



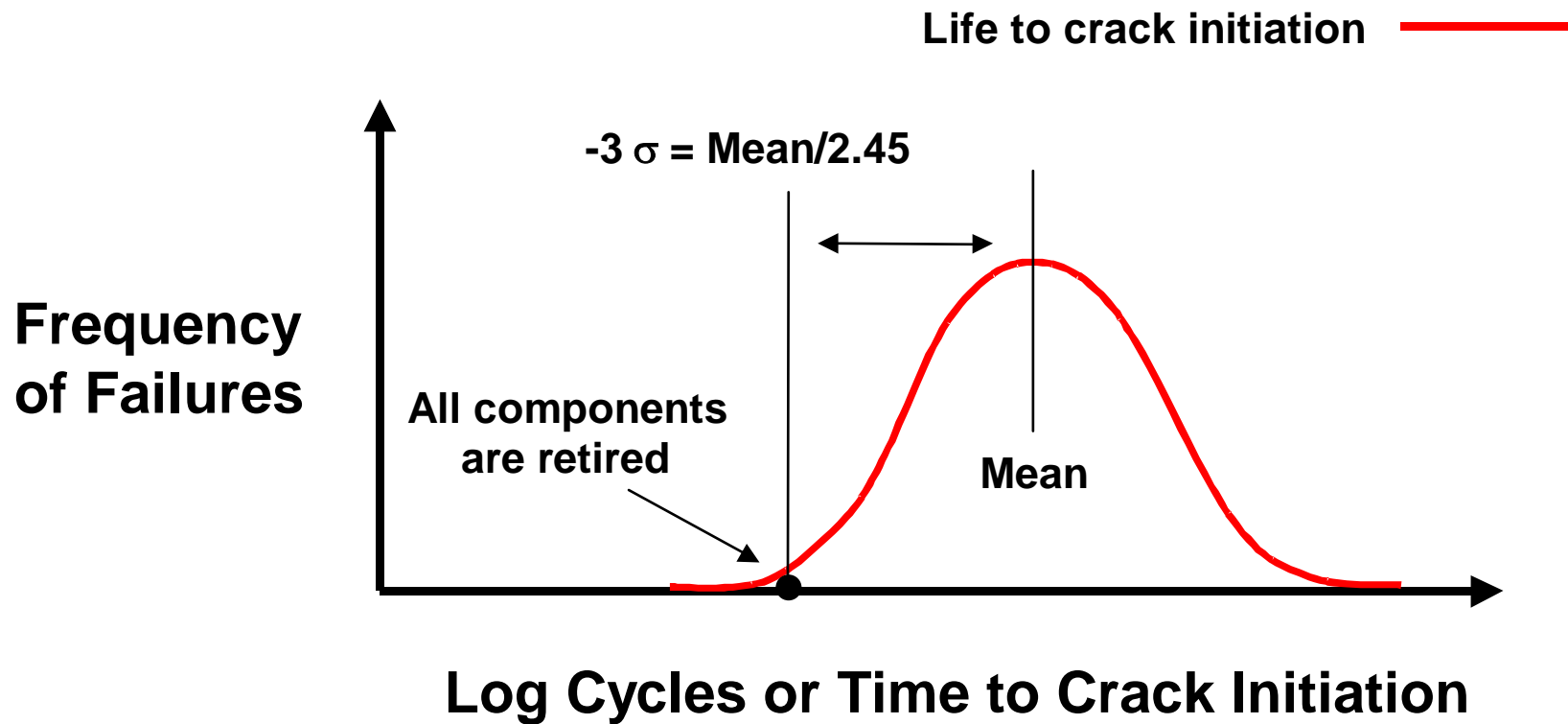
Life Prediction of Critical Turbine Components Using Prognosis

By

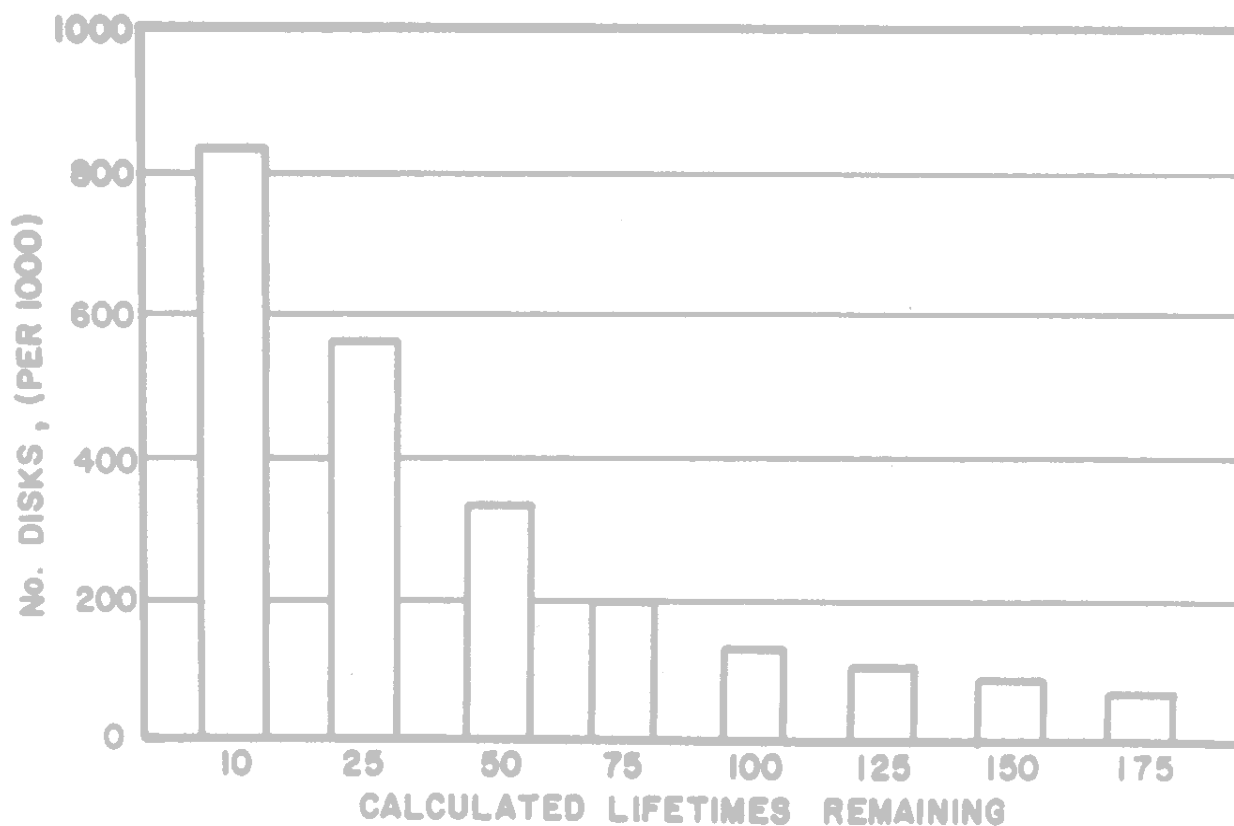
Ashok K Koul

***Life Prediction Technologies Inc.,
23-1010 Polytek Street,
Ottawa, Ontario, Canada K1J 9J1***

Safe Life LCMM Philosophy (1/1000 probability of crack initiation)

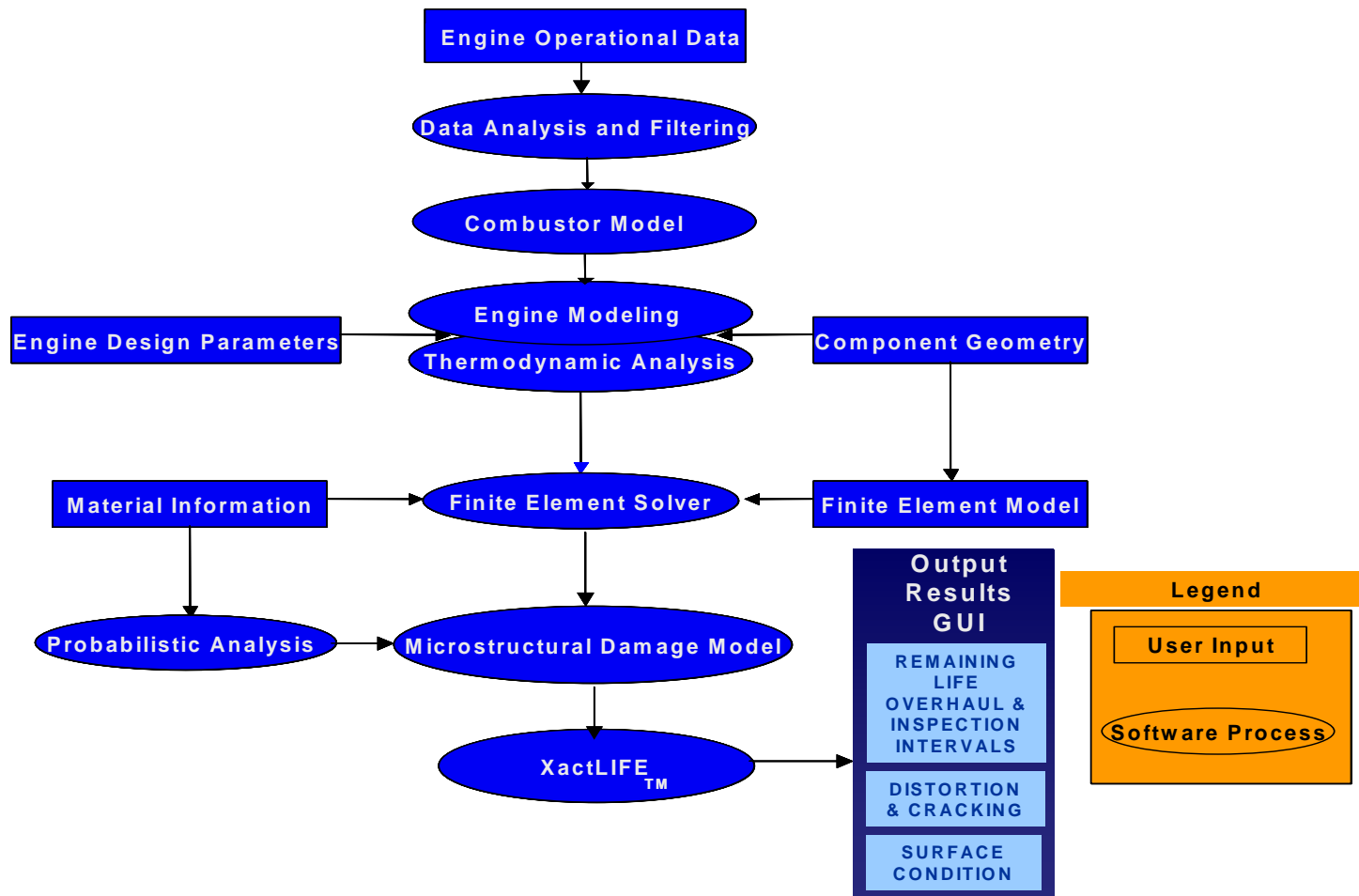


Remaining Life of Safe Life Expired Discs



USAF Study

XactLIFE System Architecture (Patent Pending)



Prognosis Based Engineering Assessment of Rotors

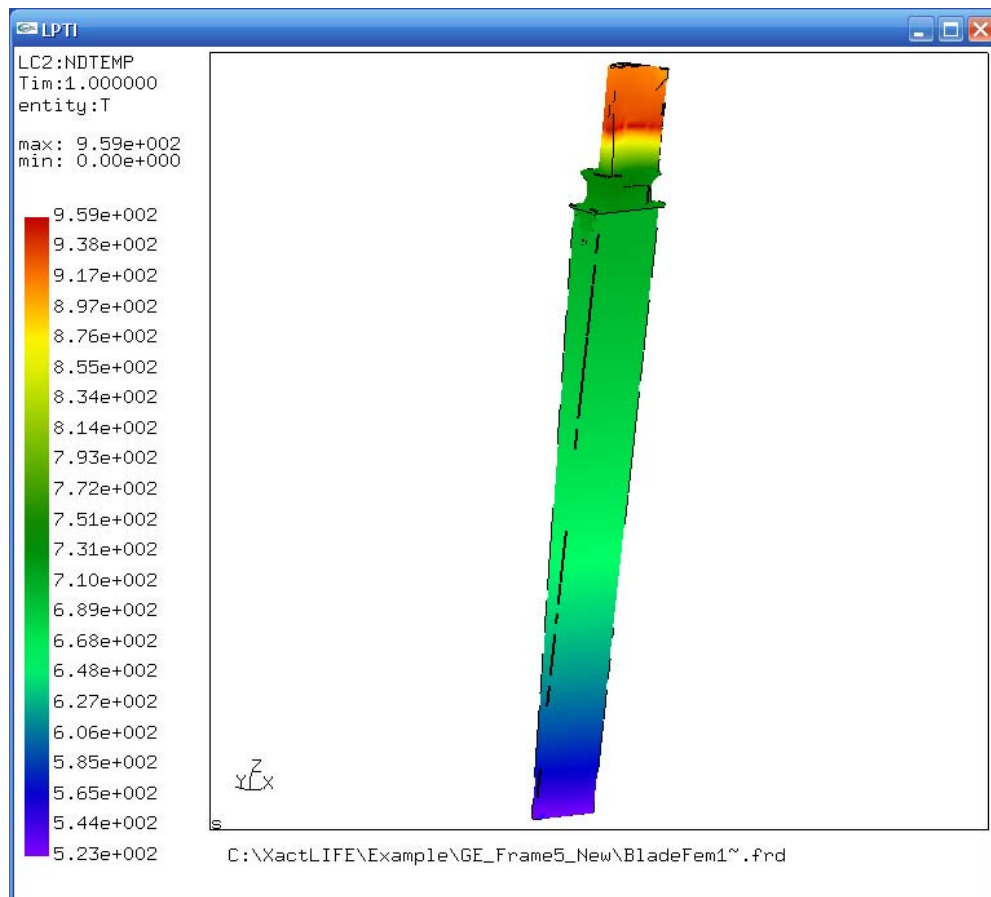


- Determine the temperature and stress profiles that the components are exposed to in a specific engine operating environment
- Consider the effect of operating loads on crack initiation life in different rotors
- Identify the most fracture prone rotor
- Determine the primary and secondary fracture critical locations of the rotor(s) and predict a safe operating interval for the engine
- Consider the effect of NDI detection limit/uncertainties on the future safe operation of the engine and select the most appropriate NDI technique for inspecting the individual rotor locations.
- Design new NDI probes if necessary

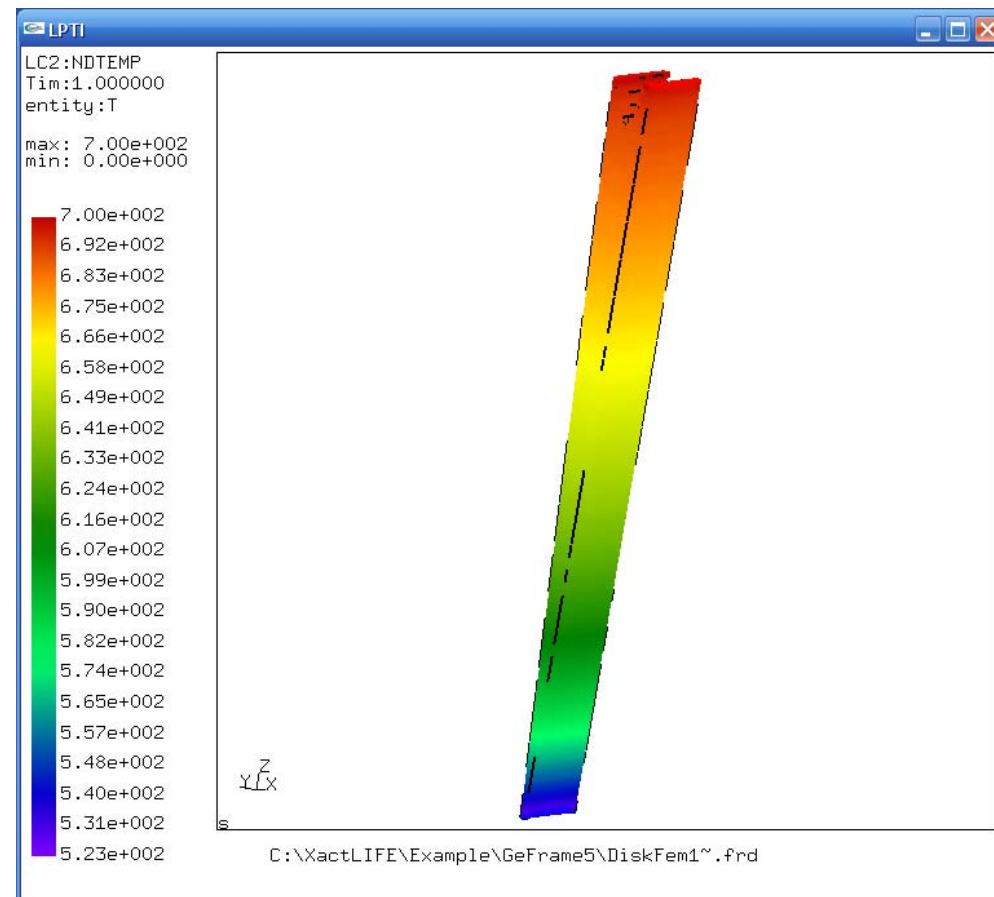
Temperature Profile Frame 5 blade/Disc Combined



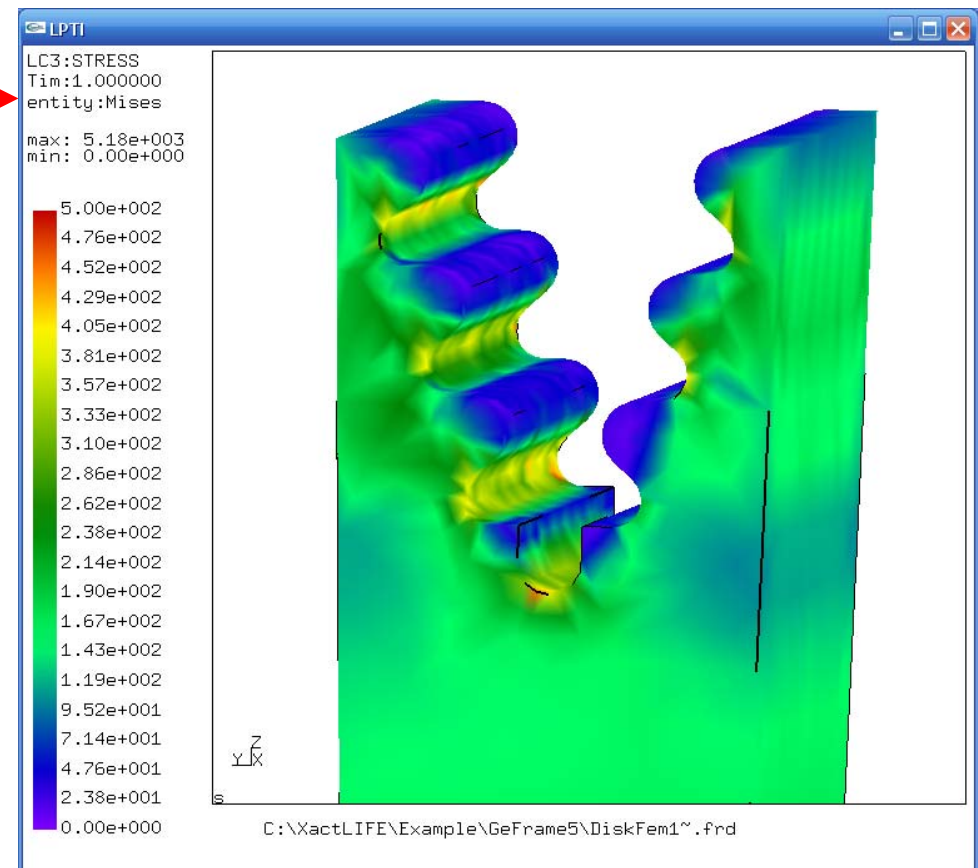
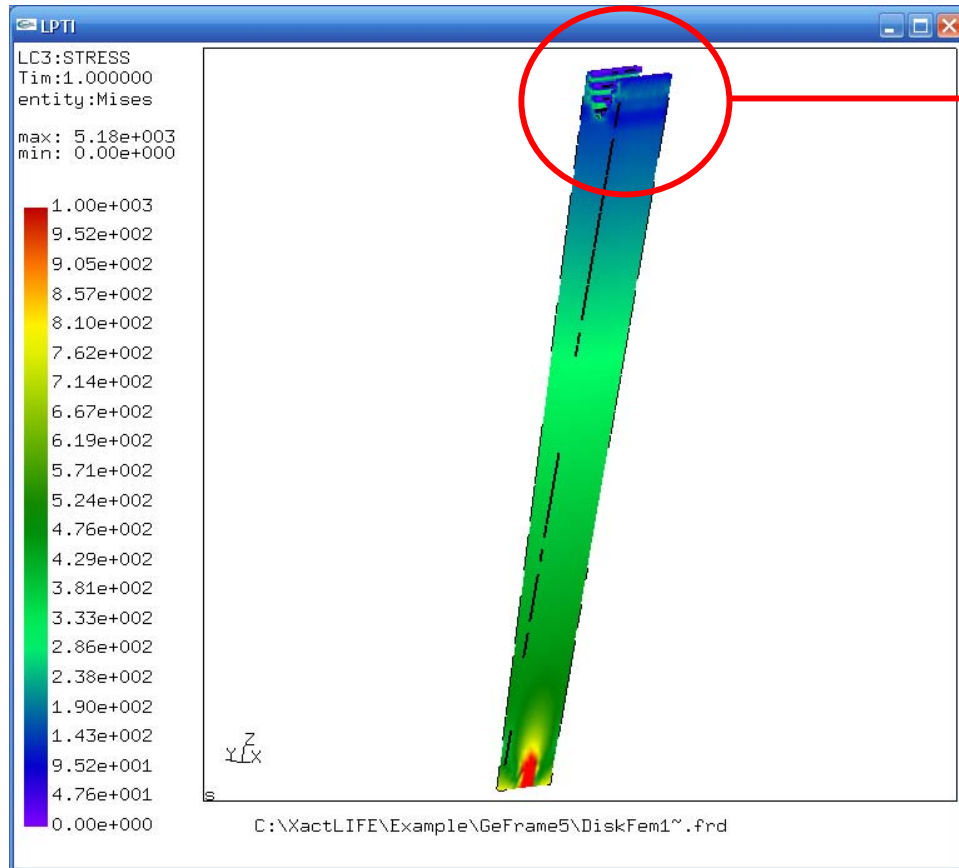
Bladed Disc



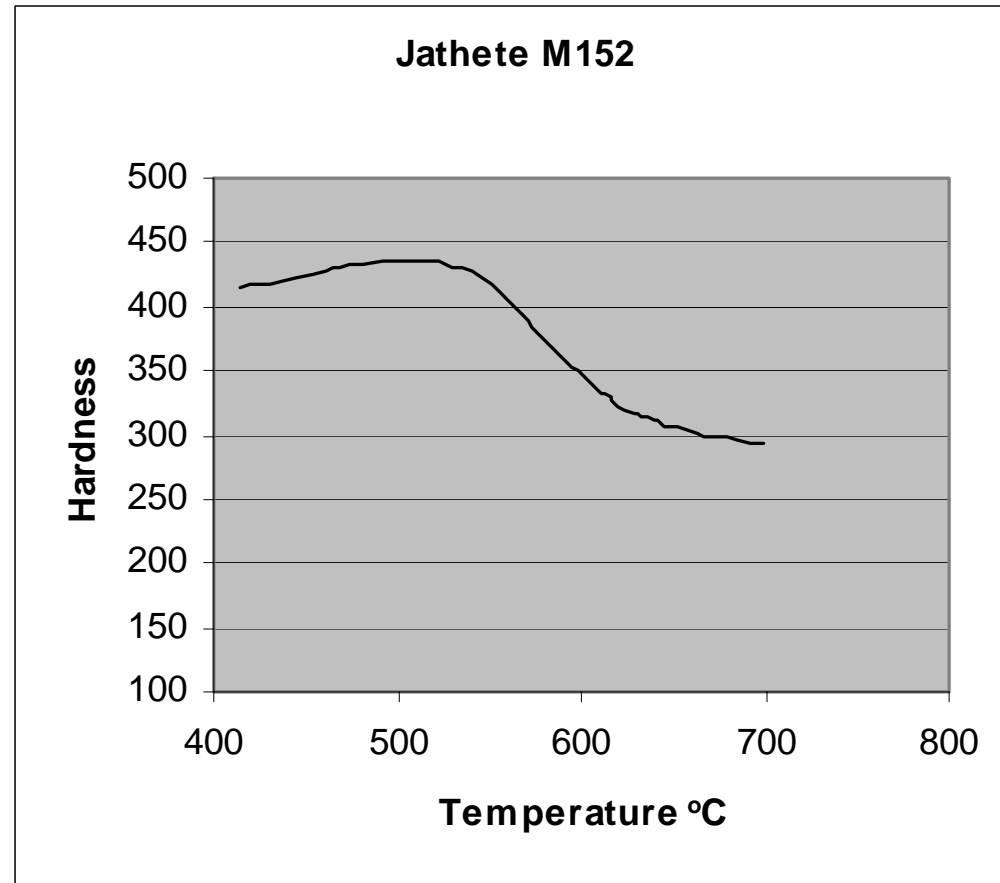
Stand Alone Disc



Stress Profile Frame5



Jethete M152 Hardness Behavior

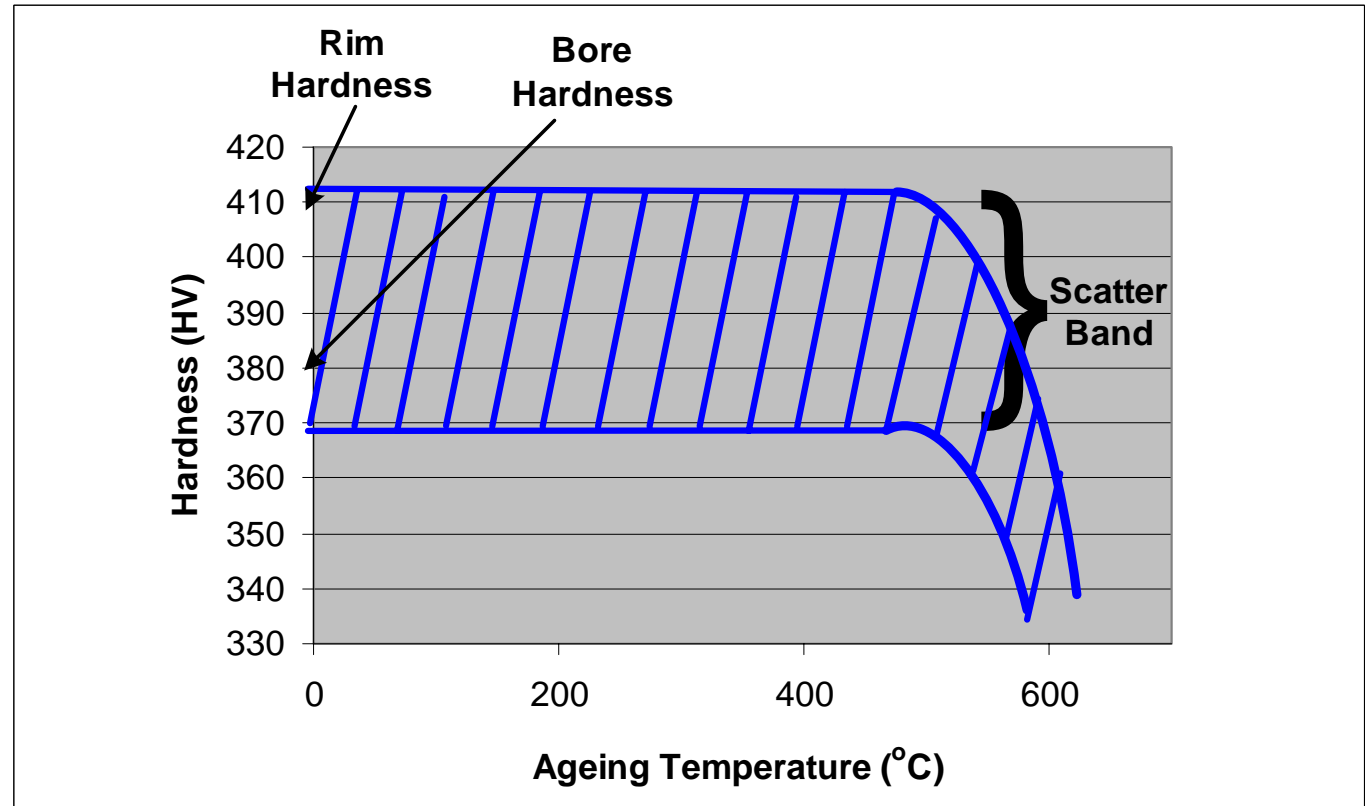


Frame 5 M152 Hardness Based Metallurgical Assessment of Discs



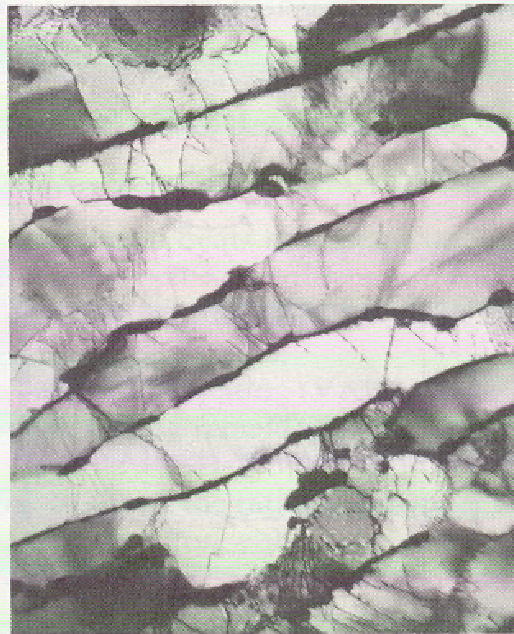
Jethete M 152 Stainless Steel

C 0.08 – 0.13 wt%
Cr 11.0 – 12.5 wt%
Mo 1.5 – 2.0 wt%
V 0.25 – 0.4 wt%
Ni 2.0 – 3.0 wt%



*Disc bore hardness was 380HV and all that can be concluded is
that the disc rim temperature is less than 500°C*

Electron microscopy based assessment of M152 Turbine Disc Steel (Thin foil / Replica)



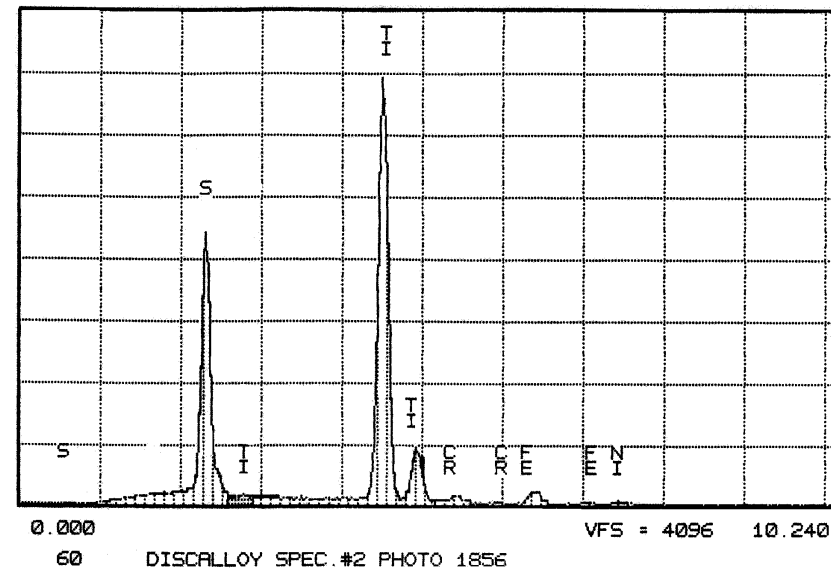
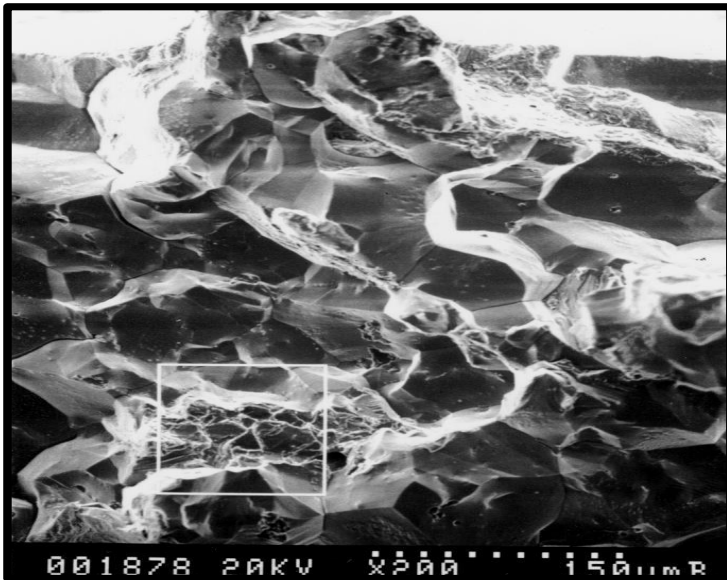
30,000h/500°C



30,000h/550°C

No noticeable change in secondary carbide morphology below 500°C (Schinkel et. al., ASM International, 1983). No over-temperature exposure or rim temperature does not exceed 500°C during service

Sub-Surface Crack Nucleation Sites in Turbine Discs



Replica technique is only good if surface connected cracks are present

XactLIFE Based Frame 5001P Disc Creep Life (Mid-East Client)



Rim Temp. °C	415 °C (predicted)	450 °C (assumed)	500 °C (assumed)
Rim Life In Hours	360,000 (predicted)	100,000 (assumed)	20,000 (assumed)

***In this case lower bound creep crack
initiation life was estimated to be
360,000 hours***

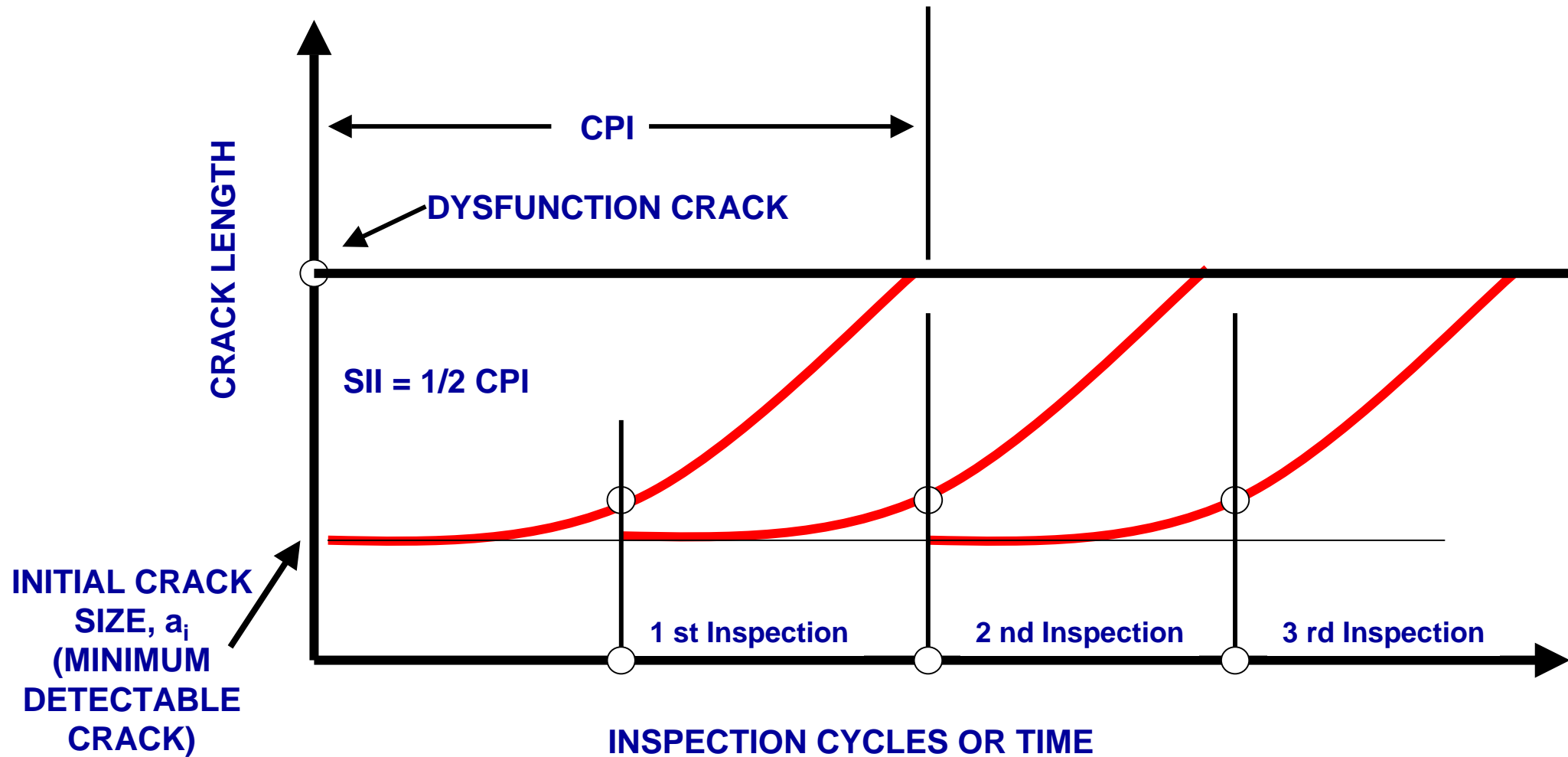
- Accurate assessment of the disc rim temperature is vital for assessing the residual life range available for life extension for a specific engine operating environment
- Rim temperature can only be determined through gas path modeling and heat transfer analysis
- Hardness checks and replica assessment only indicate whether the discs have been subjected to any over-temperature effects during service
- Replicas can only be used to detect any surface connected flaws.
- Return to service interval prediction involves detailed engineering assessment of the engine as a whole.



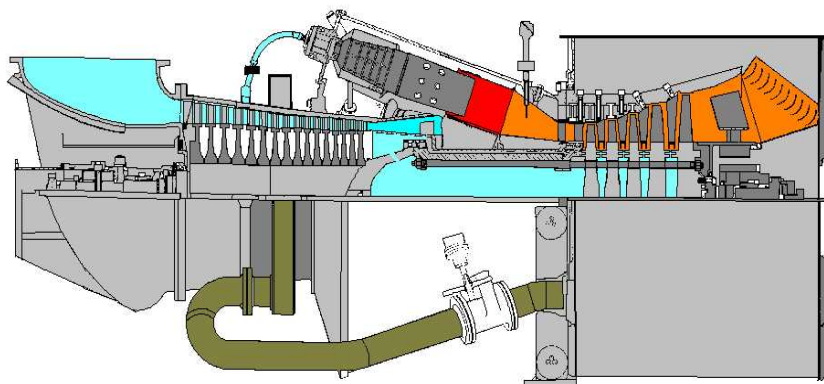
PROGNOSIS BASED SAFE INSPECTION INTERVAL OR RETURN TO SERVICE INTERVAL PREDICTION FOR W-101 ENGINES

PDVSA Fleet, Maracaibo,
Venezuela

LIFE CYCLE MANAGEMENT OF TURBINE DISCS/SPACERS MIL-STD-1783



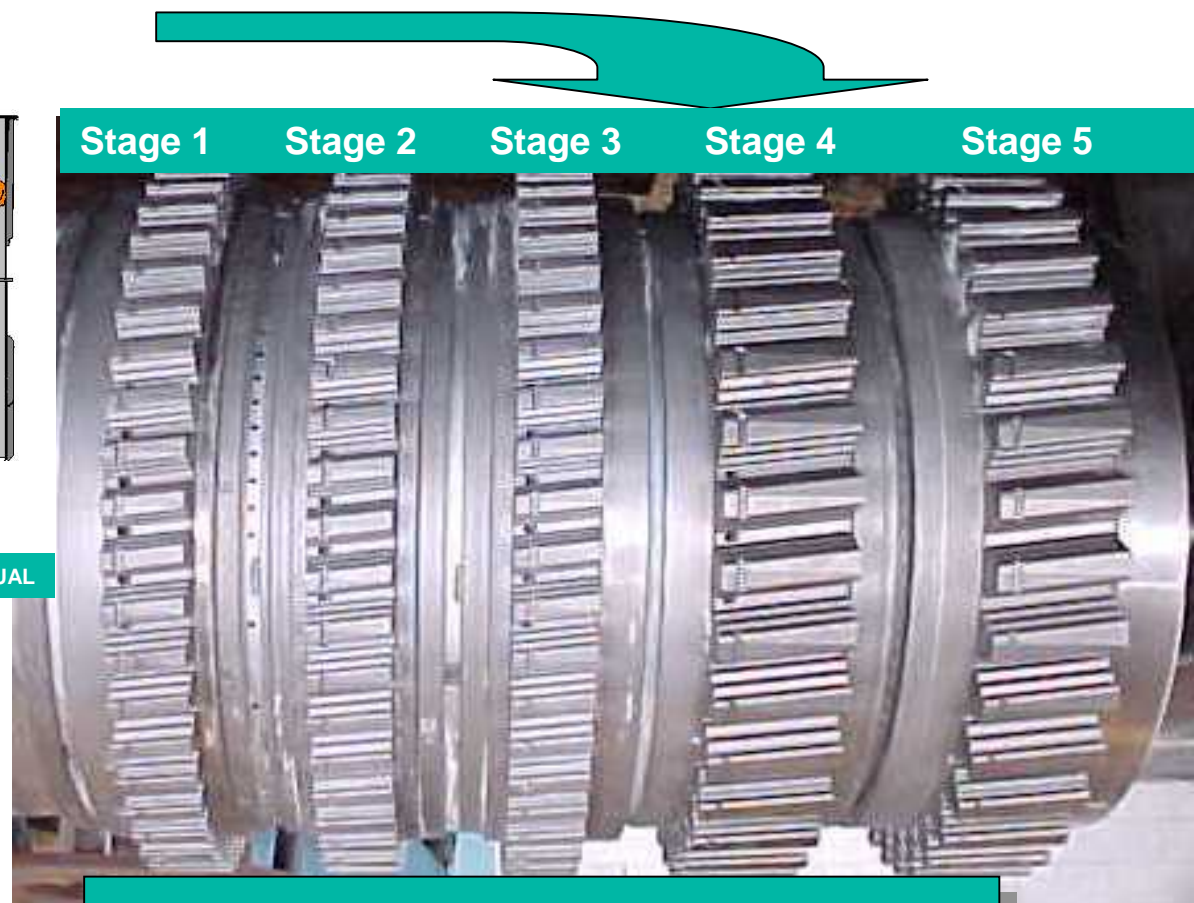
W-101 Turbine Configuration



SOURCE: TURBINE MANUAL

**SINGLE SHAFT GAS
TURBINE
9950 HP**

SPEED 6300 RPM



HOT SECTION ROTOR

LIFE CYCLE MANAGEMENT OF TURBINE DISCS



**EXTENDED DAMAGE TO THE
TURBINE HOT SECTION**

LIFE CYCLE MANAGEMENT OF TURBINE DISCS

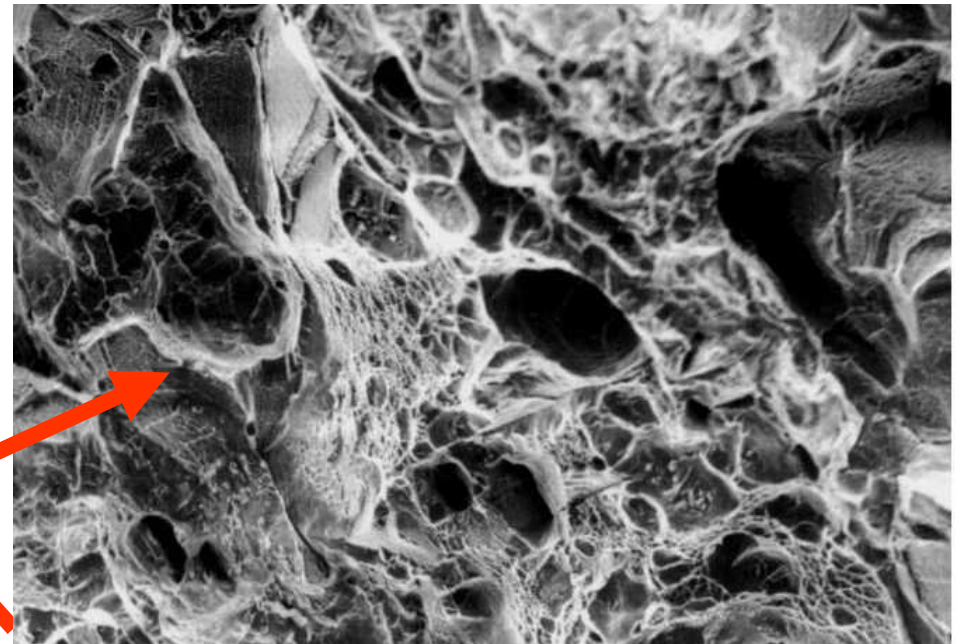
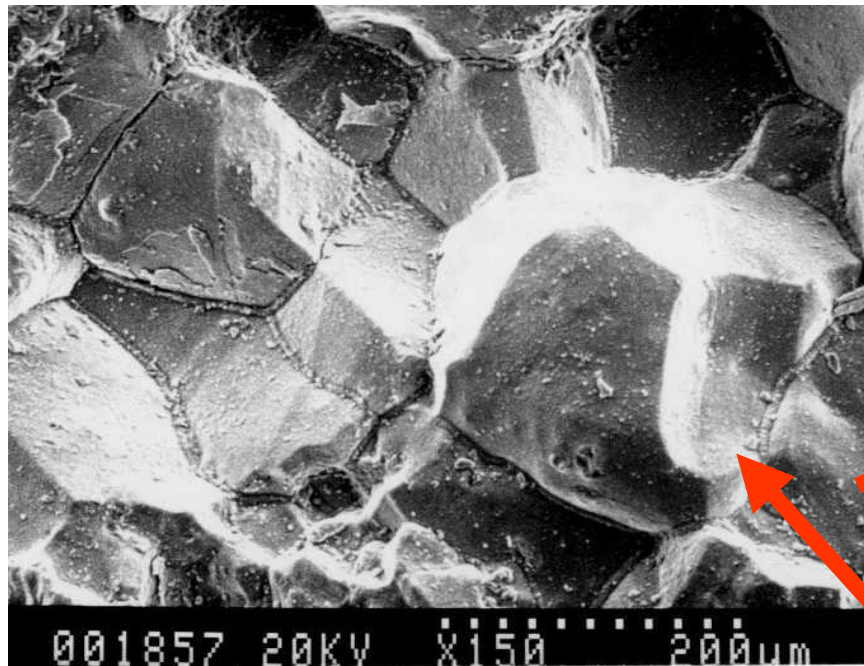


**CRACK AT THE FIR TREE BOTTOM SERRATION
DETECTED DURING OVERHAUL
INSPECTION**

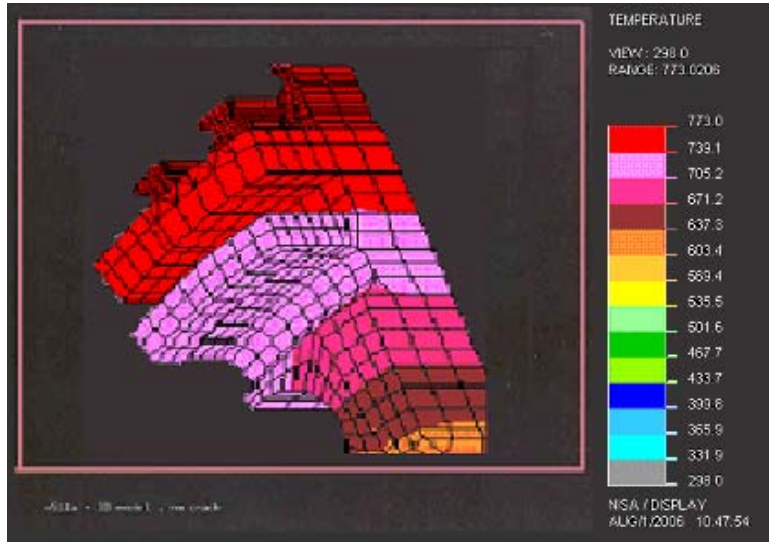
IDENTIFICATION OF CRACK INITIATION AND PROPAGATION MECHANISMS



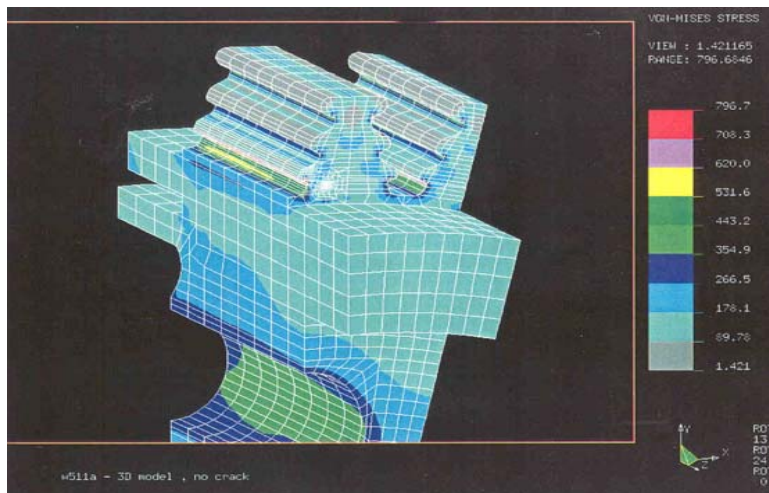
Brittle Creep Fracture in Discaloy



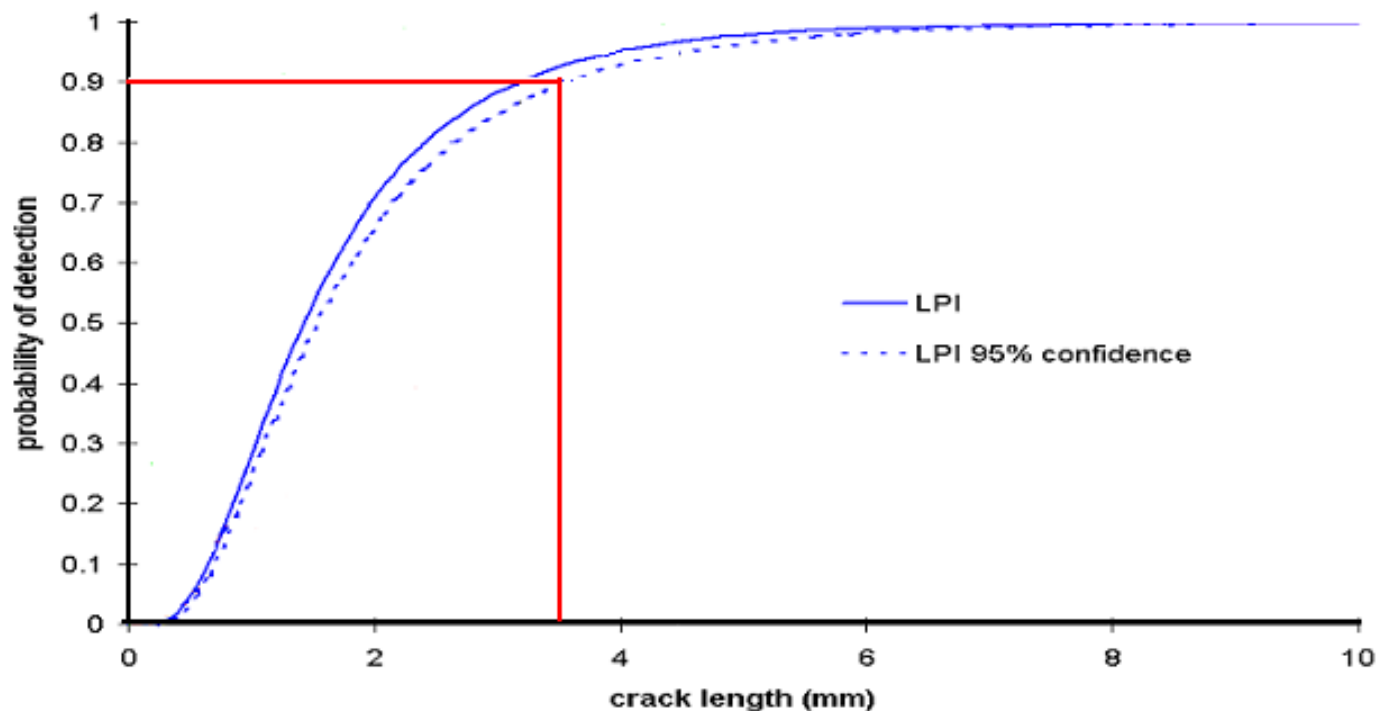
FEM Based Fracture Mechanics Analysis



- Stress-temperature profile of the part (boundary conditions)
- Identification of fracture critical location (crack nucleation / initiation based)
- Determine 'a' v/s K Correlation
- Choose damage evolution models (Short CCGR and Long CCGR)
- Conduct creep crack growth based safe inspection interval or overhaul interval prediction analysis



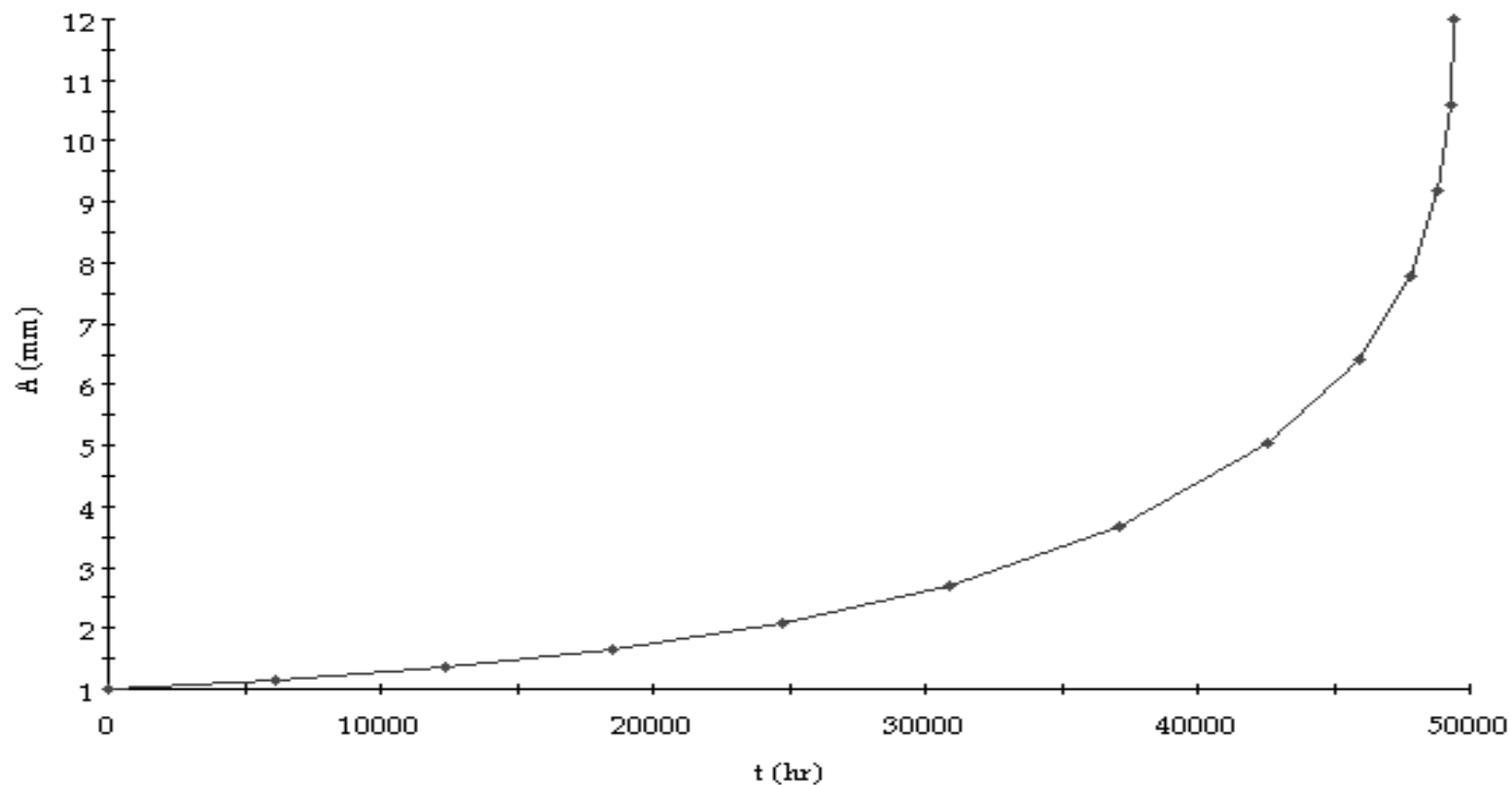
Probability of Detection Of Disc Cracks Using LPI



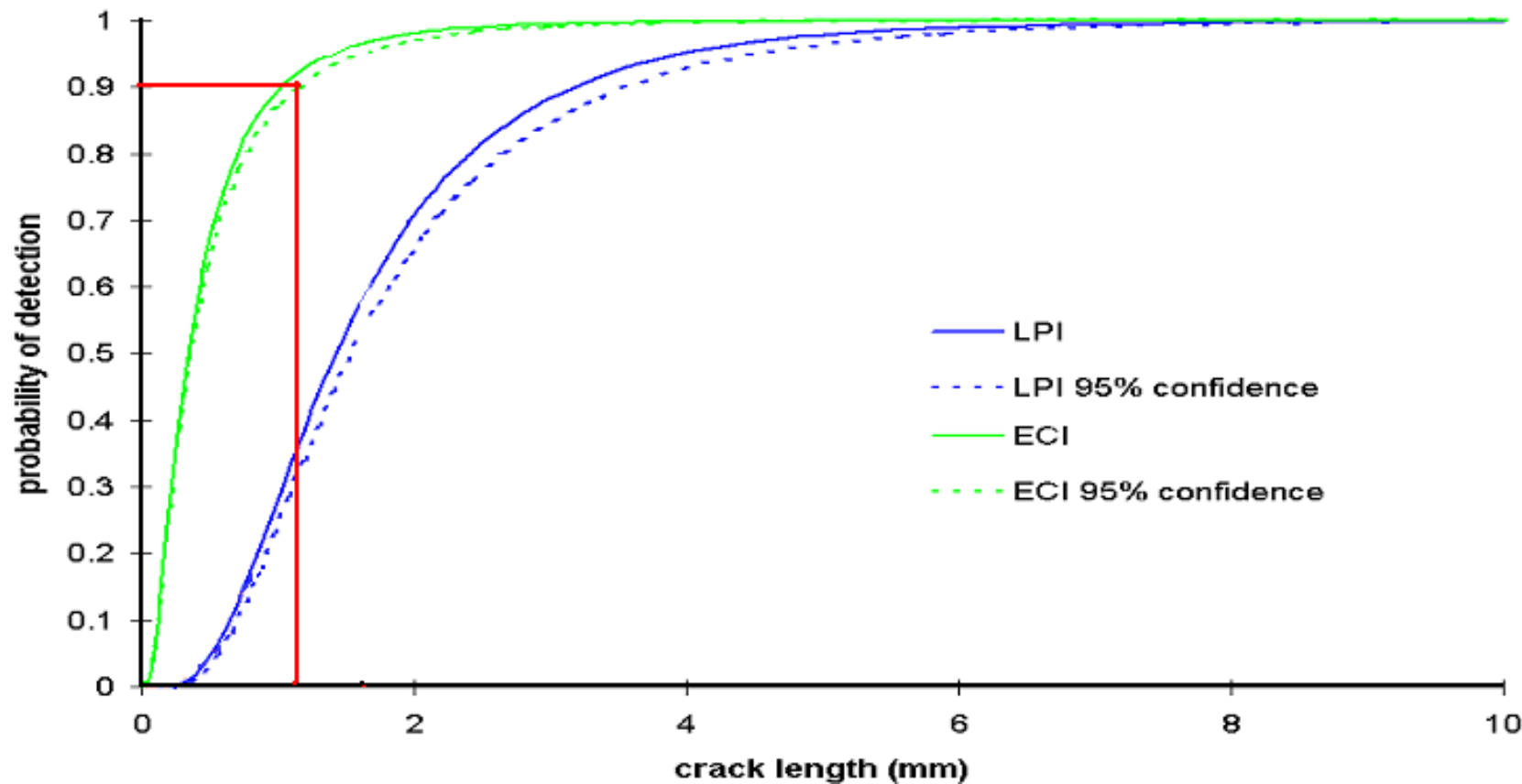
Safe Inspection Interval Prediction for LPI Technique Using Prognostics



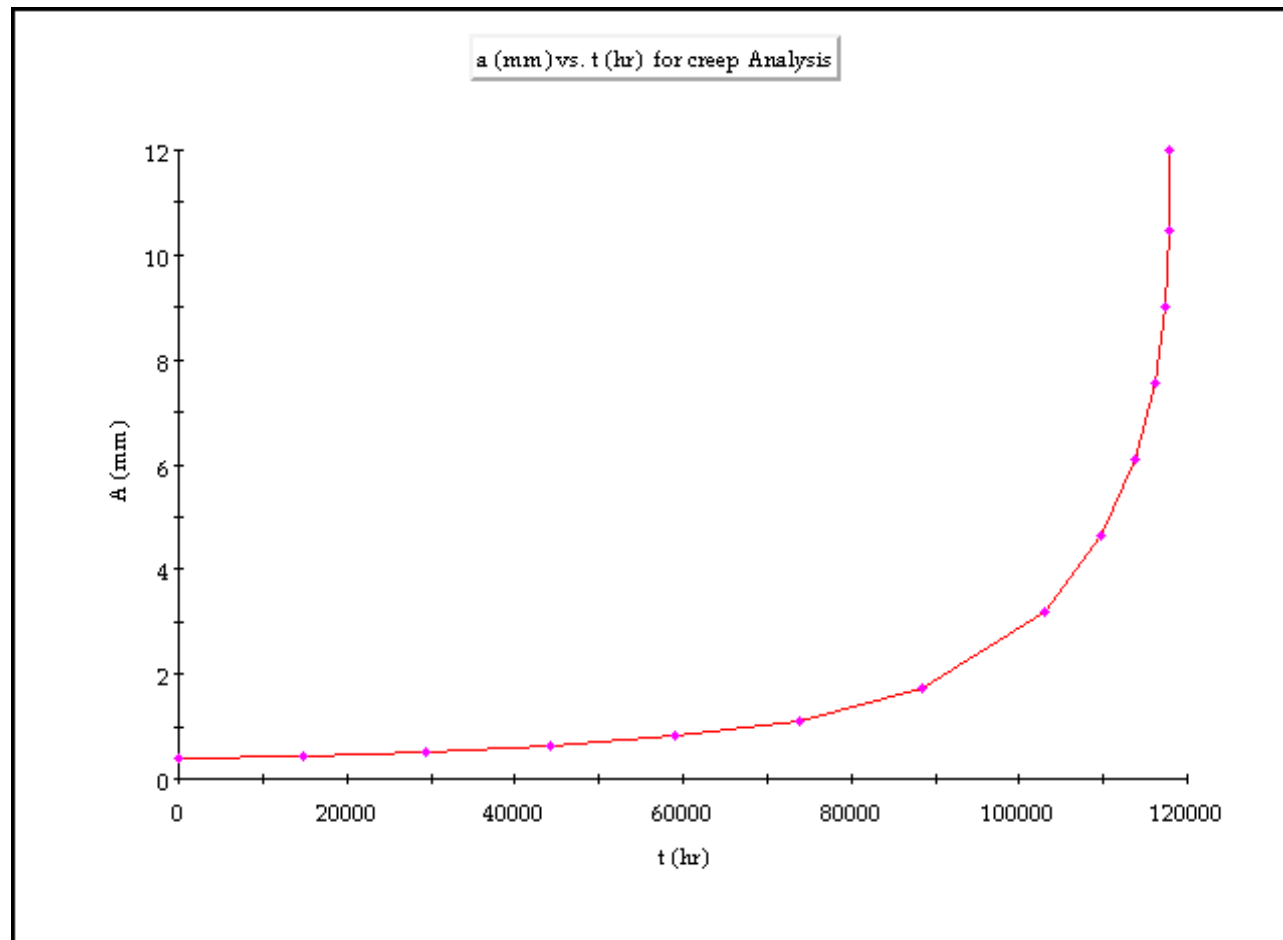
a (mm) vs. t (hr) for creep Analysis



Probability of Detection of Cracks Using LPI and ECI

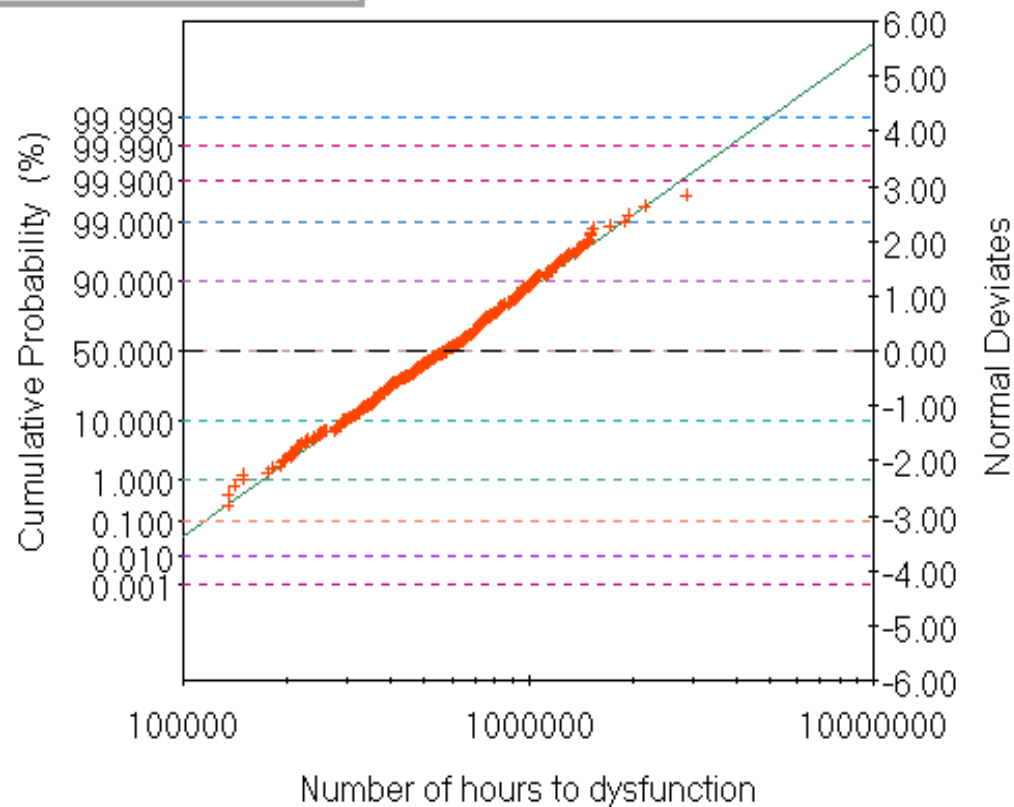


Safe Inspection Interval Prediction for ECI Technique Using Prognostics



No. of Disks = 436
Median = 565881.49 Hours
0.100 % F = 118551.60 Hours
Lambda = 13.25
dzeta = 0.51

Lognormal Analysis for Safe Inspection Intervals



W-101 Engine 10 Years of Experience Lessons Learned



- ❑ **Use RFC based life cycle management philosophy to maintain discs and NOT DESIGN LIFE APPROACH. Fleet life increased from 150,000 hours to 300,000 hours**
- ❑ **It allows utilization of the crack initiation life of each and every individual part rather than only 1 in 1000 parts allowed by the Design life approach. Only retire discs when a crack is detected or wear is excessive.**
- ❑ **Select appropriate inspection technique to suit user needs**
- ❑ **Prognosis based LCMM saves users close to 50% on overhaul costs**

What We Do Differently in Prognosis



- **Residual Life Assessment Using XactLIFE**
 - Precisely define the temperature gradient from bore to rim for the specific engine operating environment through engineering analysis
 - Use FE methods to determine thermal-mechanical stress gradients
 - Use physics based damage models to predict fracture critical locations and compute crack initiation life for a specific user
 - Use hardness checks and replication only for rough verification of T
- **Inspection Interval Prediction Using XactLIFE**
 - Predict a safe return to service interval for the specific user taking into account consequences of any flaws that may be missed during inspection or introduced during manufacturing or present as a metallurgical defect
 - Define quantitative NDI requirements for the user
 - Select most suitable NDI techniques for inspection and design probes if necessary
- **XactLIFE Based Reliability Assessment and Risk Mitigation**
 - Conduct probabilistic analysis taking into account material variability, usage variability and inspection uncertainty
 - Quantify engine reliability for future safe engine operation

Cost Savings



- Disc replacement costs are deferred on long term basis. In the case of W-101 fleet, PDVSA deferred \$40 Million investment by more than 10 years.
- Overhaul costs are reduced by as much as 45-50% of the regular overhaul costs. PDVSA saved \$2.5Million per year in reduced overhaul costs for the W-101 fleet.
- Other repair costs can also be reduced significantly



Thank You

Life Prediction Technologies Inc.

23-1010 Polytek Street

Ottawa, On, K1J 9J1

www.LifePredictionTech.com



613-744-7574

613-744-5278

