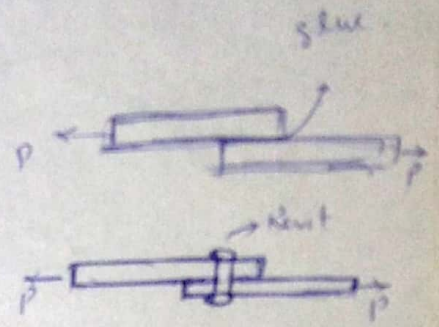


Shear stress:-

The stress produce when force acting parallel to cross sectional area.

Explanation:-

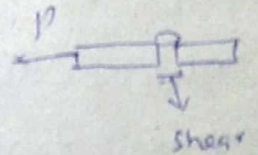
Consider the fig in which two forces are applied on opposite directions on two plates as shown.



Now let consider the two plates may be connected by pin or other glue like material in order to cause resistance to applied loads.

→ The above figure with rivet is called single shear as one surface provide resistance or crack will produce in one place.

(*) When we applied two force than the both parts will separate & rivet will break as show above parts.



Mathematically:-

Shear stress also called tangential stress.

$$\text{Shear Stress} = \tau_{ou} = \tau = \frac{V}{A}$$

Double shear:-

When two surface provide resistances as break at two places.

→ These shear (double) are mostly produce in steel and wood.



If more force applied than it will break and are shown as,



$$\text{As: } \tau = \frac{PF}{2A} = \frac{V}{A}$$

$$P = 2V$$

$$\tau = \frac{P}{2A}$$

$$V = \frac{P}{2}$$

Generally it can be written as;

$$\tau = \frac{P}{nA}$$

Bearing Stress:-

(*) Shear stress & bearing stress both arises at an object.

Consider the fig in which two surfaces are in contact through nut. Now if we applied two forces in opposite direction if the surface tears then we call it Bearing Stress.

(*) Here tearing occur at 45° .

Mathematically:-

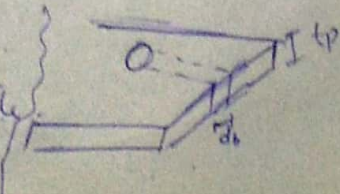
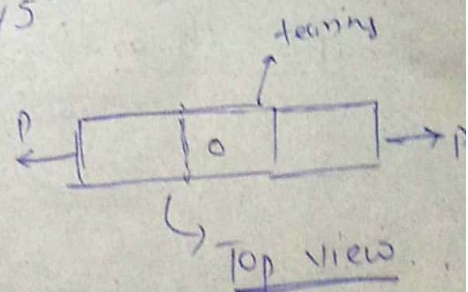
$$\sigma = \frac{P}{A}$$

$$A = d_b \times d_t$$

$$\sigma = \frac{P}{d_b \times d_t}$$

d_b = diameter of board.

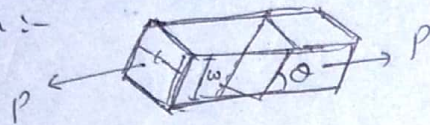
d_t = thickness of plate.



Pb-122

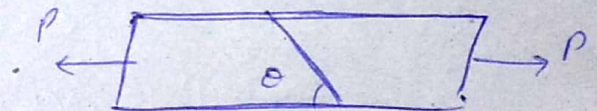
Two blocks of wood, width w and thickness t , are glued together along the joint inclined at angle θ as shown. Using free body diagram computed illustrated in fig. Show That shearing stress on glued joint is $\tau = \frac{2P \sin 2\theta}{2A}$, where A is cross sectional area.

Diagram:-



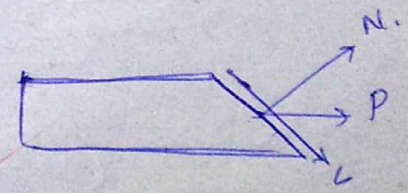
Solution:- Consider the two blocks which are joint through glue together along the joint which is inclined.

Consider ~~one~~ block after breaking.



Show the applied load, shear load & normal force.

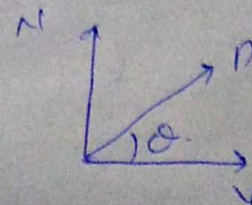
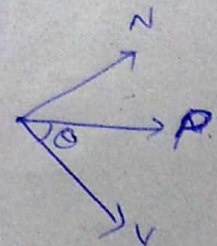
These forces can be shown as;



As we know;

$$\tau = \frac{V}{A'}$$

$\therefore A' =$ is inclined area.



from fig.

$$N = P \sin \theta$$

$$\delta = \frac{P \sin \theta}{A'} \rightarrow (2)$$

From similar triangle.

$$\frac{A'}{A} = \frac{1}{\sin \theta}$$

$$A' = \frac{A}{\sin \theta} \rightarrow (3)$$

$$\delta = \frac{P \sin \theta}{A / \sin \theta}$$

$$\boxed{\delta = \frac{P \sin^2 \theta}{A}} \rightarrow \text{Formula for normal stress.}$$

Only shear stress is required in direction?

Shear stress (required),

$$\tau = \frac{V}{A'} \quad \because A' = \text{Inclined area}$$

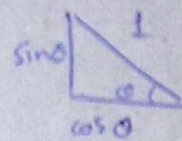
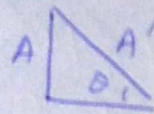
$$\tau = \frac{P \cos \theta}{A'}$$

$$= \frac{P \cos \theta}{A / \sin \theta}$$

$$= \frac{P}{A} \sin \theta \cos \theta$$

$$= \frac{P}{2A} 2 \sin \theta \cos \theta$$

$$\boxed{\tau = \frac{P \sin 2\theta}{2A}} \text{ Ans}$$



$$\because V = P \cos \theta$$

$$\because \text{Since } A' = \frac{A}{\sin \theta}$$

$$\because 2 \sin \theta \cos \theta = 2 \sin^2 \theta$$

Multiply & divide
"2"

Stress Concentration:-

We have mostly seen that crack produce at sharp edge of doors as shown in fig. But if we give some radius to its edges then no crack will produce. (filled)

It can be explain as when load of bricks upon door act downward. The stress flow similarly like water. So if there is some sharp edge in its path. It will disturb its flow and therefore crack will produce.

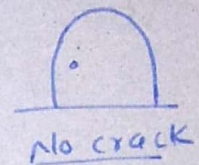
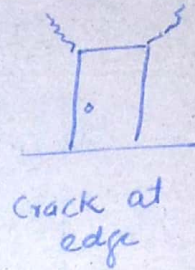
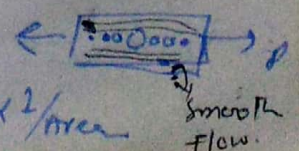
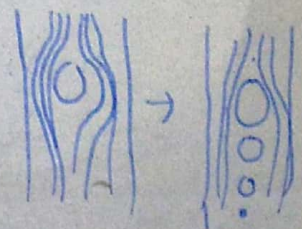
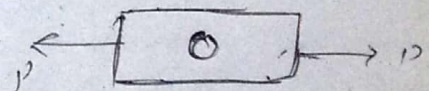
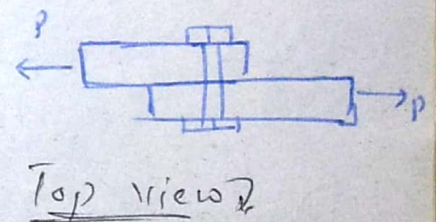


Diagram.

(2) → Consider two plates that are fixed through rivet. If load is applied it will break at hole side as at hole ^(area) is less so stress will maximum there and crack will produce there. In order to not happen we make small holes in lines with that hole so that to reduce ^{stress} as in water flow we provide more obstructs in order to smoothen flow of water as shown in fig.



stress $\propto 1/\text{area}$

* Stress does not like sharp edges.

→ In our daily life we can reduce stress concentration by avoiding sharp edges and giving fillet to its edges.

Stress Concentration factor:-

It is define as;

It is the ratio of maximum stress which is present at notch (groove or hole) to nominal stress (real stress) present at same section.

Formula:-

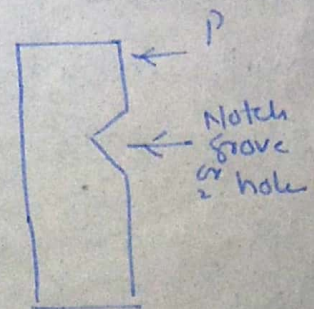
$$\text{Stress concentration factor} = K = \frac{\text{Maximum Stress}}{\text{Nominal stress.}}$$

Diagram:-

In order to understand stress concentration factor consider the diagram.

At notch section or portion the stress will maximum as area decrease since stress $\propto \frac{1}{A}$.
So the ratio of this maximum stress to average or nominal

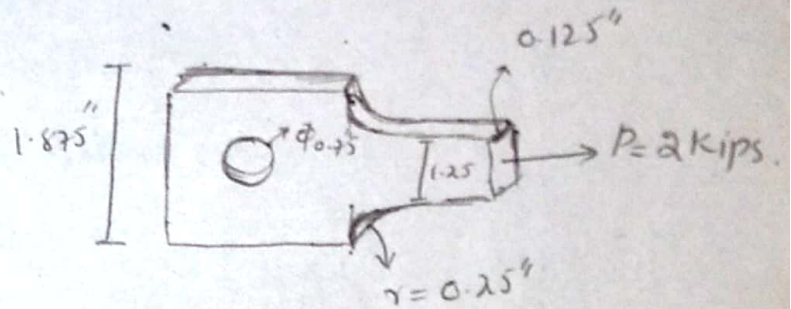
stress (which is produce due normal force at any section without groove)



Here K is called stress concentration factor whose value can be determined from the graph between w/h and r/h .

Example on Stress Concentration factor:-

Find The maximum normal stress?



Solution:-

Given data

Force applied = $P = 2$ Kips.

Fillet radius = $r = 0.25$ inches.

Height of small portion = $h = 1.25$ inches.

Height of larger portion = $H = 1.875$ inches.

Diameter of hole = $\phi = 0.75$ inches.

Width of small portion = $w = 0.125$ inches.

Required data:-

Maximum Stress = ?

At small portion:-

Since we know that;

$$K = \frac{\sigma_{\max}}{\sigma}$$

$$\sigma_{\max} = K \sigma \rightarrow \text{①}$$

$$\delta = \frac{F}{A} = \frac{P}{A}$$

$$\therefore P = 2 \text{ Kips}$$

$$A = l \times w$$

$$= \frac{2 \text{ Kips}}{l \times w} = \frac{2 \text{ Kips}}{(0.125'')(1.25'')}$$

$$\delta = 12.8 \text{ Ksi} \rightarrow \textcircled{2}$$

To find $k = ?$

Since, k value determine from graph of w/h and r/h . So we have first find w/h and r/h .

$$\frac{r}{h} = \frac{0.25''}{1.25''} = 0.20$$

$\therefore r =$ Filled radius

$h =$ height of small portion.

$$\frac{w}{h} = \frac{1.875''}{1.25''} = 1.5$$

$\therefore w =$ height of large portion

By locating such points on graph we get $k = 1.75$.

$$\text{So } \textcircled{1} \Rightarrow \sigma_{\max} = \delta \times k$$

$$= 12.8 \text{ Ksi} \times (1.75)$$

$$\boxed{\sigma_{\max} = 22.4 \text{ Ksi}}$$

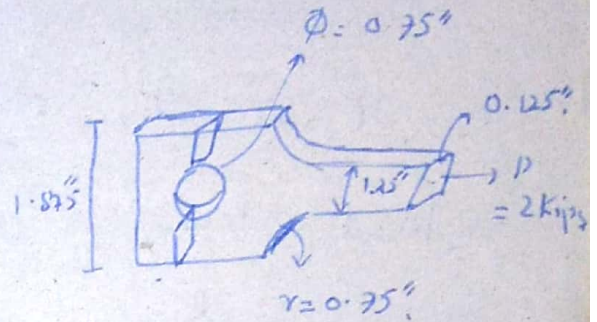
Now for Circle (hole):-
Now calculate the stress on hole portion.

Since;

$$\frac{\sigma_{max}}{\sigma} = K$$

$$\sigma_{max} = \sigma K \rightarrow (ii)$$

$$\text{As } \sigma = \frac{P}{A} = \frac{2}{(1.875 - 0.75)(1.25)}$$



$$\sigma = 14.22 \text{ Kips}$$

To find K first we have
to find w/h & r/h .

∴ For hole.

$$\frac{w}{h} = \frac{\phi}{h}$$

$$\frac{w}{h} = \frac{\phi}{h} = \frac{0.75}{1.875} = 0.4$$

$$\frac{r}{h} = \frac{0.75}{1.25} = 1.5$$

For these value $K = 2.2$.

∴ As stress is to
find in hole
i.e. why we subtract
the above & below
area to get
area of hole.

$$\begin{aligned} \text{Eq (i)} \Rightarrow \sigma_{max} &= \sigma K \\ &= 14.22 \text{ Kips} (2.2) \end{aligned}$$

$$\sigma_{max} = 31.3 \text{ Kips}$$

So the max. stress will be at hole section
which will be controlling factor.

Factor of Safety:-

factor of safety can be defined as;

"It is the ratio of maximum stress to the permissible or working stress"

OR

It is the ratio of failure stress to actual stress."

OR

It is the ratio of strength of material to maximum computed stress."

Formula:- (Generally)

$$\text{Factor of Safety} = \frac{\text{Actual failure stress}}{\text{Actual stress}}$$

For ductile Material:-

$$F.O.S = \frac{\text{Yield stress}}{\text{Working stress}}$$

For Brittle material:-

$$F.O.S = \frac{\text{Ultimate stress}}{\text{Working stress}}$$

(*) Factor of safety is just a number which should always be greater than 1.

• For Example:-

In case of building

$$F.O.S \geq 2.$$

In case of automobile

$$f.o.s \geq 3.5$$

In case of air / space craft:-

$$f.o.s \geq 1.2 - 2.5$$

In case of boiler:-

$$f.o.s \geq 8.5$$

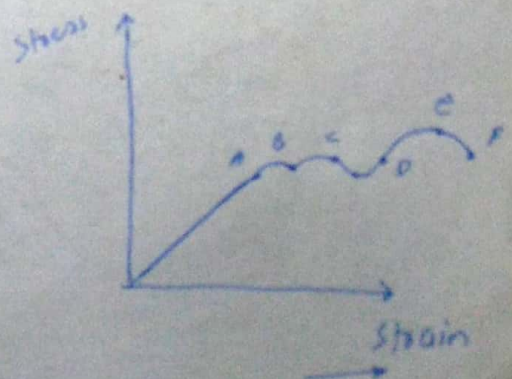
In case of bolts:-

$$f.o.s \geq 8.5$$

Importance of Factor of Safety:-

Factor of safety help us in designing of material.

Consider the stress - strain curve for ductile material. If the yield stress is 400 N/mm^2 & factor of safety is 4. So the allowable stress that material can carry will be; $\frac{400}{4} = 100 \text{ N/mm}^2$.



Since the maximum stress after that material break is 400 N/mm^2 but because of F.O.S we reduce the stress on material and now we can safely operate material, as stress now will not increase from 100 N/mm^2 .

Factors on which F.O.S depend:-

i. Material selection:-

We will see whether it is ductile or brittle.

ii. Type of loading:-

Whether the load is static, variable or shock.

iii. Cost:-

If we want to increase the F.O.S we have to make more dimension which will cost more.

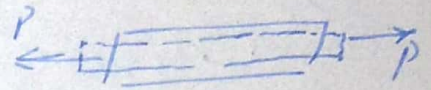
Poisson's Ratio:-

It is defined as;

"When the material is loaded within elastic limit, the ratio of lateral strain to longitudinal strain remain constant and is called Poisson's ratio."

Explanation:-

Consider the fig. In which load is applied on two parts of object. So here elongation is produced, at the same time the width of bar will also reduce.



Simeon D. Poisson showed in 1811, that the ratio of unit deformation or strain in these directions is constant for stresses within the proportional limit.

Denotation:-

It is denoted by μ i.e. ν .

Mathematically:-

$$\text{Poisson's ratio} = - \frac{\text{Lateral Strain}}{\text{Longitudinal Strain}}$$

We take negative sign to get positive value for Poisson ratio. As Poisson ratio is always positive except in case of honey comb where it is negative.

Poisson's ratio can be written as;

$$\nu = - \frac{\epsilon'}{\epsilon} \rightarrow \textcircled{1}$$

ϵ' = Lateral strain.

$$\epsilon' = \frac{\delta \text{ lateral}}{W_0}$$

ϵ = Longitudinal Strain.

$$\epsilon = \frac{\delta_{\text{long.}}}{L_0}$$

Unit:-

Poisson ratio has no unit.

Value range:-

Typical value of Poisson's ratio ranges from 0.1 to 0.5.

Some values of Poisson's ratio:-

For cork $\nu = 0.5$

For Rubber = 0.48-0.5

For Copper = 0.37.

For Aluminum = 0.35.

For Stainless Steel = 0.3.

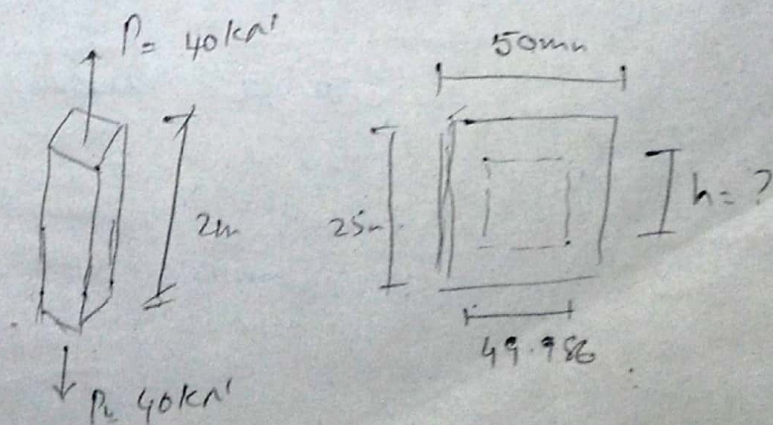
Importance of Poisson's Ratio:-

(*) Poisson's ratio is used for analysis of structural components of different members of building.

Prob on Poisson Ratio:-

Find the Poisson's ratio in the following fig.
When $P = 40 \text{ kN}$ applied at both sides. Its original longitudinal length is 2m and linear deformation is 2mm. The breadth & width is 50mm x 25mm. The deformation in breadth is 49.986mm. Find the width & height also.

So Diagram:-



The material is isotropic.

Q:- As Poisson's ratio is given as;

$$\nu' = \frac{\text{Lateral strain}}{\text{Longitudinal strain}} = \frac{\epsilon'}{\epsilon}$$

$$\nu = -\frac{\epsilon'}{\epsilon} \rightarrow \textcircled{1}$$

First we have to find lateral & longitudinal strain.

→ Lateral strain:-

$$\epsilon' = \frac{\delta (\text{Change in length})}{\text{original length (L)}}$$

$$= \frac{49.986 - 50}{50}$$

∴ -ve sign show it reduces.

$$\epsilon' = -0.00028$$

(*) Longitudinal stress:-

$$\epsilon = \frac{\delta (\text{Change in length linearly})}{\text{original length}}$$

$$= \frac{2 \text{ mm}}{2 \times 1000 \text{ mm}}$$

$$\epsilon = 0.001$$

$$\text{Q ①} \Rightarrow \nu = -\frac{(-0.00028)}{0.001}$$

$$\boxed{\nu = 0.28}$$

As for isotropic material, the Poisson's ratio for width is equal to Poisson's ratio for height.

i.e. $\nu_w = \nu_h.$

$$\nu_h = 0.28$$

$$\therefore \nu_w = 0.28.$$

$$\frac{\epsilon'}{\epsilon} = 0.28.$$

$$\therefore \nu_h = -\frac{\epsilon'}{\epsilon}.$$

$$\epsilon' = \epsilon (0.28).$$

$$\therefore \epsilon = 0.001.$$

$$= 0.001 (0.28).$$

$$\epsilon' = 0.00028 \rightarrow \textcircled{2}.$$

Since

$$\epsilon' = \frac{\delta h}{h_0}$$

$$\epsilon' = \frac{\delta h}{h_0}$$

$$\frac{\delta h}{25} = 0.00028.$$

$$\Rightarrow \delta h = 25(0.00028)$$

$$\boxed{\delta h = 0.007 \text{ mm}}$$

Pb-2:-

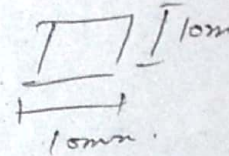
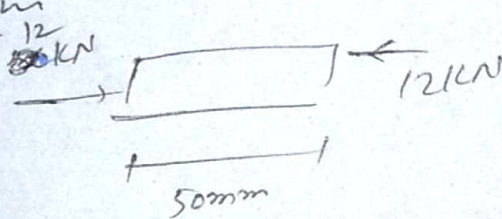
Consider Magnesium alloy on which 12kN forces apply at both sides. The original length is 50mm. The width & height is 10mm respectively. Longitudinal strain is 0.0015 and Poisson's ratio is 0.35 Find.

(a) Stress.

(b) Change in longitudinal length $\delta_{long} = ?$

(c) $h', w' = ?$

Diagram



Solution:-

(a) Since
$$\text{Stress} = \frac{\text{Force}}{\text{Area}}$$
$$= \frac{12 \text{ kN}}{100 \text{ mm}^2}$$

$$\sigma = 120 \text{ MPa}$$

(b) As we know That;

$$\epsilon = - \frac{\delta_{long}}{\text{Original length}}$$

$$\delta_{long} = -\epsilon (L_0)$$
$$= -0.0015 (50 \text{ mm})$$

$$\delta_{long} = -0.075 \text{ mm}$$

② $\delta h = ?$

As; $\epsilon' = \frac{\delta h}{h_0} \rightarrow \text{①}$

W find $\epsilon' = ?$

Since; $-\frac{\epsilon'}{\epsilon} = \nu$

$\nu = 0.35$

$\epsilon = -0.0015$

$\epsilon' = -\epsilon \nu$

$= -0.35(-0.0015)$

$\epsilon' = 0.000525$

Put in ①

$\delta h = \epsilon' h_0$

$= 0.000525(10)$

$\delta h = 0.00525 \text{ mm}$

$\delta \omega = ?$

Since; $\epsilon' = \frac{\delta \omega}{\omega_0}$

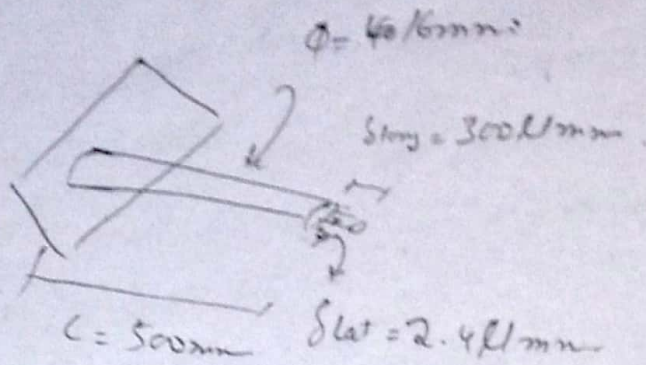
$\delta \omega = \epsilon'(\omega_0)$

$= 0.000525(10)$

$\delta \omega = 0.00525 \text{ mm}$

2) Find the Poisson ratio in the following Problem.

Figure:-



Solution:- As we know;

$$\nu = -\frac{\epsilon'}{\epsilon} \rightarrow \text{①}$$

Longitudinal

$$\text{Longitudinal strain} = \epsilon = \frac{\delta L}{L_0} = \frac{300 \text{ N/mm}}{500 \text{ mm}}$$

$$\epsilon = \frac{3}{5} \mu \Rightarrow \frac{3 \times 10^{-6}}{5}$$

$$\text{Lateral strain} = \epsilon' = -\frac{\delta d}{d_0} = -\frac{2.4 \times 10^{-6} \text{ mm}}{40 \text{ mm}}$$

$$\epsilon' = -0.15 \times 10^{-6}$$

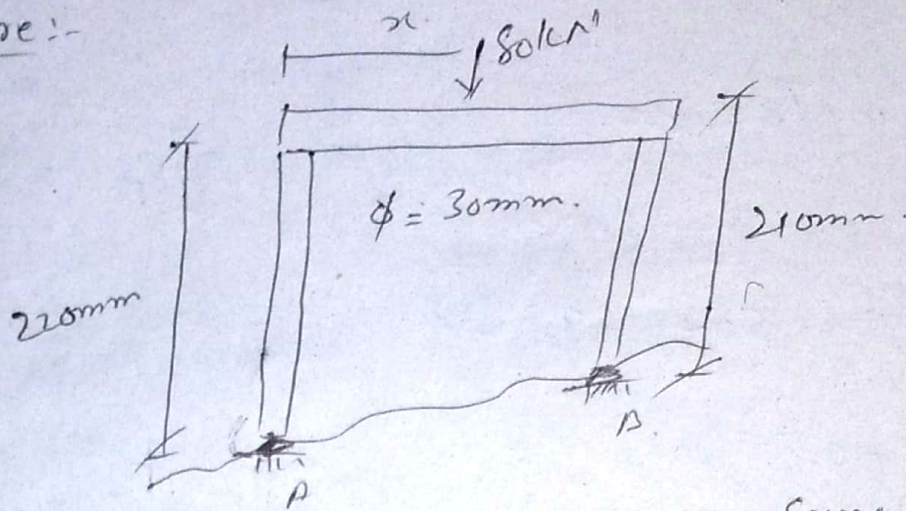
Put in eq ①.

$$\nu = \frac{-(-0.15 \times 10^{-6})}{\frac{3}{5} \times 10^{-6}}$$

$$\boxed{\nu = 0.25} \quad \text{Ans}$$

Pb-4 The Poisson ratio is 0.35, modulus of elasticity is 73.10 GPa . Find the x so that load P equally distribute.

Figure:-



Both columns are made of same materials.

Solution:-

As the columns are made of same material so, deformation will be same in column A & B.

$$\Delta_A = \Delta_B.$$

$$\frac{P_L}{AE} = \frac{P_L}{AE}$$

$\therefore$ As Area & E is same for both column.

$$P_A L_A = P_B L_B.$$

So,

$$P_A = \frac{P_B L_B}{L_A} \rightarrow \text{①}$$

By putting values.

$$P_A = \frac{210}{220} P_B.$$

$$\boxed{P_A = 0.95 P_B} \rightarrow \textcircled{2}$$

Apply condition of equilibrium.

$$(\uparrow) \sum F_y = 0$$

$$P_A + P_B = 80.$$

$$\therefore P_A = 0.95 P_B$$

$$P_B = 80 - P_A$$

$$= 80 - (0.95 P_B)$$

$$P_B + 0.95 P_B = 80$$

$$1.95 P_B = 80$$

$$\boxed{P_B = 41.02 \text{ kN}}$$

$$\text{Eq } \textcircled{2} \Rightarrow P_A = 0.95 P_B$$

$$= 0.95 (41.02)$$

$$\boxed{P_A = 38.97 \text{ kN}}$$

$$\text{Now, } \sum M_A = 0$$

$$80 \times x = 41.02 \times 3$$

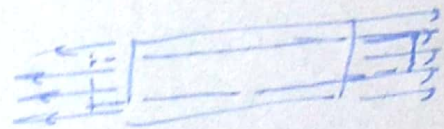
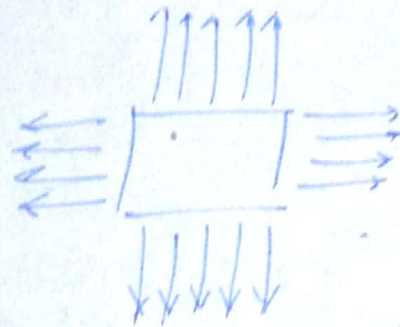
$$x = \frac{41.02 \times 3}{80}$$

$$\boxed{x = 1.53 \text{ mm}}$$

At this value the load will equally distribute in both column.

Biaxial stresses:- (Generalized Hooke's Law)

Upto now we consider the stresses produce due to one side force applied. Now in order to study the stresses produce when force is applied from both sides as shown in fig. We will divide such problem into two parts. First to solve it in x-direction and then solve for y-direction.



along x-axis:-

The strain is given as;

$$\epsilon_x = \frac{\delta x}{E} \quad \text{--- (1)} \quad \text{From Hooke's Law.}$$

(stress = E strain.)

Since from Poisson ratio;

$$\nu = - \frac{\epsilon_y}{\epsilon_x}$$

$$\epsilon_y = -\nu \epsilon_x \quad \text{--- (2)}$$

Put values in eq (1).

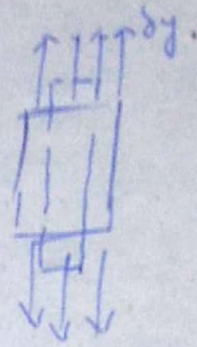
$$\left[\epsilon_y = -\nu \frac{\delta x}{E} \right] \quad \text{--- (3)}$$

Along y-axis :-

from Hook's law;

$$\epsilon_{y2} = \frac{\delta y}{E}$$

$$\epsilon_{x2} = -\nu \epsilon_{y2}$$



$$\epsilon_{x2} = -\frac{\nu \delta y}{E} \rightarrow (4)$$

Now add those who along x-axis
 ϵ along y-axis with corresponding.

$$\epsilon_x = \epsilon_{x1} + \epsilon_{x2}$$

$$\epsilon_x = \frac{\delta x}{E} - \frac{\nu \delta y}{E} \rightarrow (A)$$

$$\epsilon_y = \epsilon_{y1} + \epsilon_{y2}$$

$$\epsilon_y = \frac{\delta y}{E} - \frac{\nu \delta x}{E} \rightarrow (B)$$

Multiply eq (A) by $E \epsilon_x$ EV with
eq (B) we get

$$E \epsilon_x = \delta x - \nu \delta y \rightarrow (C)$$

$$\epsilon_y E \nu = \nu \delta y - \nu^2 \delta x \rightarrow (D)$$

Add eq (C) & (D)

$$\epsilon_x E = \delta x - \nu \delta y$$

$$\epsilon_y \times \nu E = \nu \delta y - \nu^2 \delta x$$

$$\epsilon_x E + \epsilon_y \nu E = \delta x - \nu^2 \delta x$$

$$E(\epsilon_x + \epsilon_y \nu) = \delta x(1 - \nu^2)$$

$$\boxed{\delta x = \frac{E(\epsilon_x + \nu \epsilon_y)}{1 - \nu^2}}$$

Similarly;

$$\boxed{\delta y = \frac{E(\epsilon_y + \nu \epsilon_x)}{1 - \nu^2}}$$

Relation between G , E & ν .

The relation between modulus of rigidity (G) Young's modulus (E) and Poisson's ratio is given as;

$$G = \frac{E}{2(1 + \nu)} \quad \text{--- (1)}$$

For steel this ratio is;

$$\boxed{G = 0.38 E}$$

for concrete;

$$\boxed{G = 0.4 E}$$

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Problem 223

A rectangular steel block is 3" long in x -direction, 2 inches in y -direction and 4 inches in z -direction. The block is subjected to triaxial loading

of 3 uniformly forces (distributed) as follows; 48 kips tension in x -direction, 60 kips compression in y -direction and 54 kips in z -direction. If $\nu = 0.30$ & $E = 29 \times 10^6$ psi determine the single uniformly distributed load in the x -direction that would produce the same deformation in y -direction as original loading.

Solution:-

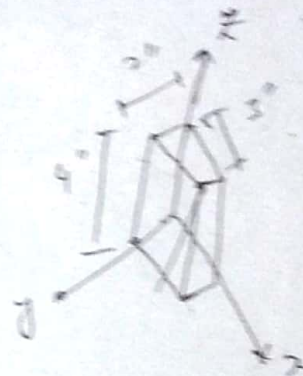
Given data;

Dimension of block;

x -direction = 3 inches.

y -direction = 2 inches.

z -direction = 4 inches.



Triaxial loads;

x -direction = 48 kips.

y -direction = 60 kips.

z -direction = 54 kips.

Poisson ratio = 0.30.

Modulus of elasticity = $E = 29 \times 10^6$ Psi.

Required data;

$P_x = ?$ that produce same deformation in y -direction.

As we know that;

$$\delta x = \frac{P}{A} \rightarrow \text{①}$$

$$\text{② } P = \delta x A \rightarrow \text{②}$$

As for x-direction.

$$\text{Area} = 4(2) = 8 \text{ in}^2 \quad \therefore \text{From fig.}$$

To find $\delta x = ?$

Since;

$$\delta x = E \epsilon_x \rightarrow \textcircled{A}$$

From Poisson's ratio;

$$\nu = -\frac{\epsilon_y}{\epsilon_x}$$

$$\epsilon_y = -\epsilon_x \nu \rightarrow \textcircled{B}$$

As; $\epsilon_x = \frac{\delta x}{L} \quad \therefore \text{From } \textcircled{A}$

$$\epsilon_y = -\frac{\delta x}{L} \nu \rightarrow \textcircled{C}$$

Now find ϵ_y :-

Since;

$$\epsilon_y = \frac{1}{E} (\delta y - \nu (\delta x + \delta z)) \rightarrow \textcircled{D}$$

$$\delta x = \frac{P_x}{A} = \frac{48}{2 \times 4} \Rightarrow 6 \text{ Ksi (Tension)}$$

$$\delta y = \frac{P_y}{A} = \frac{60}{4 \times 3} \Rightarrow 5 \text{ Ksi (Compression)}$$

$$\delta z = \frac{P_z}{A} = \frac{54}{2 \times 3} \Rightarrow 9 \text{ Ksi (Tension)}$$

$\textcircled{D} \Rightarrow$

$$\epsilon_y = \frac{1}{29000 \times 10^4} [-5000 - 0.30(6000 + 9000)]$$

$$\epsilon_y = -3.276 \times 10^{-4} \rightarrow (5)$$

Put values in eq (3)

$$-3.276 \times 10^{-4} = \frac{-\delta_x (0.36)}{29 \times 10^6}$$

$$\delta_x = 31666.67 \text{ Psi} \rightarrow (6)$$

Put values in eq (1)

$$\delta_x = \frac{P_x}{A} \Rightarrow P_x = \delta_x A$$

$$= 31666.67 (8)$$

$$= 253333.33 \text{ lb (Tension)}$$

$$\boxed{P_x = 235.33 \text{ Kips.}} \quad \underline{\text{Ans}}$$

Q224:- For the load block loaded triaxially described in Prob 223, Find the uniformly distributed load that must be added in x-direction to produce no deformation in the x-direction.

Solution:-

Since we know that;

$$\epsilon_z = \frac{1}{E} (\delta_z - \nu (\delta_x + \delta_y)) \rightarrow (1)$$

As from problem 223.

$$\delta_z = 9 \text{ Ksi}$$

$$\delta_x = 6 \text{ Ksi}$$

$$\delta_y = 5 \text{ Ksi} \quad \& \quad E = 29 \times 10^6 \text{ Ksi}$$

By putting values.

$$\epsilon_z = \frac{1}{29 \times 10^6} [9000 - (6000 - 5000)]$$

$$= \frac{1}{29 \times 10^6} (9000 - 3000)$$

$$\epsilon_z = 3 \times 10^{-3}$$

As,

$$\epsilon_z = \frac{1}{E} (\delta_z - \nu (\delta_x + \delta_y))$$

As in z-direction no deformation produced
So, $\epsilon_z = 0$.

$$0 = \frac{1}{E} (\delta_z - \nu (\delta_x + \delta_y))$$

$$\delta_z = \nu (\delta_x + \delta_y)$$

$$9000 = 0.3 (\delta_x - 5000)$$

$$\delta_x - 5000 = \frac{9000 \times 10}{0.3}$$

$$= 30000$$

$$\delta_x = 35000 \text{ psi}$$

$$\delta_{\text{added}} + 6000 = 38000$$

$$\delta_{\text{added}} = 29000 \text{ psi}$$

$$\delta_{\text{add}} = \frac{P}{A} \Rightarrow P_{\text{add}} = \delta_{\text{add}} \times A$$

$$= 2 \times 4 = A$$

$$= 29000(8)$$

$$= 232000 \text{ lb}$$

$$P_{\text{add}} = 232 \text{ kips} \quad \text{Ans}$$

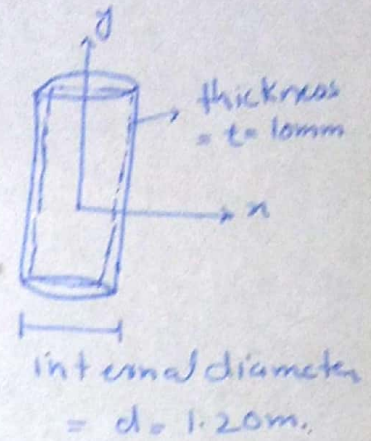
Pb-225:- A welded steel cylindrical drum made of 10mm plate has internal diameter of 1.20m. Compute the change in diameter that would be caused by an internal pressure of 1.5MPa. Assume that Poisson's ratio 0.30 & $E = 200 \text{ GPa}$.

Ans:- Given data;
Internal diameter
 $= d = 1.2 \text{ m}$.

Internal Pressure;
 $p = 1.5 \text{ MPa}$

$$\nu = 0.30.$$

$$E = 200 \text{ GPa}.$$



Required data;

$$\text{Change in diameter} = \Delta d = ?$$

Solution:-

As we know that;

$$\epsilon_x = \frac{\Delta D}{D} \rightarrow (1).$$

To find $\epsilon_x = ?$

Since;

$$\epsilon_x = \frac{\delta x}{E} - \nu \frac{\delta y}{E} \rightarrow (2).$$

finding δx & $\delta y = ?$

$$\text{As; } \delta x = \frac{P}{A}$$

$$\therefore \frac{\pi}{4} D^2 = A.$$

Strain

Chapter # 02

Definition:-

Strain can be define as;

"Change in length per unit original length is called strain."

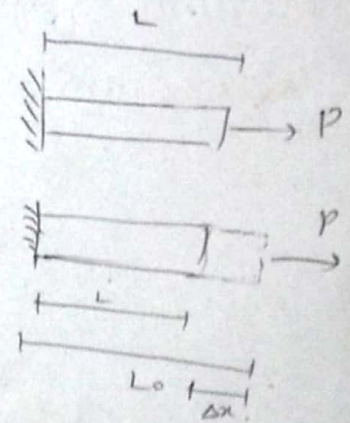
Explanation:-

When a load is applied on a body, there deformation (Change in length occur).

Consider the figure in which a cylinder of length is fixed on one side and on other side a load P is shown.

Its original length is L , but as load is applied so it extends as shown.

Now this change in length over original length is called strain.



Denotation:-

It is denoted by ϵ . (Epsilon)

Formula:-

$$\epsilon = \frac{\delta}{L}$$

$$\therefore \delta = L_{\text{Final}} - L_{\text{Original}}$$

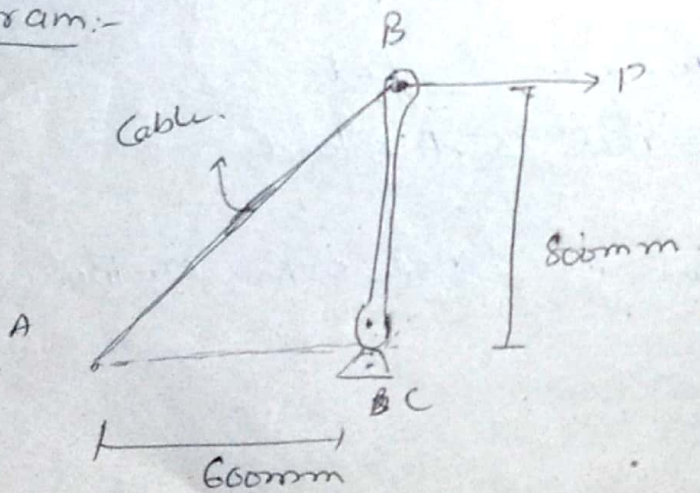
(*) Stress is dimensionless as it is ratio of two ^{same} units (lengths)

In case of elongation (tension) strain will be positive.

→ Where in case of compression i.e. (Compression) strain will be negative.

Prob:- Consider the fig. Find the strain in cable AB if force is applied P which inclined the rod at 5° . The length (height) of rod is 800mm and connected pin at B. The distance from A where cable fixed and BC is 600mm.

Diagram:-



Ans:- Since we know that;

$$\epsilon_{AB} = \frac{L_{\text{Final}} - L_{\text{Initial}}}{\text{Original length}} \rightarrow \textcircled{1}$$

2b

$$L_{\text{final}} = ?$$

and $L_{\text{initial}} = 1000 \text{ mm}$. = original length.

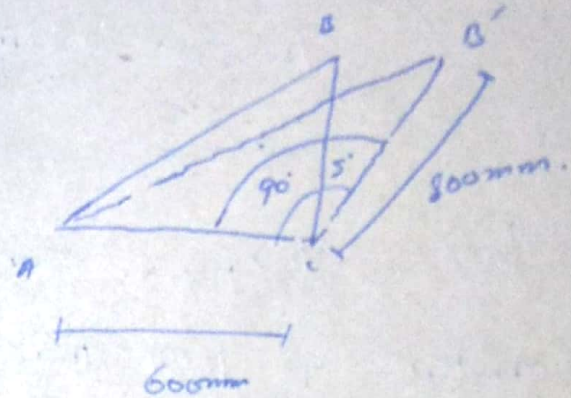
to find $L_{\text{final}} = ?$

Consider the diagram.

Using law of cosine
to find L_{final} .

i.e. length of AB

after applying a load P.



$$(AB)^2 = (600)^2 + (800)^2 - 2(600)(800) \cos(95^\circ)$$

$$AB = L_{\text{final}} = \sqrt{360000 + 640000 - 700966.618}$$

$$L_{\text{final}} = 1040.99 \text{ mm}$$

$\Rightarrow$

$$\epsilon_{AB} = \frac{1040.99 - 1000}{1000}$$

$$\epsilon_{AB} = 0.041 \frac{\text{mm}}{\text{mm}}$$

Hooke's Law:-

This law states that;

Stress is directly proportional to strain within elastic limit.

Mathematically:-

Stress $\propto$ Strain.

$$\sigma \propto \epsilon.$$

But Thomas Young in 1807 introduced a constant of proportionality that came to be known as Young's modulus (Modulus of elasticity).

$$\sigma \propto \epsilon.$$

$$\sigma = E \epsilon.$$

$$\boxed{E = \frac{\sigma}{\epsilon}} \rightarrow (1).$$

Since we know that;

$$\sigma = \frac{P}{A} \rightarrow (2).$$

$$\epsilon = \frac{\delta}{L} \rightarrow (3).$$

$\therefore$ Put eq (2) & (3) in (1).

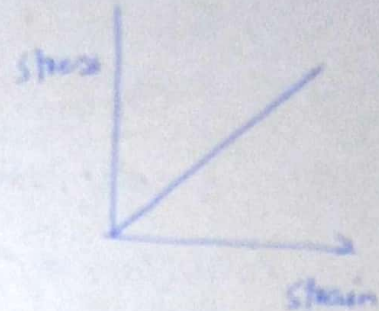
$$\text{So } \Rightarrow E = \frac{P/A}{\delta/L}.$$

$$\boxed{E = \frac{PL}{A\delta}}$$

Modulus of elasticity has same unit as stress i.e. N/mm^2 or Pascal.

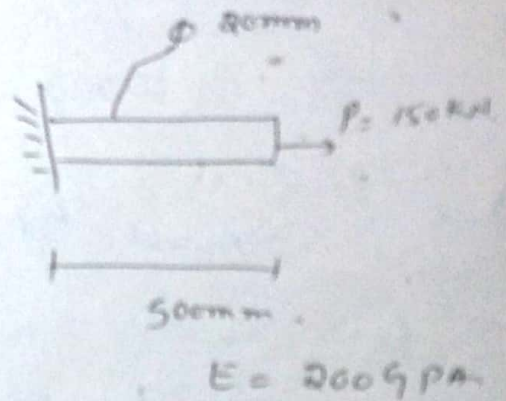
(*) Hooke's law can be explain from the graph as shown, which is graph of stress & strain.

With "proportional limit" stress is increasing with increase in strain. This graph follow Hooke's law.



Problem # 1.

Find the change in length when a load P of 150 kN is acted as shown. If diameter of rod is 20 mm & length (initial) is 500 mm . Modulus of elasticity is 200 GPa .



Solution:-

Since we know that;

$$\delta = \frac{PL}{AE}$$

By putting values.

$$= \frac{(150 \text{ kN})(500 \text{ mm})}{\frac{\pi}{4} (20 \text{ mm})^2 (200 \text{ kN/mm}^2)}$$

$$\boxed{\delta = 1.19 \text{ mm}}$$

$$= 16 \text{ Pa} \cdot \text{mm}^2$$

Deformation per unit length is;

$$\epsilon = \frac{\delta}{L} = \frac{1.19 \text{ mm}}{500 \text{ mm}}$$

$$\boxed{\epsilon = 2.3 \times 10^{-3} \frac{\text{mm}}{\text{mm}}}$$

Shearing Strain:-

It is defined as;

Angular change between two perpendicular faces of differential elements."

OR

"It is change in right angle"

Denotation:-

→ Denoted by Gamma " γ "

Formula:-

$$\gamma = \frac{\pi}{2} - \theta$$

Unit:- It is measure in radian.

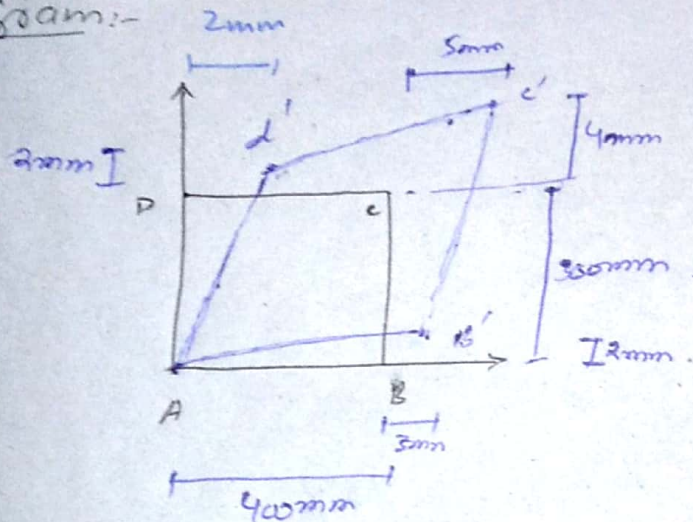
Explanation:-

Consider the figure in which a force P act on it in such a way which is parallel to its cross sectional area. Such force does not compress or elongate but change the shape of object as shown. Such change in shape or deformation is called shearing strain measure in radian.

Problem on Shearing Strain:-

Consider The fig in which some force is applied due to which its shape change. The deformations & its all lengths are given. Find The shear strain at A & D.

Diagram:-



Solution:-

We have to find shearing strain at A & D.

Shearing strain at A:-

$$\gamma_A = \alpha + \beta \rightarrow \text{①}$$

$$\tan \alpha = \frac{2 \text{ mm}}{400 \text{ mm}}$$

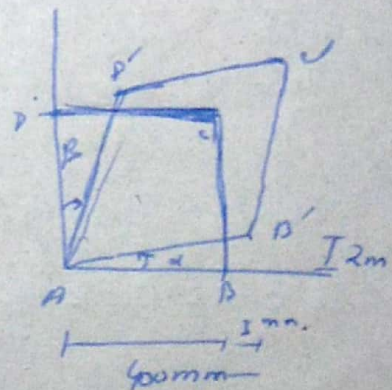
$$\alpha = \tan^{-1}(0.0049)$$

$$\alpha = 6.284^\circ$$

In radian

$$\alpha = 0.284 \times \frac{\pi}{180}$$

$$\alpha = 4.96 \times 10^{-3} \text{ rad}$$



Now For β :-

$$\tan \beta = \frac{2 \text{ mm}}{302 \text{ mm}}$$

$$= 0.3794^\circ$$

In radian:-

$$= 0.3794 \times \frac{\pi}{180}$$

$$= 6.62 \times 10^{-3} \text{ radian}$$

Now eq ①:-

Shearing strain at A = $\gamma_A = \alpha + \beta$

$$= 4.96 \times 10^{-3} \text{ rad} + 6.62 \times 10^{-3} \text{ rad}$$

$$= 11.58 \times 10^{-3} \text{ rad}$$

Shearing strain at D:-

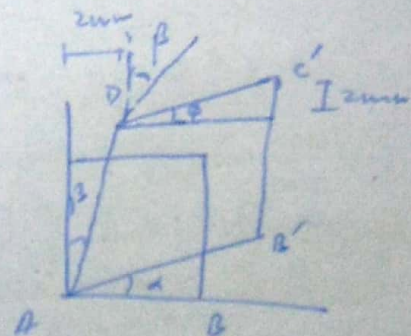
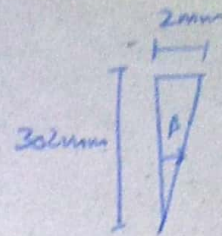
$$\gamma_D = \phi + \beta \rightarrow \text{②}$$

Since β is same as corresponding angles.

$$\tan \phi = \frac{2 \text{ mm}}{402 \text{ mm}}$$

$$\tan \phi = 0.285 (0.0049)$$

$$\phi = \tan^{-1} (0.0049)$$



$$\phi = 0.2807$$

In radian

$$\phi = 0.2807 \times \frac{\pi}{180}$$

$$\phi = 4.9 \times 10^{-3} \text{ rad.}$$

$\therefore$ Here we take its negative value because here compression of angle occur, i.e. angle reduced.

So; $\gamma_D = \phi + \beta$

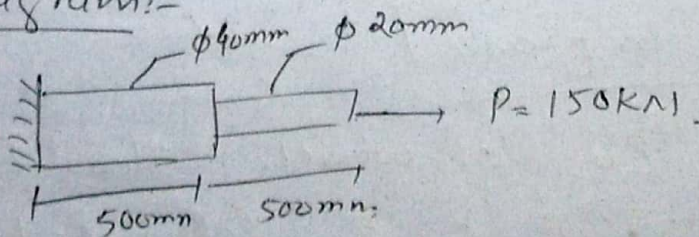
$$= 4.9 \times 10^{-3} + 6.62 \times 10^{-3} \text{ rad.}$$

$$\gamma_D = -11.58 \times 10^{-3} \text{ rad.}$$

Problem $\rightarrow$ Composite

Consider the fig which consist of different diameter i.e. 40mm and 20mm. A force of 150kN force is applied. The length of first section is 500mm & other length is also 500mm. Calculate change in length. (deformation)

Diagram:-



Solution:- Given lengths;

$$l_1 = 500 \text{ mm}$$

$$l_2 = 500 \text{ mm}$$

Diameter I = 40mm.

Diameter II = 20mm.

Force = $P = 156 \text{ kN}$.

Modulus of Elasticity = $E = 200 \text{ GPa}$

Formula:-

$$\text{Deformation} = \sum_{i=1}^n \frac{P_i L_i}{A_i E_i}$$

Putting values. to get total deformation.

$$\delta_{\text{Total}} = \frac{P_1 L_1}{A_1 E_1} + \frac{P_2 L_2}{A_2 E_2}$$

$$\delta_{\text{Total}} = \frac{150 \text{ (kN)} \times (500 \text{ mm})}{\frac{\pi}{4} (40)^2 \times 200 \text{ (kN/mm}^2\text{)}} + \frac{156 \text{ kN} \times 500 \text{ mm}}{\frac{\pi}{4} (20)^2 \times (200 \text{ kN/mm}^2)}$$

$$\delta = 0.298 \text{ mm} + 1.194 \text{ mm}$$

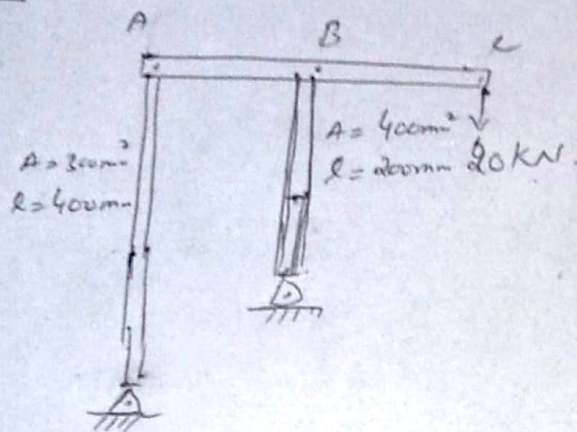
$$\boxed{\delta_{\text{Total}} = 1.492 \text{ mm}}$$

Prob - Deflection in frame

Consider the fig. In which - two columns are provided at A & B. Column at A has area of 300 mm^2 & length 4000 mm while length of column at B is 2000 mm where its area is 400 mm^2 . At C load of 20 kN act. The

between A & B = 400mm & B and C is 500mm. Calculate deformation at C. Modulus of elasticity is 500 GPa.

Diagram:-



Solution:-

First we have to find load at A & B. So draw FBD of beam.

Ⓢ Ⓢ Ⓢ

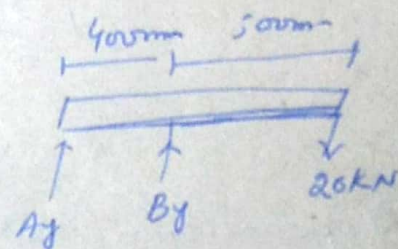
$$\sum M_A = 0$$

$$-B_y(400) + 20\text{kN} \times 900 = 0$$

$$400B_y = (200 \times 900)\text{kN}$$

$$B_y = \frac{1}{20} \times 900\text{kN}$$

$$B_y = 45\text{kN}$$



$$(\uparrow) \sum F_y = 0 (\downarrow -)$$

$$A_y + B_y - 20 = 0$$

$$A_y + 45 - 20 = 0$$

$$A_y = -25\text{kN}$$

$\therefore$ -ve sign shows our assume direction is wrong.

Deformation at A

$$\delta_A = \frac{PL}{AE}$$

$$= \frac{25 \times 40}{300 \times 200}$$

$$\boxed{\delta_A = 0.167 \text{ mm}}$$

Deformation at B:-

$$\delta_B = \frac{PL}{AE}$$

$$= \frac{45 \times 200}{400 \times 200}$$

$$\boxed{\delta_B = -0.11}$$

As length of column B will decrease due to compression.

Deformation at C

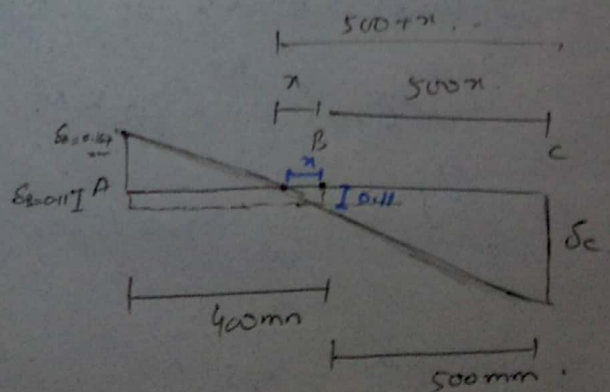
From Δ similarity of triangles.

$$\frac{\delta_C}{500+x} = \frac{0.11}{x} \rightarrow \text{①}$$

Now Consider triangles at Left side

$$\frac{0.167 \text{ mm}}{400 \text{ mm}} = \frac{0.11 \text{ mm}}{x}$$

$$\boxed{x = 263.5 \text{ mm}}$$



Consider the two triangles.

