TROLLTUNE CORPORATION 2710 Thomes Avenue Cheyenne, WY 82001

SUPPLEMENT NO. SFM7501-SW-R

FAA APPROVED

SUPPLEMENTAL AIRPLANE FLIGHT MANUAL

FOR

CESSNA 182P s/n 18262251 through 18263475

STC \$A03608AT Maximum Gross Takeoff Weight Increase

Registration	No.	
Serial No		 ***************************************

This supplement must be used in conjunction with existing placards and material required to be furnished to the pilot under CAR Part 3 (as found in Cessna Owner's Manual for the 1973 to 1974 model years) whenever this aircraft is operated at weights above 2950 lbs. in accordance with Trolltune Corporation STC SA03608AT or EASA STC 10026913. The information contained in this document supplements or supersedes the Owner's Manual or placards only in those areas listed. For limitations, procedures and performance information not contained in this supplement, consult the basic Owner's Manual, markings and operating placards.

FAA Approved

Manager, Flight Test Branch, ANM-160L Federal Aviation Administration

Los Angeles Aircraft Certification Office

Transport Airplane Directorate

Date: Muy 13, 2011

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RECORD OF REVISIONS

Rev	Page			
No.	No.	Date	Description	FAA Approved
I/R	All 1-14	22/08/2008	Maximum Takeoff Gross Weight Increase to 3100 lbs. Initial Release	David Crew Manager, Flight Test Branch Federal Aviation Administration Atlanta Aircraft Certification Office Date: 22-August-2008
	2	13/05/2011	Added reference to EASA STC and changed FAA approval block and approval date.	Manager, Flight Test Branch, ANM-160L
1	3	13/05/2011	Revision page. Added performance and limitations note.	Federal Aviation Administration Los Angeles Aircraft Certification office Transport Airplane Directorate Date:
	4	13/05/2011	Added limitation statement.	
	1-16	13/05/2011	Reformatted and incremented pages numbers.	N O O
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SECTION I - OPERATING CHECKLIST

TAKE-OFF:

MAXIMUM PERFORMANCE TAKE-OFF, Flaps 20° , 3100 lbs.: Climb Speed -- (until all obstacles are cleared): ... 63 MPH IAS

ENROUTE CLIMB:

SECTION II - DESCRIPTION AND OPERATING DETAILS

NOTE: Changes in loadings, limitations, airspeeds, and other performance data due to the gross weight increase described in this SAFM were developed based upon the original airplane configuration as found in the basic Cessna Owner's Manual (OM) and approved placards. If other STCs (e.g., autopilot, aux fuel tanks, engine upgrades, etc.) have been incorporated, it is possible that their associated flight manual supplements describe different limitations or performance data from that shown here.

TAKE-OFF:

Airspeed - As per SECTION I of this Supplement

ENROUTE CLIMB:

CRUISE:

Performance - See SECTION VI of this Supplement

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SECTION II - DESCRIPTION AND OPERATING DETAILS (continued)

NOISE ABATEMENT:

The certificated noise level for the Model 182P at 3100 pounds maximum weight is 85.5 dB(A), determined according to Appendix G of 14 CFR Part 36 through Amendment 28. No determination has been made by the Federal Aviation Administration that the noise levels of this airplane are or should be acceptable or unacceptable for operation at, into, or out of, any airport.

SECTION III - EMERGENCY PROCEDURES

ENGINE FAILURE:

· ///_		
ENGINE FAILURE AFTER TAKE-OFF, 3100 lbs.:		
Wing Flaps 0° - 20°:	MPH	IAS
ENGINE FAILURE DURING FLIGHT, 3100 lbs.:		
Maximum Glide: 86	MPH	IAS
^		
FORCED LANDINGS:		
O _X		
EMERGENCY LANDING WITHOUT ENGINE POWER, 3100 lbs.:		
Wing Flaps Up:		
Wing Flaps Down:	MPH	IAS
PRECAUTIONARY LANDING WITH ENGINE POWER, 3100 lbs.:		
Wing Flaps Down:	MPH	IAS
%		
70.		

SECTION IV - OPERATING LIMITATIONS:

The limitations in this section may be further restricted by other installed STC(s), FAA Form 337, associated AFMS(s), or SAFM(s).

MANEUVERS - NORMAL CATEGORY:

Maximum Ramp Weight: 3110	lbs.
Maximum Takeoff Weight: 3100	lbs.
Maximum Landing Weight:	lbs.

Note: A normal start, taxi and run-up time of ten minutes will consume approximately 10 lbs. of fuel. Normal landings must not be made at weights in excess of 2950 lbs. For a typical 3100 lbs. takeoff, climb, and cruise profile, this equates to a minimum flight duration of approximately one hour and forty-five minutes.

AIRSPEED LIMITATIONS:

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SECTION IV - OPERATING LIMITATIONS (continued)

WEIGHT AND BALANCE:

Center of gravity limitations and envelopes are changed for operation at weights above 2950 lbs. to and including 3100 lbs.

CENTER OF GRAVITY LIMITS:

Forward: 33.0 inches aft of datum at 2250 lbs. or less, with

straight line variation to 40.9 inches aft of datum at

3100 lbs.

Aft: 48.5 inches aft of datum at all weights except 46.0 inches aft of datum at weights above 2950 lbs. to 3100

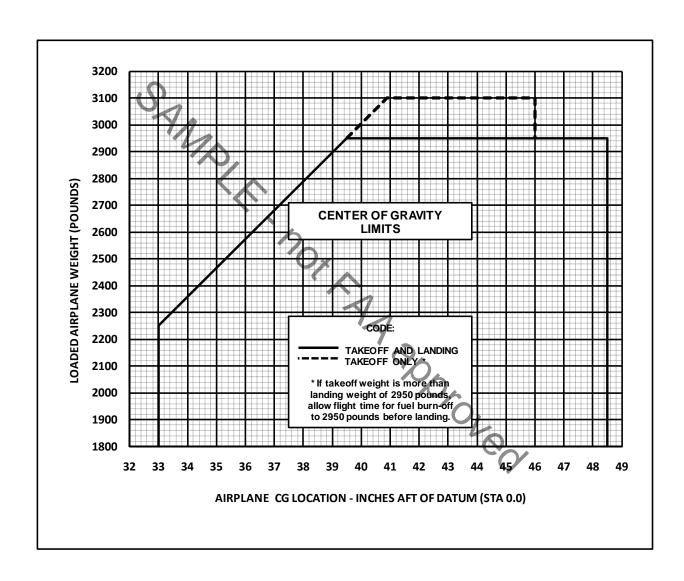
Tps.

Use the following CG limit and moment envelopes:

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SECTION IV - OPERATING LIMITATIONS (continued)

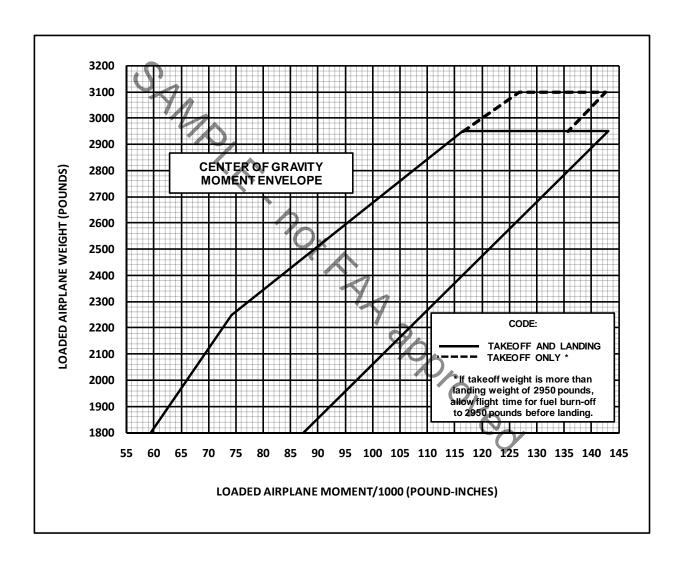
CENTER OF GRAVITY LIMITS: (continued)



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SECTION IV - OPERATING LIMITATIONS (continued)

CENTER OF GRAVITY LIMITS: (continued)



SECTION V - CARE OF THE AIRPLANE

NO CHANGES

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SECTION VI - OPERATIONAL DATA

	STALL SPEEDS - MPH CAS (IAS)										
			ANGLE OF BANK								
	CONDITION	0°	30°	60°							
	FLAPS UP	67 (58)	73 (66)	94 (92)							
3100 LBS. GROSS WEIGHT	FLAPS 20°	61 (53)	66 (60)	86 (84)							
	FLAPS 40°	60 (50)	64 (56)	85 (82)							
	40 .		•								
		POWER OFF - AFT	CG								

	TAKE-OFF DATA TAKE-OFF DISTANCE WITH 20° FLAPS FROM HARD SURFACE RUNWAY												
GROSS	IAS	HEAD	AT SEA	LEVEL & 59°F.	AT 250	0 FT. & 50°F.	AT 5000 FT. & 41°F.		AT 7500 FT. & 32°F.				
WEIGHT	@ 50'	WIND	GROUND	TOTAL TO	GROUND	TOTAL TO	GROUND	TOTAL TO	GROUND	TOTAL TO			
LBS.	MPH	KNOTS	RUN	CLEAR 50' OBS	RUN	CLEAR 50' OBS	RUN	CLEAR 50' OBS	RUN	CLEAR 50' OBS			
3100	63	0 10 20	805 560 355	1540 1170 845	965 680 440	1850 1420 1035	1155 825 550	2265 1765 1310	1415 1025 700	2945 2325 1760			

NOTES: 1. Increase distances 10% for each 25°F above standard temperature for particular altitude.

2. For operation on a dry, grass runway, increase distances (both "ground run" and "total to clear 50 ft. obstacle") by 7% of the "total to clear 50 ft. obstacle" figure.

	MAXIMUM RATE-OF-CLIMB DATA														
	AT SEA LEVEL & 59°F.			AT 5	AT 5000 FT. & 41°F.		AT 10	AT 10,000 FT. & 23°F.		AT 15,000 FT. & 5°F.		ֆ 5°F.	AT 20,000 FT. & -12°F.		-12°F.
GROSS WEIGHT LBS.	IAS MPH	RATE OF CLIMB FT/MIN	GAL. OF FUEL USED	IAS MPH	RATE OF CLIMB FT/MIN	From SL FUEL USED	IAS MPH	RATE OF CLIMB FT/MIN	From SL FUEL USED	IAS MPH	RATE OF CLIMB FT/MIN	From SL FUEL USED	IAS MPH	RATE OF CLIMB FT/MIN	From SL FUEL USED
3100	91	755	1.5	89	540	4.2	87	330	7.8	85	110	13.2			

- NOTES: 1. Flaps up, full throttle, 2600 RPM, mixture leaned for smooth operation above 5000 ft.
 2. Fuel used includes warm-up and take-off allowance.

 - For hot weather, decrease rate of climb 30 ft./min. for each 10°F above standard day temperature for particular altitude.

SECTION VI - OPERATIONAL DATA (continued)

			CRUI	SE PE	RFORMA	ANCE								
			EXT	ENDED F	RANGE MIXT	URE								
	Standa	rd Condit	ions -	Zero W	ind - Gr	oss Weight -	3100 Pound	ls						
	2500 FEET													
					56 GAL (NO	RESERVE)	75 GAL (NO	RESERVE)						
RPM	MP	% BHP	TAS MPH	GAL/ HOUR	ENDR. HOURS	RANGE MILES	ENDR. HOURS	RANGE MILES						
2450	23	76	154	14.2	3.9	600	5.3	810						
	22	72	151	13.4	4.2	630	5.5	835						
	21	68	146	12.7	4.4	635	5.8	850						
	20	63	141	12.0	4.7	655	6.2	870						
2300	23	71	148	(3.1)	, 4.3	630	5.6	830						
	22	67	145	12.2	4.6	660	6.1	885						
	21	62	140	11.5	4.8	675	6.5	905						
	20	59	137	11.0	5.1	700	6.8	925						
2200	23	67	145	12.1	4.7	670	6.1	885						
	22	63	141	11.4	4.9	690	6.5	910						
	21	59	137	10.8	5.2	710	6.9	940						
	20	55	132	10.2	5.5	720	R	955						
2000*	20	47	119	8.7	6.4	760	8.6	1015						
	19	43	113	8.2	6.8	765	9.0	1015						
	18	39	106	7.5	7.4	785	9.9	1045						
	17	35	96	7.0	8.0	765	10.6	1015						

^{*}Power settings in this block represent maximum range settings

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SECTION VI - OPERATIONAL DATA (continued)

			CRUI	SE PE	RFORMA	ANCE							
	EXTENDED RANGE MIXTURE Standard Conditions - Zero Wind - Gross Weight - 3100 Pounds 5000 FEET												
			56 GAL (NO RESERVE) 75 GAL (NO RESER										
RPM	MP	% BHP	TAS MPH	GAL/ HOUR	ENDR. HOURS	RANGE MILES	ENDR. HOURS	RANGE MILES					
2450	23	78	158	14.5	3.8	600	5.1	800					
	22	73	154	13.6	4.1	630	5.5	835					
	21	70	150	13.0	4.3	640	5.7	860					
	20	65	146	12.2	4.6	665	6.1	890					
2300	23	73	154	3.4	4.2	640	5.5	850					
	22	69	149	12.6	4.5	665	5.9	880					
	21	64	145	11.9	4.7	670	6.2	895					
	20	60	141	11.2	5.0	705	6.7	940					
2200	23	68	148	12.4	4.5	660	6.0	890					
	22	64	145	11.7	4.7	685	6.4	925					
	21	60	141	11.0	5.1	720	6.8	950					
	20	57	136	10.5	5.3	720		955					
2000*	20	48	124	9.0	6.2	770	8.3	1025					
	19	45	117	8.5	6.6	770	8.7	1020					
	18	41	109	7.9	7.1	770	9.4	1020					
	17	37	100	7.3	7.6	760	10.2	1015					

^{*}Power settings in this block represent maximum range settings

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SECTION VI - OPERATIONAL DATA (continued)

			CRUI	SE PE	RFORMA	ANCE								
	EXTENDED RANGE MIXTURE Standard Conditions - Zero Wind - Gross Weight - 3100 Pounds 7500 FEET													
	0				56 GAL (NO	RESERVE)	75 GAL (NC	RESERVE)						
RPM	MP	% BHP	TAS MPH	GAL/ HOUR	ENDR. HOURS	RANGE MILES	ENDR. HOURS	RANGE MILES						
2450	21	71	154	13.1	4.3	655	5.6	865						
	20	67	151	12.4	4.5	670	6.0	905						
	19	62	146	11.7	4.7	690	6.4	930						
	18	58	140	11.0	5.1	715	6.8	945						
2300	21	66	150	(2.2 ×	4.6	680	6.1	915						
	20	62	146	11.6	4.8	705	6.4	930						
	19	58	140	11.0	5,1	715	6.8	945						
	18	54	135	10.5	5.3	715	7.1	950						
2200	21	62	146	11.4	4.9	715	6.5	945						
	20	58	140	10.7	5.2	725	7.0	970						
	19	54	135	10.2	5.5	740	7.2	975						
	18	51	129	9.7	5.8	740	7.6	980						
2000*	20	50	128	9.2	6.0	770	8.1	1030						
	19	47	122	8.7	6.4	780	8.6	1040						
	18	43	113	8.1	6.9	775	9.2	1040						
	17	39	105	7.6	7.3	770	9.8	1025						

^{*}Power settings in this block represent maximum range settings

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SECTION VI - OPERATIONAL DATA (continued)

			CRUI	SE PE	RFORMA	ANCE								
	EXTENDED RANGE MIXTURE Standard Conditions - Zero Wind - Gross Weight - 3100 Pounds 10,000 FEET													
	0				,	RESERVE)	,	RESERVE)						
RPM	MP	% BHP	TAS MPH	GAL/ HOUR	ENDR. HOURS	RANGE MILES	ENDR. HOURS	RANGE MILES						
2450	19	63	151	11.9	4.7	700	6.2	935						
	18	60	145	11.2	5.0	725	6.7	965						
	17	55	139	10.6	5.3	735	7.1	975						
	16	51	133	10.0	5.6	740	7.4	985						
2300	19	60	145	(1) x	5.0	725	6.7	965						
	18	56	140	10.5	5.3	740	7.1	985						
	17	51	133	9.8	5,7	750	7.6	1010						
	16	47	125	9.2	6.0	755	8.1	1010						
2200	19	56	140	10.4	5.4	755	7.1	1000						
	18	52	134	9.8	5.7	760	7.6	1020						
	17	49	127	9.3	6.0	765	8.0	1010						
	16	45	119	8.7	6.4	760	8.6	1015						
2000*	19	48	125	8.9	6.2	775	8.4	1045						
	18	44	118	8.4	6.6	775	8.8	1040						
	17	40	110	7.8	7.2	785	9.5	1040						
	16	38	95	7.4	7.5	715	10.1	955						

^{*}Power settings in this block represent maximum range settings

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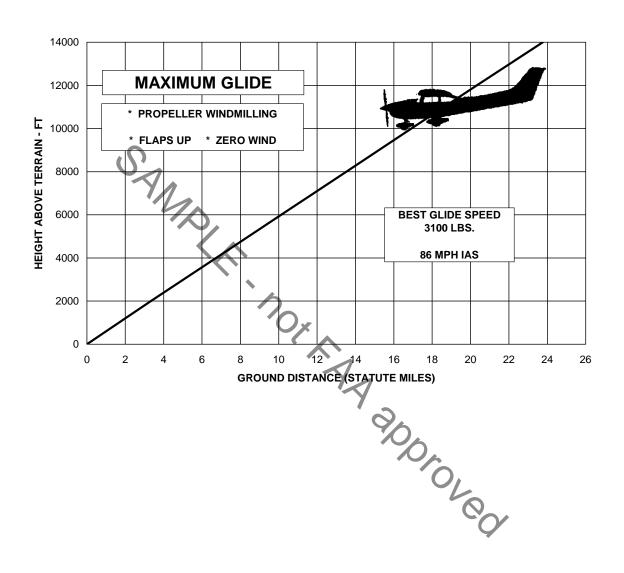
SECTION VI - OPERATIONAL DATA (continued)

			CRUI	SE PE	RFORMA	ANCE								
	EXTENDED RANGE MIXTURE Standard Conditions - Zero Wind - Gross Weight - 3100 Pounds 15,000 FEET													
RPM	SME	% BHP	TAS MPH	TAS GAL/ ENDR. RANGE ENDR.										
2450	16	54	140	10.4	5.4	755	7.1	1000						
	15 14	50 46	134 120	9.8 9.2	5.7 6.0	760 725	7.6 8.1	1020 970						
2300	16	50	134	9.6	5.8	770	7.7	1030						
	15	47	125	9.1	6.1	765	8.2	1020						
	14	42	107	8.5	6.6	705	8.7	935						
2200	16	47	125	9.1	6.1	765	8.2	1020						
	15	44	114	8.6	6.5	740	8.6	985						
	14	40	99	8.0	7.0	690	9.3	920						
2000	16	40	99	7.8	7.2	705	9.5	935						
	15	37	90	7.3	7.6	685	10.2	910						

Note: Range and endurance values in the cruise performance tables above take into account corrections to the basic Cessna Owner's Manual per AD 75-16-01 and Cessna Service Letter SE 75-7 regarding amendment of usable fuel quantities.

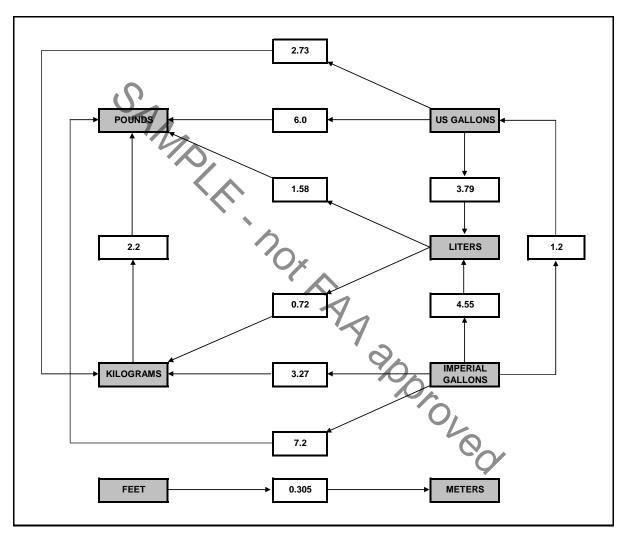
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SECTION VI - OPERATIONAL DATA (continued)



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SECTION VI - OPERATIONAL DATA (continued)



Metric / Imperial / US Units Conversion Chart

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SECTION VII - OPTIONAL SYSTEMS

NO CHANGES

ALPHABETICAL INDEX

NO CHANGES

THANGL

ING REQUIREMENTS

NO CHANGES

ARRANGE

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