



**CONFIDENTIAL & PROPRIETARY** Distribution Limited to Authorized HPEPL/DCSL/QDE **Personnel Only** Quantum Dynamic Enterprises Inc. **Centripetal Impulse Drive Prototype** Final Report – June 2019 David R. Jovel, Arega Margousian, Adrian Vicente Prof. Mitchell L. R. Walker **High-Power Electric Propulsion** Laboratory June 9th, 2019 **CONFIDENTIAL & PROPRIETARY** 

Distribution Limited to Authorized HPEPL/DCSL/QDE Personnel Only The technical data in this document (or file) is controlled for export under the International Traffic in Arms Regulations (ITAR).

22 CFR 120-130. Violations of these laws may be subject to fines and penalties under the Arms Export Control Act, 22 U.S.C. 2778.





**Goal:** Characterize the thrust performance of the Centripetal Impulse Drive (CID) thruster at a specific design condition

# **Objectives:**

- 1. Measure the thrust of the CID over the range of operating conditions in the test matrix
- 2. Capture high-resolution video of the thruster in operation

### **Deliverables:**

- 1. Documentation of experimental setup and operating conditions
- 2. Documentation of the experiment's results including the following performance metrics:
  - a) Thrust
  - **b)** Thrust-to-Power ratio
  - c) Electrical Input Power

### 6/10/2019





# Phase II Pre-Check conducted on Thursday, May 30th, 2019 with project Sponsor at the Dynamics and Controls Systems Laboratory (DCSL)

- 1. Final CID thruster configuration and operating conditions were approved by QDE personnel prior to performance testing on May 30th, 2019
- 2. A few adjustments needed to provide longer battery life for testing and capture accurate RPM data

# Final test campaign conducted on Friday, May 31st, 2019

- 1. Project sponsor was invited to witness the execution of the test operations
- 2. A total of 12, three-minute trials were performed in which input power and translational motion of CID was captured
- 3. Video of each trial was also captured by the project sponsor

### 6/10/2019





# Phase II CID Configuration and Test Plan

• Friday, 05/31/19

6/10/2019

#### Georgia Tech **ConOps of CID Test Campaign**





#### Georgia Generalized Sequence of Events



- <sup>1.</sup> Perform visual inspection and check of CID hardware.
- 2. Setup water table in DCSL ASTROS platform.
- 3. CID is equipped with VICON motion sensor markers (5 total).
- 4. VICON software program is calibrated and the geometric center of the CID is defined.
- 5. Current and voltage probes are calibrated using the Fluke wireless software.
- 6. Water table is filled with water and final visual inspection of CID and foam platform is performed.
- One test operator holds the CID in place, while the other test operator adjusts the RPM values to meet operational requirements
- 8. Once RPM value is set, test operator releases the CID
- Test trial begins for approximately 2 3 minutes: DCSL personnel begin motion sensor software to capture the translation of CID, HPEPL personnel capture input current and voltage into CID motor
- <sup>10.</sup> RPM data is captured periodically to verify constant value operations
- <sup>11.</sup> Once trial is over, CID is switched off and data logging is completed.
- <sup>12.</sup> Repeat sequences 7 and 11 over the course of several trials to obtain a performance profile over predetermined RPM values.



A total of 5 VICON markers were arbitrarily placed along three perpendicular planes defining the CID's body



- 1) The VICON motion sensor camera network identifies the 5 markers on the CID
- 2) Using the known dimensions of the CID, the geometric center of the CID is found
- 3) The geometric center is then referenced with respect to the center of gravity of the CID
- 4) Approximated as acting right in the center of the CID geometry

# 6/10/2019

٠



The geometric center is visually monitored as the CID moves during each test operation.

The VICON software package employs an intermediate step to transform the position data to the Center of Gravity of the CID.

#### Georgia Tech Phase II Test Matrix



P C	erformance haracteristic					(	Data Type				
			Time	Voltage	Current		Position		Attitude	RPM	
	Input Power		Х	Х	Х						
	Acceleration		Х				Х		Х		
Rotor Spin (RPM)			х							х	
Trial	Input Power	Ac	celeration	Rotor Spin (RPM)	Date	Du	ration (min)		Filename		
1	Х		Х	240	5/31/19		2:43	Wa	lker_Thrust_Test2019	90531T110137.mat	
2			Х	N/A	5/31/19		N/A	Wa	lker_Thrust_Test201	90531T111435.mat	
3	Х		Х	240	5/31/19		6:48	Wa	lker_Thrust_Test201	90531T113044.mat	
4	Х		Х	242	5/31/19		7:19	Wa	lker_Thrust_Test2019	90531T120225.mat	
5	Х		Х	238	5/31/19		3:23	Wa	lker_Thrust_Test201	90531T122746.mat	
6	Х		Х	240	5/31/19		4:05	Wa	lker_Thrust_Test2019	90531T123935.mat	
7	Х		Х	240	5/31/19		2:33	Wa	lker_Thrust_Test2019	90531T144046.mat	
8	Х		Х	235	5/31/19		3:06	Wa	lker_Thrust_Test201	90531T144814.mat	
9	Х		Х	238	5/31/19		4:33	Wa	lker_Thrust_Test2019	90531T150131.mat	
10	Х		Х	238	5/31/19		2:45	Wa	lker_Thrust_Test2019	90531T151059.mat	
11				N/A	5/31/19		N/A	Wa	lker_Thrust_Test201	90531T152925.mat	
12	Х		Х	240	5/31/19		3:56	Wa	lker_Thrust_Test2019	90531T155103.mat	
	0, 20, 2020	-		Institute of Technol	ogy					9	





# Data Analysis





# Power Input

- 1. Product of the input voltage and current:  $P_{input} = V \times I$
- 2. Wireless voltage and current probes used to measure the CID rotor input power

# Acceleration/Thrust

- 1. VICON dataset consists of an 11 x N array with X-Y-Z position data with respect to the laboratory reference frame
- 2. 100 data samples for CID position are collected per second (sampling frequency = 100 Hz)

	4795	4796	4797	4798	4799	4800	4801
ne (sec) 🚺	47.7751	47.7850	47.7951	47.8044	47.8147	47.8245	47.8348
2	0.6315	0.6315	0.6316	0.6316	0.6315	0.6313	0.6313
3	0.0063	0.0066	0.0065	0.0066	0.0065	0.0065	0.0066
4	0.0142	0.0142	0.0143	0.0143	0.0142	0.0141	0.0142
5	-0.7752	-0.7752	-0.7751	-0.7752	-0.7753	-0.7754	-0.7754
6	-0.0140	-0.0137	-0.0140	-0.0139	-0.0137	-0.0136	-0.0137
7	0.0277	0.0281	0.0282	0.0283	0.0280	0.0279	0.0281
8	-1.7747	-1.7747	-1.7744	-1.7744	-1.7747	-1.7752	-1.7752
inate (m) 🛛 🧧 9	0.6281	0.6282	0.6282	0.6282	0.6282	0.6283	0.6283
nate (m) 🕴 10	0.8666	0.8665	0.8665	0.8664	0.8665	0.8666	0.8666
dinate (m) 11	-0.3259	-0.3260	-0.3258	-0.3259	-0.3259	-0.3259	-0.3259

6/10/2019



3. Thrust is calculated by first determining the acceleration in each X-Y-Z direction, which is approximated by numerically differentiating the position coordinate over the sampling time interval of 0.01 seconds. So for the acceleration in the x-direction we have:

$$a_x = \frac{dV_x}{dt} \approx \frac{\Delta V_x}{\Delta t} = \frac{\left(V_x(t_{i+1}) - V_x(t_i)\right)}{t_{i+1} - t_i}, \quad where \quad V_x \approx \frac{\Delta x}{\Delta t} = \frac{\left(x(t_{i+1}) - x(t_i)\right)}{t_{i+1} - t_i}$$

**Data Analysis Summary** 

4. Using this acceleration data, we can compute Thrust in each dimension via the following equation, which, through the VICON system, is given in the *Inertial Frame of Reference (IF)* 

$$\text{Thrust}_{Nx} = F_{Nx} = m * a_x$$

Thrust<sub>Ny</sub> =  $F_{Ny} = m * a_y$ 

$$\text{Thrust}_{NZ} = F_{NZ} = m * a_Z \approx 0$$

6/10/2019

Georgia Tech



5. The CID rotated about its Center of Mass throughout each of the trials. From an external frame of reference, this would cause the thrust direction to appear to be rotating as well. To compensate for this rotation, we determine the thrust vector produced by the CID in the *Body-Fixed Frame of Reference*, rather than the *Inertial Frame* it was measured from.



Here we consider a scenario where the CID, the *Body-Fixed Frame*, has rotated 30o in the VICON *Inertial Frame (IF)* during operation.

As a result, when viewed from the *Inertial Frame*, the thrust appears to be applying on the CID at a different angle that it actually is. We must correct this affect by representing thrust in the Body-Fixed Frame.

6/10/2019



In mathematics, the quaternions are a number system frequently used to define orientation and mechanics in 3-dimensional space. We can convert from one frame of reference to another by means of quaternion multiplication with the Unit Quaternion Rotation Matrix, Q(**q**)

The Unit Quaternion Rotation Matrix, and its conjugate, are composed of vector components *q0* through *q3*, which each relate to rotation about an axis. These components are all tracked by the VICON system to allow the user to transform from the Inertial Frame to the Body-Fixed Frame.

$$Q(\mathbf{q}) = \begin{bmatrix} q_0 & -q_1 & -q_2 & -q_3 \\ q_1 & q_0 & q_3 & -q_2 \\ q_2 & -q_3 & q_0 & q_1 \\ q_3 & q_2 & -q_1 & q_0 \end{bmatrix} \qquad \bar{Q}(\mathbf{q}) = \begin{bmatrix} q_0 & -q_1 & -q_2 & -q_3 \\ q_1 & q_0 & -q_3 & q_2 \\ q_2 & q_3 & q_0 & -q_1 \\ q_3 & -q_2 & q_1 & q_0 \end{bmatrix}$$

#### 6/10/2019



• Having composed our Unit Quaternion Rotation Matrix and its conjugate, we multiply them by the Thrust values in the *Inertial Frame of Reference* to compute them for the *Body-Fixed Frame of Reference*:

**Data Analysis Summary** 

$$\begin{bmatrix} 0 \\ Thrust_B \end{bmatrix} = \bar{Q}(q)^T Q(q) \begin{bmatrix} 0 \\ Thrust_N \end{bmatrix}$$

6. The magnitude of the thrust is calculated as defined below. Note that because the CID motion is predominantly two-dimensional along the XY-plane, the acceleration (and therefore thrust) in the z-axis is approximately zero and the thrust presented in this report is approximated as:

$$|\overline{T_B}| = \sqrt{T_{B_x}^2 + T_{B_y}^2 + T_{B_z}^2} \qquad |\overline{T_B}| \approx \sqrt{T_{B_x}^2 + T_{B_y}^2}$$

6/10/2019

Georgia Tech



7. The numerically differentiated data was noisy and two filtering techniques were employed to smoothen out the data. Below is a snapshot of the X- and Y-acceleration values for a specific test run.



6/10/2019



# Data Filtering Techniques: Two methods were employed to filter and smooth the acceleration data

- Savitsky-Golay Filtering
  - Applies a finite impulse response (FIR) smoothing filter of defined polynomial order and frame length to the displacement data.
  - <sup>b)</sup> Results provided in the following slides.

#### Moving Average Filter

- a) Consistent results for all test runs,
- <sup>b)</sup> Outputs a lower mean value for the acceleration on all test runs compared to the Savitsky-Golay filter,
- Governed by the following equation:

$$y_s(i) = \frac{1}{2N+1}(y(i+N) + y(i+N-1) + \dots + y(i-N))$$

#### 6/10/2019



### Acceleration Vector field for each run

- 1. Once the X- and Y-coordinate acceleration values were calculated, it is possible to generate a vector field of the thrust vector at each CID position
- 2. Results generally indicate that the thrust vector is constantly changing in a swirl pattern



4/30/20



# Acceleration Vector field for each run

1. The thrust vector is the direction in which the CID's center of gravity accelerates linearly at a constant value around 240 RPM's







# **Phase II Test Results**

• Friday, 05/31/19

6/10/2019



Nominal voltage value of **5.19 VDC** for constant 240 RPMs

Voltage input decreases over the course of each run.

Batteries were replaced regularly between trials to avoid meeting the minimum threshold of 8.5 VDC.

Voltage data was not captured for trial 2 and 11 due to signal loss



# Georgia Current Input Summary



Nominal current value of **3.50 A** for constant 240 RPMs

Current input was about constant for all test runs, with the exception of Trial 12; the CID's pistons taped down, resulting in a current output of 2.96 A.

Overall CID demonstrated an average input current of 3.50 A every test run.



4/30/20

•

#### Georgia Tech Power Input Summary



Input power was calculated for each test run as:

Pin = V x I

The average input power was **18.15 Watts** under the normal conditions of 240 RPMs

As expected, power decreased as the voltage input decreased with each test run.



4/30/20





#### 4/30/20



# Trial 1 5/31/2019 10:57

•

# XY position as a function of time



4/30/20

#### Georgia Tech Force Summary

![](_page_25_Picture_1.jpeg)

# Trial 1 5/31/2019 10:57

- 1. Force value obtained: Numerical Differentiation.
- 2. Plot Filtering: Moving Mean & Savitsky-Golay.
- 3. Average Magnitude of Force = 3.746 mN

![](_page_25_Figure_6.jpeg)

#### Georgia Tech Force Summary

![](_page_26_Picture_1.jpeg)

# Trial 1 5/31/2019 10:57

٠

1. Force in the X and Y directions.

![](_page_26_Figure_4.jpeg)

4/30/20

![](_page_27_Picture_1.jpeg)

![](_page_27_Figure_2.jpeg)

#### 4/30/20

![](_page_28_Picture_1.jpeg)

![](_page_28_Figure_2.jpeg)

#### 4/30/20

![](_page_29_Picture_1.jpeg)

# Trial 3 5/31/2019 11:24

•

#### XY Position as a function of time

![](_page_29_Figure_4.jpeg)

4/30/20

#### Georgia Tech Force Summary

![](_page_30_Picture_1.jpeg)

# Trial 3 5/31/2019 11:24

- 1. Force value obtained: Numerical Differentiation
- 2. Plot Filtering: Moving Mean & Savitsky-Golay
- 3. Average Magnitude of Force = 4.503 mN

![](_page_30_Figure_6.jpeg)

4/30/20

#### Georgia Tech Force Summary

![](_page_31_Picture_1.jpeg)

# Trial 3 5/31/2019 11:24

1. Force in the X and Y directions.

![](_page_31_Figure_4.jpeg)

4/30/20

![](_page_32_Picture_1.jpeg)

![](_page_32_Figure_2.jpeg)

#### 4/30/20

![](_page_33_Picture_1.jpeg)

# Trial 4 5/31/2019 11:53

٠

![](_page_33_Figure_3.jpeg)

4/30/20

#### Georgia Tech Force Summary

![](_page_34_Picture_1.jpeg)

# Trial 4 5/31/2019 11:53

- 1. Force value obtained: Numerical Differentiation
- 2. Plot Filtering: Moving Mean & Savitsky-Golay Filtering
- 3. Average Magnitude of Force = 4.229 mN

![](_page_34_Figure_6.jpeg)

4/30/20

#### Georgia Tech Force Summary

![](_page_35_Picture_1.jpeg)

# Trial 4 5/31/2019 11:53

1. Force in the X and Y directions.

![](_page_35_Figure_4.jpeg)

4/30/20




#### 4/30/20



### Trial 5 5/31/2019 12:22

•



4/30/20



## Trial 5 5/31/2019 12:22

- 1. Force value obtained: Numerical Differentiation
- 2. Plot Filtering: Moving Mean & Savitsky-Golay Filtering
- 3. Average Magnitude of Force = 4.027 mN



4/30/20



## Trial 5 5/31/2019 12:22

٠

1. Force in the X and Y directions.



4/30/20



## **Trial 6** demonstrated net translational + rotational motion.



Pool Y CID X

Average Thrust	3.412 mN
Average magnitude of Force	4:05
Target Speed	240 RPM
Speed Range	240-234 RPM

Edge of CID rotated in a way that caused it to minority graze the side of the pool within the first 30 seconds of testing.

4/30/20



### Trial 6 5/31/2019 12:34

•



4/30/20



## Trial 6 5/31/2019 12:34

٠

- 1. Force value obtained: Numerical Differentiation
- 2. Plot Filtering: Moving Mean & Savitsky-Golay Filtering







## Trial 6 5/31/2019 12:34

1. Force in the X and Y directions.



4/30/20







### Trial 7 5/31/2019 14:34

•



4/30/20



## Trial 7 5/31/2019 14:34

- 1. Force value obtained: Numerical Differentiation.
- 2. Plot Filtering: Moving Mean & Savitsky-Golay Filtering.
- 3. Average Magnitude of Force = 4.498 mN.





## Trial 7 5/31/2019 14:34

1. Force in the X and Y directions.



4/30/20





4/30/20



### Trial 8 5/31/2019 14:44

٠



4/30/20



## Trial 8 5/31/2019 14:44

- 1. Thrust value obtained: Numerical Differentiation
- 2. Plot Filtering: Moving Mean & Savitsky-Golay Filtering
- 3. Average Magnitude of Force = 3.853 mN



4/30/20



## Trial 8 5/31/2019 14:44

1. Force in the X and Y directions.



4/30/20





#### 4/30/20



#### Trial 9 5/31/2019 14:55

•



4/30/20



## Trial 9 5/31/2019 14:55

- 1. Force value obtained: Numerical Differentiation
- 2. Plot Filtering: Moving Mean & Savitsky-Golay Filtering
- 3. Average Magnitude of Force = 4.292 mN



4/30/20



## Trial 9 5/31/2019 14:55

1. Force in the X and Y directions.



4/30/20





4/30/20



## Trial 10 5/31/2019 15:07

•



4/30/20



## Trial 10 5/31/2019 15:07

- 1. Force value obtained: Numerical Differentiation.
- 2. Plot Filtering: Moving Mean & Savitsky-Golay Filtering.
- 3. Average Magnitude of Force = 4.685 mN.



4/30/20



## Trial 10 5/31/2019 15:07

1. Force in the X and Y directions.



4/30/20





#### 4/30/20







#### Pistons were all taped down,

this will be the "control". Started on near the wall of the pool. Took significantly less Current to operate. Produced significantly less average thrust.

4/30/20



## Trial 12 5/31/2019 15:45

•



4/30/20



## Trial 12 5/31/2019 15:45

- 1. Force value obtained: Numerical Differentiation.
- 2. Plot Filtering: Moving Mean & Savitsky-Golay Filtering.
- 3. Average Magnitude of Force = 0.295 mN.



4/30/20



## Trial 12 5/31/2019 15:45

1. Force in the X and Y directions.



4/30/20





## Phase II Conclusion

6/10/2019



Average **input power is 18.15 W** for about 3 minutes of continuous operation. Average force magnitude is 4.14 millinewtons (0.00414 N), but **nominal thrust vector could not be ascertained.** 

No net linear thrust can be discerned due to the constantly changing direction and magnitude of the thrust vector.

The CID's repeated pattern of curved movement indicates a thrust bias in a certain direction, but the magnitude and direction of this cannot be concluded from this test. Constraining movement in some way may reveal more information in this regard.

## General rotational motion of CID in water table indicates there may be a coupling between linear acceleration and the observed spin rates.

Trial 12, having produced a mere average thrust magnitude of 0.295 mN, points to the importance of the magnet "pistons" in regards to thrust production.

Test runs are limited to <10 mins by the CID battery charge capacity.

Wireless input power is in agreement with previously measured data.

Inner four magnet "piston" were susceptible to disassembly after each run due to vibrations, as was the case in Trial 11. Suggested hardware modifications are:

Change cross-head screw to hex

Custom tooling to tighten screws holding pistons in slot

Increase clearance for manual screw adjustment

4/30/20

# Georgia Results and Observations



- Linear acceleration coupling concept
- 1. A component of the measured acceleration goes to generating motion about the CID's bodyfixed axes
- 2. Ex: If the acceleration vector acts at an offset distance from the CID's center of gravity, there the acceleration will generate a moment about the body-fixed z-axis



#### 4/30/20





## Concluding Slide / Questions?

## David R. Jovel, Arega Margousian, Adrian Vicente Prof. Mitchell L. R. Walker High-Power Electric Propulsion Laboratory

The technical data in this document (or file) is controlled for export under the International Traffic in Arms Regulations (ITAR), 22 CFR 120-130. Violations of these laws may be subject to fines and penalties under the Arms Export Control Act, 22 U.S.C. 2778.

May contain Georgia Tech proprietary information; Not for public dissemination.





## Back Up



# Georgia Water Table Planar Motion



### Objective: Visualize planar motion of CID on water table



of Technology

Travel	Distance
Х	0.80 m
Y	0.80 m
Z	0.01 m

Assuming a 66 gal. filled pool:

Total Water Mass =  $\left(1000 \frac{kg}{m^3}\right)(0.2498 m^3)$ = 249.8 kg

Weight of Pool + CID =  $(249.8 kg + 25.1 kg)(9.81 \frac{m}{s^2})$ 

Total Loading = 2.696.77 N (606.26 lbf)



71

#### Georgia Tech Test Equipment

#### Current Probe: Fluke a3002 FC Wireless AC/DC Current Meter

- Utilizes Fluke i410/i1010 AC/DC Current Clamp
- Current range: 1 400 A DC
- Accuracy: 1%
- Total mass: 1 kg (Clamp + Meter)
- Transmits data to computer
- Voltage Probe: Fluke V3001 FC Wireless DC Voltage Meter
  - Voltage range DC: 100mVDC 1000 VDC
  - Max DC Volts: 1000 V
  - Accuracy: 0.15%
  - Total mass: 0.3 kg
  - Transmits data to computer
- Wireless Transmitter & Software: Fluke Connect sw3000FC
  - Fluke software application and laptop USB adapter to collect V/I data via wireless Bluetooth connection
  - Mobile App also available

#### **Manual Switch**

- Type P/N: Eaton E10E120AS
- Max Current = 6 A
- Purpose: Safety switch / mechanical switch to enable operations









4/30/20




Speed Controller: 65E60-12 – Dart Controls battery control

- Operating range: 12 48 VDC
- Current range: 0 20 ADC
- Speed adjustment: 5K ohm potentiometer
- Accuracy: ±1% of base speed
- Total mass: 1.27 kg
- Controls speed of CID motor
- Tachometer
  - Reading range : 2 99,999 RPM
  - RPM Accuracy: ± 0.05%
  - Non-contact laser readings using reflective tape
  - Reads the speed of the CID motor



High-Power Electric Propulsion Laboratory | Georgia Institute of Technology







4/30/20

High-Power Electric Propulsion Laboratory | Georgia Institute of Technology