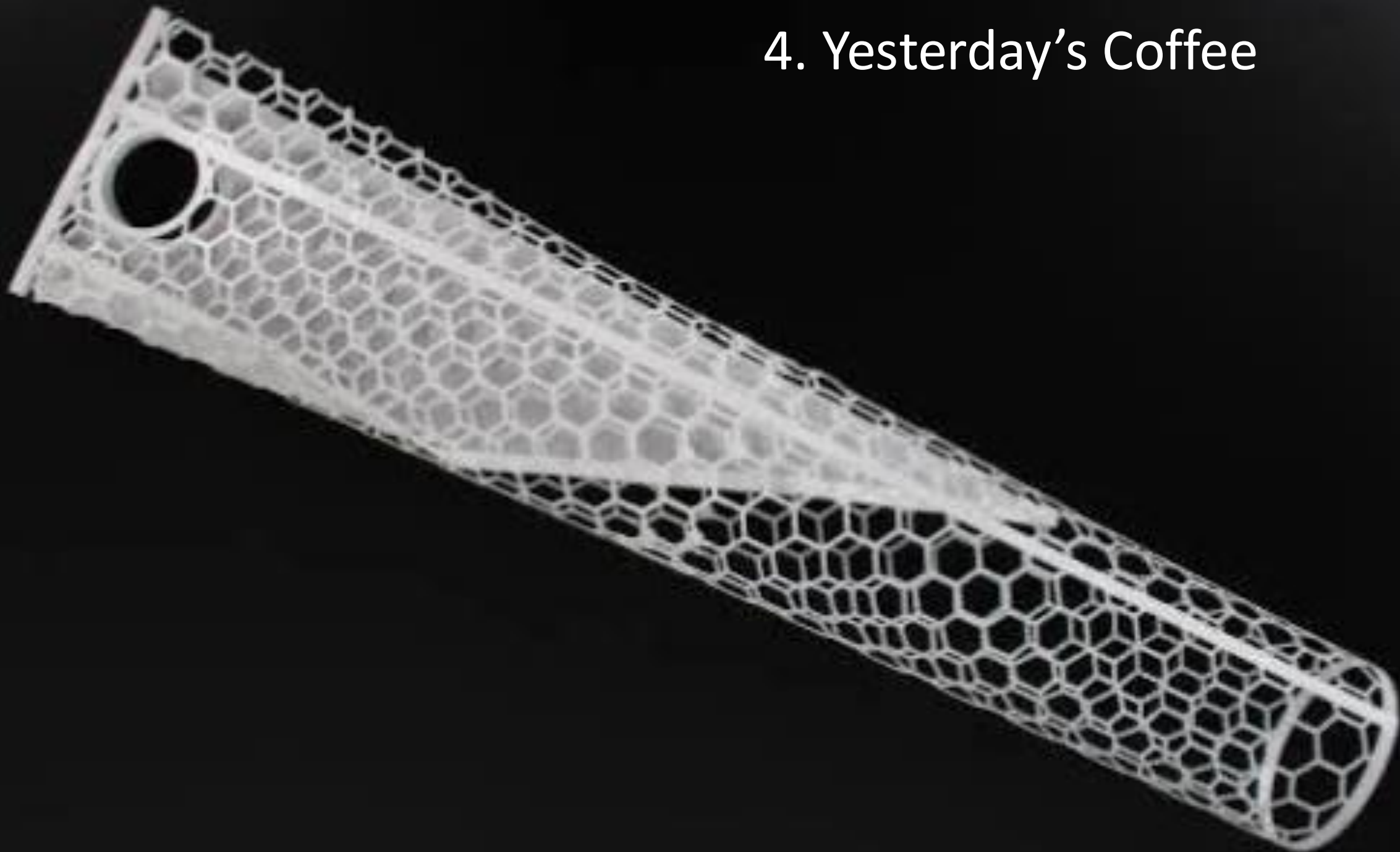




Don't Spill the Water!

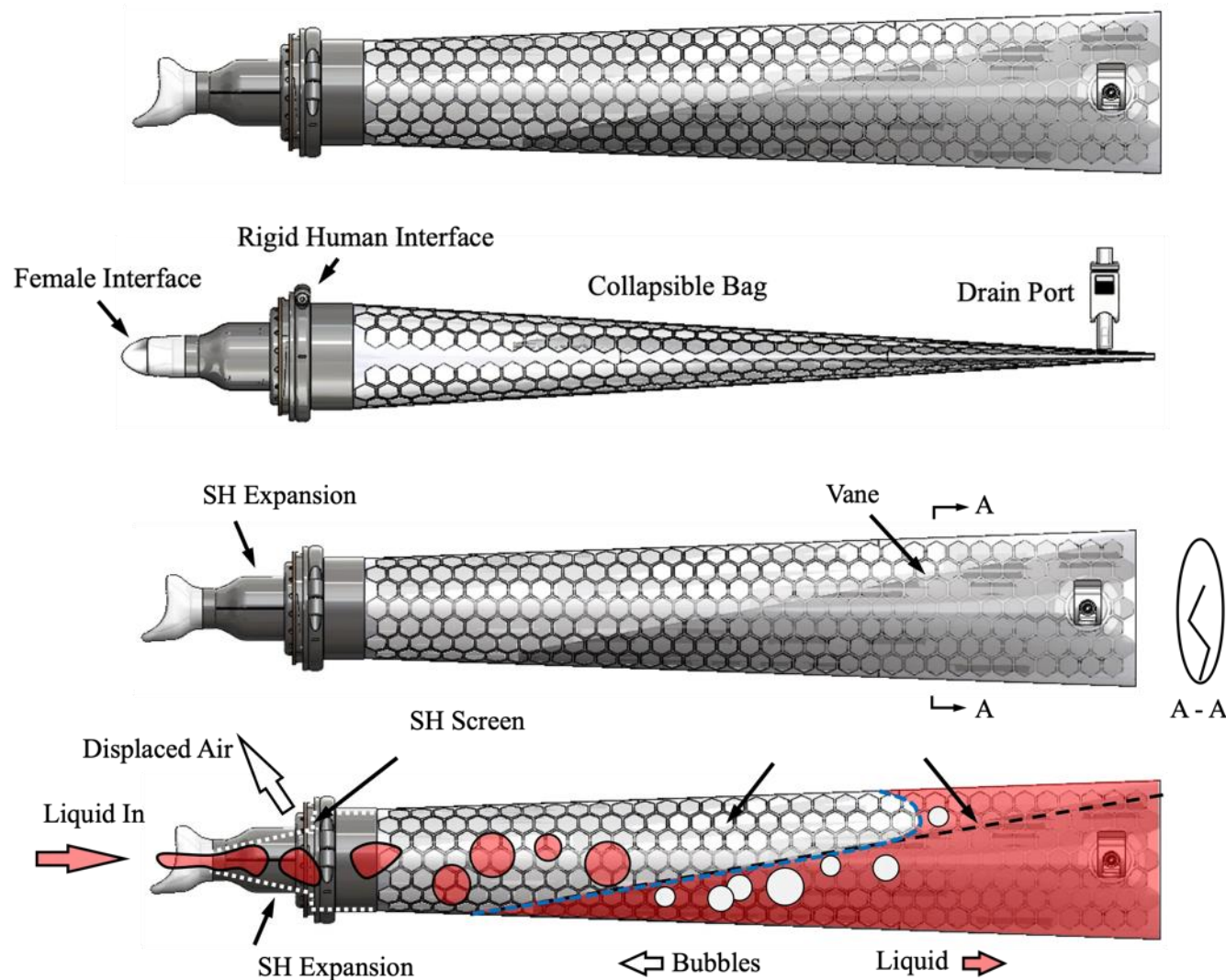


4. Yesterday's Coffee



CCU Saves the Day

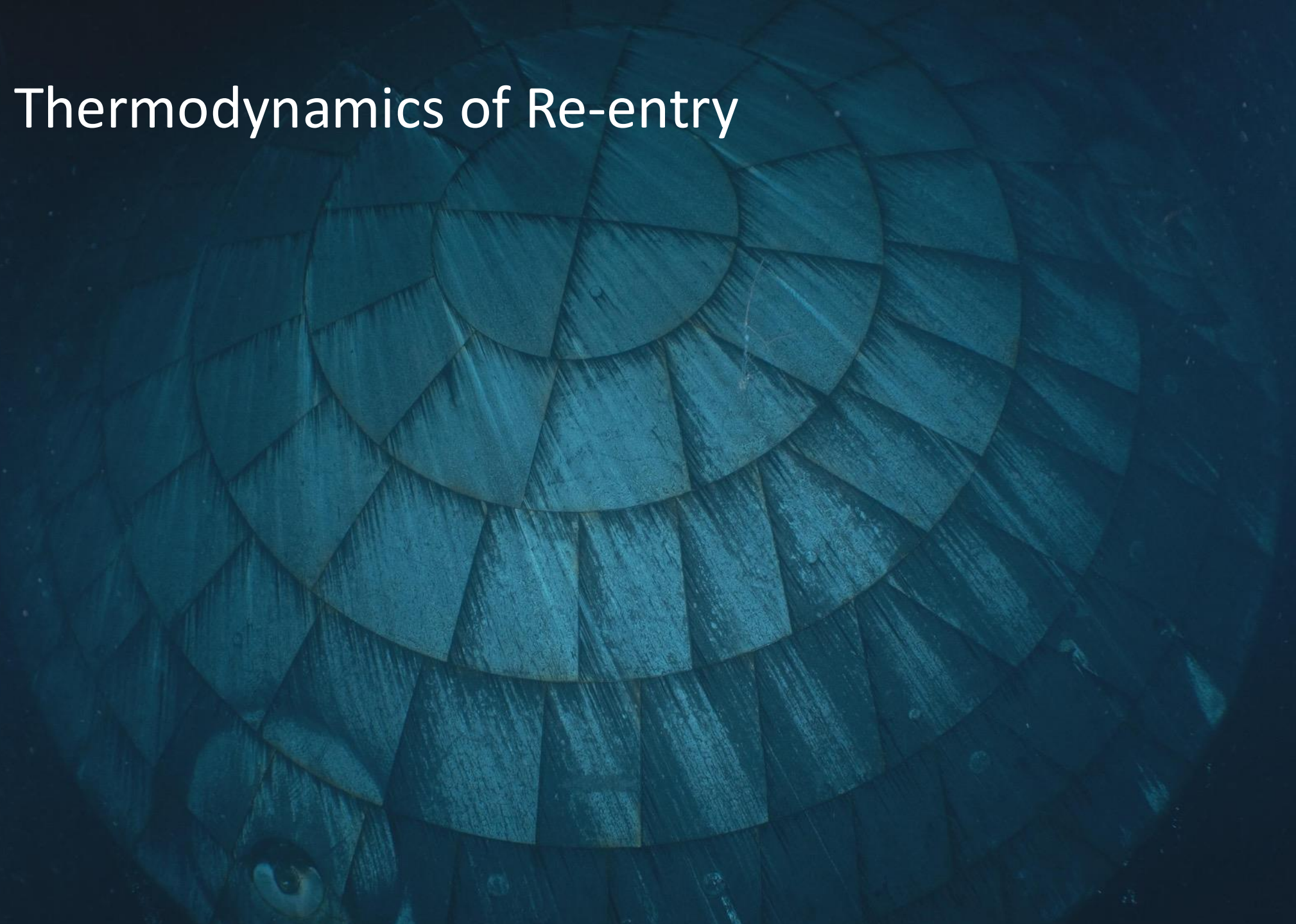
- The CCU is a passive microgravity fluid device that replaces gravity with surface chemistry and geometry:
- Water-repelling surfaces push urine away from the body, water-loving surfaces wick it into the bag,
- The tapered honeycomb-supported liner gives the liquid a preferred "downhill" direction.







5. The Thermodynamics of Re-entry







Artemis I Heat Shield

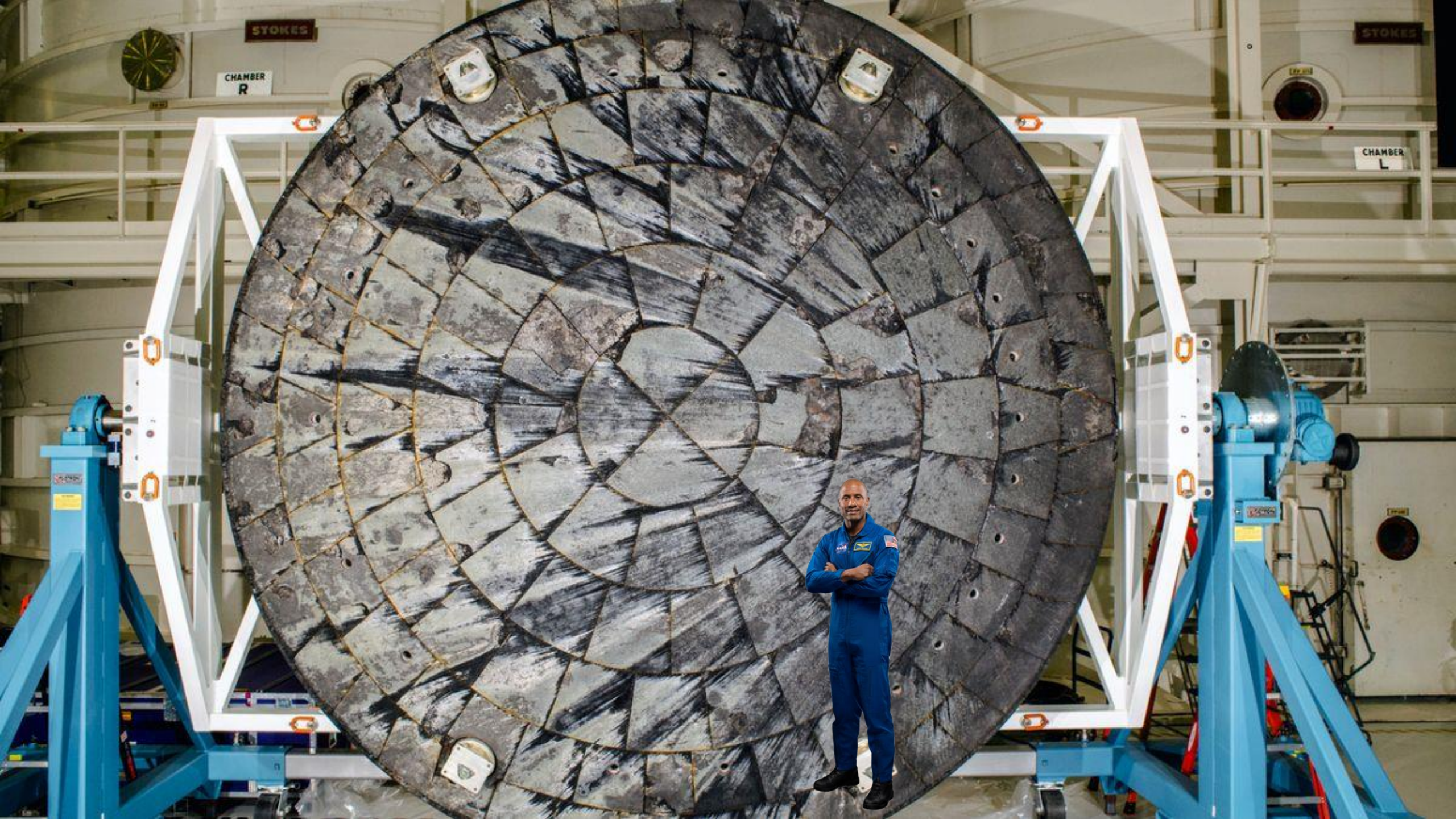
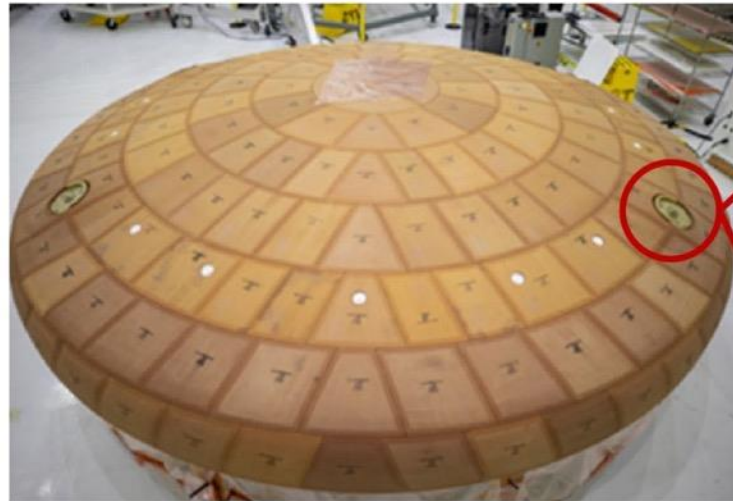


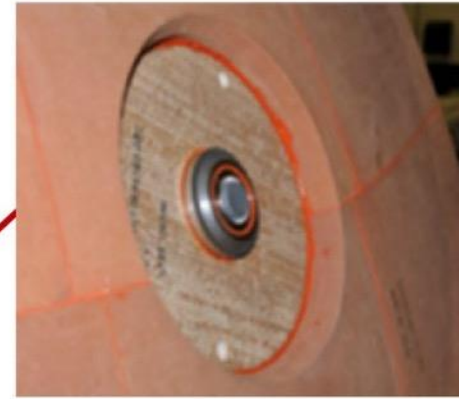




Figure 4: Orion Crew Module/Service Module Separation Bolt Melting and Erosion



Orion heat shield prior to the Artemis I mission, showing the separation bolt circled in red



Close up of Orion separation bolt prior to Artemis I mission



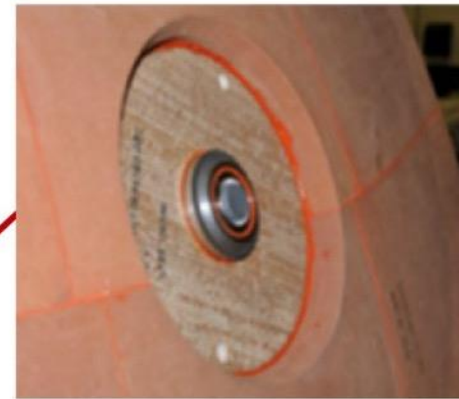
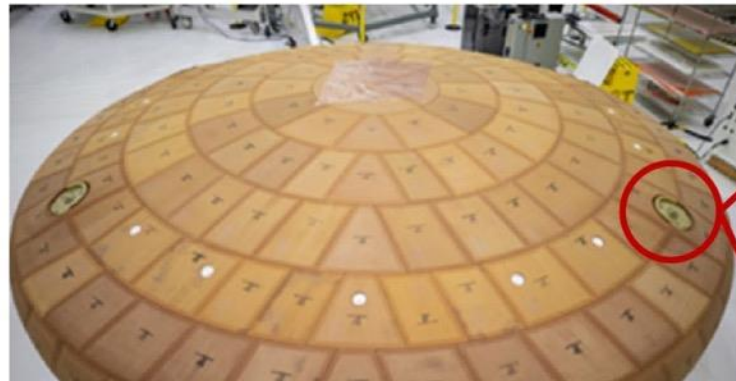
Separation bolt melting and erosion post-Artemis I reentry

Source: NASA.





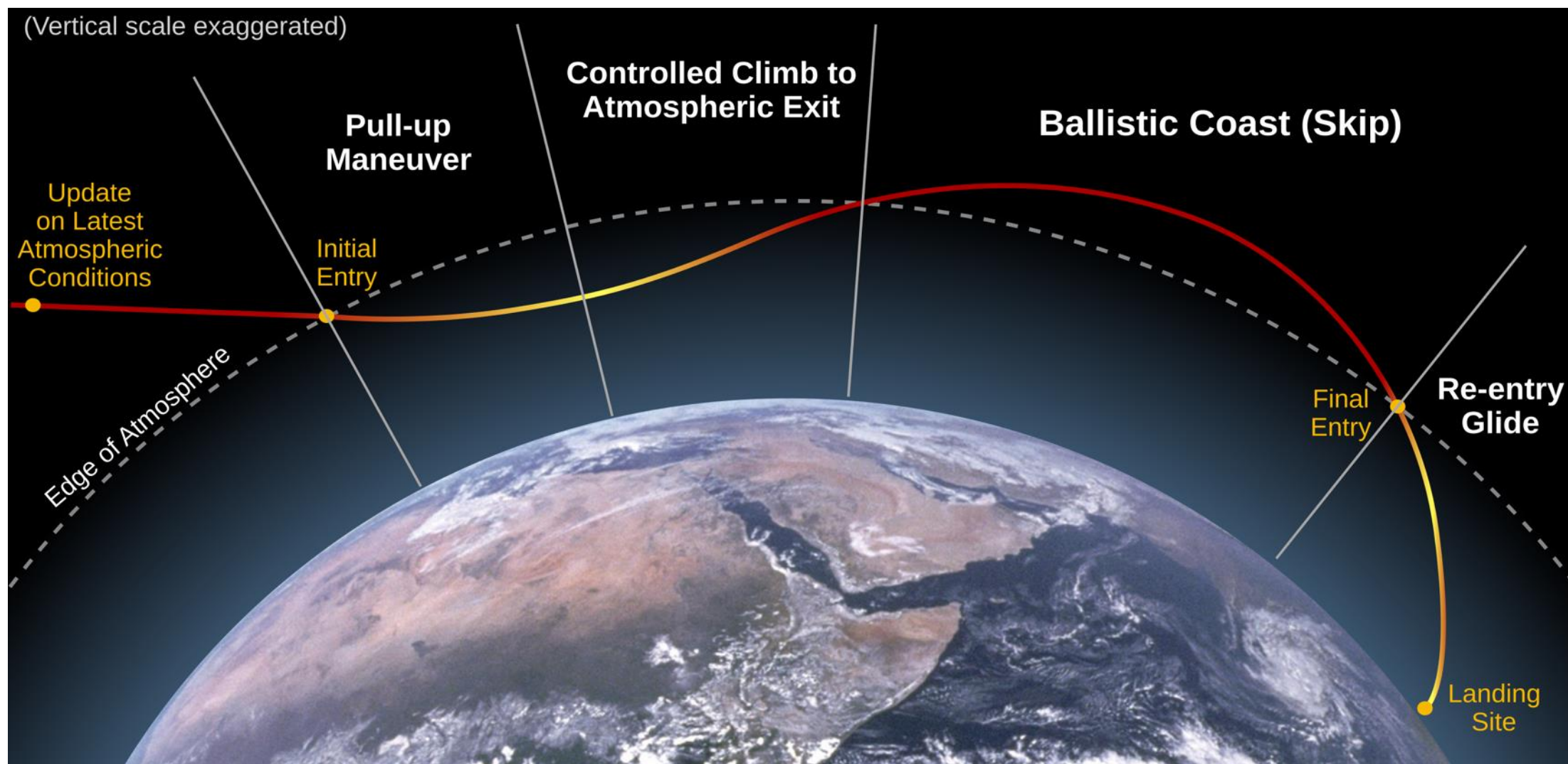
Figure 4: Orion Crew Module/Service Module Separation Bolt Melting and Erosion



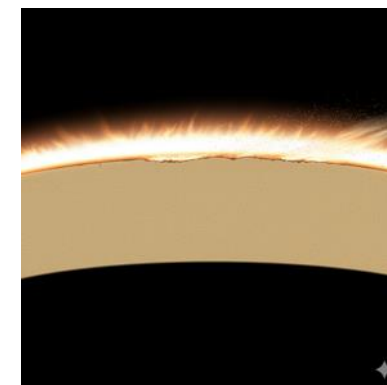
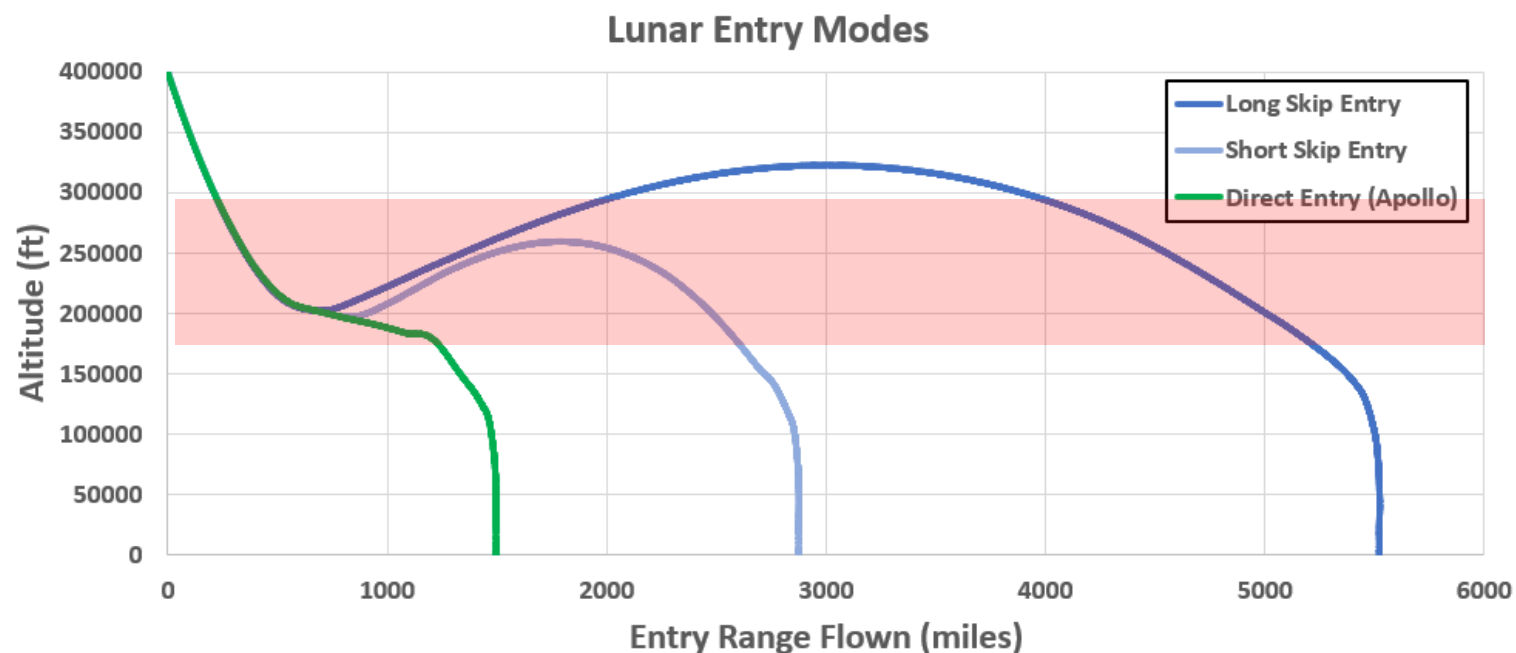
Close up of Orion separation bolt prior to Artemis I mission



Skip-Entry

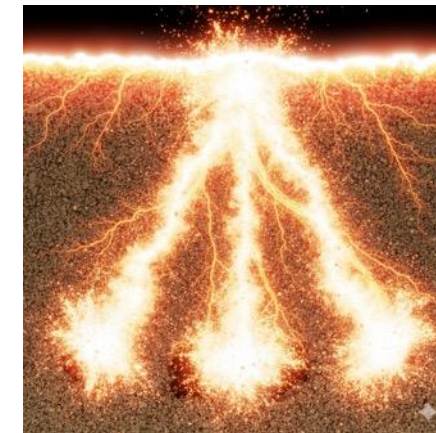
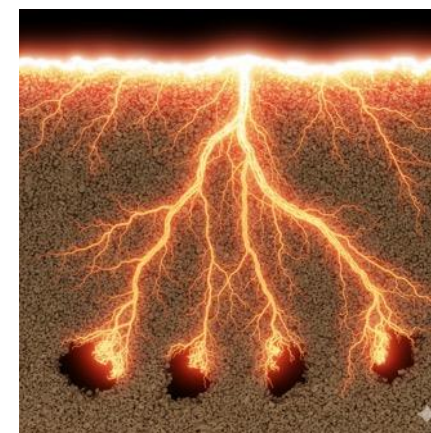
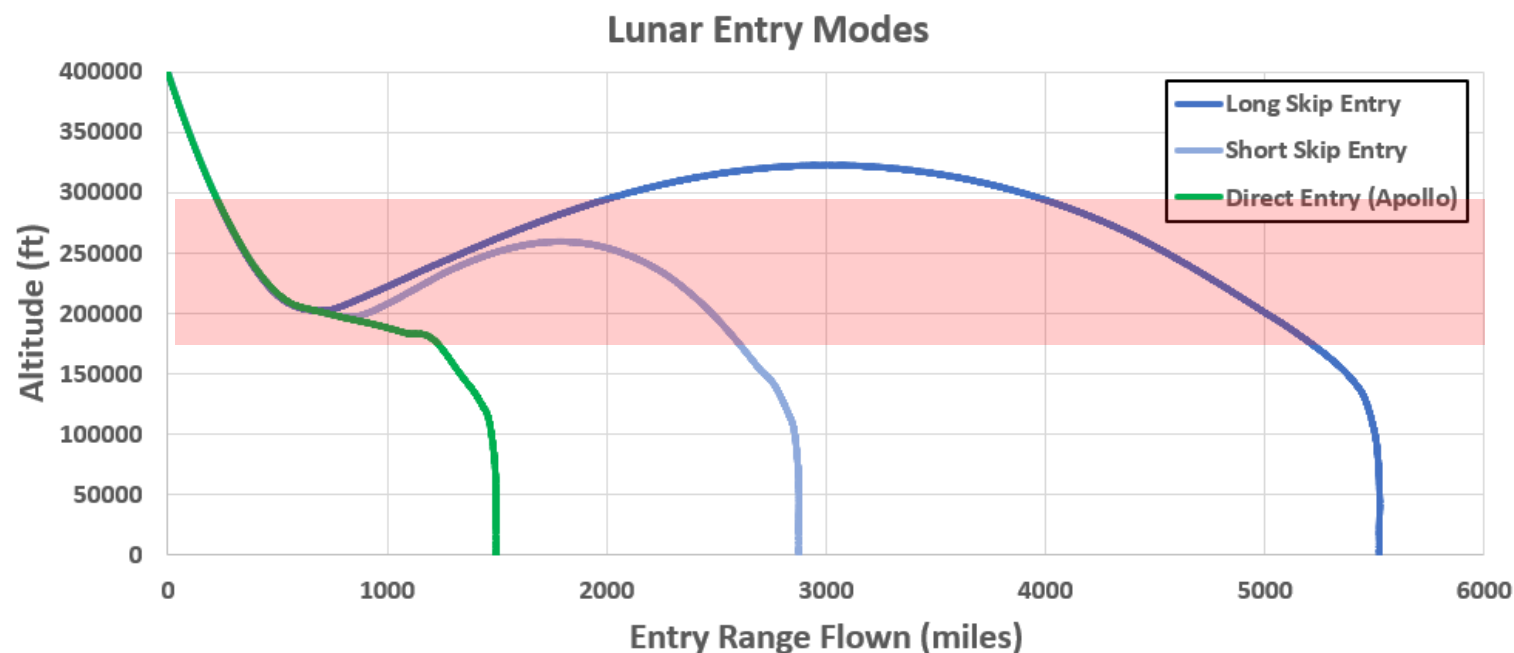


Culprit: Skip-Entry



Heat Shield Cross-section
Intended behavior

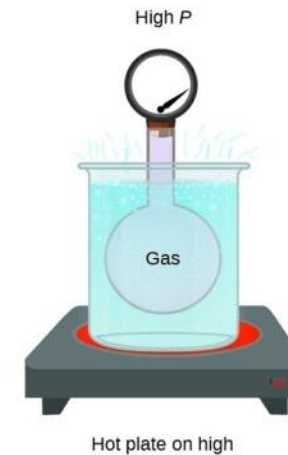
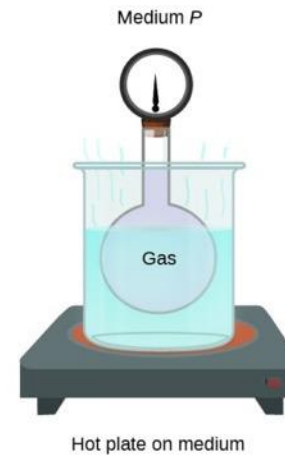
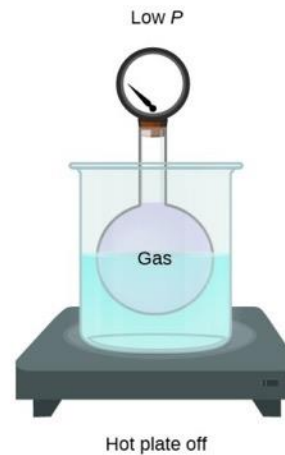
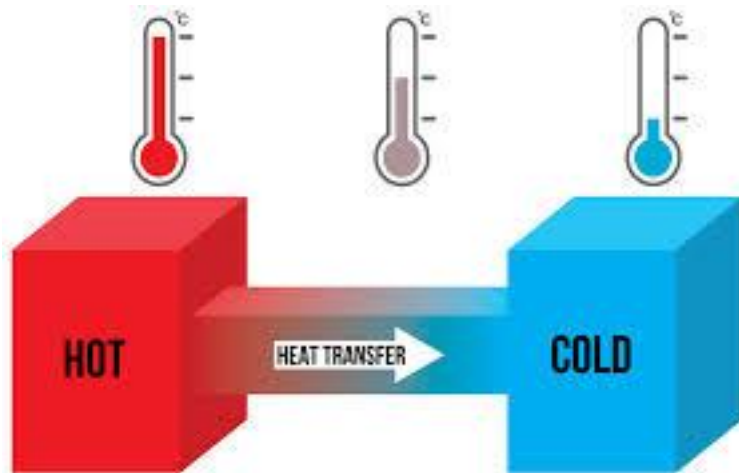
Culprit: Skip-Entry



Heat Shield Cross-section: void formation during excessive heating from skip-entry

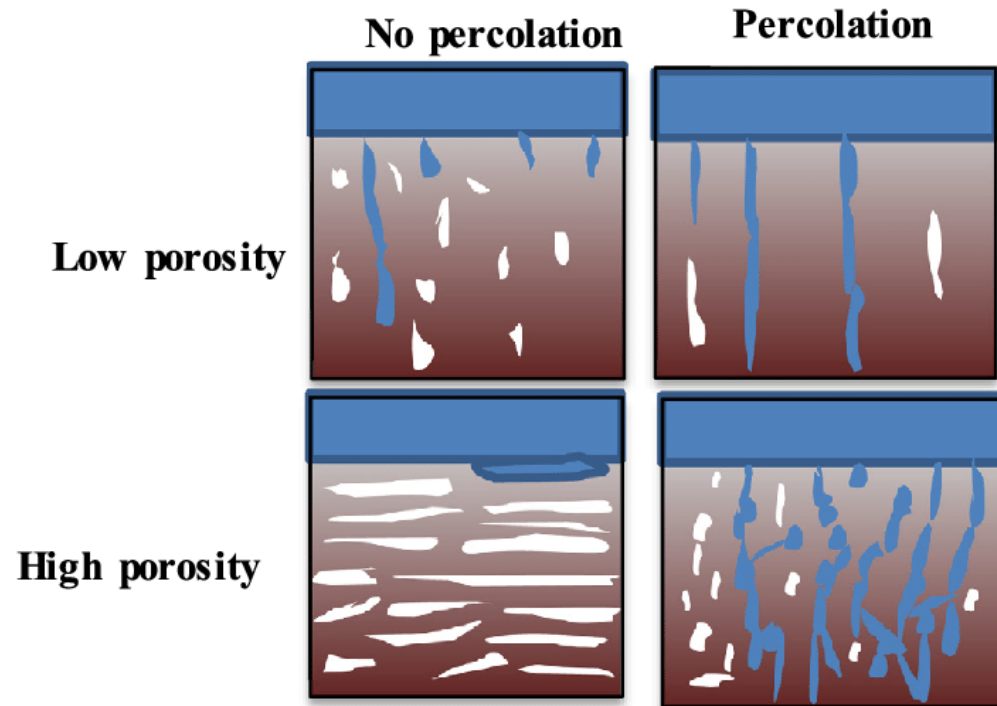
Classroom Connections

- Transport and Ideal Gas Law

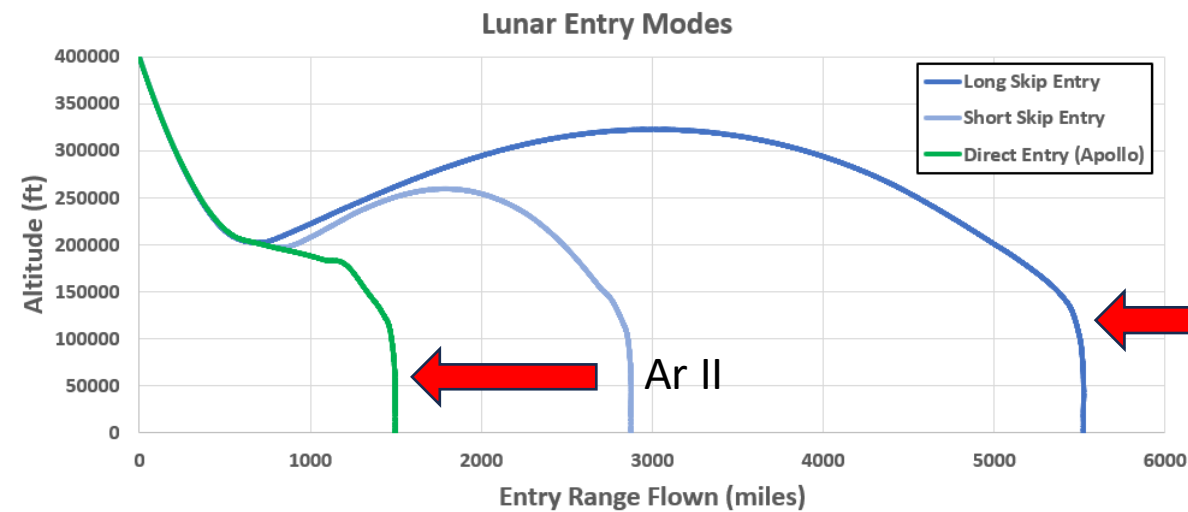


Classroom Connections

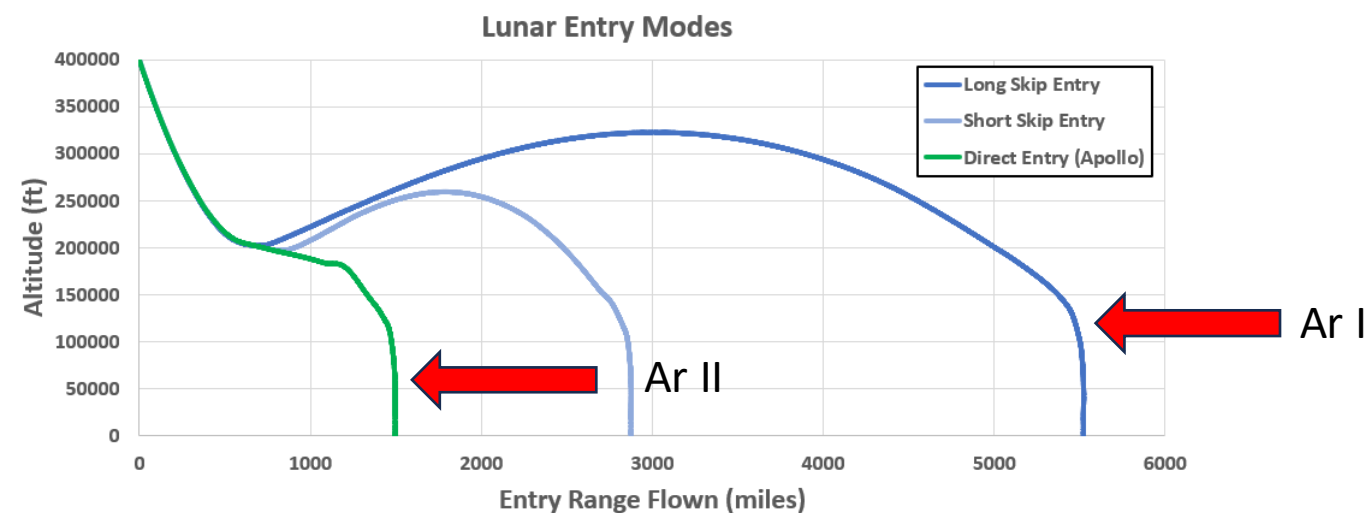
- Percolation Theory



ECLSS System + Heat Shield Issue



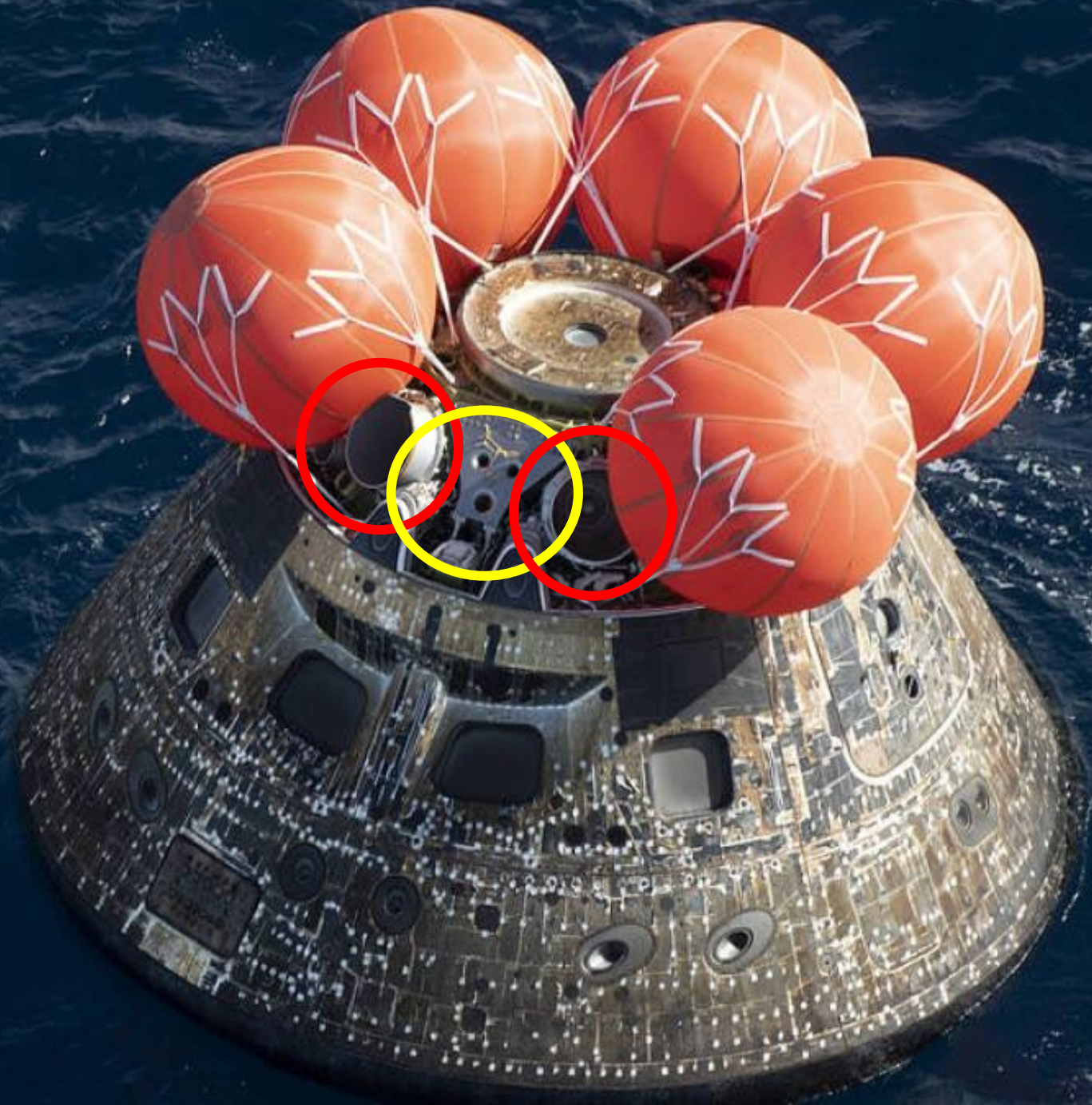
ECLSS System + Heat Shield Issue



Reduced splashdown range; higher uncertainty in Ar II Orion splashdown





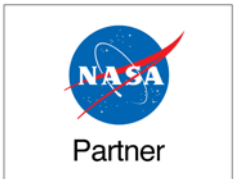


More Artemis Challenges

- Landing Large Masses on the Moon
- Dust Mitigation and Surface Operations
- Human-Robot Interaction in Lunar Operations
- Habitat Scalability and Power Distribution
- Medical Autonomy
- The list goes on ... and each is directly related to basic concepts we teach.

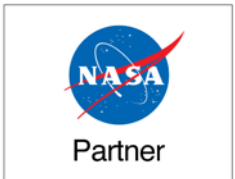


Harrison Schmitt retrieves lunar samples during Apollo 17. Lunar dust impacts suits, mechanisms, lungs, and solar panels.



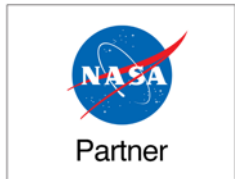
Summary

- Progress in exploration is often gated by *the pace of science*, not just policy and funding
- The pace of discovery depends on people who think from first principles.
- How we teach shapes the questions students learn to ask and the problems they believe are solvable.



Gratitude

- As with all science, the students do the actual work.
- Over 50 Carthage College, Purdue, and Georgia Tech students have contributed to the projects described here.
- NASA has supported this work through SMD MOSAICS, STMD Flight Opportunities Program, and the TechLeap Program.
- Thank you to collaborators at Johnson Space Center, Kennedy Space Center, and Georgia Tech.



Thank you!

