TROLLTUNE CORPORATION 2710 Thomes Avenue Cheyenne, WY 82001

SUPPLEMENT NO. SFM7499-SW-R

FAA APPROVED

SUPPLEMENTAL AIRPLANE FLIGHT MANUAL

FOR

CESSNA 182P s/n 18260826 through 18261425

STC SA03608AT Maximum Gross Takeoff Weight Increase

Registratio	on No.	
Serial No.		
, N		

This supplement must be used in conjunction with existing placards and material required to be furnished to the pilot under CAR Part 3 (as found in Cessna Owner's Manual for the 1972 model year) whenever this aircraft is operated at weights above 2950 lbs. in accordance with Trolltune Corporation FAA STC SA03608AT or EASA STC 10026913. The information contained in this document supplements or supersedes the Owner's Manual or placards only in those areas listed. For limitations, procedures and performance information not contained in this supplement, consult the basic Owner's Manual, markings and operating placards.

FAA Approved

Manager, Flight Test Branch, ANM-160L

Federal Aviation Administration

Los Angeles Aircraft Certification Office

Transport Airplane Directorate

Date: May 13, 2011

SUPPLEMENT AIRPLANE FLIGHT MANUAL FOR CESSNA 182P MAXIMUM GROSS WEIGHT INCREASE STC No. SA03608AT

SUPPLEMENT NO. SFM7499-SW-R

RECORD OF REVISIONS

Rev	Page			77000011000111000
No.	No.	Date	Description	FAA Approved
I/R	All 1-14	22/08/2008	Maximum Takeoff Gross Weight Increase to 3100 lbs. Initial Release	David Crew Manager, Flight Test Branch Federal Aviation Administration Atlanta Aircraft Certification Office Date: 22-August-2008
	1	13/05/2011	Added reference to EASA STC and changed FAA approval block and approval date.	A. R.
	2	13/05/2011	Added Record of Revision page.	Manager, FlYght Test Branch, ANM-160L Federal Aviation Administration Los Angeles Aircraft Certification office
1	3	13/05/2011	Added performance and limitations note.	Transport Airplane Directorate Date: 5/13/1011
	4	13/05/2011	Added limitation statement.	Ty
	1-16	13/05/2011	Reformatted and incremented pages numbers.	900
				No.

SUPPLEMENT AIRPLANE FLIGHT MANUAL FOR CESSNA 182P
MAXIMUM GROSS WEIGHT INCREASE STC No. SA03608AT

SUPPLEMENT NO. SFM7499-SW-R

SECTION I - OPERATING CHECKLIST

TAKE-OFF:

MAXIMUM PERFORMANCE TAKE-OFF, Flaps 20° , 3100 lbs.: Climb Speed -- (until all obstacles are cleared): ... 66 MPH IAS

ENROUTE CLIMB:

SECTION II - DESCRIPTION AND OPERATING DETAILS

NOTE: Changes in loadings, limitations, airspeeds, and other performance data due to the gross weight increase described in this SAFM were developed based upon the original airplane configuration as found in the basic Cessna Owner's Manual (OM) and approved placards. If other STCs (e.g., autopilot, aux fuel tanks, engine upgrades, etc.) have been incorporated, it is possible that their associated flight manual supplements describe different limitations or performance data from that shown here.

TAKE-OFF:

Airspeed - As per SECTION I of this Supplement

ENROUTE CLIMB:

Airspeed - Best angle of climb, flaps up:73 MPH IAS

CRUISE:

Performance - See SECTION VI of this Supplement

FAA Approved Date: May 13, 2011 Page 3 of 16

SUPPLEMENT AIRPLANE FLIGHT MANUAL FOR CESSNA 182P MAXIMUM GROSS WEIGHT INCREASE STC No. SA03608AT

Page 4 of 16

SUPPLEMENT NO. SFM7499-SW-R

SECTION II - DESCRIPTION AND OPERATING DETAILS (continued)

NOISE ABATEMENT:

The certificated noise level for the Model 182P at 3100 pounds maximum weight is 85.5 dB(A), determined according to Appendix G of 14 CFR Part 36 through Amendment 28. No determination has been made by the Federal Aviation Administration that the noise levels of this airplane are or should be acceptable or unacceptable for operation at, into, or out of, any airport.

SECTION III EMERGENCY PROCEDURES

ENGINE FAI

	ENGINE FAILURE AFTER TAKE-OFF, 3100 lbs.:	
	Wing Flaps 0° - 20° 85 MPH	IAS
	ENGINE FAILURE DURING FLIGHT, 3100 lbs.:	
	Maximum Glide: 86 MPH	IAS
	FORCED LANDINGS:	
	′O _×	
	EMERGENCY LANDING WITHOUT ENGINE POWER, 3100 lbs.:	
	Wing Flaps Up: 85 MPH	IAS
	Wing Flaps Down:	IAS
	PRECAUTIONARY LANDING WITH ENGINE POWER, 3100 lbs.:	
	Wing Flaps Down: 80 MPH	IAS
	40	
:c	ON IV - OPERATING LIMITATIONS	
	'	

SECTIO

The limitations in this section may be further restricted by other installed STC(s), FAA Form 337, associated AFMS(s), or SAFM(s).

MANEUVERS - NORMAL CATEGORY:

Maximum Ramp Weight: 3110	lbs.
Maximum Takeoff Weight: 3100	lbs.
Maximum Landing Weight:	lbs.

Note: A normal start, taxi and run-up time of ten minutes will consume approximately 10 lbs. of fuel. Normal landings must not be made at weights in excess of 2950 lbs. For a typical 3100 lbs. takeoff, climb, and cruise profile, this equates to a minimum flight duration of approximately one hour and forty-five minutes.

AIRSPEED LIMITATIONS:

Maneuvering	Speed	127	MPH	CAS	(128	MPH	IAS)
-------------	-------	-----	-----	-----	------	-----	------

FAA Approved Date: May 13, 2011

SUPPLEMENT AIRPLANE FLIGHT MANUAL FOR CESSNA 182P
MAXIMUM GROSS WEIGHT INCREASE STC No. SA03608AT

SUPPLEMENT NO. SFM7499-SW-R

SECTION IV - OPERATING LIMITATIONS (continued)

WEIGHT AND BALANCE:

Center of gravity limitations and envelopes are changed for operation at weights above 2950 lbs. to and including 3100 lbs.

CENTER OF GRAVITY LIMITS:

Forward:

33.0 inches aft of datum at 2250 lbs. or less, with straight line variation to 40.9 inches aft of datum at 3100 lbs.

Aft:

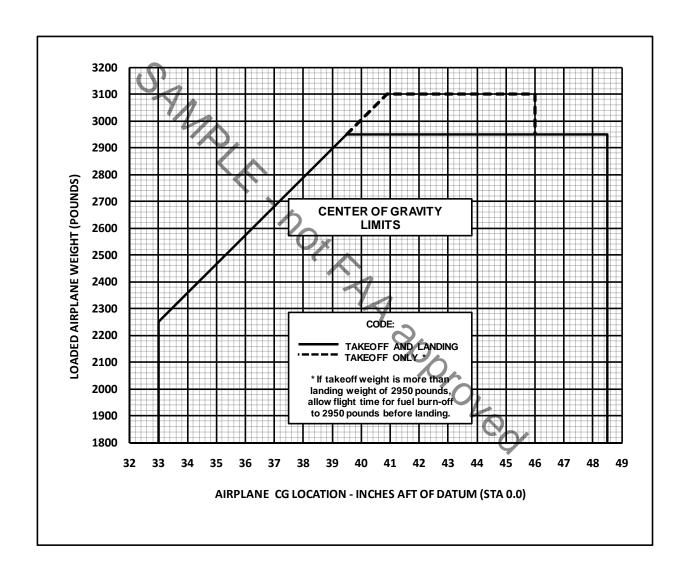
48.5 inches aft of datum at all weights except 46.0 inches aft of datum at weights above 2950 lbs. to 3100 lbs.

Use the following CG limit and moment envelopes:

FAA Approved Date: May 13, 2011 Page 5 of 16

SECTION IV - OPERATING LIMITATIONS (continued)

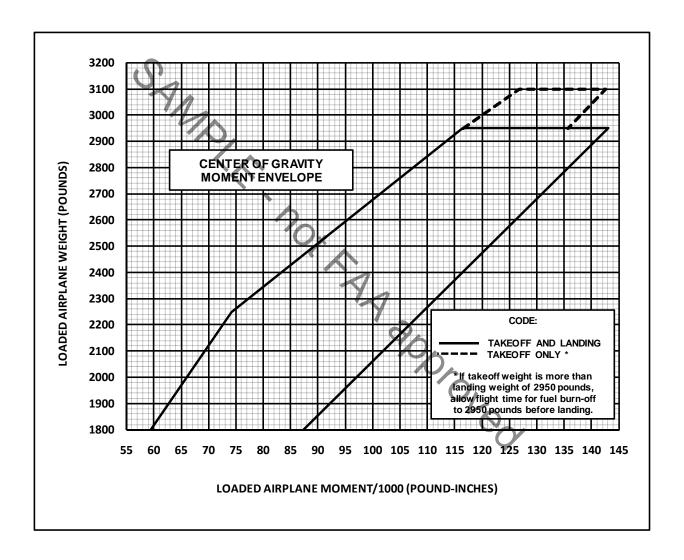
CENTER OF GRAVITY LIMITS: (continued)



FAA Approved Date: May 13, 2011 Page 6 of 16

SECTION IV - OPERATING LIMITATIONS (continued)

CENTER OF GRAVITY LIMITS: (continued)



SECTION V - CARE OF THE AIRPLANE

NO CHANGES

FAA Approved Date: May 13, 2011 Page 7 of 16

SECTION VI - OPERATIONAL DATA

STALL SPEEDS - MPH CAS (IAS)										
		ANGLE OF BANK								
	CONDITION	0°	30°	60°						
	FLAPS UP	67 (58)	73 (66)	94 (92)						
3100 LBS. GROSS WEIGHT	FLAPS 20°	61 (48)	66 (57)	86 (84)						
	FLAPS 40°	60 (45)	64 (52)	85 (82)						
		POWER OFF - AFT (CG							

	TAKE-OFF DATA TAKE-OFF DISTANCE WITH 20° FLAPS FROM HARD SURFACE RUNWAY												
GROSS	IAS	HEAD	AT SEA	LEVEL & 59°F.	AT 250	00 FT. & 50°F.	AT 500	00 FT. & 41°F.	AT 7500 FT. & 32°F.				
WEIGHT	@ 50'	WIND	GROUND	TOTAL TO	GROUND	TOTAL TO	GROUND	TOTAL TO	GROUND	TOTAL TO			
LBS.	MPH	KNOTS	RUN	CLEAR 50' OBS	RUN	CLEAR 50' OBS	RUN	CLEAR 50' OBS	RUN	CLEAR 50' OBS			
3100	66	0 10 20	805 560 355	1540 1170 845	965 680 440	1850 1420 1035	1155 825 550	2265 1765 1310	1415 1025 700	2945 2325 1760			

NOTES: 1. Increase distances 10% for each 25°F above standard temperature for particular altitude.

2. For operation on a dry, grass runway, increase distances (both "ground run" and "total to clear 50 ft. obstacle") by 7% of the "total to clear 50 ft. obstacle" figure.

	MAXIMUM RATE-OF-CLIMB DATA														
	AT SEA LEVEL & 59°F.		k 59°F.	AT 5	000 FT. &	41°F.	AT 10	AT 10,000 FT. & 23°F.		AT 15,000 FT. & 5°F.			AT 20	,000 FT. &	-12°F.
GROSS WEIGHT LBS.	IAS MPH	RATE OF CLIMB FT/MIN	GAL. OF FUEL USED	IAS MPH	RATE OF CLIMB FT/MIN	From SL FUEL USED	IAS MPH	RATE OF CLIMB FT/MIN	From SL FUEL USED	IAS MPH	RATE OF CLIMB FT/MIN	From SL FUEL USED	IAS MPH	RATE OF CLIMB FT/MIN	From SL FUEL USED
3100	91	755	1.5	89	540	4.2	87	330	7.8	85	110	13.2	-		

NOTES: 1. Flaps up, full throttle, 2600 RPM, mixture leaned for smooth operation above 5000 ft.

 Fuel used includes warm-up and take-off allowance.
 For hot weather, decrease rate of climb 30 ft./min. for each 10°F above standard day temperature for particular altitude.

FAA Approved Date: May 13, 2011 Page 8 of 16

SECTION VI - OPERATIONAL DATA (continued)

			CRUI	SE PE	RFORMA	NCE								
	LEAN MIXTURE Standard Conditions - Zero Wind - Gross Weight - 3100 Pounds 2500 FEET													
	(60 GAL (NO	RESERVE)	79 GAL (NC	RESERVE)						
RPM	MP	% BHP	TAS MPH	GAL/ HOUR	ENDR. HOURS	RANGE MILES	ENDR. HOURS	RANGE MILES						
2450	23	76	154	14.2	4.2	645	5.6	860						
	22	72	151	13.4	4.5	675	5.9	890						
	21	68	146	12.7	4.7	685	6.2	905						
	20	63	141	12.0	5.0	705	6.6	930						
2300	23	71	148	13.1	4.6	680	6.0	885						
	22	67	145	12.2	4.9	710	6.5	940						
	21	62	140	11.5	5.2	725	6.9	965						
	20	59	137	11.0	5.5	750	7.2	985						
2200	23	67	145	12.1	5.0	725	6.5	940						
	22	63	141	11.4	5.3	745	6.9	970						
	21	59	137	10.8	5.6	765	7.3	1000						
	20	55	132	10.2	5.9	775		1015						
2000*	20	47	119	8.7	6.9	820	9.1	1080						
	19	43	113	8.2	7.3	820	9.6	1080						
	18	39	106	7.5	8.0	845	10.5	1110						
	17	35	96	7.0	8.6	825	11.3	1080						

^{*}Power settings in this block represent maximum range settings

FAA Approved Date: May 13, 2011 Page 9 of 16

SECTION VI - OPERATIONAL DATA (continued)

	CRUISE PERFORMANCE												
	LEAN MIXTURE Standard Conditions - Zero Wind - Gross Weight - 3100 Pounds 5000 FEET												
					60 GAL (NO	RESERVE)	79 GAL (NC	RESERVE)					
RPM	MP	% BHP	TAS MPH	GAL/ HOUR	ENDR. HOURS	RANGE MILES	ENDR. HOURS	RANGE MILES					
2450	23	78	158	14.5	4.1	645	5.4	850					
	22	73	154	13.6	4.4	675	5.8	890					
	21	70	150	13.0	4.6	690	6.1	915					
	20	65	146	12.2	4.9	715	6.5	945					
2300	23	73	154	3.4	4.5	690	5.9	905					
	22	69	149	12.6	4.8	715	6.3	935					
	21	64	145	11.9	5,0	725	6.6	955					
	20	60	141	11.2	5.4	760	7.1	1000					
2200	23	68	148	12.4	4.8	710	6.4	945					
	22	64	145	11.7	5.1	735	6.8	985					
	21	60	141	11.0	5.5	775	7.2	1015					
	20	57	136	10.5	5.7	775	55	1020					
2000*	20	48	124	9.0	6.7	830	8.8	1090					
	19	45	117	8.5	7.1	830	9.3	1085					
	18	41	109	7.9	7.6	825	10.0	1090					
	17	37	100	7.3	8.2	820	10.8	1080					

^{*}Power settings in this block represent maximum range settings

FAA Approved Date: May 13, 2011 Page 10 of 16

SECTION VI - OPERATIONAL DATA (continued)

			CRUI	SE PE	RFORMA	NCE								
	LEAN MIXTURE Standard Conditions - Zero Wind - Gross Weight - 3100 Pounds 7500 FEET													
	0				60 GAL (NO	RESERVE)	79 GAL (NO	RESERVE)						
RPM	MP	% BHP	TAS MPH	GAL/ HOUR	ENDR. HOURS	RANGE MILES	ENDR. HOURS	RANGE MILES						
2450	21	71	154	13.1	4.6	705	6.0	920						
	20	67	151	12.4	4.8	720	6.4	965						
	19	62	146	11.7	5.1	740	6.8	990						
	18	58	140	11.0	5.5	770	7.2	1005						
2300	21	66	150	22	4.9	735	6.5	975						
	20	62	146	11.6	5.2	755	6.8	990						
	19	58	140	11.0	5,5	770	7.2	1005						
	18	54	135	10.5	5.7	765	7.5	1010						
2200	21	62	146	11.4	5.3	770	6.9	1005						
	20	58	140	10.7	5.6	780	7.4	1035						
	19	54	135	10.2	5.9	795	7.7	1035						
	18	51	129	9.7	6.2	795		1040						
2000*	20	50	128	9.2	6.5	830	8.6	1100						
	19	47	122	8.7	6.9	840	9.1	1110						
	18	43	113	8.1	7.4	835	9.8	1105						
	17	39	105	7.6	7.9	825	10.4	1090						

^{*}Power settings in this block represent maximum range settings

FAA Approved Date: May 13, 2011 Page 11 of 16

SECTION VI - OPERATIONAL DATA (continued)

			CRUI	SE PE	RFORMA	NCE								
				LEAN	MIXTURE									
	Standa	rd Condit	ions -	Zero W	ind - Gr	oss Weight -	3100 Pound	s						
	10,000 FEET 60 GAL (NO RESERVE) 79 GAL (NO RESERVE)													
	0					RESERVE)	·							
RPM	MP	% BHP	TAS MPH	GAL/ HOUR	ENDR. HOURS	RANGE MILES	ENDR. HOURS	RANGE MILES						
2450	19	63	151	11.9	5.0	755	6.6	995						
	18	60	145	11.2	5.4	780	7.1	1025						
	17	55	139	10.6	5.7	790	7.5	1040						
	16	51	133	10.0	6.0	795	7.9	1050						
2300	19	60	145	(1) x	5.4	780	7.1	1025						
	18	56	140	10.5	5.7	795	7.5	1050						
	17	51	133	9.8	6,1	810	8.1	1075						
	16	47	125	9.2	6.5	810	8.6	1075						
2200	19	56	140	10.4	5.8	810	7.6	1060						
	18	52	134	9.8	6.1	815	8.1	1085						
	17	49	127	9.3	6.5	825	8.5	1075						
	16	45	119	8.7	6.9	820	9,1	1080						
2000*	19	48	125	8.9	6.7	835	8.9	1110						
	18	44	118	8.4	7.1	835	9.4	1105						
	17	40	110	7.8	7.7	845	10.1	1110						
	16	38	95	7.4	8.1	765	10.7	1015						

^{*}Power settings in this block represent maximum range settings

FAA Approved Date: May 13, 2011 Page 12 of 16

SUPPLEMENT AIRPLANE FLIGHT MANUAL FOR CESSNA 182P MAXIMUM GROSS WEIGHT INCREASE STC No. SA03608AT

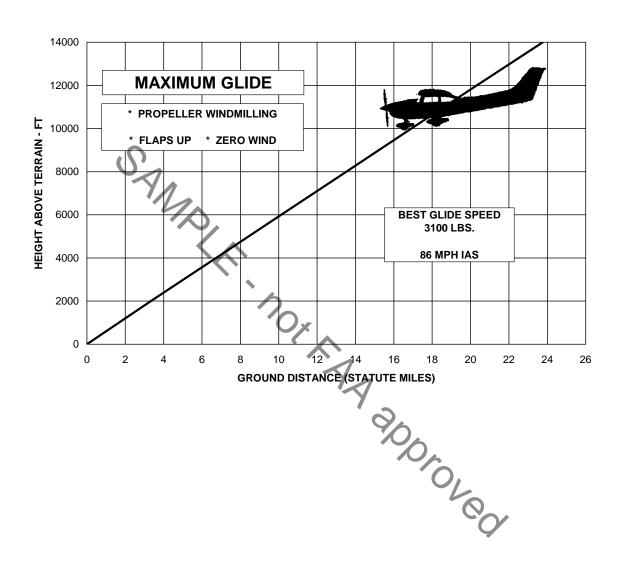
SUPPLEMENT NO. SFM7499-SW-R

SECTION VI - OPERATIONAL DATA (continued)

CRUISE PERFORMANCE								
LEAN MIXTURE Standard Conditions - Zero Wind - Gross Weight - 3100 Pounds 15,000 FEET								
RPM	SME	% BHP	TAS MPH	GAL/ HOUR	60 GAL (NO ENDR. HOURS	RESERVE) RANGE MILES	79 GAL (NO ENDR. HOURS	RESERVE) RANGE MILES
2450	16	54	140	10.4	5.8	810	7.6	1060
	15	50	134	9.8	6.1	815	8.1	1085
	14	46	120	9.2	6.5	780	8.6	1030
2300	16	50	134	9.6	6.2	830	8.2	1095
	15	47	125	9.1	6.6	825	8.7	1085
	14	42	107	8.5	7.1	755	9.3	995
2200	16	47	125	9.1	6,6	825	8.7	1085
	15	44	114	8.6	7.0	795	9.2	1045
	14	40	99	8.0	7.5	740	9.9	980
2000	16	40	99	7.8	7.7	760	10.1	995
	15	37	90	7.3	8.2	735	10.8	970

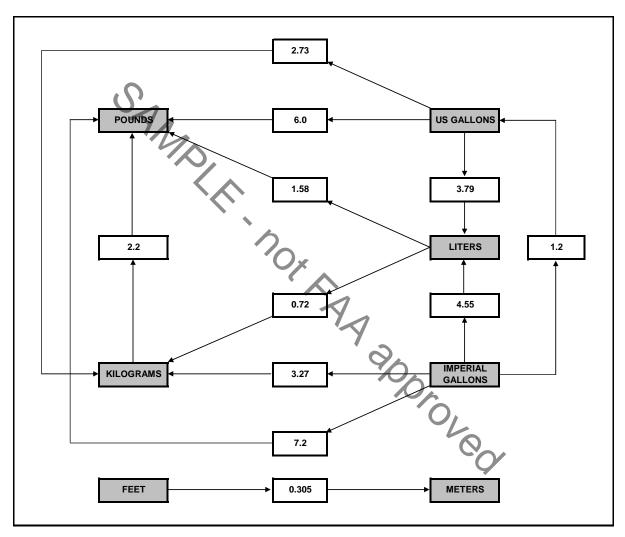
FAA Approved Date: May 13, 2011 Page 13 of 16

SECTION VI - OPERATIONAL DATA (continued)



FAA Approved Date: May 13, 2011 Page 14 of 16

SECTION VI - OPERATIONAL DATA (continued)



Metric / Imperial / US Units Conversion Chart

FAA Approved Date: May 13, 2011 Page 15 of 16

SUPPLEMENT AIRPLANE FLIGHT MANUAL FOR CESSNA 182P MAXIMUM GROSS WEIGHT INCREASE STC No. SA03608AT

SUPPLEMENT NO. SFM7499-SW-R

SECTION VII - OPTIONAL SYSTEMS

NO CHANGES

ALPHABETICAL INDEX

NO CHANGES

CHANG.

ING REQUIREMENTS

NO CHANGES

TOKAN

ADDROVED

TOKAN

TOKAN

ADDROVED SERVICING REQUIREMENTS

FAA Approved Date: May 13, 2011 Page 16 of 16