



SECTION VI OPERATIONAL DATA

The operational data charts on the following pages are presented for two purposes; first, so that you may know what to expect from your airplane under various conditions; and second, to enable you to plan your flights in detail and with reasonable accuracy.

A power setting selected from the range charts usually will be more efficient than a random setting, since it will permit accurate fuel flow settings and your fuel consumption can be estimated closely. You will find that using the charts and your Cessna Model 310 Power Computer will pay dividends in over-all efficiency.

The data in the charts has been compiled from actual flight tests with the airplane and engines in good condition, and using average piloting techniques. Note also that the range charts make no allowances for wind, navigational errors, warm-up, takeoff, climb, etc. You must estimate these variables for yourself and make allowances accordingly.

AIRSPED CORRECTION TABLE					
Flaps 0°		Flaps 15°*		Flaps 35°**	
IAS, MPH	CAS, MPH	IAS, MPH	CAS, MPH	IAS, MPH	CAS, MPH
80	78	80	78	70	68
100	98	90	88	80	78
120	118	100	98	90	88
140	138	110	108	100	98
160	158	120	118	110	107
180	177	130	128	120	117
200	197	140	138	130	127
220	216	150	148	140	137
240	235	160	158	150	146
		170	168	160	156
		180	177		

* Maximum Flap Speed 180 MPH (15°) ** Maximum Flap Speed 160 MPH (35°)

Figure 6-1

STALL SPEED CHART								
MPH								
5200 POUNDS GROSS WEIGHT, IDLE POWER								
CONFIGURATION	ANGLE OF BANK							
	0°		20°		40°		60°	
	IAS	CAS	IAS	CAS	IAS	CAS	IAS	CAS
Gear and Flaps Up	84	82	87	85	96	94	117	116
Gear Down and Flaps 15°	79	77	82	80	90	88	111	109
Gear Down and Flaps 35°	75	73	77	75	85	83	105	103

Figure 6-2

NORMAL TAKEOFF DISTANCE

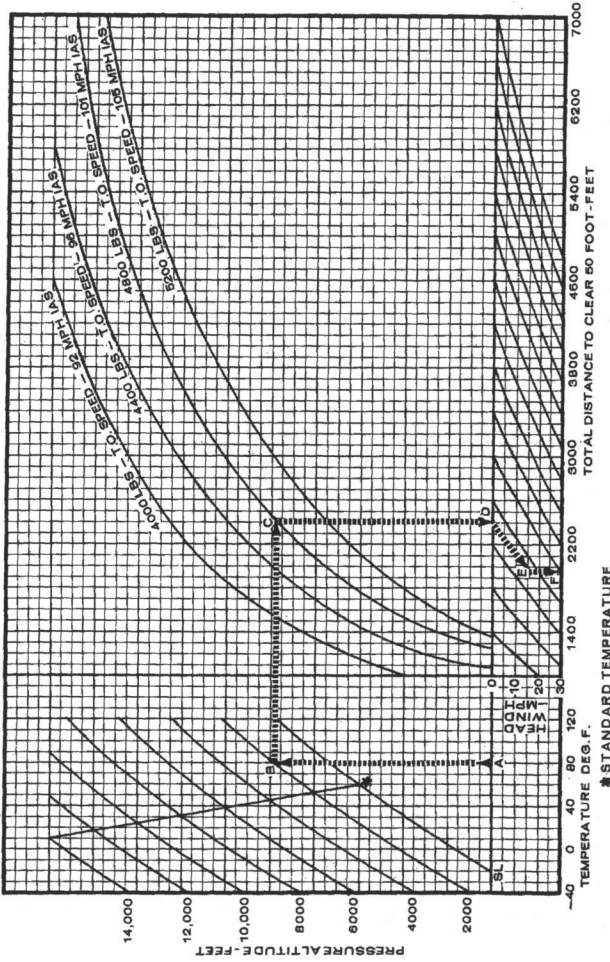


Figure 6-3

CONDITIONS:

1. Level Hard Surface Runway.
2. Wing Flaps UP.
3. Full Throttle and 2825 RPM Before Releasing Brakes.
4. Mixture at Recommended Fuel Flow.
5. Maintain Speed to 50 Feet.

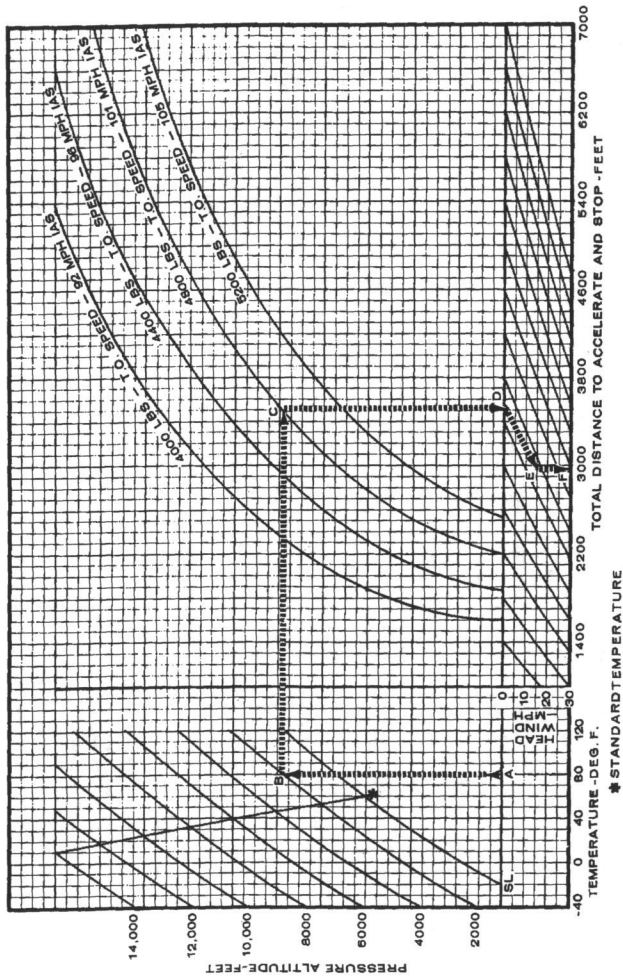
NOTE: Ground Run Is 80% of Total Distance.

NOTE: Increase total distance by 3.5% for operation on firm dry sod runways.

EXAMPLE:

- A. Temperature - 80°F.
- B. Pressure Altitude - 2000 Ft.
- C. Gross Weight - 4800 Lbs.
- D. Total Distance to Clear 50 Ft. (No Wind) - 2410 Ft.
- E. Headwind - 15 MPH.
- F. Total Distance to Clear 50 Ft. (15 MPH Headwind) - 1960 Ft.

ACCELERATE STOP DISTANCE



CONDITIONS:

1. Level Hard Surface Runway.
2. Wing Flaps UP.
3. Full Throttle and 2625 RPM Before Releasing Brakes.
4. Mixture at Recommended Fuel Flow.
5. Engine Failure at Takeoff Speed.
6. Heavy Braking After Engine Failure.

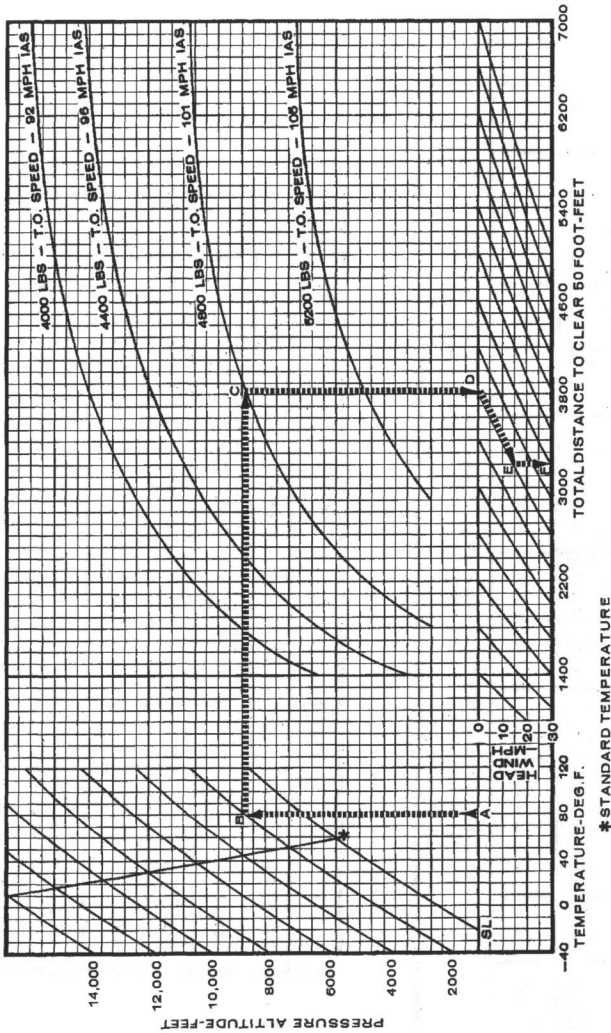
EXAMPLE:

- A. Temperature - 80° F.
- B. Pressure Altitude - 2000 Ft.
- C. Gross Weight - 4800 Lbs.
- D. Total Distance to Stop (No Wind) - 3525 Ft.
- E. Headwind - 15 MPH.
- F. Total Distance to Stop (15 MPH Headwind) - 2960 Ft.

NOTE: Accelerating Distance is Approximately 59% of the Total Distance.

Figure 6-4

SINGLE ENGINE TAKEOFF DISTANCE



CONDITIONS:

1. Level Hard Surface Runway.
2. Wing Flaps UP.
3. Full Throttle and 2625 RPM Before Releasing Brakes.
4. Mixture at Recommended Fuel Flow.
5. Engine Failure at Takeoff Speed.
6. Propeller Feathered and Gear Retracted During Climb.
7. Maintain Speed to 50 Feet.

EXAMPLE:

- A. Temperature - 80° F.
- B. Pressure Altitude - 2000 Ft.
- C. Gross Weight - 4800 Lbs.
- D. Total Distance to Clear 50 Ft. (No Wind) - 3830 Ft.
- E. Headwind - 15 MPH.
- F. Total Distance to Clear 50 Ft. (15 MPH Headwind) - 3200 Ft.

Figure 6-5

TWIN ENGINE CLIMB DATA AT 5200 POUNDS

MAXIMUM PERFORMANCE

SEA LEVEL 59°F		5000 FT. 41°F			10000 FT. 23°F			15000 FT. 5°F			20000 FT. -12°F			
Best Climb IAS MPH	Rate of Climb Ft/Min	Gal of Fuel Used	Best Climb IAS MPH	Rate of Climb Ft/Min	From S.L. Fuel Used	Best Climb IAS MPH	Rate of Climb Ft/Min	From S.L. Fuel Used	Best Climb IAS MPH	Rate of Climb Ft/Min	From S.L. Fuel Used			
125	1542	4	123	1178	6.3	122	818	9.0	121	458	12.7	120	93	20.5

NOTE: FULL THROTTLE, 2625 RPM, MIXTURE AT RECOMMENDED FUEL FLOW, FLAPS AND GEAR UP. FUEL USED INCLUDES WARM-UP AND TAKEOFF ALLOWANCE

POWER SETTING		CRUISE CLIMB											
		2450 RPM, 24" MP to 5000 FT. FULL THROTTLE AFTERWARDS											
RPM	M.P.	5000 FT. 41°F			10000 FT. 23°F			15000 FT. 5°F			FROM SEA LEVEL		
		Dist. Miles	Time Min.	Fuel Used Gal.	Dist. Miles	Time Min.	Fuel Used Gal.	Dist. Miles	Time Min.	Fuel Used Gal.	Dist. Miles	Time Min.	Fuel Used Gal.
2450	24	11.0	4.8	6.5	25.9	11.3	9.7	56.4	24.7	15.0			

NOTE: WARM-UP AND TAKEOFF ALLOWANCE 4 GALLONS AT SEA LEVEL. MIXTURE AT RECOMMENDED FUEL FLOW, FLAPS AND GEAR UP.

Figure 6-6

MAXIMUM PERFORMANCE TAKEOFF 15° FLAPS											
DENSITY ALTITUDE											
Gross Weight Pounds	IAS at Takeoff MPH	IAS at Obstacle MPH	Head Wind MPH	SEA LEVEL		2500 FT.		5000 FT.		7500 FT.	
				Ground Run	Total Distance over 50 Ft Obstacle	Ground Run	Total Distance over 50 Ft Obstacle	Ground Run	Total Distance over 50 Ft Obstacle	Ground Run	Total Distance over 50 Ft Obstacle
5200	93	93	0 15 30	1451 1120 824	1716 1345 1009	1808 1410 1050	2119 1676 1272	2340 1842 1392	2711 2162 1660	3061 2435 1864	3509 2823 2193

Figure 6-7

SINGLE ENGINE CLIMB DATA											
Gross Weight Pounds	SEA LEVEL 59°F		2500 FT 50°F		5000 FT 41°F		7500 FT 32°F		10000 FT 23°F		
	Best Climb IAS MPH	Rate of Climb Ft/Min	Best Climb IAS MPH	Rate of Climb Ft/Min	Best Climb IAS MPH	Rate of Climb Ft/Min	Best Climb IAS MPH	Rate of Climb Ft/Min	Best Climb IAS MPH	Rate of Climb Ft/Min	
											Best Climb IAS MPH
4400	114	534	113	420	111	303	110	188	108	72	
4800	117	433	115.5	325	114	215	112	106	111	-5	
5200	119	330	117	227	116	125	114	22	113	-80	

NOTE: Flaps and gear up, inoperative propeller-feathered, wing banked 5° toward operating engine, full throttle, 2625 RPM and mixture at recommended leaning schedule. Decrease rate of climb 10 FT/MIN for each 10°F above standard temperature for particular altitude.

Figure 6-8

INDICATED ENROUTE TERRAIN CLEARANCE ALTITUDES							
BEST CLIMB SPEED APPROXIMATELY 120 MPH IAS							
Gross Weight Pounds	OUTSIDE AIR TEMPERATURE °F						
	-10	0	10	20	30	40	50
	ALTITUDE-FEET						
5200	8600	8100	7500	6800	6300	5700	5250
4800	10200	9600	9100	8500	7900	7300	6800
4400	11500	10900	10300	9750	9200	8700	8100

NOTE: The terrain clearance altitude as defined by FAR 121.181 is the highest ground elevation that can be cleared by 1000 feet on single engine with rate-of-climb of 50 ft/min. Increase indicated enroute terrain clearance altitudes 100 feet for each 0.10 inch Hg. altimeter setting greater than 29.92 for new indicated altitudes. Decrease indicated enroute terrain clearance altitudes 100 feet for each 0.10 inch Hg. altimeter setting less than 29.92 for new indicated altitudes.

Figure 6-9

CRUISE PERFORMANCE WITH NORMAL LEAN MIXTURE AT 2,500 FT.

RPM	MP	% BHP	TAS	Total Gals./Hr.	Endurance 100 Gals.	Range 100 Gals.	Endurance 140 Gals.	Range 140 Gals.
2450	24	74	210	28.0	3.6	750	5.0	1050
	23	70	204	26.3	3.8	775	5.3	1085
	22	66	198	24.5	4.1	808	5.7	1131
	21	62	193	23.2	4.3	833	6.0	1164
2300	24	68	202	25.5	3.9	789	5.5	1109
	23	64	196	23.8	4.2	823	5.9	1152
	22	60	190	22.5	4.4	836	6.2	1182
	21	56	184	21.2	4.7	869	6.6	1214
2200	23	58	187	21.8	4.6	858	6.4	1201
	22	55	182	20.7	4.8	878	6.8	1230
	21	50	175	19.3	5.2	907	7.3	1269
	20	47	169	18.3	5.5	923	7.7	1293
2100	22	49	172	18.8	5.3	915	7.5	1281
	21	46	166	17.8	5.6	933	7.9	1306
	20	43	158	16.8	5.9	941	8.3	1316
	19	40	151	16.2	6.2	932	8.6	1306
	18	37	140	15.0	6.7	932	9.3	1306

CRUISE PERFORMANCE IS BASED ON STANDARD CONDITIONS, ZERO WIND, NORMAL LEAN MIXTURE, 100 AND 140 GALLONS OF FUEL (NO RESERVE), AND 5200 POUNDS GROSS WEIGHT.

CRUISE PERFORMANCE WITH NORMAL LEAN MIXTURE AT 5,000 FT.

RPM	MP	% BHP	TAS	Total Gals./Hr.	Endurance 100 Gals.	Range 100 Gals.	Endurance 140 Gals.	Range 140 Gals.
2450	24	77	218	29.0	3.4	752	4.8	1053
	23	73	212	27.3	3.7	777	5.1	1085
	22	68	208	25.7	3.9	809	5.5	1134
	21	64	202	24.2	4.1	834	5.8	1170
2300	24	70	208	26.2	3.8	797	5.3	1111
	23	66	204	24.7	4.0	825	5.7	1157
	22	62	197	23.2	4.3	851	6.0	1188
	21	58	192	21.8	4.6	877	6.4	1233
2200	23	61	196	22.7	4.4	866	6.2	1209
	22	57	189	21.3	4.7	886	6.6	1242
	21	53	184	20.2	5.0	914	6.9	1275
	20	50	177	19.2	5.2	924	7.3	1290
2100	22	51	180	19.7	5.1	914	7.1	1280
	21	48	173	18.5	5.4	934	7.6	1310
	20	45	167	17.7	5.7	943	7.9	1321
	19	42	157	16.5	6.1	951	8.5	1331
	18	39	148	15.7	6.4	943	8.9	1320

CRUISE PERFORMANCE IS BASED ON STANDARD CONDITIONS, ZERO WIND, NORMAL LEAN MIXTURE, 100 AND 140 GALLONS OF FUEL (NO RESERVE), AND 5200 POUNDS GROSS WEIGHT.

Figure 6-10 (Sheet 1 of 3)

CRUISE PERFORMANCE WITH NORMAL LEAN MIXTURE AT 7,500 FT.

RPM	MP	%BHP	TAS	Total Gals./Hr.	Endurance 100 Gals.	Range 100 Gals.	Endurance 140 Gals.	Range 140 Gals.
2450	22	71	214	26.5	3.8	807	5.3	1130
	21	67	209	25.0	4.0	837	5.6	1170
	20	63	203	23.5	4.3	865	6.0	1210
	19	58	196	22.0	4.5	891	6.4	1247
2300	22	64	204	24.0	4.2	852	5.8	1189
	21	60	199	22.7	4.4	878	6.2	1228
	20	56	192	21.2	4.7	907	6.6	1267
	19	53	186	20.2	4.9	921	6.9	1289
2200	22	58	196	22.0	4.5	891	6.4	1247
	21	55	191	21.0	4.8	910	6.7	1274
	20	52	185	19.8	5.1	934	7.1	1308
	19	48	177	18.7	5.4	947	7.5	1326
2100	21	50	181	19.2	5.2	944	7.3	1319
	20	47	175	18.3	5.5	957	7.7	1339
	19	44	166	16.8	5.9	988	8.3	1383
	18	40	155	16.2	6.2	958	8.6	1339
	17	37	132	15.3	6.5	862	9.1	1208

CRUISE PERFORMANCE IS BASED ON STANDARD CONDITIONS, ZERO WIND, NORMAL LEAN MIXTURE, 100 AND 140 GALLONS OF FUEL (NO RESERVE), AND 5200 POUNDS GROSS WEIGHT.

CRUISE PERFORMANCE WITH NORMAL LEAN MIXTURE AT 10,000 FT.

RPM	MP	% BHP	TAS	Total Gals./Hr.	Endurance 100 Gals.	Range 100 Gals.	Endurance 140 Gals.	Range 140 Gals.
2450	20	65	211	24.4	4.1	865	5.7	1211
	19	61	204	22.8	4.4	892	6.1	1253
	18	57	197	21.5	4.7	915	6.5	1282
	17	53	189	20.0	5.0	945	7.0	1323
2300	20	59	201	22.0	4.5	913	6.4	1278
	19	55	194	20.8	4.8	931	6.7	1306
	18	51	186	19.7	5.1	944	7.1	1322
	17	48	179	18.5	5.4	966	7.6	1355
2200	20	54	191	20.5	4.9	932	6.8	1305
	19	50	185	19.3	5.2	958	7.3	1341
	18	47	176	18.2	5.5	967	7.7	1353
	17	44	167	17.3	5.8	964	8.1	1351
2100	20	49	181	18.8	5.3	963	7.5	1348
	19	46	172	17.8	5.6	967	7.9	1354
	18	43	163	16.8	5.9	970	8.3	1358
	17	40	146	16.0	6.3	913	8.7	1278

CRUISE PERFORMANCE IS BASED ON STANDARD CONDITIONS, ZERO WIND, NORMAL LEAN MIXTURE, 100 AND 140 GALLONS OF FUEL (NO RESERVE), AND 5200 POUNDS GROSS WEIGHT.

Figure 6-10 (Sheet 2 of 3)

CRUISE PERFORMANCE WITH NORMAL LEAN MIXTURE AT 15,000 FT.								
RPM	MP	% BHP	TAS	Total Gals./Hr.	Endurance 100 Gals.	Range 100 Gals.	Endurance 140 Gals.	Range 140 Gals.
2450	16	53	196	20.2	5.0	972	6.9	1358
	15	48	183	18.7	5.4	979	7.5	1371
	14	44	168	17.3	5.8	972	8.1	1359
2300	16	48	183	18.7	5.4	979	7.5	1371
	15	44	165	17.3	5.8	952	8.1	1335
2200	16	44	168	17.3	5.8	972	8.1	1359

CRUISE PERFORMANCE IS BASED ON STANDARD CONDITIONS, ZERO WIND, NORMAL LEAN MIXTURE, 100 AND 140 GALLONS OF FUEL (NO RESERVE), AND 5200 POUNDS GROSS WEIGHT.

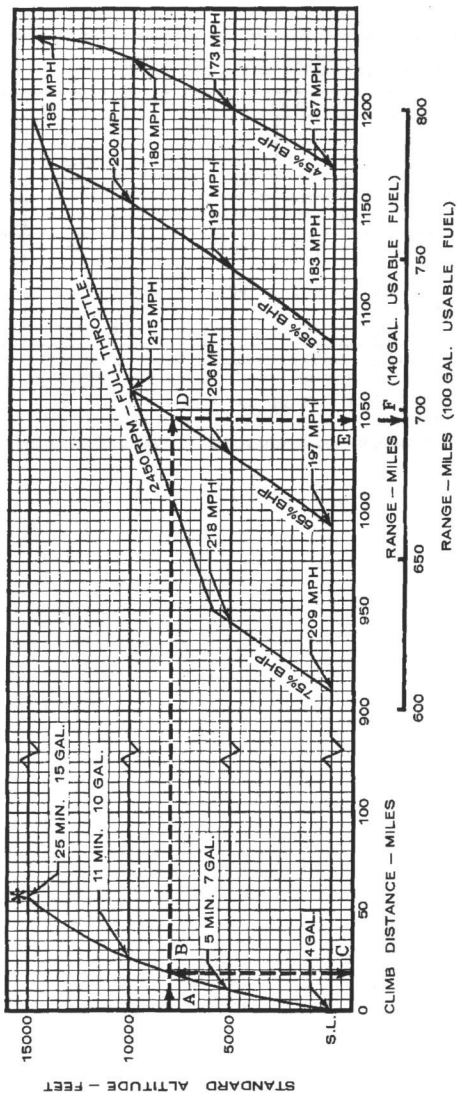
CRUISE PERFORMANCE WITH NORMAL LEAN MIXTURE AT 20,000 FT.								
RPM	MP	% BHP	TAS	Total Gals./Hr.	Endurance 100 Gals.	Range 100 Gals.	Endurance 140 Gals.	Range 140 Gals.
2450	13.5	46	171	18.0	5.6	949	7.8	1330

CRUISE PERFORMANCE IS BASED ON STANDARD CONDITIONS, ZERO WIND, NORMAL LEAN MIXTURE, 100 AND 140 GALLONS OF FUEL (NO RESERVE), AND 5200 POUNDS GROSS WEIGHT.

Figure 6-10 (Sheet 3 of 3)

RANGE PROFILE

*Cruise Climb at 2450 RPM, 24.0 In. Hg. MP (Full Throttle Above 5000 Ft.) and 140 MPH IAS



NOTES:

1. Maximum range is not changed appreciably with variations in climb power setting and climb speed.
2. Range includes distance to alternate destination.

CONDITIONS:

1. Starting Weight - 5200 Lbs.
2. Cruise Climb to Desired Cruise Altitude
3. Cruise Fuel Flow Normal Lean Mixture
4. Zero Wind
5. 45 Min. Reserve Fuel (13 Gals.) at 45% BHP

EXAMPLE:

- A. Cruising Altitude - 8000 Ft.
- B. Time and Fuel Used to Climb From Sea Level to 8000 Ft. - 8.5 Min. and 9 Gals.
- C. Climb Distance - 18 Miles
- D. Cruise Power and Speed - 65% and 211 MPH TAS
- E. Range - 1045 Mi. (140 Gals. Usable Fuel - Optional)
- F. Range - 697 Mi. (100 Gals. Usable Fuel - Standard)

Figure 6-11

LANDING PERFORMANCE									
Gross Weight Pounds	IAS at Obstacle MPH	SEA LEVEL 59° F		2500 FT. 50° F		5000 FT. 41° F		7500 FT. 32° F	
		Ground Run	Total Distance Over 50 Foot Obstacle	Ground Run	Total Distance Over 50 Foot Obstacle	Ground Run	Total Distance Over 50 Foot Obstacle	Ground Run	Total Distance Over 50 Foot Obstacle
4400	97	693	1276	748	1327	806	1385	872	1448
4800	101	840	1420	902	1483	973	1552	1052	1632
5200	105	1002	1582	1077	1631	1160	1743	1254	1842
NOTE: WING FLAPS 35°, POWER OFF, HARD SURFACE RUNWAY, ZERO WIND, MAXIMUM BRAKING EFFORT. REDUCE LANDING DISTANCE 10% FOR EACH 10 MPH HEADWIND.									
NOTE: INCREASE DISTANCE BY 25% OF THE GROUND RUN FOR OPERATION ON FIRM DRY SOD RUNWAYS.									

Figure 6-12